



US00RE47101E

(19) **United States**
(12) **Reissued Patent**
McClure et al.

(10) **Patent Number: US RE47,101 E**
(45) **Date of Reissued Patent: Oct. 30, 2018**

(54) **CONTROL FOR DISPENSING MATERIAL FROM VEHICLE**

(71) Applicant: **AgJunction LLC**, Hiawatha, KS (US)
(72) Inventors: **John A. McClure**, Scottsdale, AZ (US);
Aaron C. Stichter, Chandler, AZ (US)

(73) Assignee: **AGJUNCTION LLC**, Hiawatha, KS (US)

(21) Appl. No.: **15/003,756**

(22) Filed: **Jan. 21, 2016**

Related U.S. Patent Documents

Reissue of:

(64) Patent No.: **8,634,993**
Issued: **Jan. 21, 2014**
Appl. No.: **13/217,839**
Filed: **Aug. 25, 2011**

U.S. Applications:

(63) Continuation-in-part of application No. 12/355,776, filed on Jan. 17, 2009, now Pat. No. 8,140,223, which
(Continued)

(51) **Int. Cl.**
A01B 79/00 (2006.01)
G01S 19/14 (2010.01)
(Continued)

(52) **U.S. Cl.**
CPC **A01B 79/005** (2013.01); **G01C 15/00**
(2013.01); **G01S 5/0063** (2013.01); **G01S 19/04** (2013.01);
(Continued)

(58) **Field of Classification Search**
CPC **A01B 79/005**; **A01D 41/127**
(Continued)

(56) **References Cited**

U.S. PATENT DOCUMENTS

3,585,537 A 6/1971 Rennick
3,596,228 A 7/1971 Reed, Jr.
(Continued)

FOREIGN PATENT DOCUMENTS

AU 2002244539 10/2002
AU 2002325645 9/2007
(Continued)

OTHER PUBLICATIONS

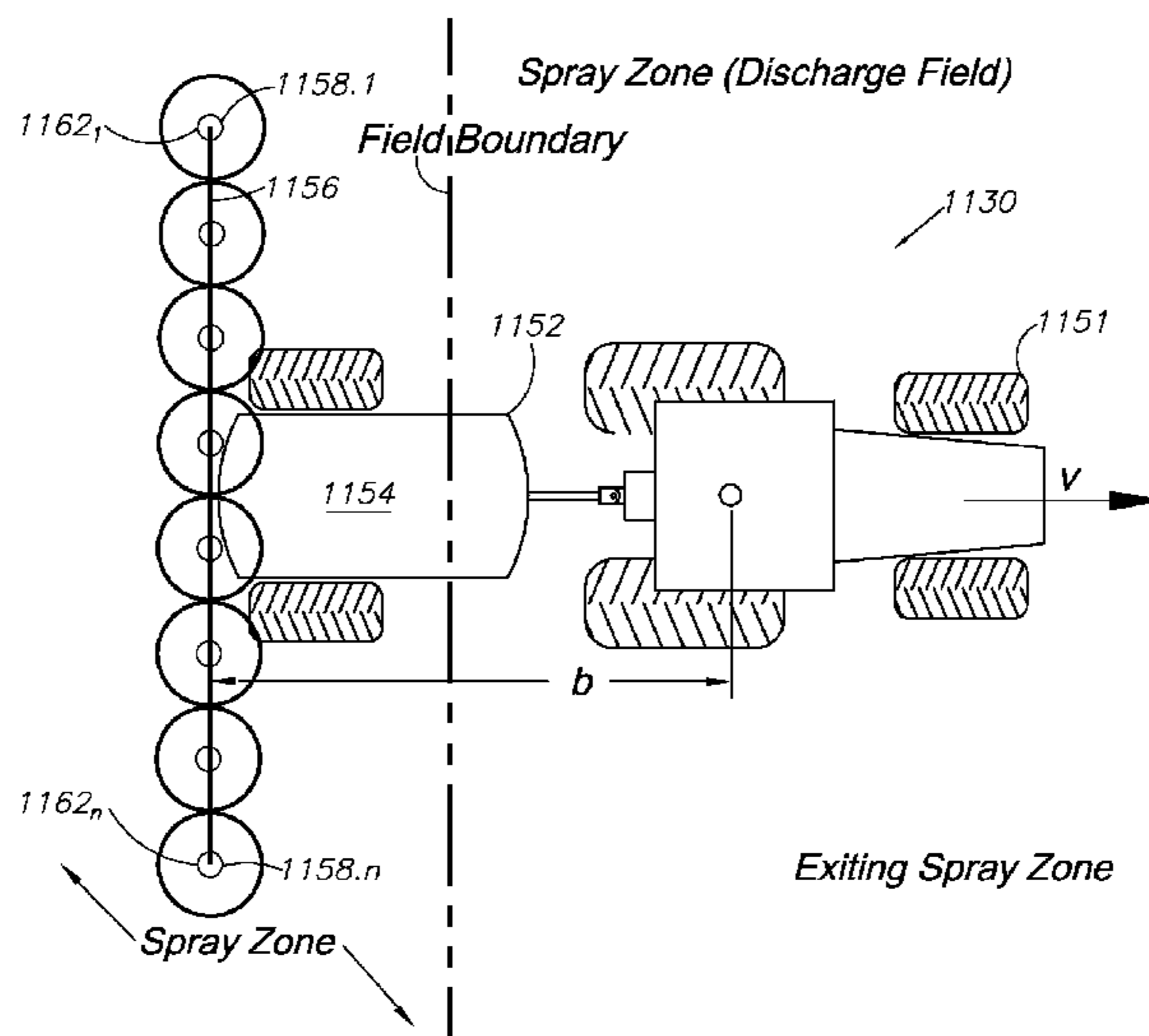
“ARINC Engineering Services, Interface Specification IS-GPS-200, Revision D”, Online [retrieved on May 18, 2010]. Retrieved from the Internet: <<http://www.navcen.uscg.gov/gps/geninfo/IS-GPS-200D.pdf>> (Dec. 7, 2004) p. 168, para [0001].
(Continued)

