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Masters et al.

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(54) **TWIN FIN FAIRING**

(71) Applicant: **Asset Integrity Management Solutions, L.L.C.**, Houston, TX (US)

(72) Inventors: **Rodney H. Masters**, Houston, TX (US); **Randy W. Masters**, Houston, TX (US); **Michael Sykes**, Spring, TX (US); **Steven J. Leverette**, Richmond, TX (US); **Kenneth J. Schaudt**, Katy, TX (US)

(73) Assignee: **Asset Integrity Management Solutions, L.L.C.**, Houston, TX (US)

(21) Appl. No.: **15/255,460**

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Related U.S. Patent Documents

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Appl. No.: **11/684,861**
Filed: **Mar. 12, 2007**

U.S. Applications:

(63) Continuation-in-part of application No. 11/463,413, filed on Aug. 9, 2006, now Pat. No. 7,337,742.

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F15D 1/10 (2006.01)
F16L 1/12 (2006.01)
E21B 17/01 (2006.01)

(52) **U.S. Cl.**
CPC **F15D 1/10** (2013.01); **E21B 17/01** (2013.01); **F16L 1/123** (2013.01)

(58) **Field of Classification Search**
CPC F15D 1/10; F16L 1/123; E21B 17/01
See application file for complete search history.

(56) **References Cited**

U.S. PATENT DOCUMENTS

3,209,718 A *	10/1965	Rather et al.	114/243
3,241,513 A	3/1966	Rather et al.	
3,454,051 A	7/1969	Goepfert et al.	
3,859,949 A	1/1975	Toussaint et al.	
4,078,605 A *	3/1978	Jones	E21B 17/01 166/359
4,365,574 A	12/1982	Norminton et al.	
4,398,487 A	8/1983	Ortloff et al.	B63B 1/32 114/243
4,474,129 A	10/1984	Watkins et al.	
4,542,708 A	9/1985	Holcombe et al.	
4,655,682 A *	4/1987	Kunz et al.	415/119
4,829,929 A	5/1989	Kerfoot	
5,410,979 A	5/1995	Allen et al.	
5,456,199 A	10/1995	Kernkamp	

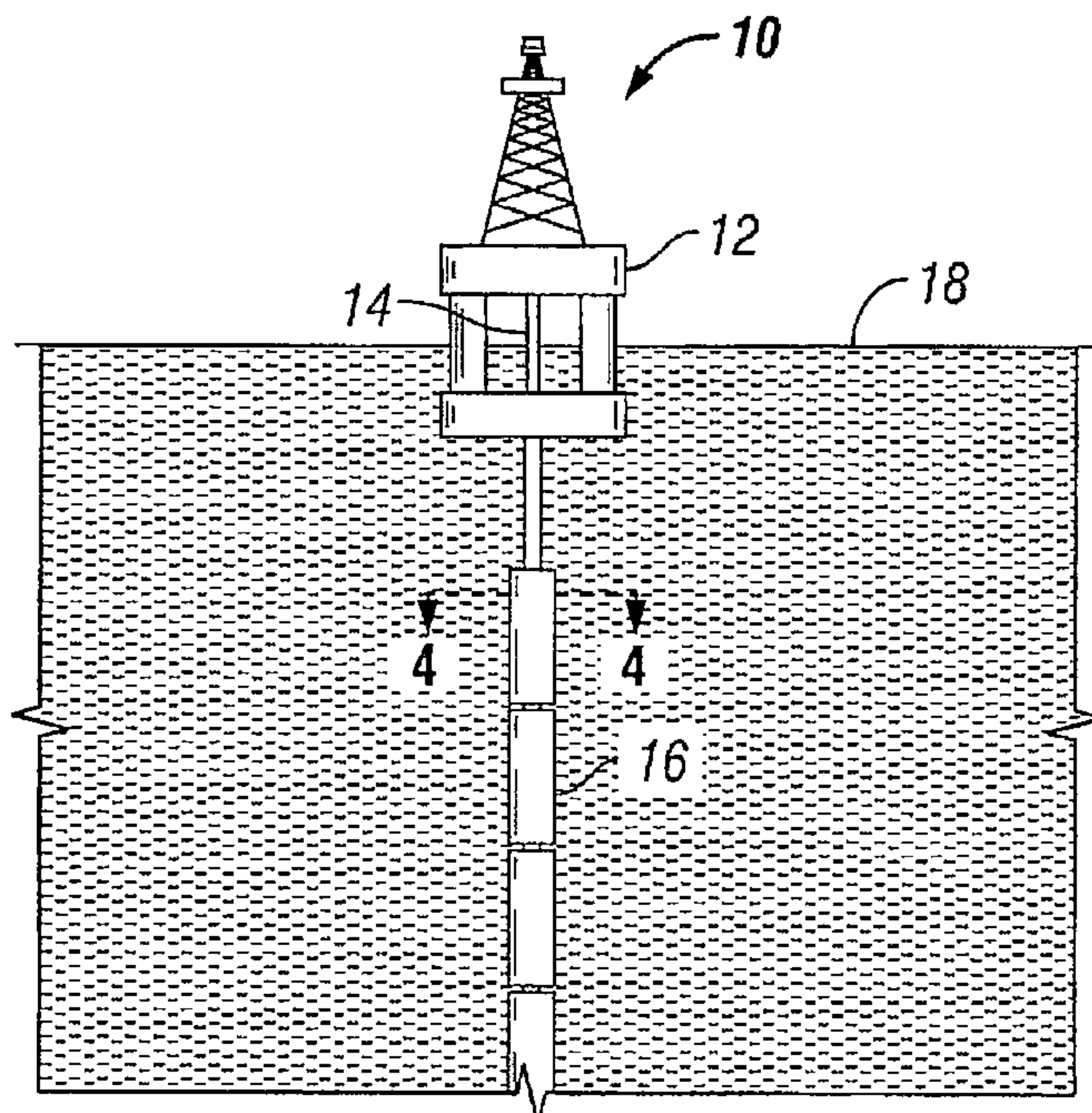
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Primary Examiner — Patricia L Engle
(74) *Attorney, Agent, or Firm* — Edmonds & Cmaidalka, P.C.

(57) **ABSTRACT**

A fairing for the reduction of vortex-induced vibration and the minimization of drag about a substantially cylindrical element immersed in a fluid medium. The fairing also eliminates the galloping phenomenon typically associated with a teardrop-shaped fairing. The fairing having a U-shaped cylindrical shell with opposing edges defining a longitudinal gap and parallel fins extending outwardly from the opposing edges of the shell, the parallel fins being positioned so as to reduce vortex-induced vibration, minimize drag and to eliminate the galloping phenomenon on the cylindrical element.

32 Claims, 12 Drawing Sheets



(56)

References Cited

U.S. PATENT DOCUMENTS

5,984,584 A * 11/1999 McMillan E21B 17/01
114/243
6,048,136 A 4/2000 Denison et al.
6,062,769 A 5/2000 Cunningham
6,067,922 A 5/2000 Denison et al.
6,092,483 A 7/2000 Allen et al.
6,223,672 B1 5/2001 Allen et al.
6,244,204 B1 6/2001 Weyman et al.
6,401,646 B1 6/2002 Masters et al.
6,517,289 B1 * 2/2003 Coakley et al. 405/216
2006/0021560 A1 * 2/2006 McMillan E21B 17/01
114/221 R
2008/0025800 A1 * 1/2008 Watkins B63B 21/502
405/216
2010/0119308 A1 * 5/2010 Somerville B63B 21/502
405/216

* cited by examiner

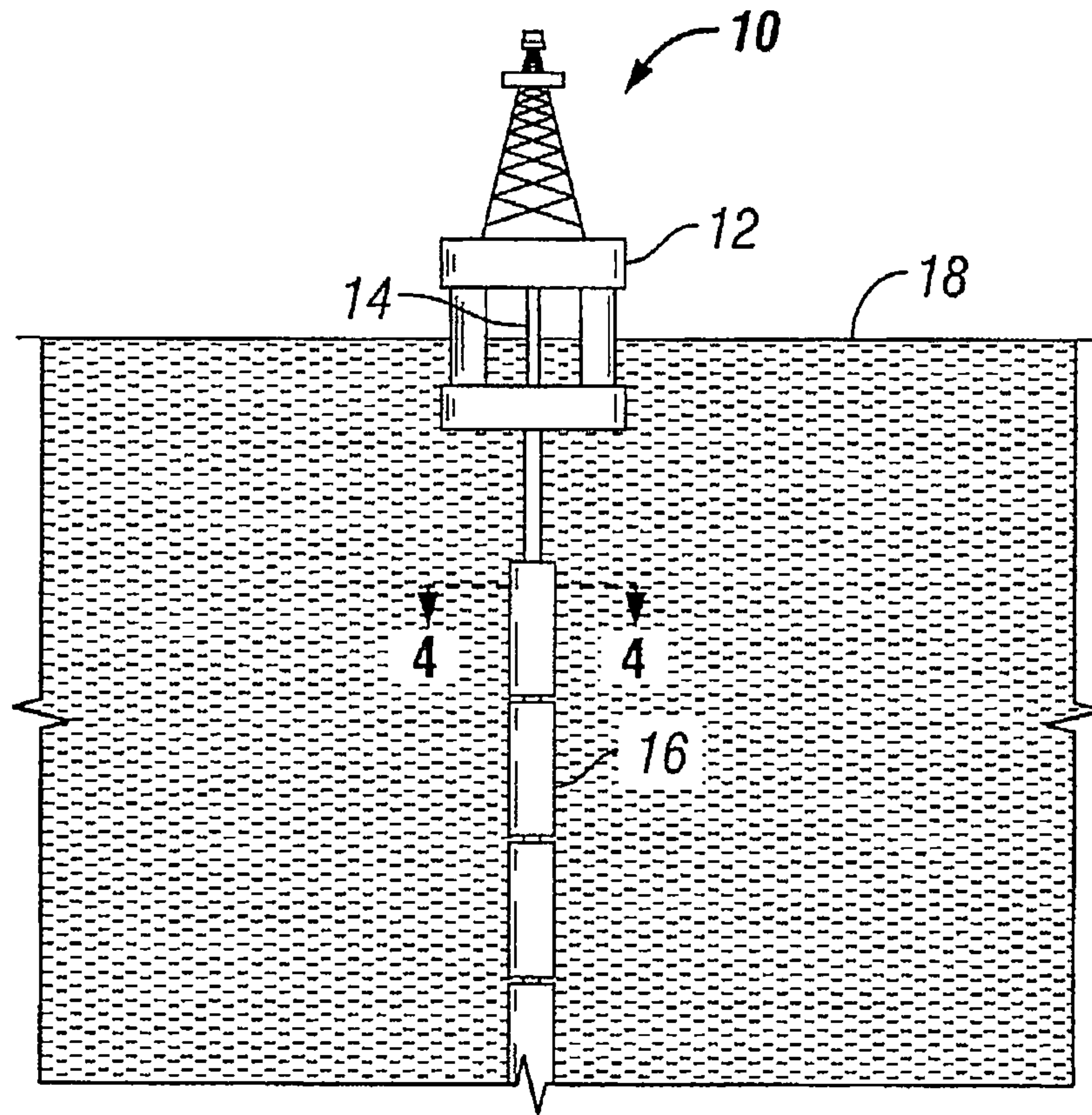


FIG. 1

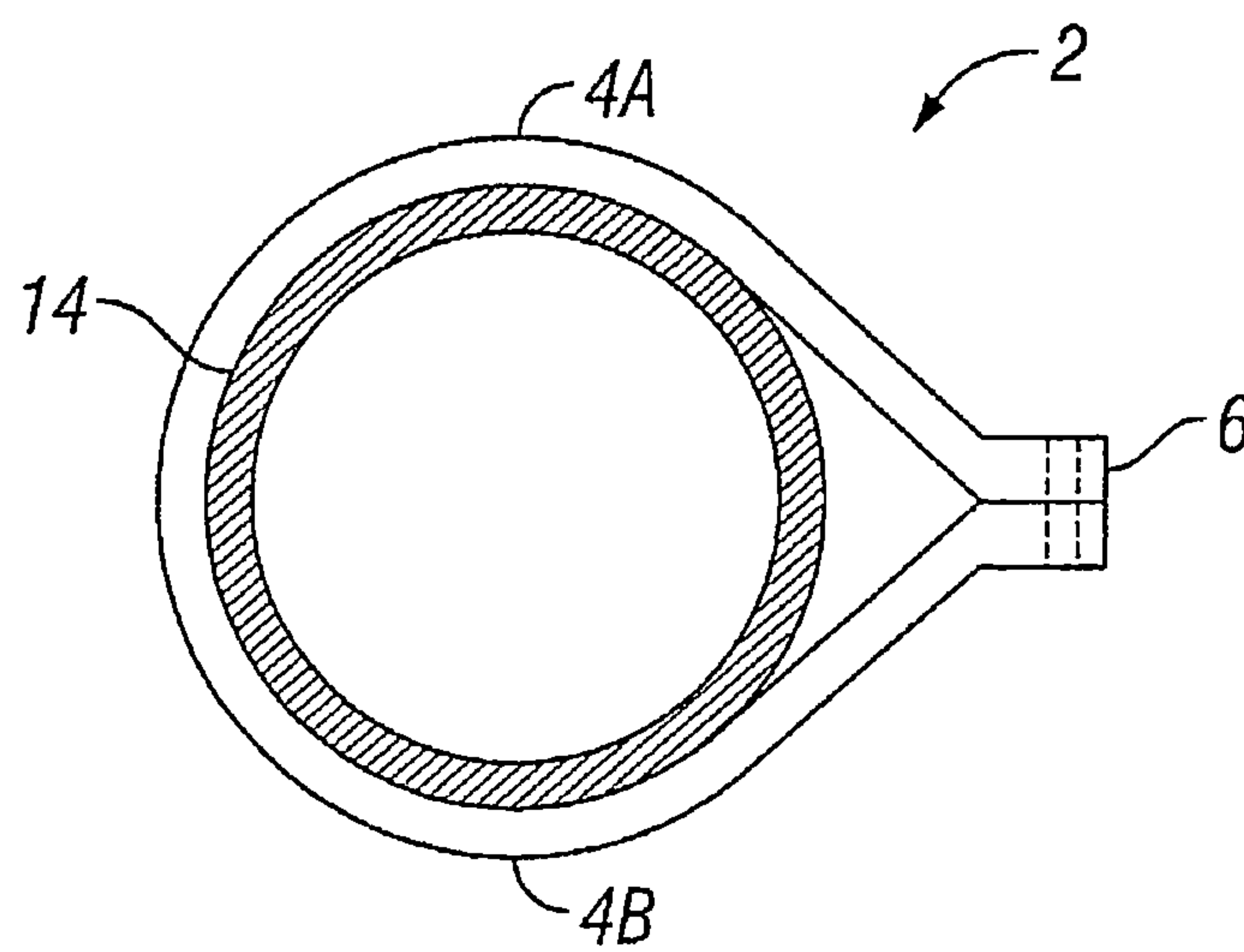


FIG. 2
(Prior Art)