Primary Examiner — Nick Corsaro
(74) *Attorney, Agent, or Firm* — Schwabe Williamson & Wyatt

(57) **ABSTRACT**

A spray control method employs a spray vehicle including a material tank, a pump communicating with the tank, and nozzles of a spray boom communicating with the pump. A GNSS receiver mounted on the vehicle and interfaced to a controller tracks its position in relation to stored position coordinates of field boundaries separating spray zones from spray exclusion zones. The tank is activated and deactivated by the controller to retain spray of the material within the spray zones and to prevent spray of the material in the exclusion zones, by processing an offset of the spray nozzles from the receiver, the spray range of the nozzles, spray turn-on and turn-off lag times, and the velocity of the spray vehicle, all in relation to the field boundaries. An alternative embodiment individually controls spray from the nozzles by using associated valves interfaced to the controller.

22 Claims, 32 Drawing Sheets



Related U.S. Application Data

is a continuation-in-part of application No. 12/171,399, filed on Jul. 11, 2008, now Pat. No. 8,265,826, which is a continuation-in-part of application No. 10/804,758, filed on Mar. 19, 2004, now Pat. No. 7,400,956, and a continuation-in-part of application No. 10/828,745, filed on Apr. 21, 2004, now abandoned, said application No. 13/217,839 is a continuation-in-part of application No. 12/750,429, filed on Mar. 30, 2010, now Pat. No. 8,214,111, which is a continuation of application No. 11/184,657, filed on Jul. 19, 2005, now Pat. No. 7,689,354.

(60) Provisional application No. 60/464,756, filed on Apr. 23, 2003, provisional application No. 60/456,146, filed on Mar. 20, 2003.

(51) **Int. Cl.**

G05D 1/02 (2006.01)
G01S 5/00 (2006.01)
G01S 19/04 (2010.01)
G01S 19/54 (2010.01)
G01C 15/00 (2006.01)

(52) **U.S. Cl.**

CPC **G01S 19/14** (2013.01); **G01S 19/54** (2013.01); **G05D 1/027** (2013.01); **G05D 1/0278** (2013.01); **G05D 1/0259** (2013.01); **G05D 2201/0201** (2013.01)

(58) **Field of Classification Search**

USPC 701/50
 See application file for complete search history.

(56) **References Cited**

U.S. PATENT DOCUMENTS

3,727,710 A 4/1973 Sanders
 3,815,272 A 6/1974 Marleau
 3,899,028 A 8/1975 Morris
 3,987,456 A 10/1976 Gelin
 4,132,272 A 1/1979 Holloway
 4,170,776 A 10/1979 Frosch
 4,180,133 A 12/1979 Collogan
 4,398,162 A 8/1983 Nagai
 4,453,614 A 6/1984 Allen
 4,529,990 A 7/1985 Brunner
 4,637,474 A 1/1987 Leonard
 4,667,203 A 5/1987 Counselman, III
 4,689,556 A 8/1987 Cedrone
 4,694,264 A 9/1987 Owens
 4,701,760 A 10/1987 Raoux
 4,710,775 A 12/1987 Coe
 4,714,435 A 12/1987 Stipanuk
 4,739,448 A 4/1988 Rowe
 4,751,512 A 6/1988 Longaker
 4,769,700 A 9/1988 Pryor
 4,785,463 A 11/1988 Janc
 4,802,545 A 2/1989 Nystuen
 4,803,626 A 2/1989 Bachman et al.
 4,812,991 A 3/1989 Hatch
 4,858,132 A 8/1989 Holmquist
 4,864,320 A 9/1989 Munson
 4,894,662 A 1/1990 Counselman
 4,916,577 A 4/1990 Dawkins
 4,918,607 A 4/1990 Wible
 4,963,889 A 10/1990 Hatch
 5,021,792 A 6/1991 Hwang
 5,031,704 A 7/1991 Fleischer
 5,100,229 A 3/1992 Lundberg
 5,134,407 A 7/1992 Lorenz
 5,144,317 A 9/1992 Duddek
 5,144,767 A 9/1992 McCloy et al.

5,148,179 A 9/1992 Allison
 5,152,347 A 10/1992 Miller
 5,155,490 A 10/1992 Spradley
 5,155,493 A 10/1992 Thursby
 5,156,219 A 10/1992 Schmidt
 5,165,109 A 11/1992 Han
 5,173,715 A 12/1992 Rodal
 5,177,489 A 1/1993 Hatch
 5,185,610 A 2/1993 Ward
 5,191,351 A 3/1993 Hofer
 5,194,851 A 3/1993 Kraning et al.
 5,202,829 A 4/1993 Geier
 5,207,239 A 5/1993 Schwitalia
 5,239,669 A 8/1993 Mason
 5,246,164 A 9/1993 McCann et al.
 5,255,756 A 10/1993 Follmer
 5,260,875 A 11/1993 Tofte et al.
 5,268,695 A 12/1993 Dentinger et al.
 5,293,170 A 3/1994 Lorenz
 5,294,970 A 3/1994 Dornbusch
 5,296,861 A 3/1994 Knight
 5,311,149 A 5/1994 Wagner
 5,323,322 A 6/1994 Mueller
 5,334,987 A 8/1994 Teach
 5,343,209 A 8/1994 Sennott
 5,344,105 A 9/1994 Youhanaie
 5,345,245 A 9/1994 Ishikawa
 5,359,332 A 10/1994 Allison
 5,361,212 A 11/1994 Class
 5,365,447 A 11/1994 Dennis
 5,369,589 A 11/1994 Steiner
 5,375,059 A 12/1994 Krytsos
 5,389,934 A 2/1995 Kass
 5,390,124 A 2/1995 Kyrtos
 5,390,125 A 2/1995 Sennott
 5,390,207 A 2/1995 Fenton
 5,404,661 A 4/1995 Sahm
 5,416,712 A 5/1995 Geier
 5,430,654 A 7/1995 Kyrtos et al.
 5,442,363 A 8/1995 Remondi
 5,444,453 A 8/1995 Lalezari
 5,451,964 A 9/1995 Babu
 5,467,282 A 11/1995 Dennis
 5,471,217 A 11/1995 Hatch
 5,476,147 A 12/1995 Fixemer
 5,477,228 A 12/1995 Tiwari
 5,477,458 A 12/1995 Loomis
 5,490,073 A 2/1996 Kyrtos
 5,491,636 A 2/1996 Robertson
 5,495,257 A 2/1996 Loomis
 5,504,482 A 4/1996 Schreder
 5,511,623 A 4/1996 Frasier
 5,519,620 A 5/1996 Talbot
 5,521,610 A 5/1996 Rodal
 5,523,761 A 6/1996 Gildea
 5,534,875 A 7/1996 Diefes
 5,539,398 A 7/1996 Hall
 5,543,804 A 8/1996 Buchler
 5,546,093 A 8/1996 Gudat
 5,548,293 A 8/1996 Cohen
 5,561,432 A 10/1996 Knight
 5,563,786 A 10/1996 Torii
 5,564,632 A 10/1996 Prohazka
 5,568,152 A 10/1996 Janky
 5,568,162 A 10/1996 Samsel
 5,583,513 A 12/1996 Cohen
 5,589,835 A 12/1996 Gildea
 5,592,382 A 1/1997 Colley
 5,596,328 A 1/1997 Stangeland
 5,600,670 A 2/1997 Turney
 5,604,506 A 2/1997 Rodal
 5,608,393 A 3/1997 Hartman
 5,610,522 A 3/1997 Locatelli
 5,610,616 A 3/1997 Vallot
 5,610,845 A 3/1997 Slabinski
 5,612,864 A 3/1997 Henderson
 5,612,883 A 3/1997 Shaffer
 5,615,116 A 3/1997 Gudat
 5,617,100 A 4/1997 Akiyoshi

(56)