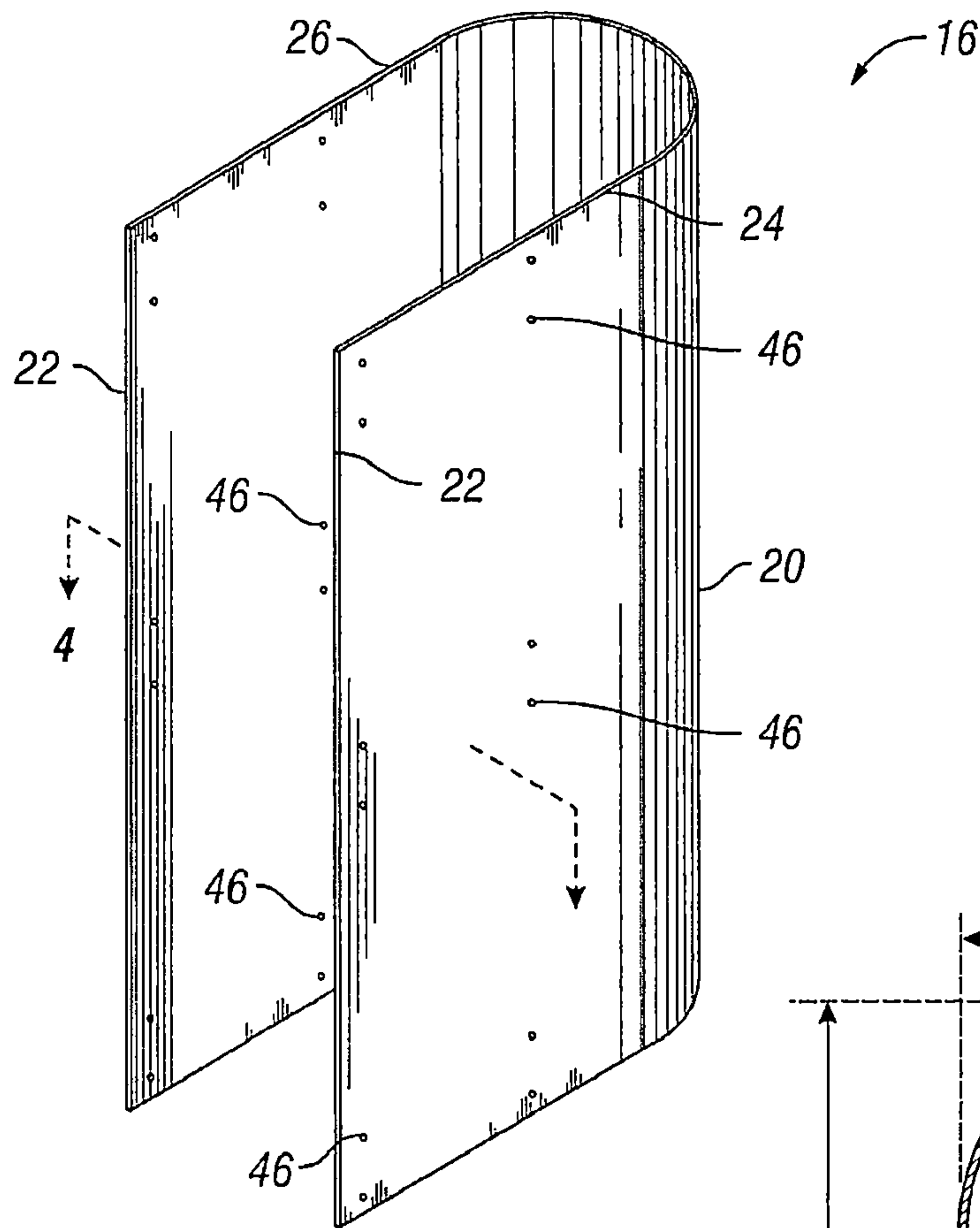


FIG. 3

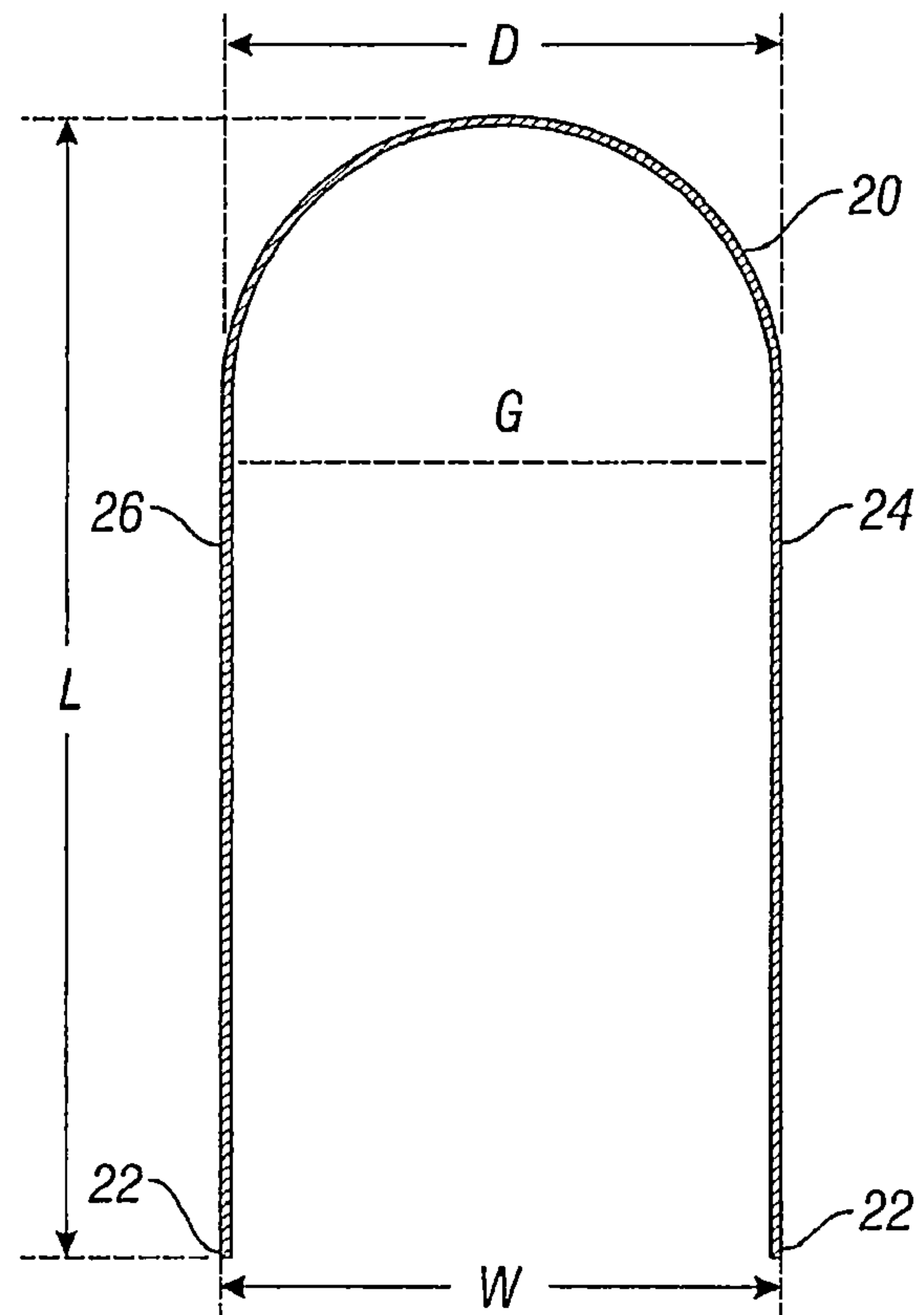


FIG. 4

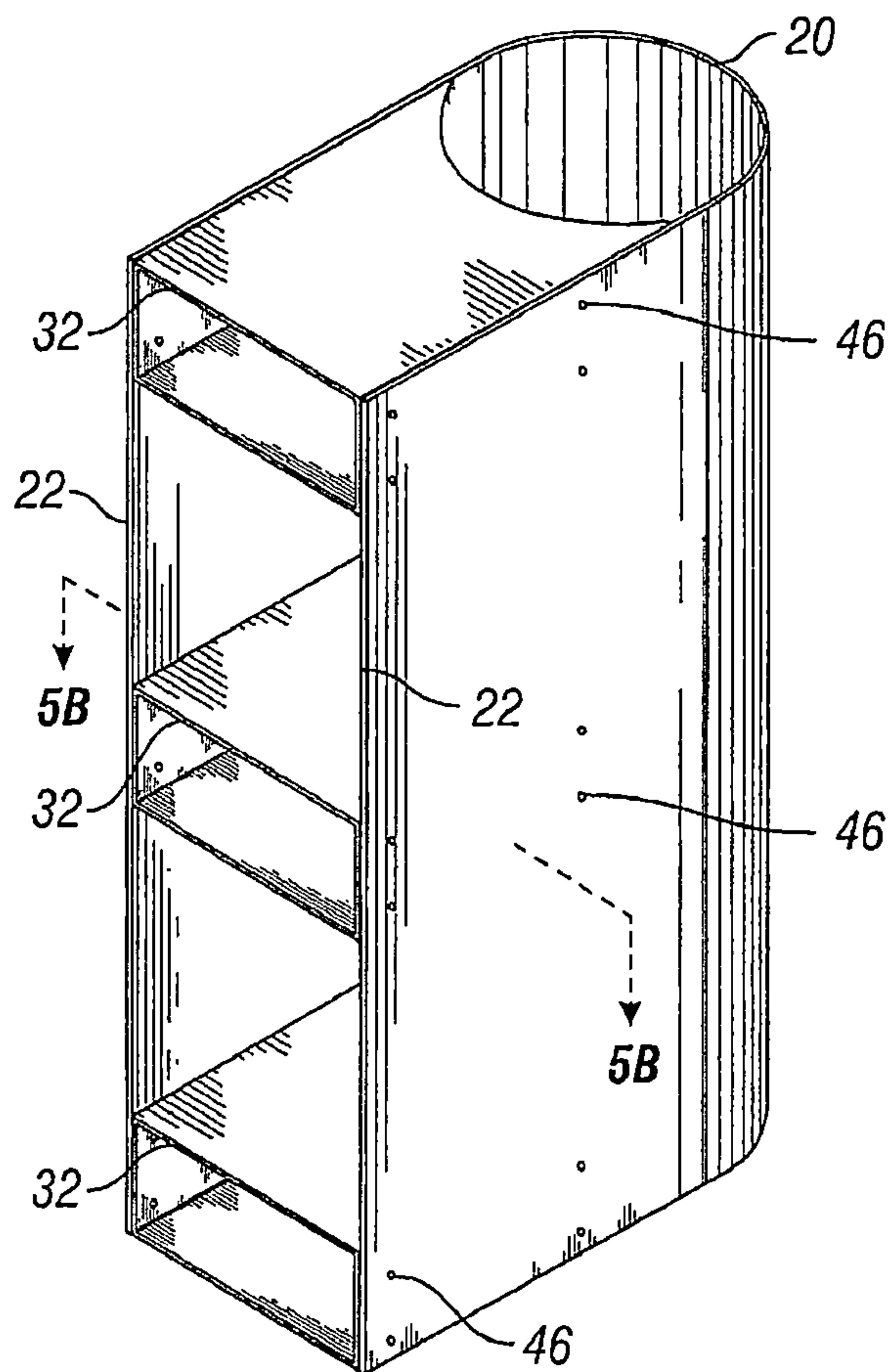


FIG. 5A

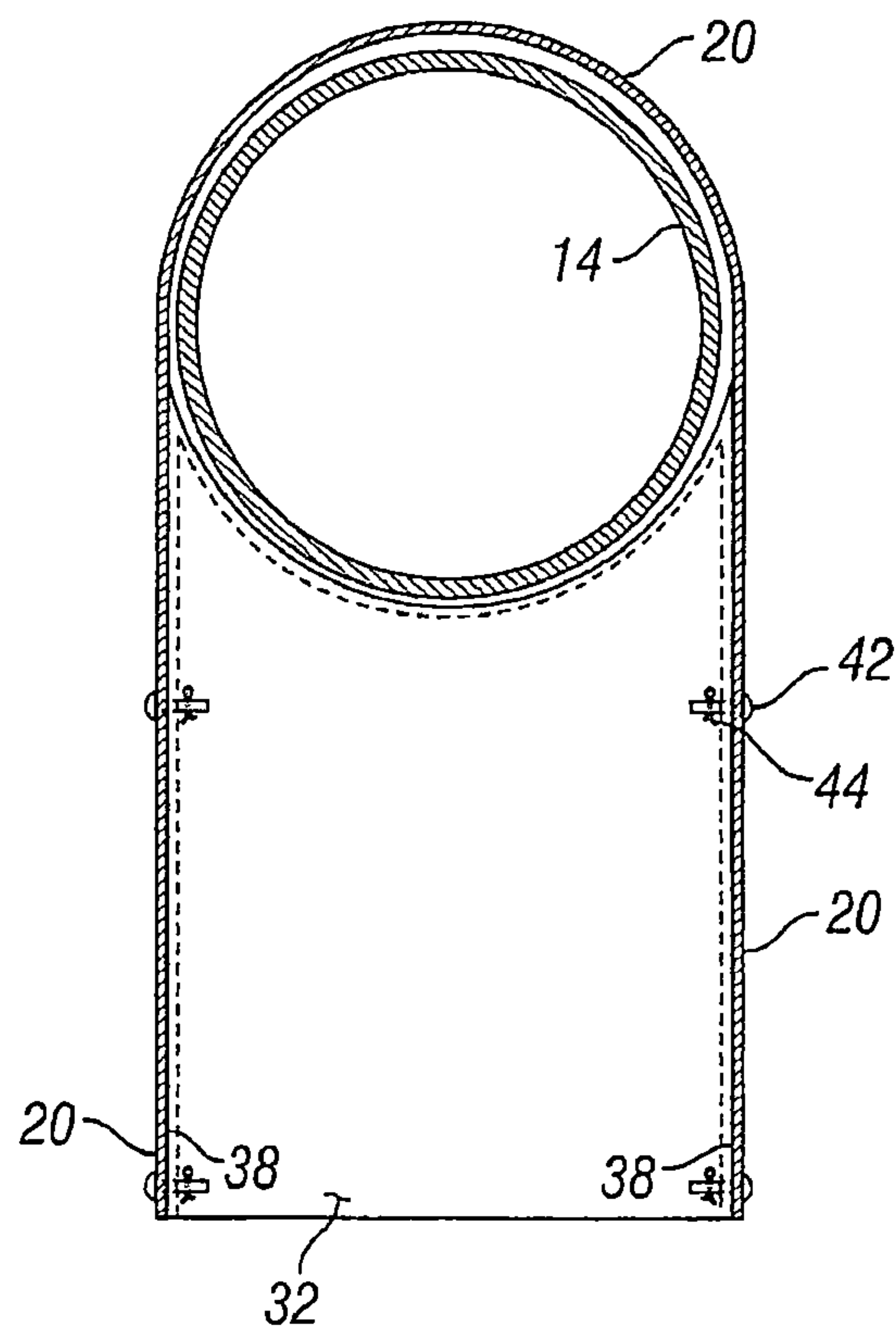


FIG. 5B

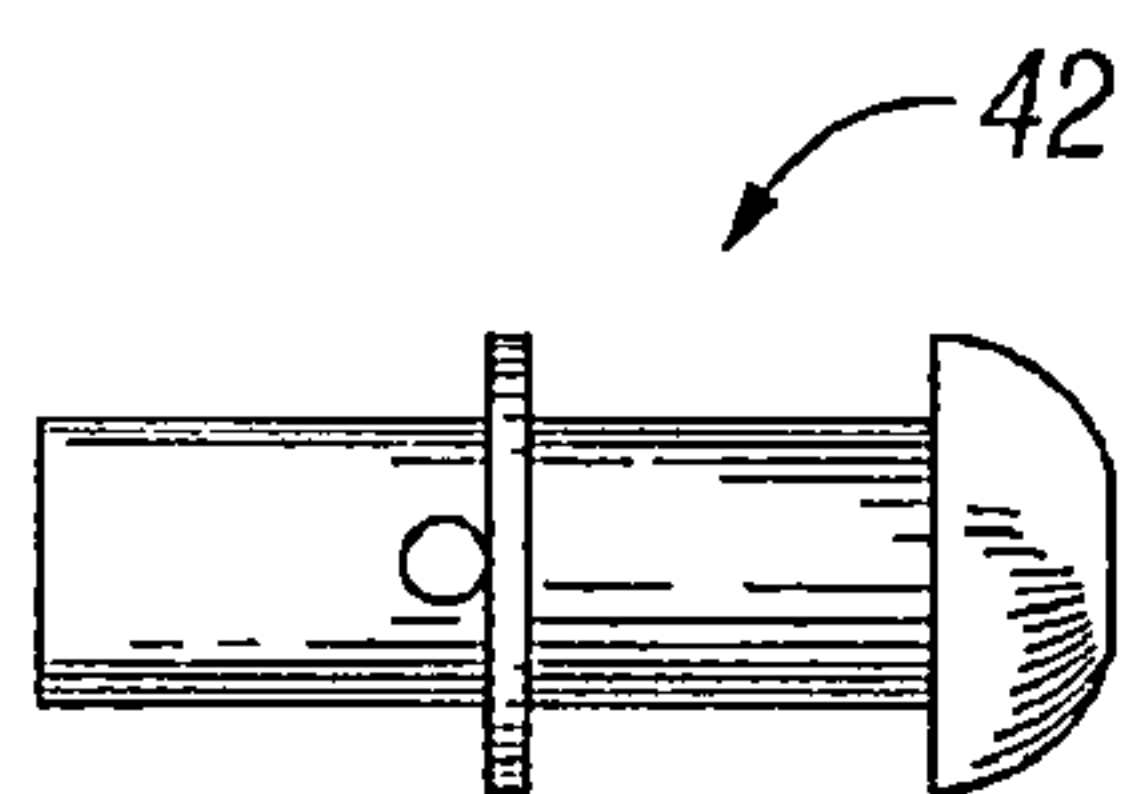


FIG. 6A

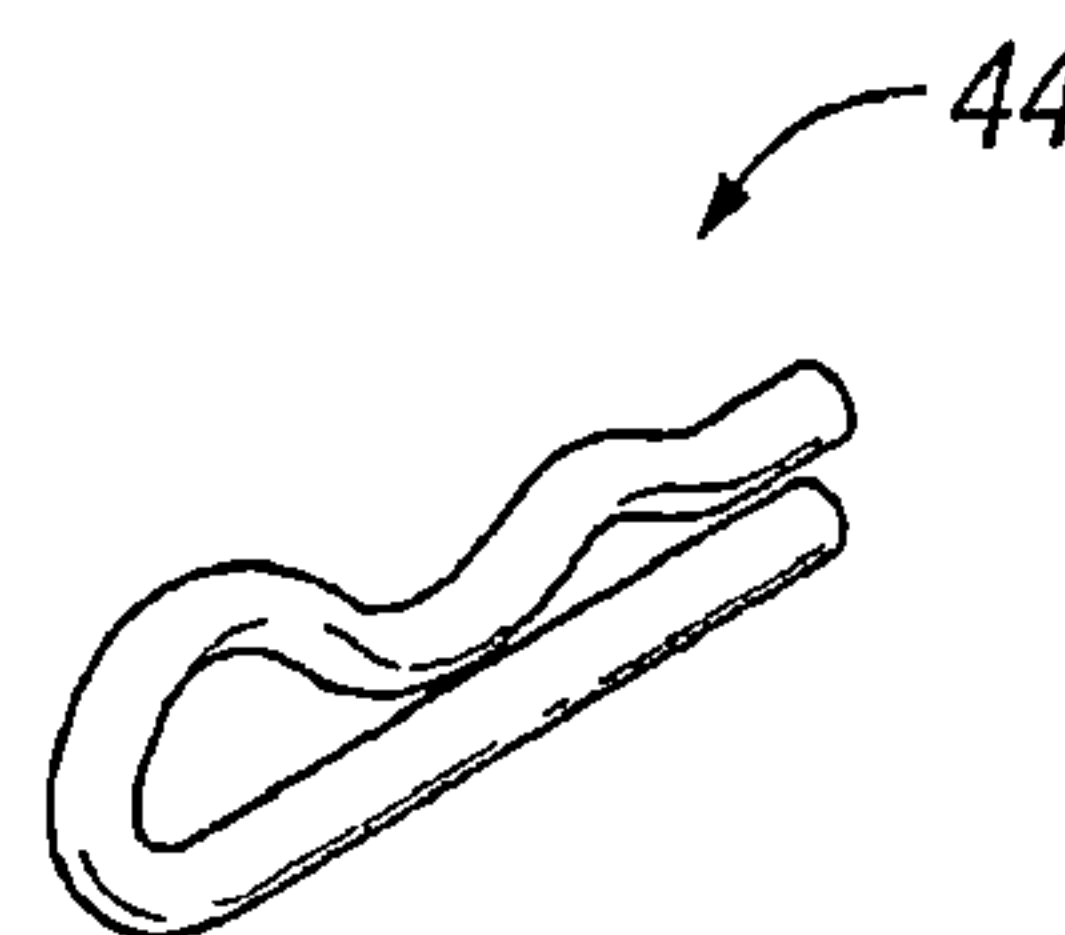


FIG. 6B

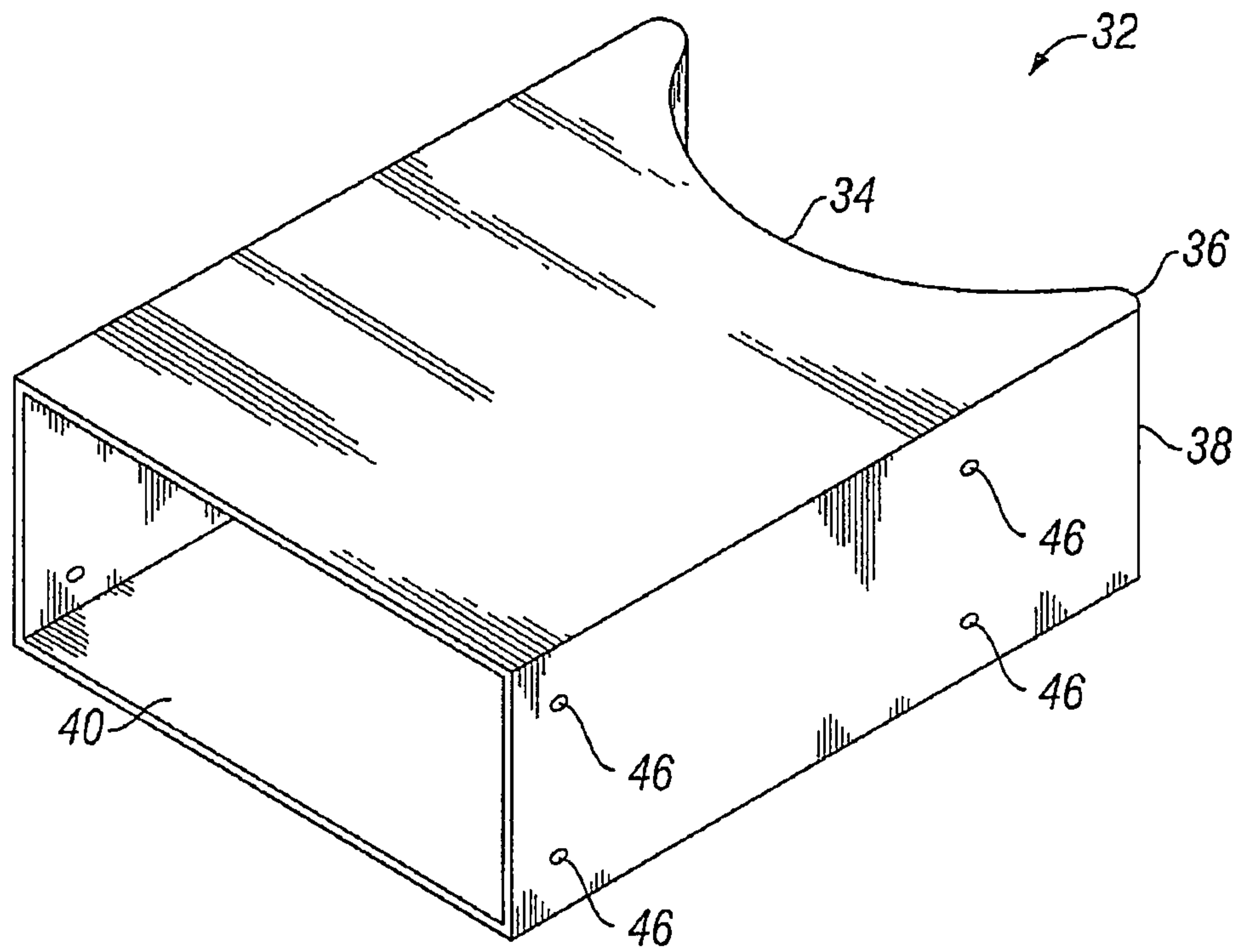


FIG. 7

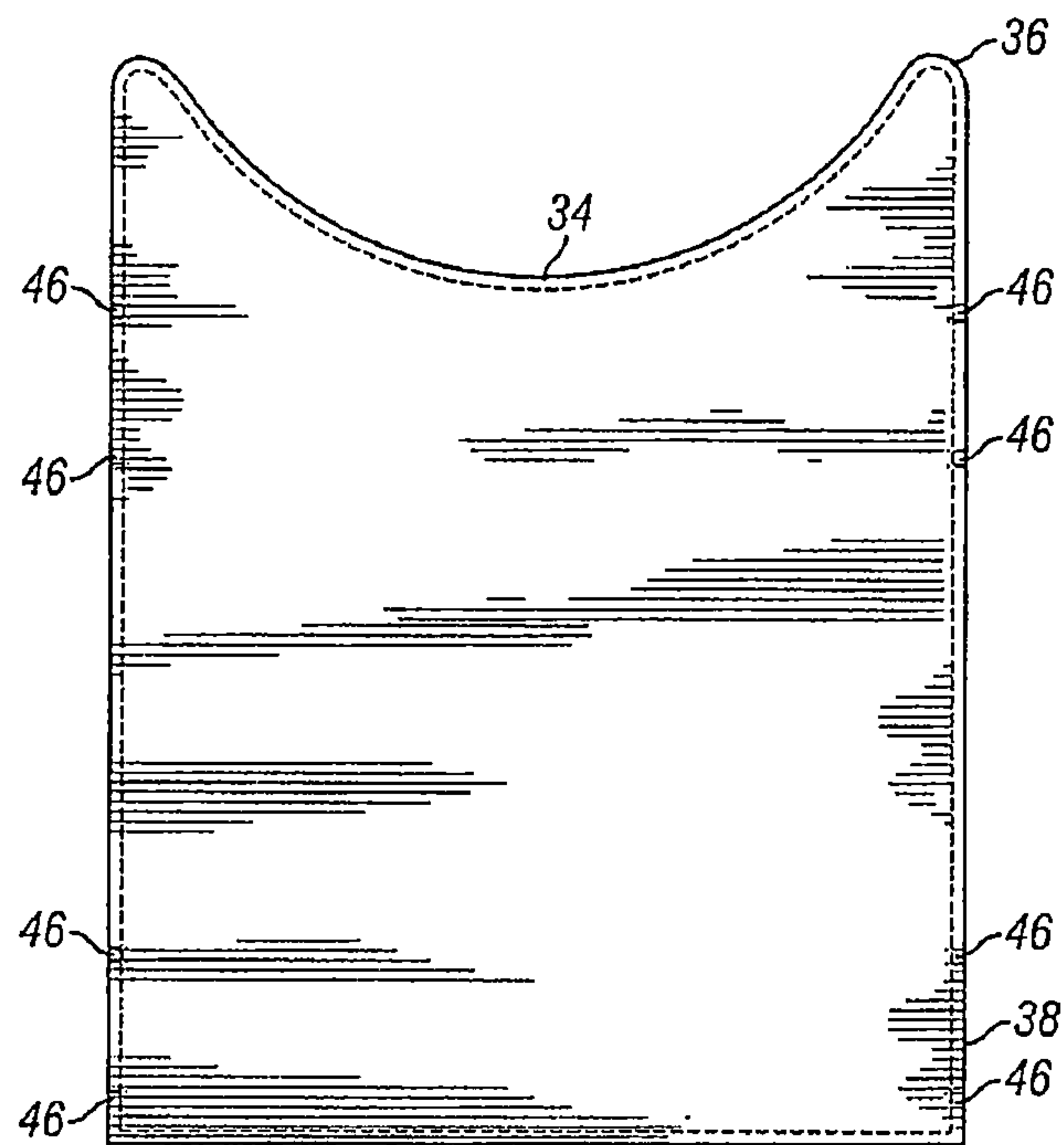


FIG. 8

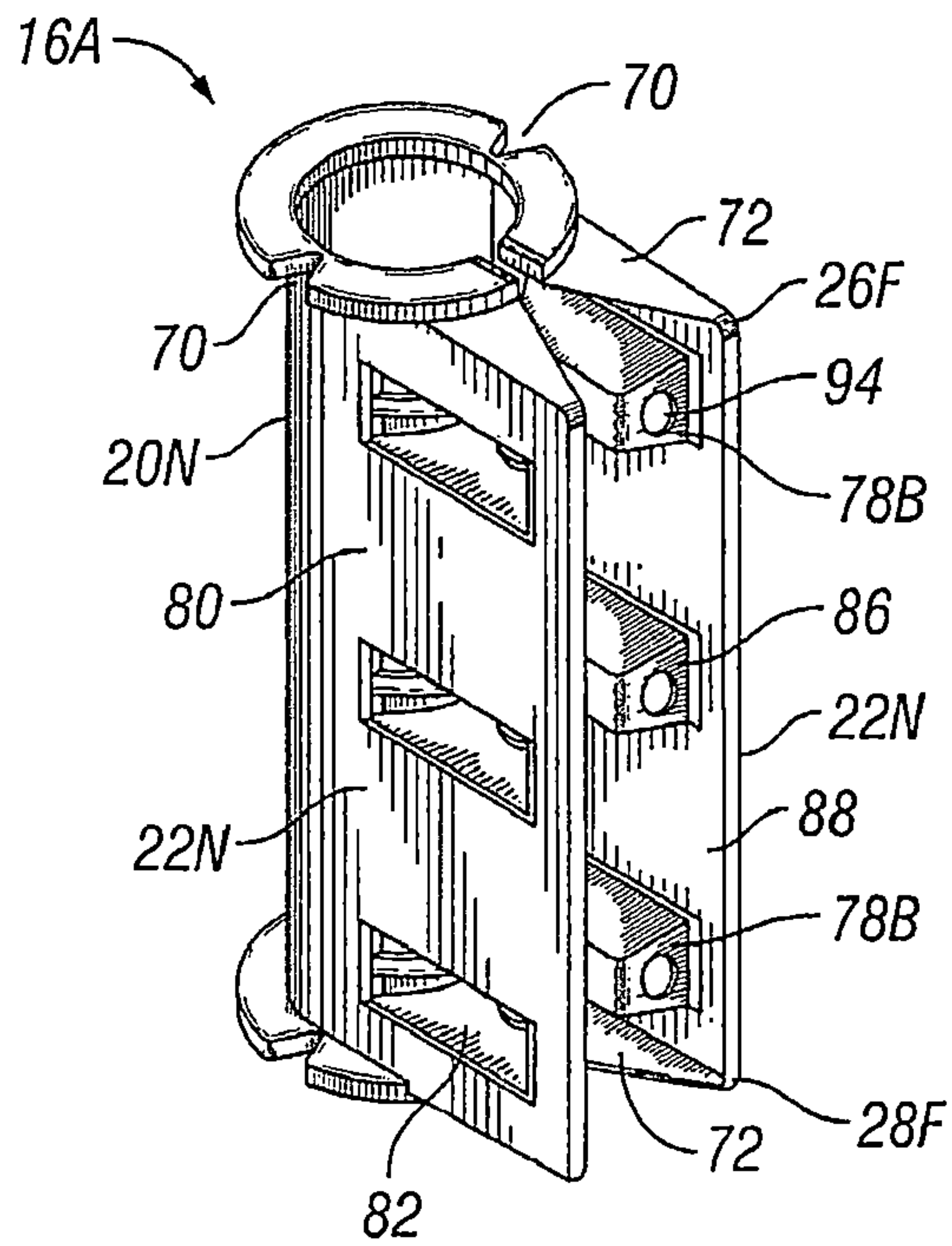


FIG. 9

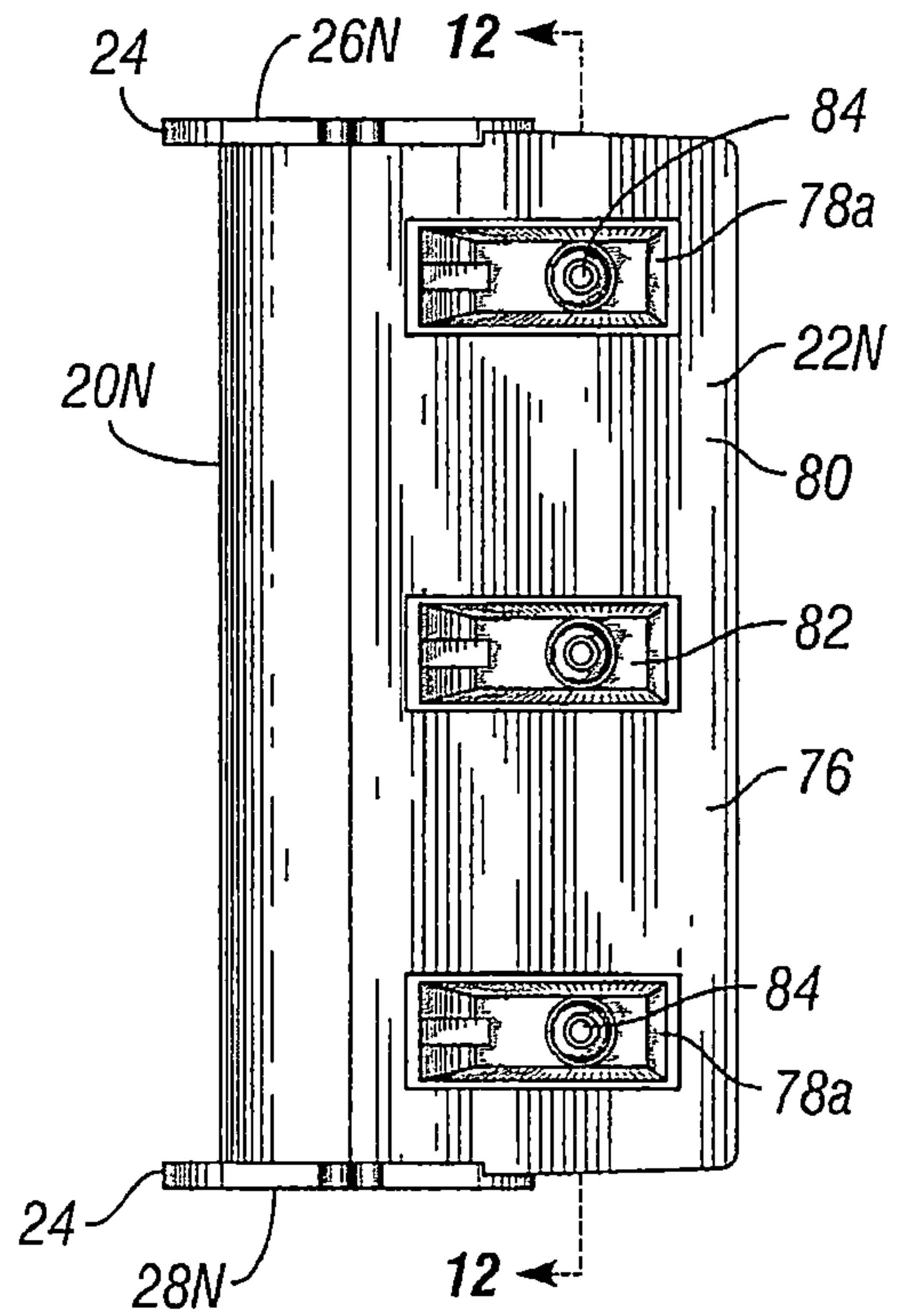


FIG. 10