References Cited

U.S. PATENT DOCUMENTS

5,617,317 A	4/1997	Ignagni	6,069,583 A	5/2000	Silvestrin
5,621,646 A	4/1997	Enge	6,070,673 A	6/2000	Wendte
5,638,077 A	6/1997	Martin	6,076,612 A	6/2000	Carr
5,640,323 A	6/1997	Kleimenhagen et al.	6,081,171 A	6/2000	Ella
5,644,139 A	7/1997	Allen	6,088,644 A	7/2000	Brandt
5,646,630 A	7/1997	Sheynblat	6,100,842 A	8/2000	Dreier
5,653,389 A *	8/1997	Henderson et al. 239/69	6,122,595 A	9/2000	Varley
5,663,879 A	9/1997	Trovato et al.	6,128,574 A	10/2000	Diekhans
5,664,632 A	9/1997	Frasier	6,144,335 A	11/2000	Rogers
5,673,491 A	10/1997	Brenna	6,191,730 B1	2/2001	Nelson, Jr.
5,680,140 A	10/1997	Loomis	6,191,732 B1	2/2001	Carlson
5,684,476 A	11/1997	Anderson	6,191,733 B1	2/2001	Dizchavez
5,684,696 A	11/1997	Rao	6,198,430 B1	3/2001	Hwang
5,702,070 A	12/1997	Waid	6,198,992 B1	3/2001	Winslow
5,704,546 A	1/1998	Henderson et al.	6,199,000 B1	3/2001	Keller et al.
5,706,015 A	1/1998	Chen	6,205,401 B1	3/2001	Pickhard
5,717,593 A	2/1998	Gvili	6,212,453 B1	4/2001	Kawagoe et al.
5,725,230 A	3/1998	Walkup	6,215,828 B1	4/2001	Signell
5,731,786 A	3/1998	Abraham	6,229,479 B1	5/2001	Kozlov et al.
5,739,785 A	4/1998	Allison	6,230,091 B1	5/2001	McQuinn
5,757,316 A	5/1998	Buchler	6,230,097 B1	5/2001	Dance
5,765,123 A	6/1998	Nimura	6,233,511 B1	5/2001	Berger
5,777,578 A	7/1998	Chang	6,236,907 B1	5/2001	Hauwiller et al.
5,810,095 A	9/1998	Orbach	6,236,916 B1	5/2001	Staub
5,821,900 A	10/1998	Kishimoto	6,236,924 B1	5/2001	Motz
5,828,336 A	10/1998	Yunck	6,253,160 B1	6/2001	Hanseder
5,838,562 A	11/1998	Gudat	6,256,583 B1	7/2001	Sutton
5,854,987 A	12/1998	Sekine	6,259,398 B1	7/2001	Riley
5,862,501 A	1/1999	Talbot	6,266,595 B1	7/2001	Greatline
5,864,315 A	1/1999	Welles	6,271,788 B1	8/2001	Longaker
5,864,318 A	1/1999	Cosenza	6,278,918 B1	8/2001	Dickson et al.
5,875,408 A	2/1999	Bendett	6,285,320 B1	9/2001	Olster
5,877,725 A	3/1999	Kalafus	6,292,132 B1	9/2001	Wilson
5,884,224 A	3/1999	McNabb et al.	6,304,210 B1	10/2001	Allison et al.
5,890,091 A	3/1999	Talbot	6,307,505 B1	10/2001	Green
5,899,957 A	5/1999	Loomis	6,313,788 B1	11/2001	Wilson
5,903,235 A	5/1999	Nichols	6,314,348 B1	11/2001	Winslow
5,906,645 A	5/1999	Kagawa	6,324,473 B1	11/2001	Eschenbach
5,912,798 A	6/1999	Chu	6,325,684 B1	12/2001	Knight
5,914,685 A	6/1999	Kozlov	6,336,051 B1	1/2002	Pangels
5,917,448 A	6/1999	Mickelson	6,336,066 B1	1/2002	Pellenc et al.
5,918,558 A	7/1999	Susag	6,345,231 B2	2/2002	Quincke
5,919,242 A	7/1999	Greatline et al.	6,356,602 B1	3/2002	Rodal
5,923,270 A	7/1999	Sampo et al.	6,371,416 B1	4/2002	Hawthorne
5,924,239 A	7/1999	Rees et al.	6,377,889 B1	4/2002	Soest
5,926,079 A	7/1999	Heine	6,380,888 B1	4/2002	Kucik
5,927,603 A	7/1999	McNabb	6,385,515 B1	5/2002	Dickson et al.
5,928,309 A	7/1999	Korver et al.	6,389,345 B2	5/2002	Phelps
5,929,721 A	7/1999	Munn	6,392,589 B1	5/2002	Rogers
5,931,882 A	8/1999	Fick et al.	6,397,147 B1	5/2002	Whitehead
5,933,110 A	8/1999	Tang	6,411,254 B1	6/2002	Moeglein
5,935,183 A	8/1999	Sahm	6,415,229 B1	7/2002	Diekhans
5,936,573 A	8/1999	Smith	6,418,031 B1	7/2002	Archambeault
5,940,026 A	8/1999	Popeck	6,421,003 B1	7/2002	Riley
5,941,317 A	8/1999	Mansur	6,424,915 B1	7/2002	Fukuda
5,943,008 A	8/1999	Van Dusseldorp	6,425,186 B1	7/2002	Oliver
5,944,770 A	8/1999	Enge	6,431,576 B1	8/2002	Viaud
5,945,917 A	8/1999	Harry	6,434,462 B1	8/2002	Bevly et al.
5,949,371 A	9/1999	Nichols	6,445,983 B1	9/2002	Dickson et al.
5,951,613 A	9/1999	Sahm et al.	6,445,990 B1	9/2002	Manring
5,955,973 A	9/1999	Anderson	6,449,558 B1	9/2002	Small
5,956,250 A	9/1999	Gudat	6,463,091 B1	10/2002	Zhodzicshsky
5,969,670 A	10/1999	Kalafus	6,463,374 B1	10/2002	Keller
5,987,383 A	11/1999	Keller	6,466,871 B1	10/2002	Reisman
6,014,101 A	1/2000	Loomis	6,469,663 B1	10/2002	Whitehead
6,014,608 A	1/2000	Seo	6,484,097 B2	11/2002	Fuchs
6,018,313 A	1/2000	Englemayer	6,501,422 B1	12/2002	Nichols
6,023,239 A	2/2000	Kovach	6,505,117 B1	1/2003	Ratert
6,049,304 A	4/2000	Rudel	6,510,367 B1 *	1/2003	McQuinn A01B 79/005 700/241
6,052,647 A	4/2000	Parkinson	6,515,619 B1	2/2003	McKay, Jr.
6,055,477 A	4/2000	McBurney	6,516,271 B2	2/2003	Upadhyaya
6,057,800 A	5/2000	Yang	6,522,948 B1 *	2/2003	Benneweis A01B 79/005 700/243
6,061,390 A	5/2000	Meehan	6,539,303 B2	3/2003	McClure
6,061,632 A	5/2000	Dreier	6,542,077 B2	4/2003	Joao
6,062,317 A	5/2000	Gharsalli	6,549,835 B2	4/2003	Deguchi
			6,553,299 B1	4/2003	Keller
			6,553,300 B2	4/2003	Ma