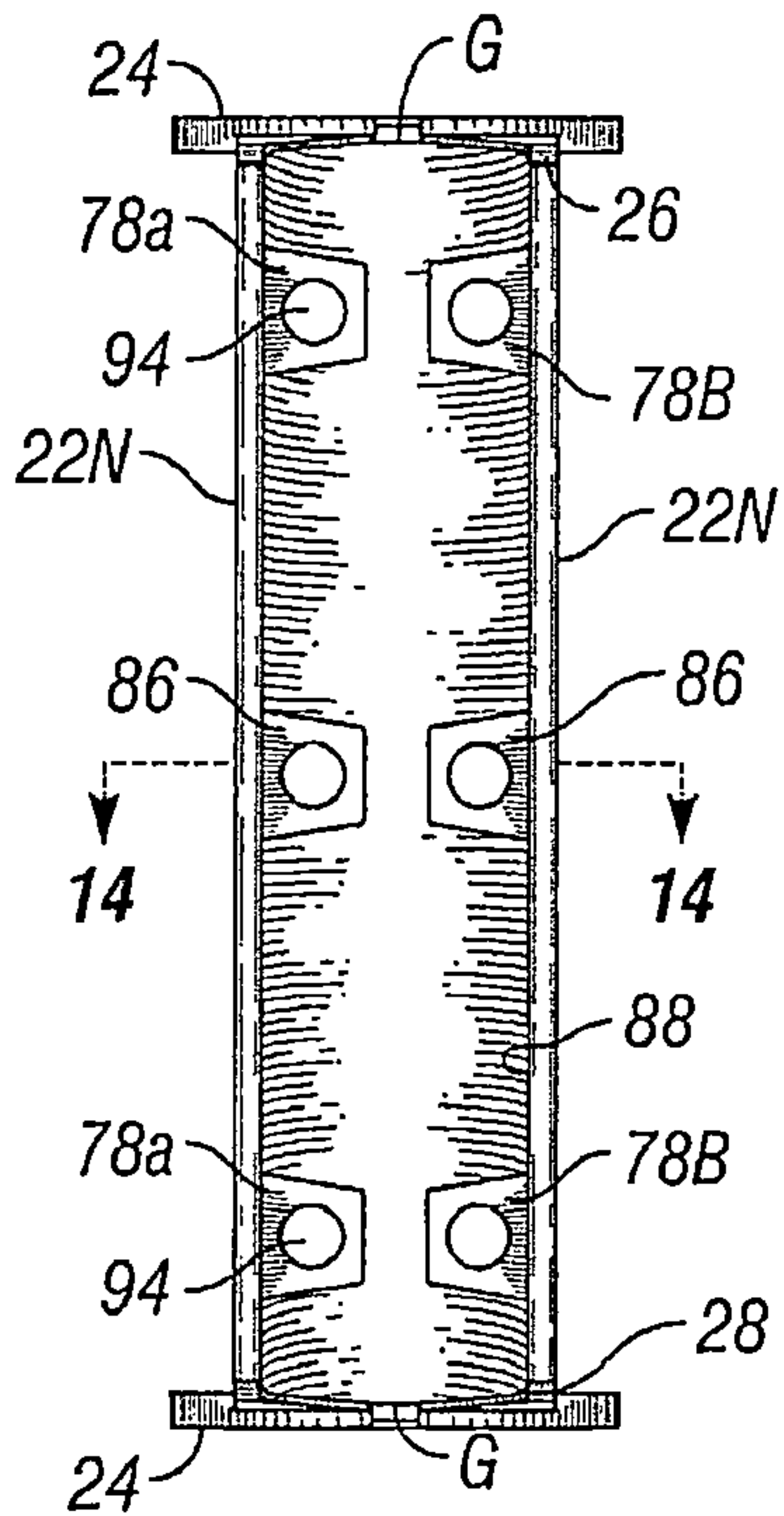


FIG. 11

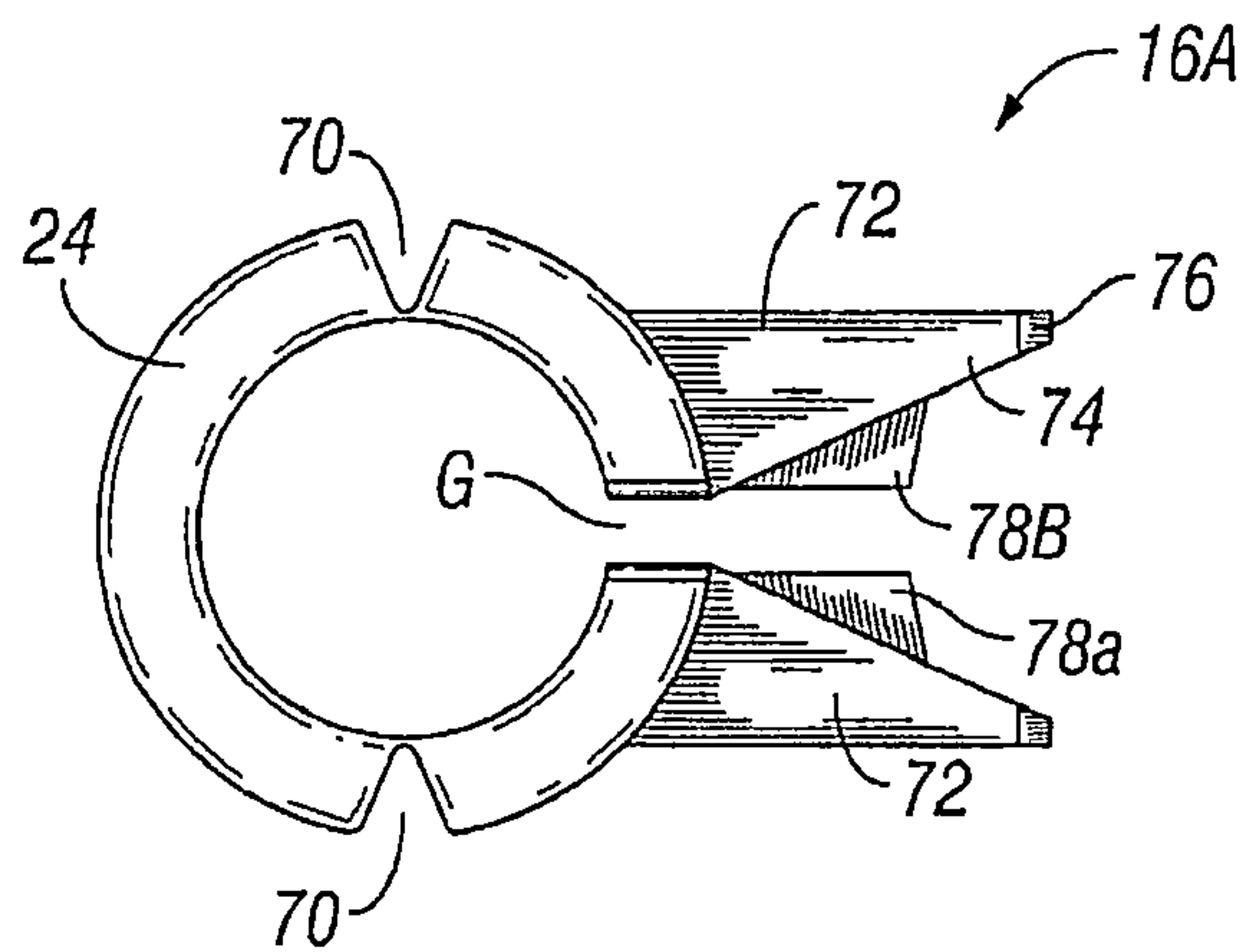


FIG. 13

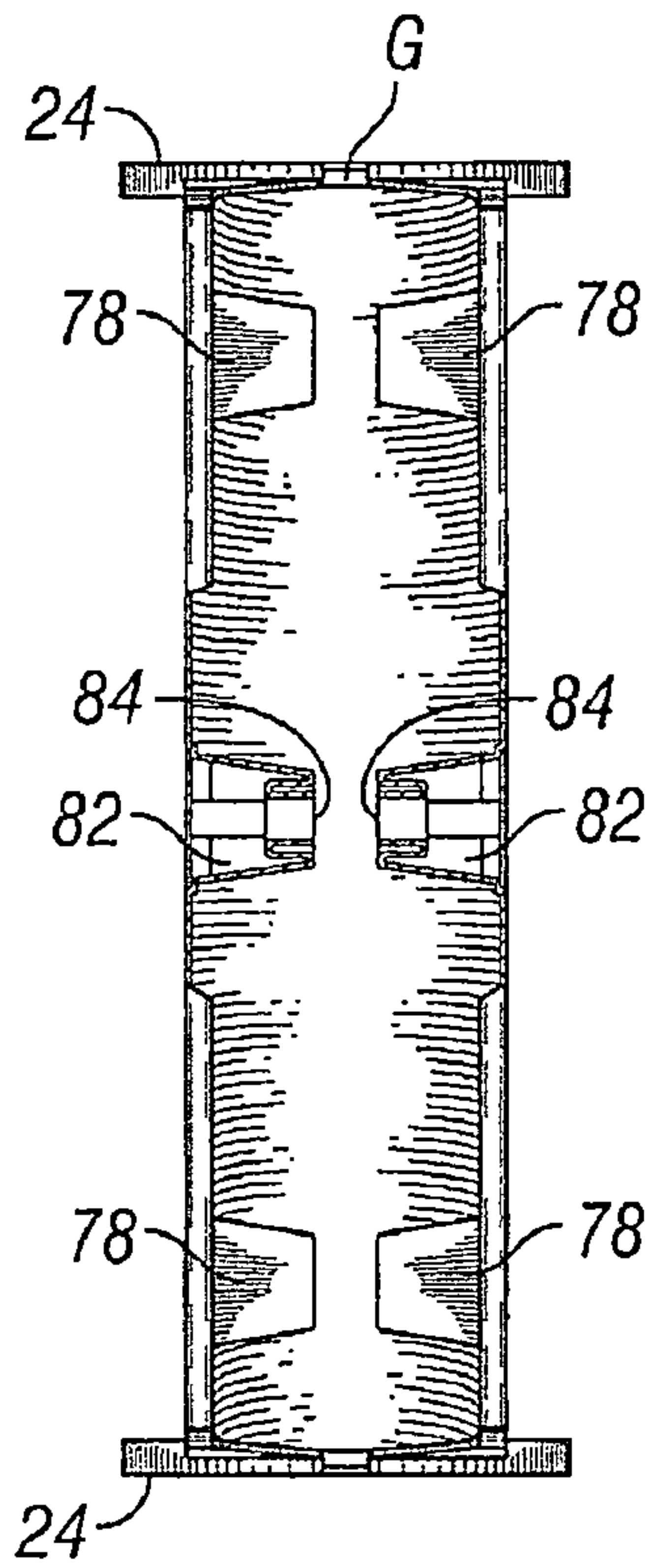


FIG. 12

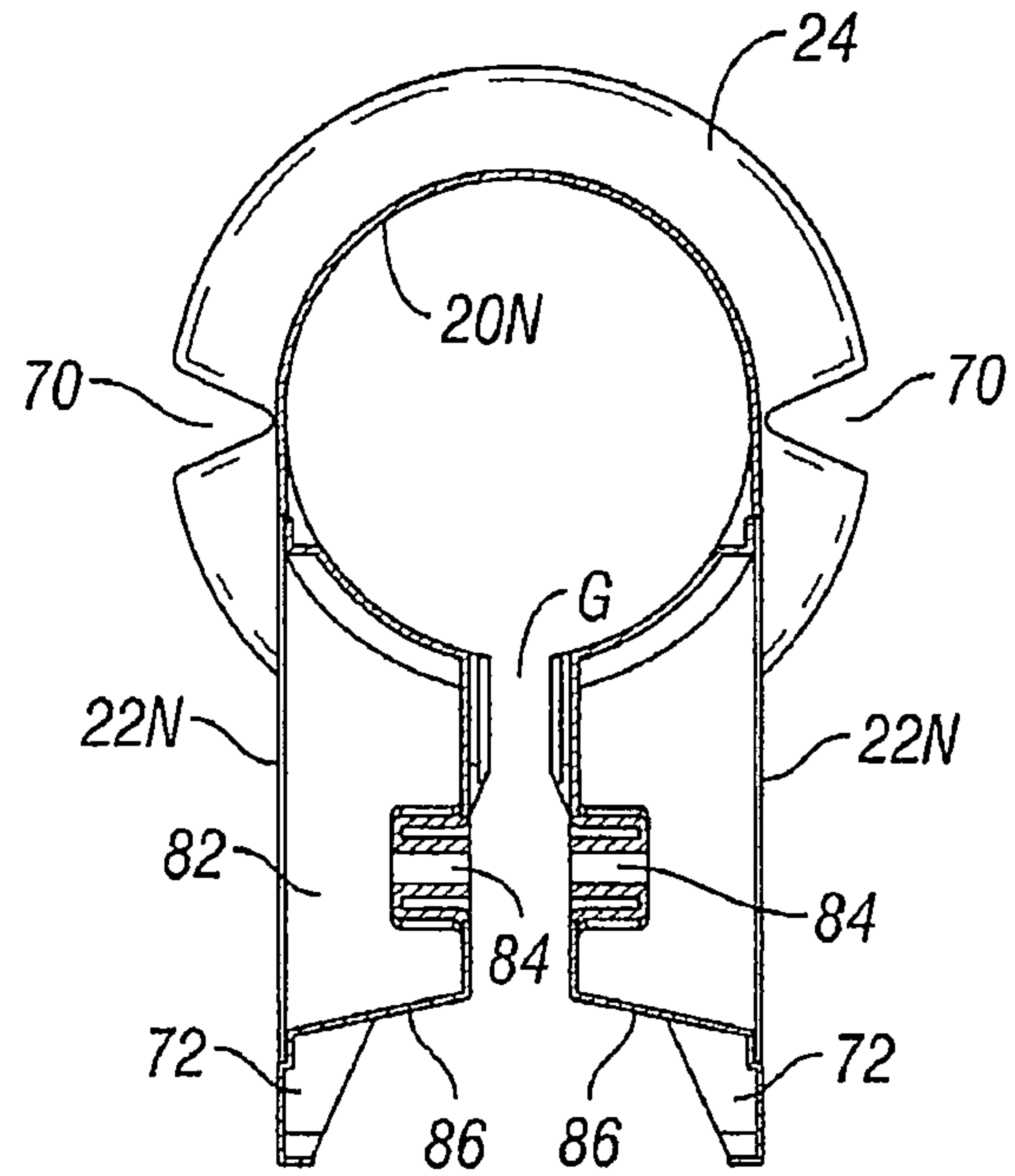


FIG. 14

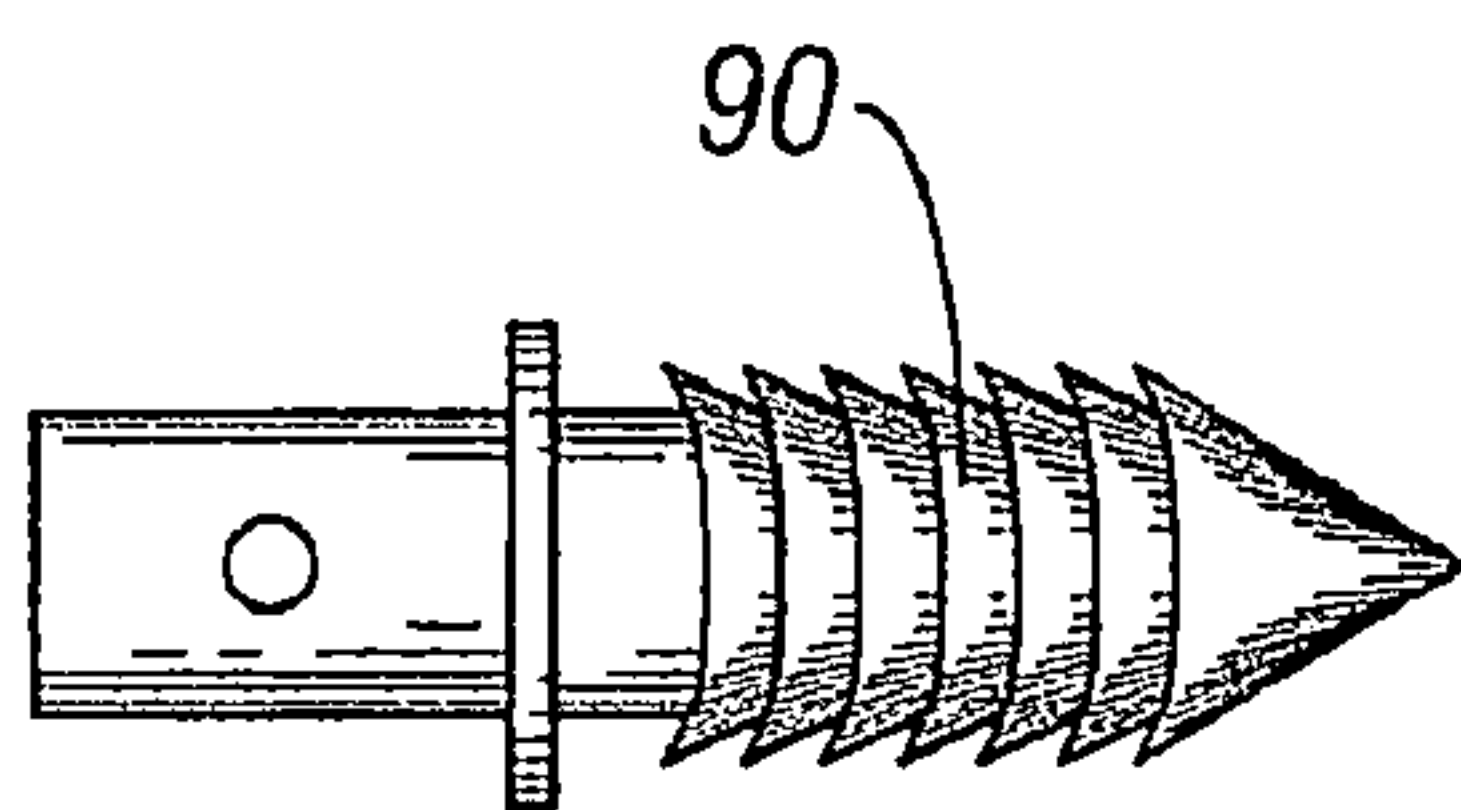


FIG. 15A

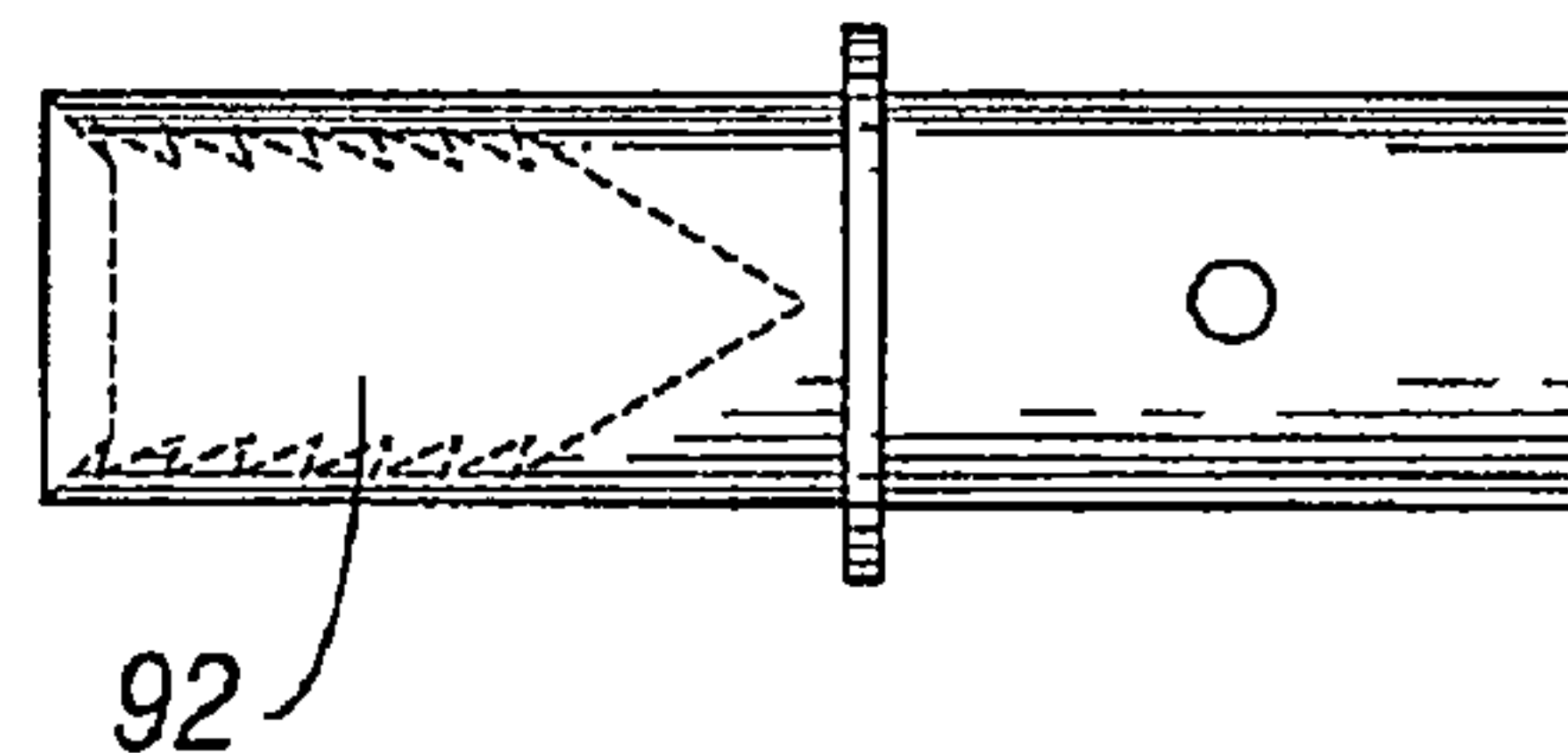


FIG. 15B

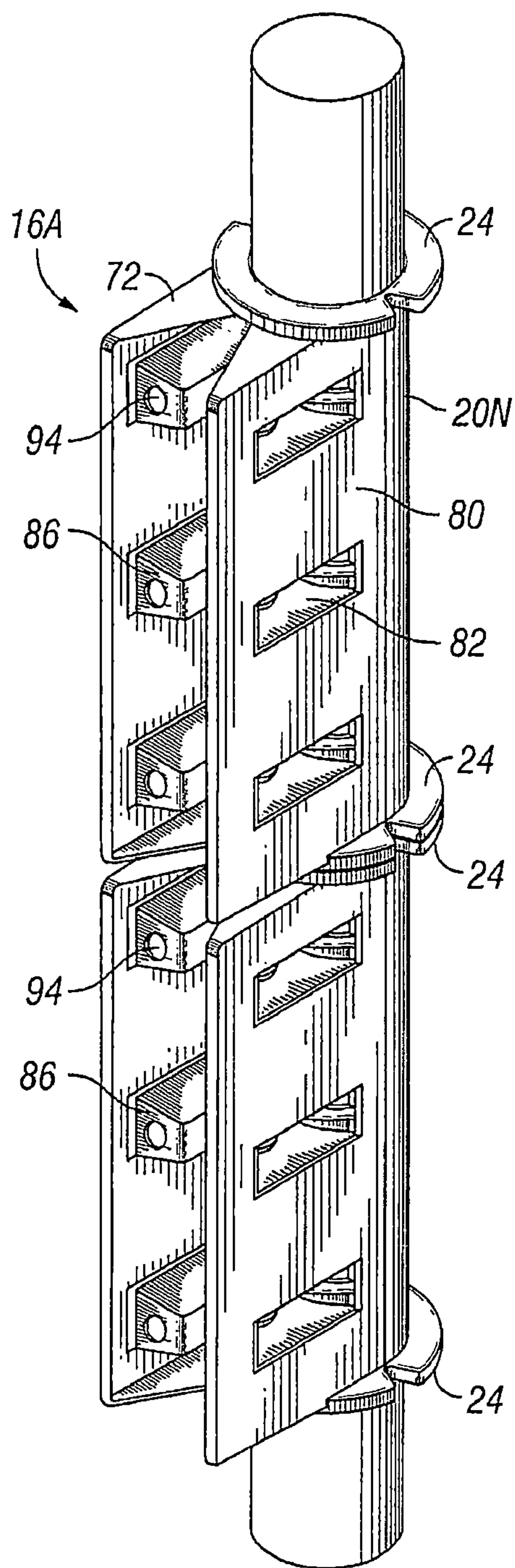


FIG. 16

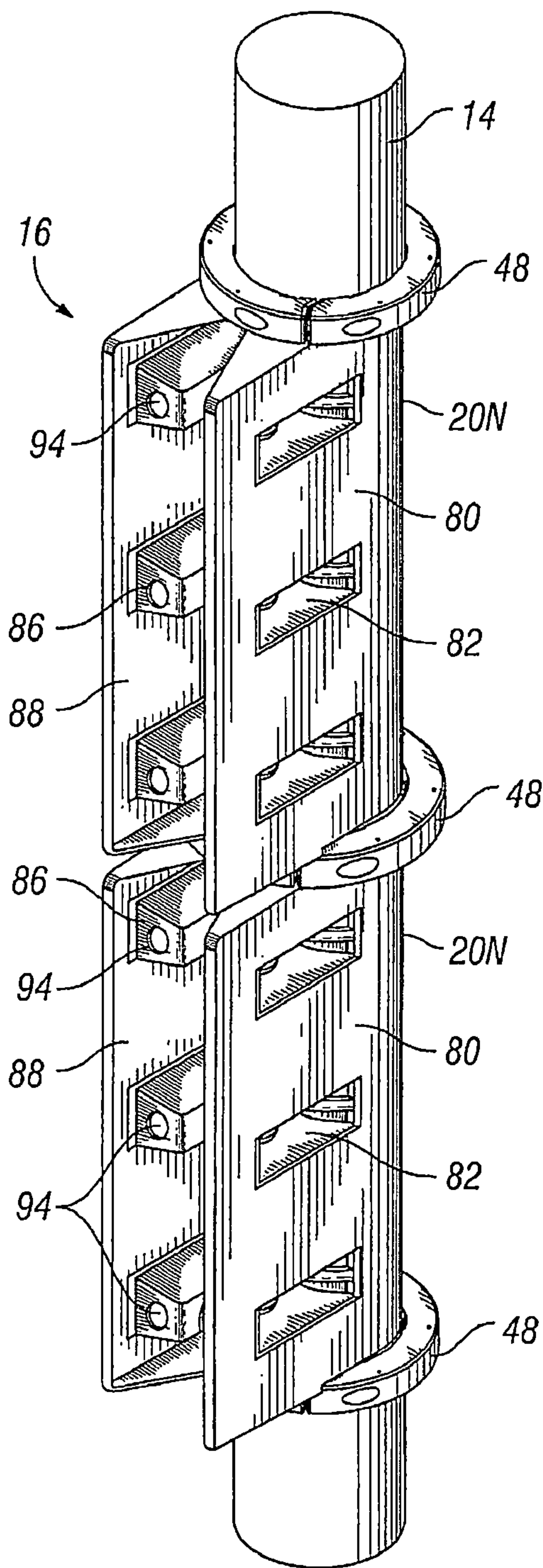


FIG. 17

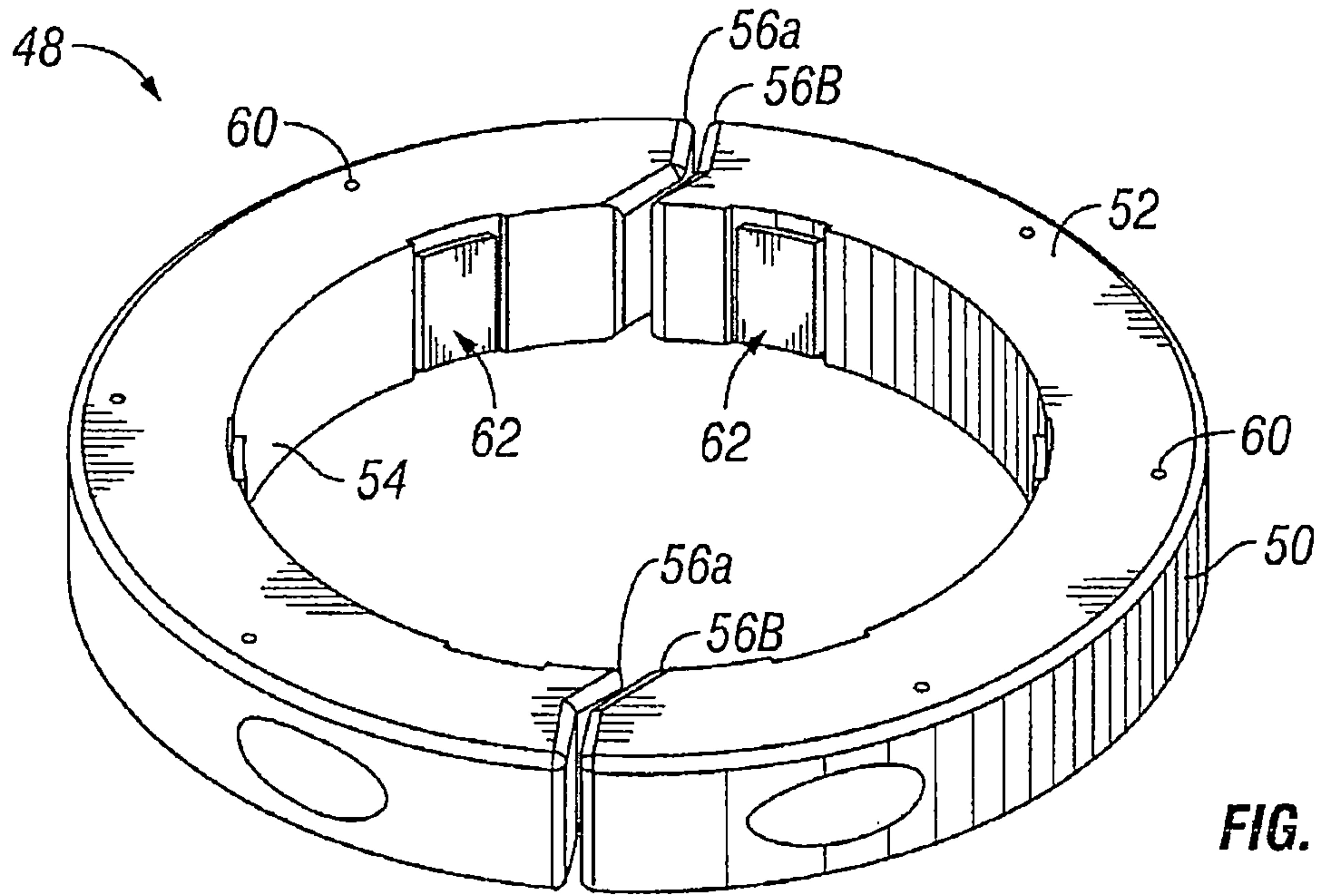


FIG. 18

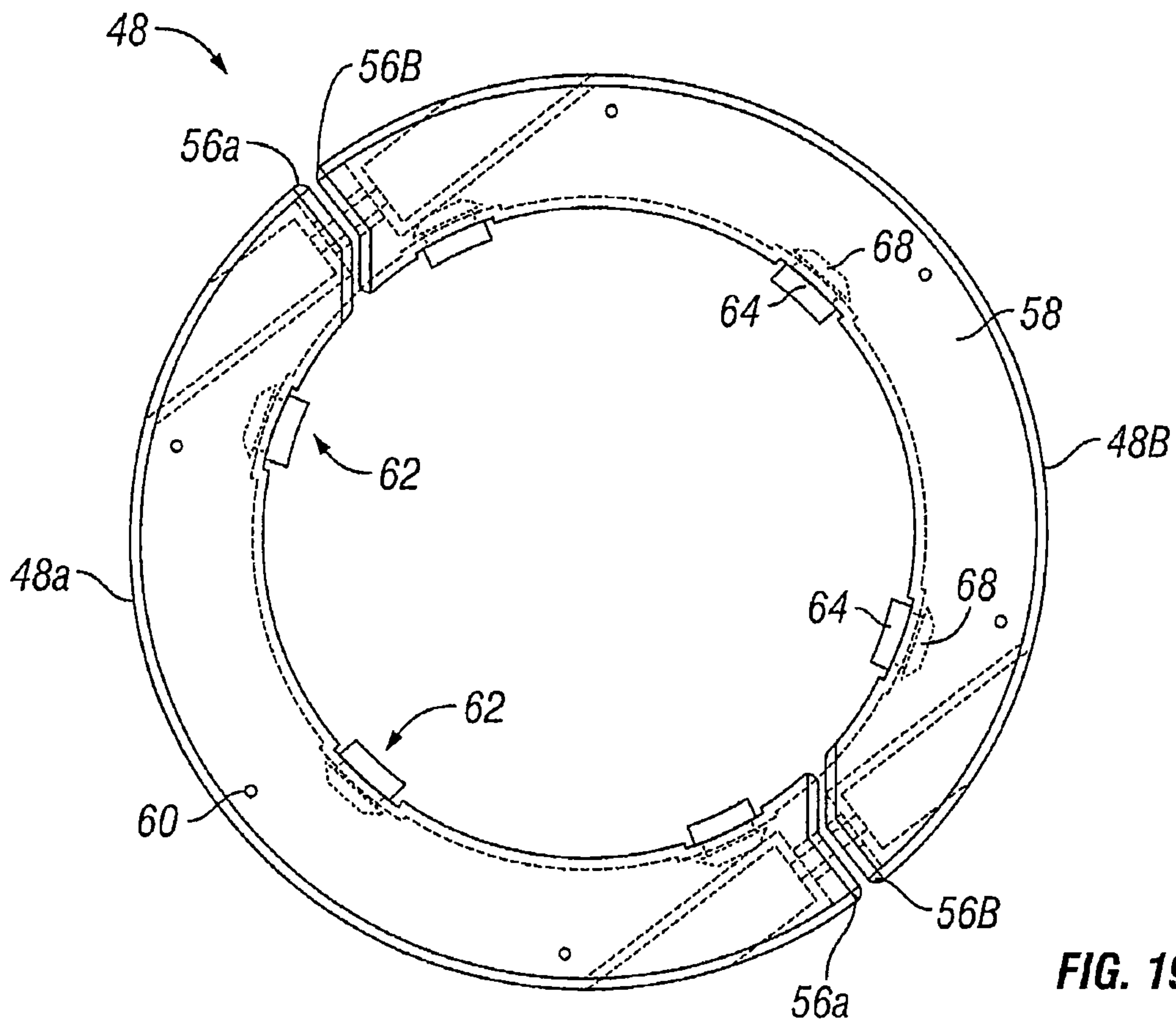


FIG. 19