(56)

References Cited

U.S. PATENT DOCUMENTS

6,553,311 B2	4/2003	Aheam	7,271,766 B2	9/2007	Zimmerman
6,567,041 B1	5/2003	O'Dell	7,277,784 B2	10/2007	Weiss
6,570,534 B2	5/2003	Cohen	7,277,792 B2	10/2007	Overschie
6,577,952 B2	6/2003	Geier	7,292,185 B2	11/2007	Whitehead et al.
6,587,761 B2	7/2003	Kumar	7,292,186 B2	11/2007	Miller
6,606,542 B2	8/2003	Hauwiller	7,324,915 B2	1/2008	Altmann
6,611,228 B2	8/2003	Toda	7,358,896 B2	4/2008	Gradincic
6,611,754 B2	8/2003	Klein	7,373,231 B2	5/2008	McClure
6,611,755 B1	8/2003	Coffee	7,388,539 B2	6/2008	Whitehead
6,618,671 B2	9/2003	Dooley	7,395,769 B2	7/2008	Jensen
6,622,091 B2	9/2003	Perlmutter	7,400,294 B2	7/2008	Whitehead et al.
6,631,394 B1	10/2003	Ronkka	7,400,956 B1	7/2008	Feller
6,631,916 B1	10/2003	Miller	7,428,259 B2	9/2008	Wang
6,633,814 B2	10/2003	Kohli	7,437,230 B2	10/2008	McClure
6,643,576 B1	11/2003	O'Connor	7,451,030 B2	11/2008	Eglington
6,646,603 B2	11/2003	Dooley	7,460,942 B2	12/2008	Mailer
6,657,585 B1	12/2003	Kucik	7,479,900 B2	1/2009	Horstemeyer
6,657,875 B1	12/2003	Zeng	7,505,848 B2	3/2009	Flann
6,671,587 B2	12/2003	Hrovat	7,522,099 B2	4/2009	Zhodzishsky
6,686,878 B1	2/2004	Lange	7,522,100 B2	4/2009	Yang
6,688,403 B2	2/2004	Bernhardt	7,571,029 B2	8/2009	Dai
6,702,200 B2	3/2004	Solie et al.	7,689,354 B2	3/2010	Heiniger
6,703,973 B1	3/2004	Nichols	7,930,085 B2	4/2011	Anderson et al.
6,711,501 B2	3/2004	McClure	7,994,971 B2	8/2011	Velde et al.
6,721,638 B2	4/2004	Zeitler	8,018,376 B2	9/2011	McClure et al.
6,744,404 B1	6/2004	Whitehead	8,140,223 B2	3/2012	Whitehead et al.
6,754,584 B2	6/2004	Pinto	8,190,337 B2	5/2012	McClure
6,756,938 B2	6/2004	Zhao	8,214,111 B2	7/2012	Heiniger
6,771,501 B2	8/2004	Coleman	8,219,926 B2	7/2012	Othmer
6,774,843 B2	8/2004	Takahashi	8,265,826 B2	9/2012	Feller et al.
6,788,951 B2	9/2004	Aoki	8,437,901 B2	5/2013	Anderson
6,789,014 B1	9/2004	Rekow et al.	8,523,085 B2	9/2013	Grimm et al.
6,792,380 B2	9/2004	Toda	8,571,764 B2	10/2013	Peterson et al.
6,810,315 B2	10/2004	Cessac	8,634,993 B2	1/2014	McClure
6,813,544 B2	11/2004	Hood et al.	8,649,930 B2	2/2014	Reeve et al.
6,819,269 B2	11/2004	Flick	2002/0029110 A1	3/2002	Fukuda
6,819,780 B2	11/2004	Benson et al.	2002/0038171 A1	3/2002	Deguchi
6,822,314 B2	11/2004	Beasom	2002/0072850 A1	6/2002	McClure
6,832,024 B2	12/2004	Gerstenberger	2002/0107609 A1*	8/2002	Benneweis 700/231