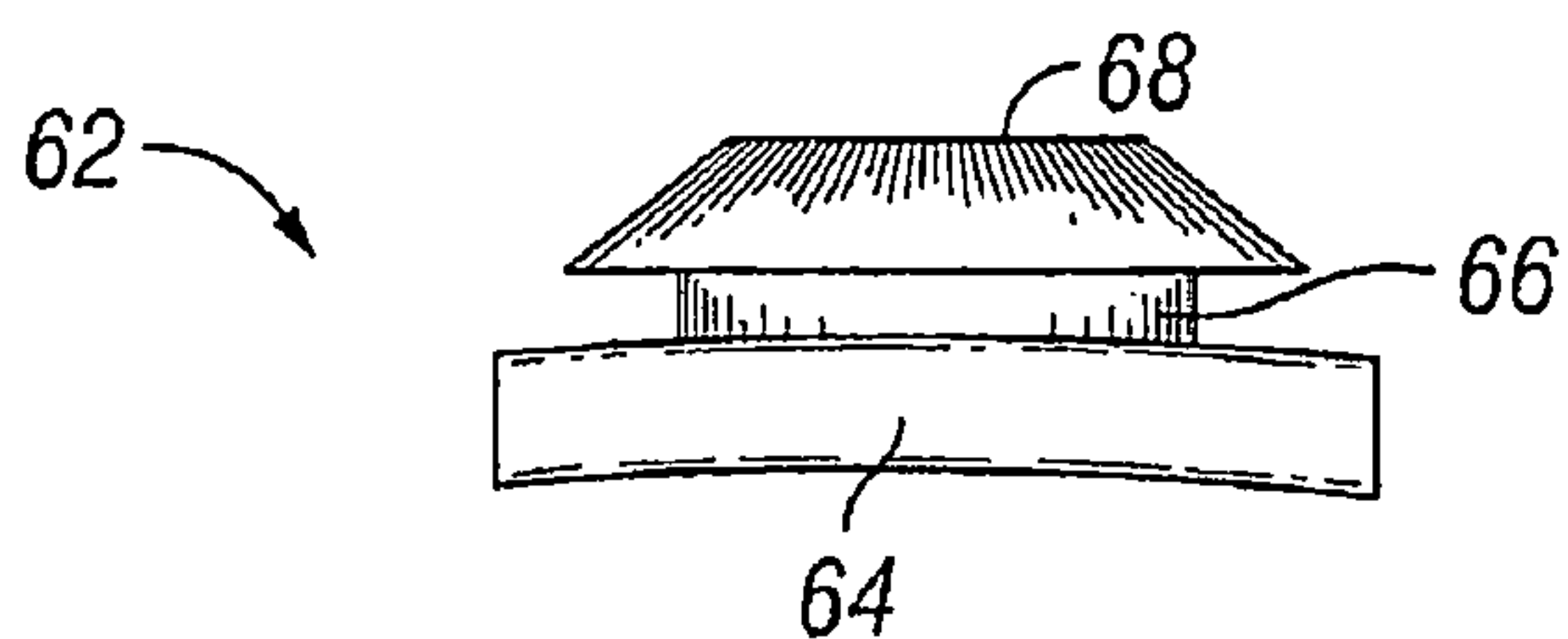


FIG. 20

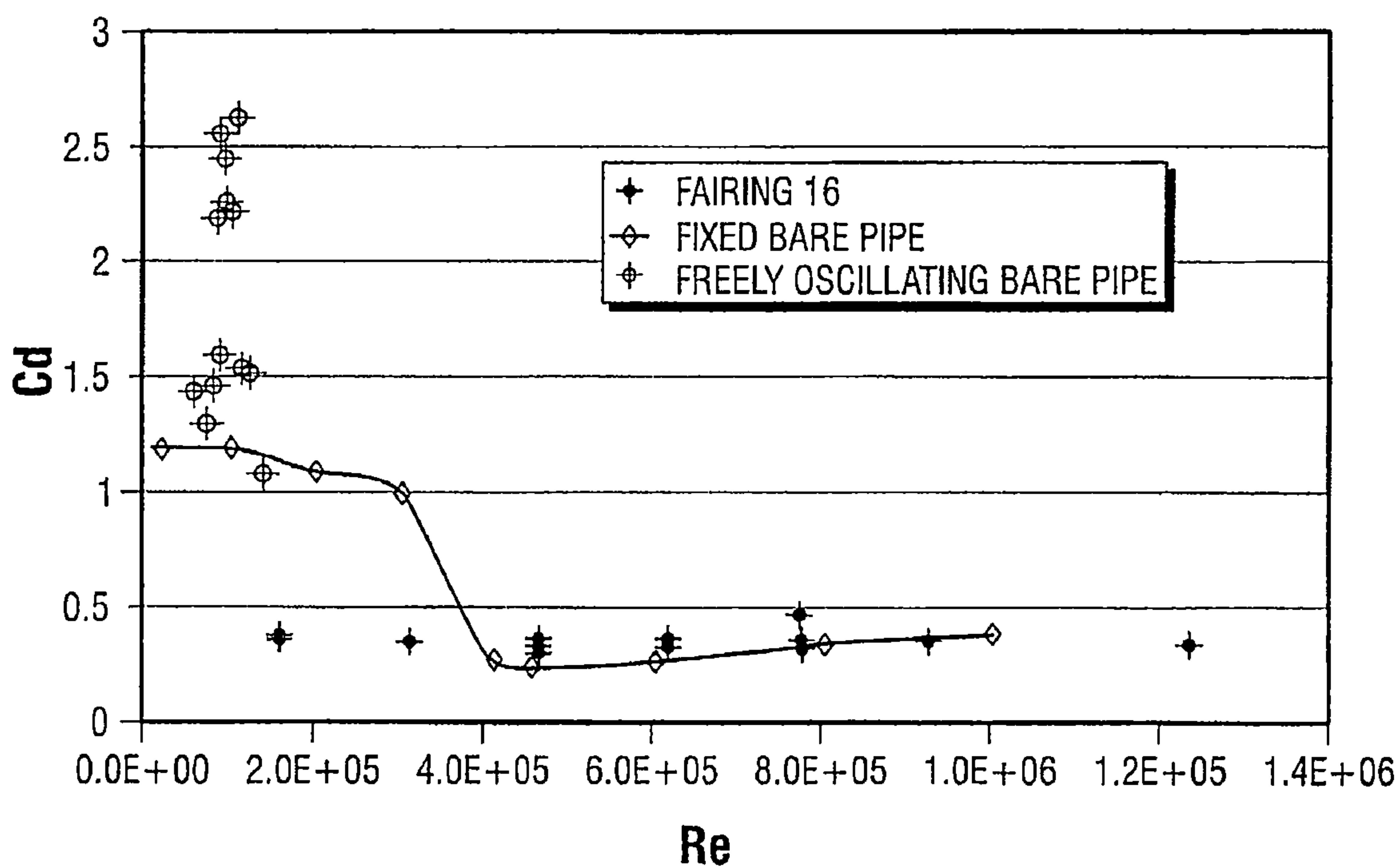


FIG. 21

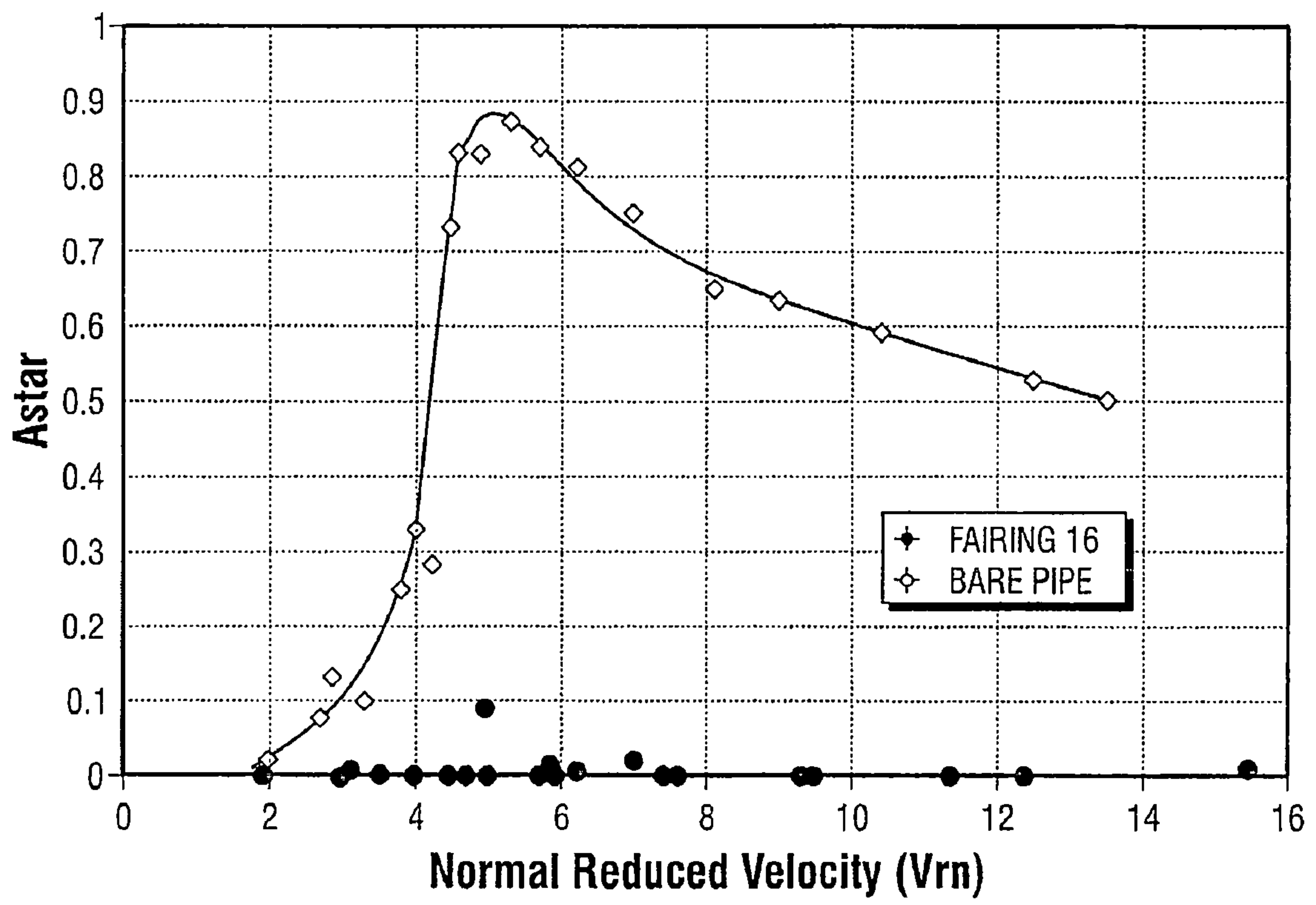


FIG. 22

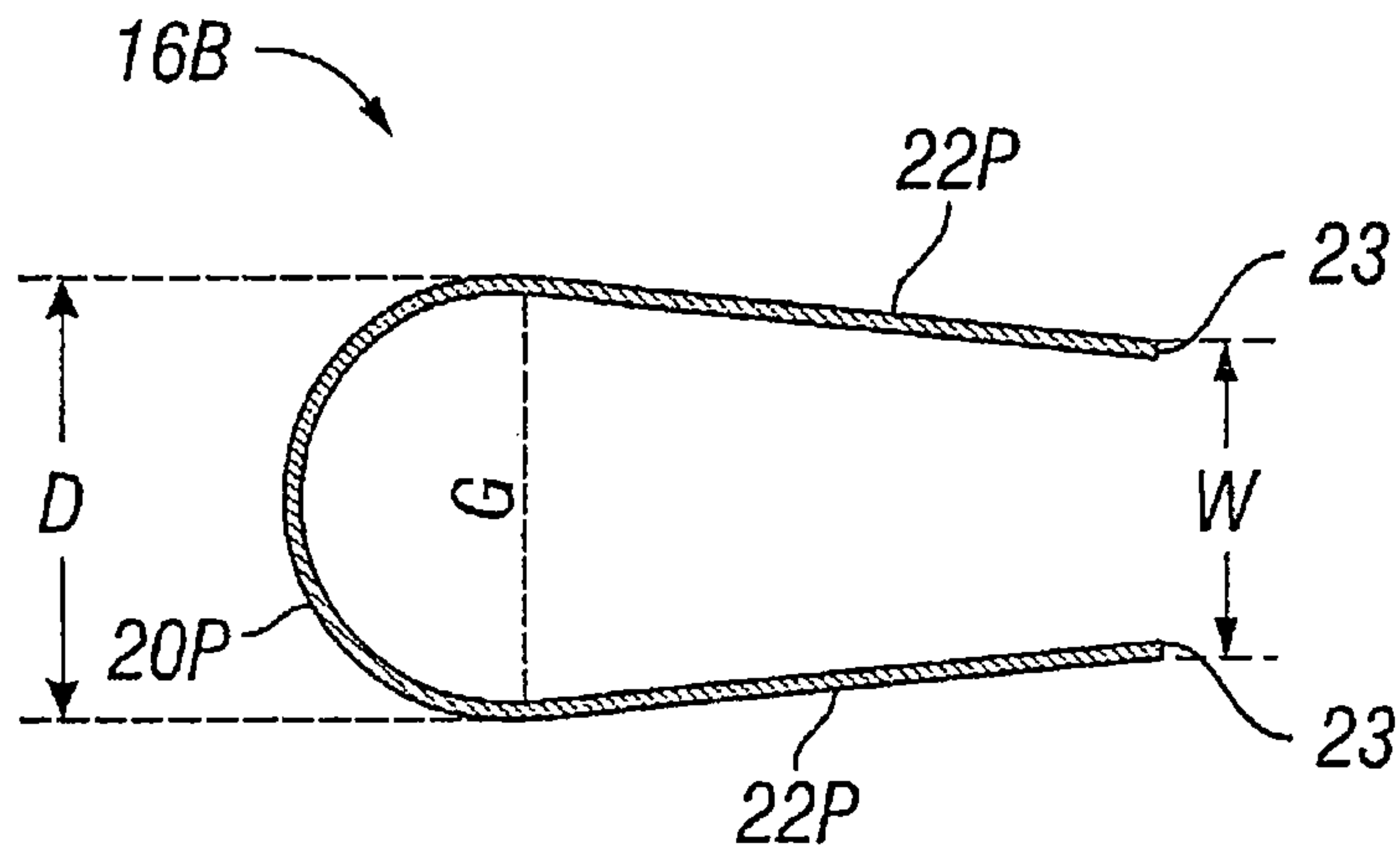


FIG. 23

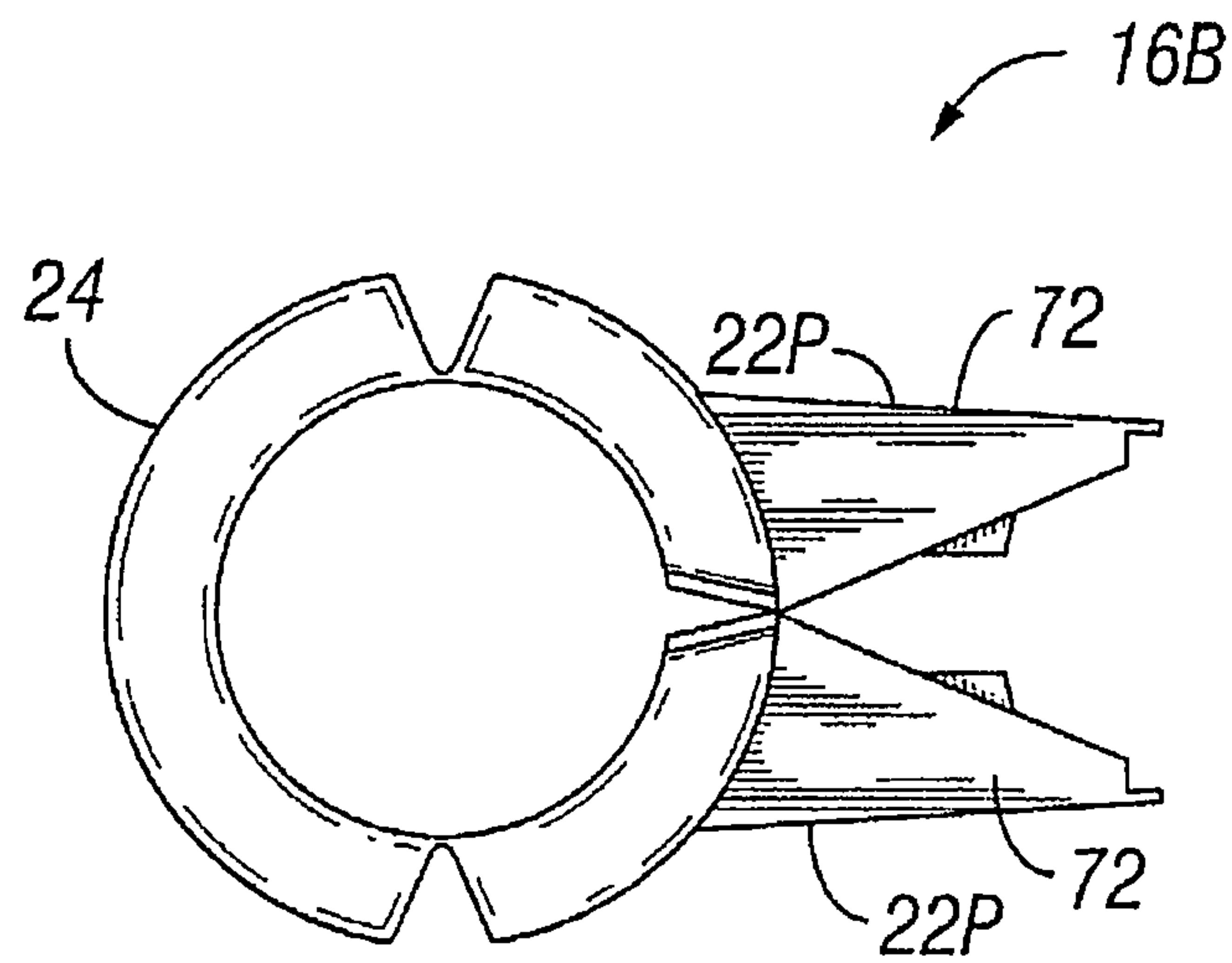


FIG. 24

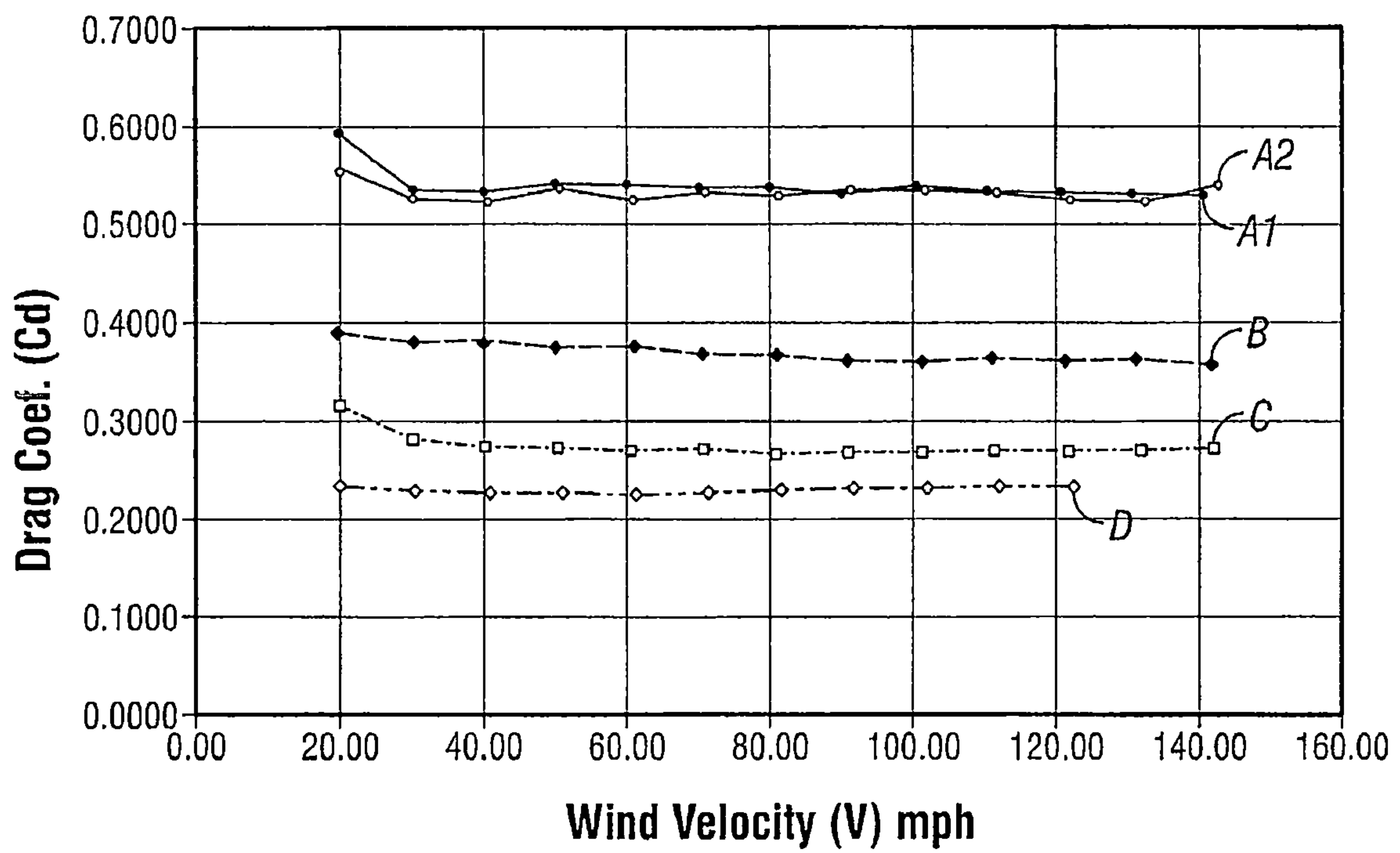


FIG. 25

TWIN FIN FAIRING

Matter enclosed in heavy brackets [] appears in the original patent but forms no part of this reissue specification; matter printed in italics indicates the additions made by reissue; a claim printed with strikethrough indicates that the claim was canceled, disclaimed, or held invalid by a prior post-patent action or proceeding.

CROSS REFERENCE TO RELATED APPLICATIONS

This application is a continuation-in-part of U.S. Ser. No. 11/463,413 filed Aug. 9, 2006, now U.S. Pat. No. 7,337,742 issued Mar. 4, 2008 which is herein incorporated in its entirety.

TECHNICAL FIELD

The present invention relates generally to the reduction of vortex-induced vibration (“VIV”) and more particularly to a fairing used for the reduction of VIV on pipes or other structural components immersed in a fluid.

BACKGROUND OF THE INVENTION

Exploration for oil and natural gas reserves led drillers offshore many years ago and as offshore exploration continues, offshore producers find themselves in deeper and deeper waters. While those waters may bring the reserves they seek, the offshore producers are also faced with stronger currents threatening the structural integrity of their risers, pipelines, and other elongated and cylindrical components involved in oil and gas production.

Stresses on the pipes or other structural components immersed in fluid, such as drilling risers, production risers, pipelines, structural tendons, etc. greatly increase as the velocity of the current increases and the stresses are magnified as the depth of the water and length of the risers at the well location increases. When operating rigs in high current areas, the riser is exposed to currents that can cause at least two kinds of stresses. The first is caused by vibration resulting from vortices shed off a component when fluid flows by. That vibration, occurring perpendicular to the current, is referred to as “vortex-induced vibration,” or “VIV.” When water flows past the riser, it may cause vortices to be alternatively shed from each side of the riser when Reynolds Numbers reach a certain range. If the frequency of this harmonic load is near the resonant frequency of the riser, large vibrations transverse to the current can occur. The second type of stress is caused by the drag forces that push the riser in the direction of the current due to the riser’s resistance to fluid flow. The drag forces are significantly amplified by the vortex-induced vibrations of the riser, i.e. a vibrating pipe. A riser pipe that is vibrating due to vortex shedding will disrupt the flow of water around it more than a stationary riser or a non-vibrating pipe. This results in more energy transfer from the current to the riser, and hence more drag.

The movement of oil and gas exploration, development and production into deep and ultra-deep waters has created unique engineering challenges requiring innovative engineering solutions. One particular challenge is the vortex-induced-vibrations (VIV) of long drilling and production risers. As discussed above, when long elements such as subsea pipelines, risers, tendons, umbilicals and cables are

affected by relatively strong currents over extended lengths along the element, the currents may cause vortices to be shed from the sides of the element in an alternating manner which can induce VIV. The resulting vibration increases drag, reduces fatigue life and left unchecked may lead to the failure of the marine element or its supports.

Shrouds, strakes and fairings have traditionally been added to risers and other submerged pipes in order to minimize the current-induced stresses on these pipes. Strakes and shrouds can be effective regardless of the current orientation, but they tend to increase the drag acting on the riser. By contrast, fairings are generally more efficient in reducing drag and VIV. Fairings generally comprise streamlined shaped bodies (such as airplane wings) that weather-vane or rotate about the riser maintaining positions substantially aligned with the water current. Fairings generally reduce vortex-induced forces and minimize drag on the riser by reducing or breaking up the low pressure areas that exist down-current of the riser.

One example of a fairing is found in U.S. Pat. No. 4,474,129 that discloses a fairing removably mountable on risers equipped with buoyancy modules that has a tail tapering aft and a fin positioned after the tail. This fairing completely surrounds the riser and is fastened together at the back portion of the fairing. Another example of a fairing is found in U.S. Pat. No. 4,398,487 which describes a streamlined symmetrical structure having a nose portion, a tail portion and two opposed side portions. This fairing is formed as two shell halves that completely surround the riser and are connected at the front end of the nose by quick release fasteners and at the end of the tail portion by hinges. U.S. Pat. No. 5,410,979 describes a small fixed teardrop-shaped fairing that surrounds a riser and is fixed to the riser so as to not rotate. U.S. Pat. No. 6,048,136 describes a fairing that is installed on a drilling riser in combination with a synthetic foam buoyancy module. This fairing is formed as two shell halves that surround the riser and attach at the front and back portions of the fairing. A rotating fairing is described in U.S. Pat. No. 6,067,922 as including a copper element mounted in the annular region of the faking to discourage marine growth. This fairing is formed as a single piece that completely surrounds the riser and is attached at the tail or flange portion with bolts. An ultrashort fairing described in U.S. Pat. No. 6,223,672 is shown in FIG. 2. This fairing 2 has a pair of shaped sides 4A, 4B departing from the circular profile of a marine riser and converging at a trailing edge 6. It should be noted that all of the above described fairings are constructed in predetermined lengths and a plurality of fairings are positioned along the length of any particular riser.

While fairings can be effective for reducing VIV, a number of problems still exist with the prior art fairings. As illustrated in the prior art, fairings have become more and more complex in design, they often require a large number of parts, and as such, they have become more costly to produce and maintain. Generally, fairings must be secured to the elongated component by bands, bolts or other fasteners that may fail. Further, the use of such fasteners adds to the cost and labor associated with the fairing’s use. Additionally, corrosion and marine growth frequently causes the rotational elements of a fairing to seize up so that it can no longer rotate and properly align with the current. Such a concern has often resulted in fairings being used only on risers or other components that remain on the risers only a short period of time, leaving those in the industry to rely upon less effective VIV reduction means such as fixed-fin vortex strakes for more permanently fixed components.

Although strakes with certain fin heights and fin periods can reduce the amplitude of the VIV induced motion by more than 90% and have been proven to be effective tools, a lower drag solution would be desirable. While conventional teardrop fairings are effective in reducing both drag and VIV, users report that these devices are subject to a “galloping” motion. The causes of this galloping motion remain unclear. Therefore, there remains a need for an improved VIV suppression device that reduces vibration, that does not increase drag and is resistant to galloping motions.

Fairings are typically applied to drilling and/or production risers in one of two ways. In one manner of installation, fairings are placed on the riser after it is in place, suspended between the platform and the ocean floor, in which divers or submersible vehicles (referred to as ROVs) are used to fasten the multiple fairings around the riser. A second method of installation is carried out as the riser is being assembled on a vessel and installed. In this method the fairings are fastened to the pipe as lengths of pipes are fitted together to form the riser. This method of installation is typically performed on a specially designed vessel, called an S-Lay, J-Lay barge or Reel Lay barge. An S-Lay barge is one that has a declining ramp, positioned along a side or rear of the vessel and descending below the ocean’s surface, that is equipped with rollers (referred to as a “stinger”). As the lengths of pipe are fitted together, fairings are attached to the connected pipe sections before the pipe is rolled down the ramp and into the ocean. One of the problems of installing fairings in this manner is that when the fins of the fairing rotate over the rollers on the ramp, the fins frequently become damaged by the rollers. In this method of installation, the completed riser laid on the ocean floor, then is pulled up to a vertical position when it reaches the appropriate length and is attached to the surface platform and the well head on the ocean floor.

It would be advantageous to provide a relatively lightweight, resilient fairing that can be easily placed on a riser rather than having to be fastened around the entire circumference.

It would be advantageous to provide a fairing that reduces vibration, that does not increase drag and is resistant to galloping motions.

BRIEF SUMMARY OF THE INVENTION

The subject invention is directed to a fairing for the reduction of vortex-induced vibration, the minimization of drag and the elimination of galloping about a substantially cylindrical element immersed in a fluid medium. The fairing has a U-shaped cylindrical shell with opposing edges defining a longitudinal gap and parallel fins extending outwardly from the opposing edges of the shell, the parallel fins being positioned so as to reduce vortex-induced vibration, minimize drag and to eliminate the phenomenon of galloping on the cylindrical element. The fairing has a length to diameter ratio of 1.5 to 2.50 and preferably a ratio of 1.75 to 2.0 where the length is measured from the leading nose of the fairing body to the end of the fins, and the diameter is the diameter of the center of the circle defining the shell.

The shell has an outer diameter D and the parallel fins have a distance W between opposing edges of the fins. The ratio of W to D is from $W=D$ to $W=75\%$ of D . The fins of the fairing can taper inward.

The fairing further includes a bearing pad configured to fit in the gap in the shell between the shell’s opposing edges and the parallel fins. The bearing pad has a curved inside

surface and side surfaces in parallel alignment with each of the fins. Each fairing includes a plurality of bearing pads for securing the fairing to a cylindrical element.

The fairing further includes at least a set of opposed connectors for securing the fairing to a cylindrical element, each connector being positioned on an inside surface of each parallel fin. The fins can include three sets of opposed connectors. Each connector includes an opening configured to receive a fastening means for securing the opposing connectors together.

The fairing further includes a flange at a top and bottom edge of the fairing, the flange extending around the circumference of the shell and outwardly from the shell. The flange can include one or more V-shaped cutouts for the creation of opening hinges.

Each fin on the fairing does not extend beyond the outer diameter of the shell.

The fairing is constructed from a non-metallic, low corrosive material selected from a group consisting of polyethylene, polyurethane, vinyl ester resin, poly vinyl chloride and fiberglass.

The invention also includes a fairing system for the reduction of vortex-induced vibration and the minimization of drag about a substantially cylindrical element immersed in a fluid medium. The fairing system has a plurality of fairings having U-shaped cylindrical shells, each shell having opposing edges defining a longitudinal gap. Parallel fins extending outwardly from the opposing edges of each of the plurality of shells, the parallel fins being positioned so as to reduce vortex-induced vibration, minimize drag and the elimination of galloping on the cylindrical element. The fairing includes means for securing each of the plurality of fairings around the cylindrical element.

The collar includes a plurality of compliant annulus spacers extending outwardly from an inside surface of the collar, the spacers being configured to induce frictional interaction between the collar and the cylindrical element.

The means for securing the fairings around the cylindrical element includes a bearing pad configured to fit in the gap in each shell between the shell’s opposing edges and the parallel fins. The bearing pad has a curved inside surface and side surfaces in parallel alignment with each of the fins. Each fairing includes a plurality of bearing pads for securing the fairing to a cylindrical element.

An alternate means for securing the fairings around the cylindrical element include at least a set of opposed connectors, each connector being positioned on an inside surface of each parallel fin. The fins can include a plurality of opposed connectors. Each connector includes an opening configured to receive a fastening means for securing the opposing connectors together.

The fairing further may include a flange at a top and bottom edge of the fairing, the flange extending around the circumference of the shell and outwardly from the shell. The flange can include at least one V-shaped cutouts. The flanges on each fairing are configured such that they allow each fairing to freely rotate on an adjoining fairing.

A circular collar can be positioned between a group of or each of the plurality of fairings and the collar being configured such that it allows each fairing to freely rotate on the collar. The collar is in two sections held together by means for securing the collar around the cylindrical element. The collar is fixed to the riser and does not rotate.

The foregoing has outlined rather broadly the features and technical advantages of the present invention in order that the detailed description of the invention that follows may be better understood. Additional features and advantages of the

invention will be described hereinafter which form the subject of the claims of the invention. It should be appreciated by those skilled in the art that the conception and specific embodiment disclosed may be readily utilized as a basis for modifying or designing other structures for carrying out the same purposes of the present invention. It should also be realized by those skilled in the art that such equivalent constructions do not depart from the spirit and scope of the invention as set forth in the appended claims. The novel features which are believed to be characteristic of the invention, both as to its organization and method of operation, together with further objects and advantages will be better understood from the following description when considered in connection with the accompanying figures. It is to be expressly understood, however, that each of the figures is provided for the purpose of illustration and description only and is not intended as a definition of the limits of the present invention.

BRIEF DESCRIPTION OF THE DRAWINGS

For a more complete understanding of the present invention, reference is now made to the following descriptions taken in conjunction with the accompanying drawing, in which:

FIG. 1 is a side elevational view of a drilling vessel illustrating the use of the inventive fairings in one of the environments in which the invention is used;

FIG. 2 is a side elevational view of a prior art fairing;

FIG. 3 is a perspective view of one embodiment of the fairing of the present invention;

FIG. 4 is a cross-sectional view of the inventive fairing taken along lines 4-4 of FIG. 3;

FIGS. 5A and B is a perspective view of the fairing of FIG. 3 with bearing pads installed; is a cross-sectional view of the fairing taken along lines 6-6 of FIG. 5;

FIGS. 6A and B illustrate a fastener for securing the bearing pads to the fairing in FIG. 5A;

FIG. 7 is a perspective view of a bearing pad of FIG. 5;

FIG. 8 is a top plan view of the bearing pad of FIG. 7;

FIG. 9 is a perspective view of an alternate embodiment of the fairing of the present invention;

FIG. 10 is a side plan view of the fairing of FIG. 9;

FIG. 11 is a front plan view of the fairing of FIG. 9;

FIG. 12 is a cross-sectional view of the fairing taken along lines 12-12 of FIG. 11;

FIG. 13 is a top plan view of the fairing of FIG. 9;

FIG. 14 is a cross-sectional view of the fairing of FIG. 9;

FIGS. 15A and B is a side plan view of a fastener for securing the fairing of FIG. 9 to a riser;

FIG. 16 is a perspective view of a series of fairing segments of FIG. 9;

FIG. 17 is an alternate embodiment of a series of fairing segments of FIG. 9;

FIG. 18 is a perspective view of one the collars separating the fairing segments of FIG. 17;

FIG. 19 is a bottom plane view of the collar of FIG. 18;

FIG. 20 is a top plan view of an annulus spacer that is inserted in the inside surface of the collar of FIG. 18;

FIG. 21 is a graph of drag coefficient (Cd) for a bare pipe and the inventive fairing by Reynolds number (Re);

FIG. 22 is a graph of A^* by nominal reduced velocity (V_m) for a bar pipe and the inventive fairing;

FIG. 23 is a cross-sectional view of an alternate embodiment of the fairing of FIG. 9;

FIG. 24 is top plan view of the fairing of FIG. 23; and

FIG. 25 is a graph of drag coefficient (Cd) compared to velocity (V) in which each line on the graph represents a different W:D ratio.

DETAILED DESCRIPTION OF THE INVENTION

The present invention is directed to rotating fairings that include specifically placed fins for the reduction of vortex-induced vibration ("VIV") on pipes or other structural components immersed in fluid. As discussed above, when a solid object is exposed to fluid flows vibration results from vortices shed off the object when the fluid flows by it. The flow pattern around a cylinder can be characterized by the Reynolds Number (Re) of the incident flow and the location where flow separates from the cylinder surface which depends on whether the boundary layer is turbulent or laminar. In the subcritical range, the Reynolds number range is $300 < Re < 1.5 \times 10^5$, the laminar boundary layers separate at about 80 degrees aft of the leading edge of the cylinder and vortex shedding is strong and periodic. The range $1.5 \times 10^5 < Re < 3.5 \times 10^6$ is called the transition region. In these regions the boundary layer becomes turbulent and the separation points move aft to 140 degrees and the cylinder drag coefficient drops abruptly.

A fairing is described in U.S. Pat. No. 6,401,646 that includes a cylindrical shell having opposing edges defining a longitudinal gap and a pair of corresponding fins extending outwardly in a vertical direction. The fins are placed generally at a 120° angle relative to the circumference of the shell. U.S. Pat. No. 6,401,646 is incorporated herein in its entirety. During hydrodynamic testing, it has been discovered that a U-shaped fairing having parallel twin fins causes a substantial reduction in VIV and drag forces and the inventive fairing is not affected by a galloping motion.

The present invention is directed to a rotatable U-shaped fairing system having parallel twin fins for the reduction of VIV on pipes or other structural components immersed in fluid. In one embodiment, the inventive fairing is installed on drilling and production risers used in offshore oil and natural gas exploration. FIG. 1 illustrates one environment in which the inventive fairing is used. A drilling vessel or platform 10 provides surface facilities 12. Riser 14 descends from beneath the deck of the surface facilities 12 and is fitted with large OD buoyancy modules which are fitted with twin fin fairings 16 below the ocean surface 18. A plurality of fairings 16 are installed along the riser 14 to reduce VIV and minimize drag on the long unsupported riser 14. This illustrative embodiment shows the fairing system installed on a drilling or production riser. However, cylindrical pipes are employed in a variety of other applications such as subsea pipelines; drilling, import and export risers; tendons for tension leg platforms; legs for traditional fixed and for compliant platforms; other mooring elements for deepwater platforms; and so forth. Those having ordinary skill in the art can readily apply these teachings to such other applications.

Fairing 16 is formed from a U-shaped shell 20 having opposing edges 24, 26 that define a longitudinal gap G and a pair of corresponding fins 22 extending outwardly from edges 24, 26 in a vertical direction (FIGS. 3 and 4). The spaced apart fins 22 are parallel to each other and extend outwardly in a direction parallel with the fluid current in order to move the boundary layer vortex eddies further away from the riser 14 without adding significant drag (FIG. 4). The fins 22 can be any length, however, regardless of the length, the fins 22 do not extend beyond the nominal outer diameter of the shell 20. Preferably, fairing 16 has dimen-

sions of length L to diameter D (shell diameter) such that the length L to diameter D ratio (aspect ratio) is in the range of 1.5 to 2.50, preferably in the range of 1.75 to 2.0. The shell's U-shape provides for a longitudinal gap G in shell **20** that allows for placement of the shell **20** around a cylindrical object such as a riser (FIG. 5A).

Fairing **16** is secured to the riser **14** with bearing pads **32** that are configured to fit in the gap G in the shell **20** between the fins **22** (FIGS. 5, 5A). Each bearing pad **32** has a curved inside surface **34**, end portions **36** and side surfaces **38**. The inside surface **34** has a curve that completes the circle of the shell's circumference. The end portions **36** are configured to fit in the space between the opposing edges **24**, **26** and the riser **14** and the side surfaces **38** are configured to align with the fins **22** (FIG. 5). In a preferred embodiment, the back portion **40** of the inside surface **34** is open (FIGS. 5, 5A, 7). The bearing pads **32** can be secured to the shell **20** by any number of means known to one skilled in the art. An example of one securing means is a threadless bolt **42** (FIG. 6A) with a washer and cotter pin **44** (FIG. 6B) in which the bolt is placed through a pair of aligned openings **46** in fins **22** and in the side surfaces **38** of the bearing pad **32** (FIG. 5A).

The number of bearing pads **32** required to secure the fairing **16** to a riser **14** will depend upon the length of the fairing and the amount of external forces being placed on the riser. For example, if a fairing has a length of about 4½ feet, three bearing pads **32** spaced about 23 inches apart, could be used to secure the fairing **16** to a section of the riser **14** as illustrated in FIG. 5.

In an alternate embodiment, fairing **16A** is formed from a cylindrical shell **20N** having opposing fins **22N**, extending outwardly in a vertical direction, that define a longitudinal gap G (FIGS. 9 and 13). The spaced apart fins **22N** are parallel to each other relative to the center of the circle defining the shell and hence the circumference of the shell **20N** (FIG. 13). The gap G in shell **20N** provides an opening that allows for placement of the shell **20N** around a cylindrical object such as a riser. Fairing **16A** also includes a flange **24** at its top **26** and bottom **28** edge, creating a top bearing surface **26N** and a bottom bearing surface **28N** for the fairing **16A** (FIGS. 9-13).

Flange **24** extends around the circumference of the shell **20N** and extend outwardly from the shell **20N** about 3 to 4 inches. Optionally, flange **24** includes at least one V-shaped cutout **70** to act as opening hinges for the fairing **16A**. In a preferred embodiment, the V-shaped cutout can be positioned at the 12:00 o'clock, 3:00 o'clock and 9:00 o'clock position of shell **20N** in relation to the gap G at the 6:00 o'clock position. Alternatively, the V-shaped cutout can be placed anywhere on the flange **24**. The top and bottom edges **26F**, **28F** of fins **22N** each include a tail section **72** that extends outwardly from the flange **24** at the gap G. The inside edge **74** of each tail section **72** is angled from the edge of the gap G to the outer edge **76** of each fin **22N**. The angle depends on the fin length.