6,862,083 B1	3/2005	McConnell et al.	2002/0165645 A1	11/2002	Kageyama
6,865,465 B2	3/2005	McClure	2002/0165648 A1	11/2002	Zeitler
6,865,484 B2	3/2005	Miyasaka	2002/0165669 A1	11/2002	Pinto
6,876,920 B1	4/2005	Mailer	2002/0169553 A1	11/2002	Perlmutter
6,879,283 B1	4/2005	Bird	2002/0175858 A1	11/2002	Takahashi
6,900,992 B2	5/2005	Kelly	2003/0009282 A1	1/2003	Upadhyaya et al.
6,922,635 B2	7/2005	Rorabaugh	2003/0014171 A1	1/2003	Ma
6,931,233 B1	8/2005	Tso	2003/0093210 A1	5/2003	Kondo
6,937,939 B1	8/2005	Shibusawa et al.	2003/0187560 A1	10/2003	Keller
6,961,018 B2	11/2005	Heppe	2003/0187577 A1	10/2003	McClure
6,967,538 B2	11/2005	Woo	2003/0191568 A1	10/2003	Breed
6,990,399 B2	1/2006	Hrazdera et al.	2003/0195008 A1	10/2003	Mohi et al.
7,006,032 B2	2/2006	King	2003/0208319 A1	11/2003	Ell
7,026,982 B2	4/2006	Toda	2004/0006426 A1	1/2004	Armstrong et al.
7,027,918 B2	4/2006	Zimmerman	2004/0039514 A1	2/2004	Steichen
7,031,725 B2	4/2006	Rorabaugh	2004/0069875 A1*	4/2004	Bui 239/533.2
7,065,440 B2	6/2006	Aral	2004/0186644 A1	9/2004	McClure
7,089,099 B2	8/2006	Shostak	2004/0210357 A1	10/2004	McKay
7,110,762 B1	9/2006	Cameron et al.	2004/0212533 A1	10/2004	Whitehead
7,110,881 B2	9/2006	Gray	2005/0043882 A1	2/2005	Takazawa
7,124,964 B2	10/2006	Bui	2005/0055147 A1	3/2005	Hrazdera
7,142,956 B2	11/2006	Heiniger	2005/0060069 A1	3/2005	Breed et al.
7,143,980 B2	12/2006	Heller et al.	2005/0080559 A1	4/2005	Ishibashi
7,155,335 B2	12/2006	Rennels	2005/0116859 A1	6/2005	Miller
7,162,348 B2	1/2007	McClure	2005/0165546 A1	7/2005	Aral
7,162,384 B1	1/2007	Browning	2005/0225955 A1	10/2005	Grebenkemper
7,184,859 B2	2/2007	Hood et al.	2005/0265494 A1	12/2005	Goodings
7,191,061 B2	3/2007	McKay	2006/0171611 A1	1/2006	Hatch
7,200,490 B2	4/2007	Lange	2006/0031664 A1	2/2006	Wilson
7,216,033 B2	5/2007	Flann	2006/0061469 A1	3/2006	Jaeger
7,221,314 B2	5/2007	Brabec	2006/0142936 A1	6/2006	Dix
7,228,214 B2	6/2007	Flann	2006/0167600 A1	7/2006	Nelson
7,231,290 B2	6/2007	Steichen	2006/0178820 A1	8/2006	Eglington
7,248,211 B2	7/2007	Hatch	2006/0178823 A1	8/2006	Eglington
7,254,485 B2	8/2007	Rooney et al.	2006/0178825 A1	8/2006	Eglington
7,256,388 B2	8/2007	Eglington	2006/0206246 A1	9/2006	Walker
			2006/0215739 A1	9/2006	Williamson
			2007/0021913 A1	1/2007	Heiniger
			2007/0078570 A1	4/2007	Dai
			2007/0088447 A1	4/2007	Stothert

(56)

References Cited

U.S. PATENT DOCUMENTS

2007/0121708 A1 5/2007 Simpson
 2007/0192024 A1 8/2007 Flann
 2007/0205940 A1 9/2007 Yang
 2007/0285308 A1 12/2007 Bauregger
 2008/0129586 A1 6/2008 Martin
 2008/0195268 A1 8/2008 Sapilewski et al.
 2008/0204312 A1 8/2008 Euler
 2008/0269988 A1 10/2008 Feller
 2009/0005990 A1 1/2009 Anderson et al.
 2009/0093959 A1 4/2009 Scherzinger et al.
 2009/0121932 A1 5/2009 Whitehead et al.
 2009/0164067 A1 6/2009 Whitehead
 2009/0171583 A1 7/2009 DiEsposti
 2009/0174597 A1 7/2009 DiLellio
 2009/0174622 A1 7/2009 Kanou
 2009/0177395 A1 7/2009 Stelpstra
 2009/0177399 A1 7/2009 Park
 2009/0259397 A1 10/2009 Stanton
 2009/0259707 A1 10/2009 Martin
 2009/0262014 A1 10/2009 DiEsposti
 2009/0262018 A1 10/2009 Vasilyev
 2009/0262974 A1 10/2009 Lithopoulos
 2009/0265054 A1 10/2009 Basnayake
 2009/0265101 A1 10/2009 Jow
 2009/0265104 A1 10/2009 Shroff
 2009/0265308 A1 10/2009 Brown
 2009/0273372 A1 11/2009 Brenner
 2009/0273513 A1 11/2009 Huang
 2009/0274079 A1 11/2009 Bhatia
 2009/0274113 A1 11/2009 Katz
 2009/0276127 A1 11/2009 Dix
 2009/0276155 A1 11/2009 Jeerage
 2009/0295633 A1 12/2009 Pinto
 2009/0295634 A1 12/2009 Yu
 2009/0299550 A1 12/2009 Baker
 2009/0322597 A1 12/2009 Medina Herrero
 2009/0322598 A1 12/2009 Fly
 2009/0322600 A1 12/2009 Whitehead
 2009/0322601 A1 12/2009 Ladd
 2009/0322606 A1 12/2009 Gronemeyer
 2009/0326809 A1 12/2009 Colley
 2010/0013703 A1 1/2010 Tekawy
 2010/0026569 A1 2/2010 Amidi
 2010/0030470 A1 2/2010 Wang
 2010/0039316 A1 2/2010 Gronemeyer
 2010/0039318 A1 2/2010 Kmiecik
 2010/0039320 A1 2/2010 Boyer
 2010/0039321 A1 2/2010 Abraham
 2010/0060518 A1 3/2010 Bar-Sever
 2010/0063649 A1 3/2010 Wu
 2010/0084147 A1 4/2010 Aral
 2010/0085249 A1 4/2010 Ferguson
 2010/0085253 A1 4/2010 Ferguson
 2010/0103033 A1 4/2010 Roh
 2010/0103034 A1 4/2010 Tobe
 2010/0103038 A1 4/2010 Yeh
 2010/0103040 A1 4/2010 Broadbent
 2010/0106414 A1 4/2010 Whitehead
 2010/0106445 A1 4/2010 Kondoh
 2010/0109944 A1 5/2010 Whitehead
 2010/0109945 A1 5/2010 Roh
 2010/0109947 A1 5/2010 Rintanen
 2010/0109948 A1 5/2010 Razoumov
 2010/0109950 A1 5/2010 Roh
 2010/0111372 A1 5/2010 Zheng
 2010/0114483 A1 5/2010 Heo
 2010/0117894 A1 5/2010 Velde
 2010/0117899 A1 5/2010 Papadmitratos
 2010/0117900 A1 5/2010 van Diggelen
 2010/0124210 A1 5/2010 Lo
 2010/0124212 A1 5/2010 Lo
 2010/0134354 A1 6/2010 Lennen
 2010/0149025 A1 6/2010 Meyers
 2010/0149030 A1 6/2010 Verma
 2010/0149033 A1 6/2010 Abraham