Fins **22N** also include a first and second connector **78a, b** that form a set of opposed connectors **78**. Each fin **22N** includes at least two sets of connectors, preferably three sets, for securing the fins **22N** together in order to attach them around a riser **14** (FIGS. 9-14). Each connector **78** extends inwardly from the outside surface **80** of each fin **22N**, creating a cavity **82** that includes an opening **84** for receiving a fastening means. The walls of each connector **78** taper inward and the opening **84** is offset from the center of the connector **78** toward the outer edge **76** of fin **22N**. The cavity **82** of each connector **78a, b** forms a rectangular shaped box

86 that extends from the inside surface **88** of the fin **22N** and is in horizontal alignment with each tail section **72**. In a preferred embodiment, a cover plate (not shown) can be secured over each of the cavity openings. Each connector **78a, b** of the set is in parallel alignment with each other. The fastening means for securing the fins **22N** together can include male-female connectors **90, 92** or studs, nuts and washers (not shown). In a preferred embodiment, the fastening means are a male and female connector **90, 92** formed from 70 Shore D polyurethane, a 90-95 Shore A polyurethane or glass reinforced polyethylene (FIGS. 15A, 15B). Alternatively, fiberglass/Inconel studs, nuts and washers can be used. The male connector **90** is placed in one **78a** of the set of connectors **78** and the female connector **92** is placed in the second **78b** of the set of connectors **78** and the fasteners **90, 92** are secured together, drawing the fins **22N** together, and thus the fairing **16N**, around riser **14**. The connectors **78** include hand holes **94** for accessing the fastening means (FIGS. 9, 11, 16, 17).

Shell **20, 20N** has an outer diameter of D and fins **22, 22N** have a distance between their ends of W. When W is equal to D the fins are parallel (FIG. 4). It has been found that if W decreases relative to D, i.e. the fins **22, 22N** are tapered, the drag is reduced. In an alternate embodiment, fairing **16B** is formed from a cylindrical shell **20P** having opposing fins **22P**, extending outwardly in a vertical direction, that define a longitudinal gap G. However, rather than being parallel, fins **22P** are tapered and thus, the distance of W is decreased between the opposing edges **23** of the fins **22P**. (FIGS. 23-24) The placement of the fins **22P** can be from W=D (parallel) to W=75% of D (tapered). Any decrease of W in relation to D, will result in tapering of the fins **22P**. A preferred ratio is W=25% of D which is about a 12.5% reduction for each fin **22P**, resulting in a 25% decrease in W with respect to D.

The fins are placed in a direction parallel with the fluid current in order to move the boundary layer vortex eddies further away from the riser **14** without adding significant drag (FIG. 13). Fins **22N, 22P** can be any length, however, regardless of the length, fins **22N, 22P** do not extend beyond the nominal outer diameter of the shell **20**. Preferably, fairing **16A, 16B** has dimensions of length to diameter (shell diameter) such that the length to diameter ratio or aspect ratio is in the range of 1.50 to 2.50, preferably equal to or greater than 1.75 to 2.0.

Fairings **16, 16A, 16B** would typically range in height from about 2 to 12 feet and would typically have a diameter of about 6 to 48 inches. The shell **20, 20N, 20P** is rotatably mounted about a substantially cylindrical element, such as the riser **14**, and rotates around the riser **14** to match the fins **22, 22N, 22P** with the direction of the current.

Shells **20, 20N** and **20P** are configured to fit around riser **14** such that they provide for pressure equalization; allow for fluid to reach the bearing face of the shell **20, 20N** for lubricating the bearing face with fluid; and to allow for the flow of fluid to retard marine growth. The configuration of the shell also assists in the directional rotation of the shell **20, 20N, 20P** around the riser **14** in order to align the fairing **16, 16N, 16P** with the current.

As shown in FIGS. 16 and 17, a fairing system is contemplated in which a number of fairing segments **16, 16A, 16B** can be installed on the riser **14** to rotate independently along a longer elongated element. With fairing segments **16A** and **16B**, the top and bottom bearing surfaces **26N, 28N** of the flanges **24** allow each fairing segment **16A, 16B** to freely rotate on the adjoining flanges **24** (FIG. 16). In an alternate embodiment, each fairing segment **16, 16A, 16B**

may be separated by a two-section collar **48** configured such that it allows each fairing **16**, **16A**, **16B** segment to freely rotate on the collar **48** (FIG. **17**). In one embodiment, the circular collar **48** has an outside surface **50**, a top surface **52**, an inside surface **54** and end sections **56a**, **b** (FIGS. **18**, **19**). The collar **48** can be any height, for example in one embodiment it can be about 3 inches. The diameter of the collar **48** will depend upon the diameter of the riser **14** it will encircle. The collar **48** also includes a plurality of annulus spacers **62** placed around the inside surface **54** of the collar **48**. The annulus spacers **62** secure the collar **48** to the riser **14** and reduce rotational and axial movement of the collar **48** by inducing hoop stress and providing a frictional surface on the riser **14**. The spacers **62** extend outwardly from the inside surface **54** such that the ID of annulus spacers **62** is smaller than the ID of the collar **48**. The number of spacers **62** on each collar will depend upon the circumference of the collar. In one embodiment, at least six spacers **62** would be used.

Each annulus spacer **62** has a spacer face **64**, and intermediate portion **66** and a spacer retainer **68** (FIG. **20**). The spacer retainer **68** is inserted through a spacer hole (not shown) in the inside surface **54** of the collar **48**. The spacer retainer **68** is sufficiently flexible that it may be elastically deformed to pass through the spacer hole in the inside surface of the collar **48**, while the spacer face **64** is sized to prevent passage through the spacer hole when a determined force is asserted against the spacer face **64**. Annulus spacers **62** are constructed of material suitable to induce frictional interaction between the collar **48** and the riser **14**. For example, the spacers **62** can be formed from a polyurethane material that provides compliance so that when the two sections **48a**, **48b** of the collar are secured around the riser **14** there is radial compression from the inside surface **54** of the collar **48** onto the riser **14**, thus compressing the polyurethane and causing friction. The two sections **48a**, **48b** of the collar are placed around the riser **14** and secured on the collar's underside **58** with a securing means such as a bolt, nut and washer threaded through an opening in each of the end sections **56a**, **b**. In a preferred embodiment, the collar will include a plurality of pressure relief holes **60** (FIGS. **14-15**).

Fairings **16**, **16A**, **16B** and collar **48** can be constructed from any non-metallic, low corrosive material such as high or low density polyethylene, polyurethane, vinyl ester resin, poly vinyl chloride (PVC), or other materials with substantially similar flexibility and durability properties or multi-layer fiberglass mat. These materials provide fairings **16**, **16A**, **16B** and collar **48** with the strength to stay on the riser **14**, but enough flex to allow it to be placed around the riser **14** during installation. The use of such materials eliminates the possibility of corrosion, which can cause the fairing shell to seize up around the elongated element it surrounds.

FIGS. **21** and **22** present test results demonstrating the surprising effectiveness of the inventive fairing **16**. FIG. **21** is a graph of drag coefficient (Cd) for a bare pipe and the ADFS by Reynolds number (Re) and FIG. **22** is a graph of A* by nominal reduced velocity (Vm), which is defined as $V_m = U / (f_n \cdot D)$ where U is the tow velocity, D the cylinder diameter and f_n is the natural frequency of the system. A* is the normalized vibration amplitude which is defined as $A^* = A / D$. For reference, the normalized vibration amplitude of a bare pipe would be expected to be in the range of 0.9 to 1.0 so this fairing decreases the amplitude of vibration by significantly more than 90%.

These tests were conducted in a tow tank with the marine element towed to develop relative motion between the test sample and the water. The test sample was allowed to freely

vibrate in the transverse direction. FIG. **21** illustrates the drag coefficient (Cd) for both a bare cylinder and a cylinder protected by the fairing **16**. FIG. **22** illustrates the velocity of both a bare cylinder and a cylinder protected by the fairing **16**. In both cases, the test sample was allowed to freely vibrate in the transverse direction. Also shown are published curves of Cd by Reynolds number for a fixed bare pipe. The transverse vibration setoff by VIV causes a several-fold increase in Cd. With the fairing **16** installed, drag coefficients are significantly reduced relative to either fixed or freely oscillating pipe to 0.4 or less. Further, the installation of the fairing **16** causes vibration to be reduced by at least 90% and by as much as 95 to 99%. FIG. **22** also illustrates the surprising superior VIV suppression performance of the inventive fairing **16**, **16A**, showing that the efficiency of the fairing **16**, **16A** is approximately 95%. A* for a bare pipe is upwards of 0.8 while a pipe equipped with the inventive fairing possesses an A* of approximately 0.01. The graph illustrates that the lateral vibrations caused by VIV are almost totally eliminated.

Tests results have shown, as illustrated in FIG. **25**, that as W decreases with respect to D, the drag coefficient is reduced. The graph of FIG. **25** shows Drag Coefficient (Cd) compared to Wind Velocity "V", where each line on the graph represents a different W:D ratio. Lines A1, A2 represents a parallel fin arrangement where W=D; line B represents the drag for a tapered fin where W=75% of D (W=0.75 of D); line C represents the drag for a tapered fin where W=50% of D (W=0.50 of D); and line D represents the drag for a tapered fin where W=5% of D (W=0.05 of D) (nearly touching). Initially as distance of W is decreased from D there is a large change in Cd. As the distance of W is decreased even further the amount of reduction in Cd is less than when initially started. As can be seen from the graph in FIG. **25**, a reduction in 25% of the distance of W in relation to D at the beginning shows a large drop in Cd (difference between lines A1, A2 and line B), while the reduction of 50% of the distance W in relation to D to the reduction of 75% of the distance of W in relation to D shows less of a change in Cd (difference between lines C and D). This trend continues till the point where the distance of W is reduced to 0 and the fins **22P** are touching.

Although the present invention and its advantages have been described in detail, it should be understood that various changes, substitutions and alterations can be made herein without departing from the spirit and scope of the invention as defined by the appended claims. Moreover, the scope of the present application is not intended to be limited to the particular embodiments of the process, machine, manufacture, composition of matter, means, methods and steps described in the specification. As one of ordinary skill in the art will readily appreciate from the disclosure of the present invention, processes, machines, manufacture, compositions of matter, means, methods, or steps, presently existing or later to be developed that perform substantially the same function or achieve substantially the same result as the corresponding embodiments described herein may be utilized according to the present invention. Accordingly, the appended claims are intended to include within their scope such processes, machines, manufacture, compositions of matter, means, methods, or steps.

What is claimed is:

1. A fairing for the reduction of vortex-induced vibration, the minimization of drag and the elimination of the galloping phenomenon about a substantially cylindrical element immersed in a fluid medium, comprising:

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a one-piece fairing having a U-shaped cylindrical shell portion [at] comprising a leading edge [of the fairing with], opposing edges [extending toward], and a trailing edge [of the fairing], the opposing edges extending from the leading edge toward the trailing edge of the shell portion, the opposing edges further defining a longitudinal gap therebetween that extends from the leading edge to the trailing edge of the shell portion, wherein the longitudinal gap provides an opening that allows for placement of the substantially cylindrical element through the longitudinal gap and into the shell portion of the fairing; and

[parallel] longitudinal fins extending outwardly from the opposing edges of the shell portion in which the [parallel] longitudinal fins taper inward toward the trailing edge of the fairing, wherein the [parallel] longitudinal fins are positioned so as to reduce vortex-induced vibration and minimize drag on the cylindrical element.

2. The fairing of claim 1, further including a [bearing pad] connector configured to fit in the gap in the shell portion between the [shell's] shell portion's opposing edges and the [parallel] longitudinal fins.

[3. The fairing of claim 2, wherein the bearing pad has a curved inside surface and side surfaces in parallel alignment with each of the fins.]

4. The fairing of claim 2, wherein each fairing includes at least one [bearing pad] connector for securing the fairing to a cylindrical element, wherein each connector has a curved inside surface and side surfaces in alignment with each of the fins.

5. The fairing of claim 4, wherein each fairing includes a plurality of [bearing pads] connectors for securing the fairing to a cylindrical element.

6. The fairing of claim 1, further including at least a set of opposed connectors for securing the fairing to a cylindrical element, each connector being positioned on an inside surface of each [parallel] longitudinal fin.

7. The fairing of claim 6, wherein the fins include a plurality of opposed connectors.

8. The fairing of claim 6, wherein each connector includes an opening configured to receive a fastening means for securing the opposing connectors together.

9. The fairing of claim 6, further including a flange at a top and bottom edge of the fairing, the flange extending around the circumference of the shell and outwardly from the shell.

10. The fairing of claim 9, wherein the flange includes at least one V-shaped cutouts.

11. The fairing of claim 1, wherein each fin does not extend beyond the outer diameter of the shell.

12. The fairing of claim 1, wherein the fairing is constructed from a non-metallic, low corrosive material selected from a group consisting of polyethylene, polyurethane, vinyl ester resin, poly vinyl chloride and fiberglass.

13. The fairing of claim 1, wherein the shell portion has an outer diameter D and the longitudinal fins have a distance W located between opposing [edges] ends of the longitudinal fins at the trailing edge of the fairing.

14. The fairing of claim 13, wherein [the] a ratio of W to D is from [W=D] $W < 100\%$ of D to $W = 75\%$ of D.

15. A fairing system for the reduction of vortex-induced vibration, the minimization of drag and the elimination of the galloping phenomenon about a substantially cylindrical element immersed in a fluid medium, the fairing system comprising:

a plurality of one-piece fairings having U-shaped cylindrical shell portions [at], each shell portion comprising

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a leading edge [of each shell portion, each shell portion having], opposing edges [extending toward], and a trailing edge [of the fairing], the opposing edges extending from the leading edge toward the trailing edge of the shell portion, the opposing edges further defining a longitudinal gap therebetween that extends from the leading edge to the trailing edge of the shell portion, wherein the longitudinal gap provides an opening that allows for placement of the substantially cylindrical element through the longitudinal gap and into the shell portion;

[parallel] longitudinal fins extending outwardly from the opposing edges of each of the plurality of shell portions in which the [parallel] longitudinal fins taper inward toward the trailing edge of the fairing, wherein the [parallel] longitudinal fins are positioned so as to reduce vortex-induced vibration and minimize drag on the cylindrical element; and

means for securing each of the plurality of fairings around the cylindrical element.

16. The fairing system of claim 15, wherein the means for securing the fairings around the cylindrical element includes a [bearing pad] connector configured to fit in the gap in each shell portion between the [shell's] shell portion's opposing edges and the [parallel] longitudinal fins.

[17. The fairing system of claim 16, wherein the bearing pad has a curved inside surface and side surfaces in parallel alignment with each of the fins.]

18. The fairing system of claim 16, wherein each fairing includes at least one [bearing pad] connector for securing the fairing to a cylindrical element, wherein each connector has a curved inside surface and side surfaces in alignment with each of the fins.

19. The fairing of claim 15, wherein the means for securing include at least a set of opposed connectors, each connector being positioned on an inside surface of each [parallel] longitudinal fin.

20. The fairing of claim 19, wherein the fins include a plurality of opposed connectors.

21. The fairing of claim 19, wherein each connector includes an opening configured to receive a fastening means for securing the opposing connectors together.

22. The fairing of claim 15, further including a flange at a top and bottom edge of the fairing, the flange extending around the circumference of the shell and outwardly from the shell.

23. The fairing of claim 22, wherein the flange includes at least one V-shaped cutout.

24. The fairing system of claim 22, wherein the flanges on each fairing are configured such that they allows each fairing to freely rotate on an adjoining fairing.

25. The fairing system of claim 15, wherein a circular collar is positioned between each of the plurality of fairings, the collar configured such that it allows each fairing to freely rotate on the collar.

26. The fairing system of claim 25, wherein the collar is in two sections held together by securing means for securing the collar around the cylindrical element.

27. The fairing system of claim 25, wherein the collar includes a plurality of compliant annulus spacers extending outwardly from an inside surface of the collar, the spacers being configured to induce frictional interaction between the collar and the cylindrical element.

28. The fairing system of claim 15, wherein each fin does not extend beyond the outer diameter of the shell.

29. The fairing system of claim 15, wherein the fairing is constructed from a non-metallic, low corrosive material

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selected from a group consisting of polyethylene, polyurethane, vinyl ester resin, poly vinyl chloride and fiberglass.

30. The fairing system of claim 15, wherein the shell portion has an outer diameter D and the [parallel] longitudinal fins have a distance W located between opposing [edges] ends of the fins at the trailing edge of the fairing. 5

31. The fairing of claim 30, wherein [the] a ratio of W to D is from [W=D] $W < 100\%$ of D to $W = 75\%$ of D.

32. The fairing of claim 13, wherein the [ration] ratio of W to D is from [W=D] $W < 100\%$ of D to $W = 50\%$ of D. 10

33. The fairing system of claim 30, wherein the [ration] ratio of W to D is from [W=D] $W < 100\%$ of D to $W = 50\%$ of D.

34. A fairing for the reduction of vortex-induced vibration, the minimization of drag and the elimination of the galloping phenomenon about a substantially cylindrical element immersed in a fluid medium, comprising: 15

a one-piece fairing having a U-shaped cylindrical shell portion [at] comprising a leading edge [of the fairing with], opposing edges [extending toward], and a trailing edge [of the fairing], the opposing edges extending from the leading edge toward the trailing edge of the 20

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shell portion, the opposing edges further defining a longitudinal gap therebetween that extends from the leading edge to the trailing edge of the shell portion, wherein the longitudinal gap provides an opening that allows for placement of the substantially cylindrical element through the longitudinal gap and into the shell portion of the fairing;

[parallel] longitudinal fins extending outwardly from the opposing edges of the shell portion in which the [parallel] longitudinal fins taper inward toward the trailing edge of the fairing, wherein the [parallel] longitudinal fins are positioned so as to reduce vortex-induced vibration and minimize drag on the cylindrical element;

a flange at a top and bottom edge of the fairing, the flange extending around the circumference of the shell portion and outwardly from the shell portion; and

at least a set of opposed connectors for securing the fairing to the cylindrical element, each connector being positioned on an inside surface of each [parallel] longitudinal fin.

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