2010/0149034 A1 6/2010 Chen
 2010/0149037 A1 6/2010 Cho
 2010/0150284 A1 6/2010 Fielder
 2010/0152949 A1 6/2010 Nunan
 2010/0156709 A1 6/2010 Zhang
 2010/0156712 A1 6/2010 Pisz
 2010/0156718 A1 6/2010 Chen
 2010/0159943 A1 6/2010 Salmon
 2010/0161179 A1 6/2010 McClure
 2010/0161211 A1 6/2010 Chang
 2010/0161568 A1 6/2010 Xiao
 2010/0171660 A1 7/2010 Shyr
 2010/0171757 A1 7/2010 Melamed
 2010/0185364 A1 7/2010 McClure
 2010/0185366 A1 7/2010 Heiniger
 2010/0185389 A1 7/2010 Woodard
 2010/0188285 A1 7/2010 Collins
 2010/0188286 A1 7/2010 Bickerstaff
 2010/0189163 A1 7/2010 Burgi
 2010/0207811 A1 8/2010 Lackey
 2010/0210206 A1 8/2010 Young
 2010/0211248 A1 8/2010 Craig
 2010/0211315 A1 8/2010 Toda
 2010/0211316 A1 8/2010 DaSilva
 2010/0220004 A1 9/2010 Malko
 2010/0220008 A1 9/2010 Conover
 2010/0222076 A1 9/2010 Poon
 2010/0225537 A1 9/2010 Abraham
 2010/0228408 A1 9/2010 Lithgow
 2010/0231443 A1 9/2010 Whitehead
 2010/0231446 A1 9/2010 Marshall
 2010/0232351 A1 9/2010 Chansarkar
 2010/0235093 A1 9/2010 Chang
 2010/0238976 A1 9/2010 Young
 2010/0241347 A1 9/2010 King
 2010/0241353 A1 9/2010 Park
 2010/0241441 A1 9/2010 Page
 2010/0241864 A1 9/2010 Kelley
 2010/0312428 A1 12/2010 Roberge et al.
 2011/0001668 A1 1/2011 Cobb et al.
 2011/0015817 A1 1/2011 Reeve
 2011/0018765 A1 1/2011 Whitehead et al.
 2011/0054729 A1 3/2011 Whitehead et al.
 2011/0231061 A1 9/2011 Reeve et al.
 2011/0264307 A1 10/2011 Guyette et al.
 2011/0266357 A1* 11/2011 Orcutt et al. 239/1
 2011/0270495 A1 11/2011 Knapp
 2011/0305260 A1 12/2011 McManus
 2011/0309974 A1 12/2011 Williamson
 2012/0034940 A1 2/2012 Ismail
 2012/0116676 A1 5/2012 Basnayake et al.
 2012/0127032 A1 5/2012 McClure
 2012/0174445 A1 7/2012 Jones et al.
 2012/0215410 A1 8/2012 McClure et al.
 2012/0306692 A1 12/2012 Werner
 2013/0069821 A1 3/2013 Jambulingam et al.
 2013/0069822 A1 3/2013 Wu et al.
 2013/0107034 A1 5/2013 Di Bernardo et al.
 2013/0124055 A1 5/2013 Baurer et al.
 2013/0179026 A1 7/2013 McClure et al.
 2013/0179204 A1 7/2013 Sabarez, II
 2015/0375247 A1 12/2015 Funseth et al.

FOREIGN PATENT DOCUMENTS

CA 2218642 4/1998
 CN 2562886 Y 7/2003
 CN 102613161 6/2013
 EP 2837285 2/2015
 GB 2347368 A * 9/2000
 JP 07244150 9/1995
 JP 2004008187 1/2004
 JP 4001933 10/2007
 WO WO9515499 6/1995
 WO 1998/036288 A1 8/1998
 WO 2000/024239 A1 5/2000
 WO 2002/080652 A1 10/2002
 WO 2005/119386 A1 12/2005
 WO 2003/019430 A1 3/2006

(56)

References Cited

FOREIGN PATENT DOCUMENTS

WO	2007/139467	A1	12/2007
WO	WO-2007139467		12/2007
WO	2008/080193	A1	7/2008
WO	WO-2008080193		7/2008
WO	2009/066183	A2	5/2009
WO	2009/082745	A1	7/2009
WO	2009/126587	A1	10/2009
WO	WO-2009126587		10/2009
WO	2009/148638	A1	12/2009
WO	WO-2010005945		1/2010
WO	2010/042131	A1	4/2010
WO	2010/005945	A1	7/2010
WO	2010/104782	A1	9/2010
WO	2011/014431	A1	2/2011
WO	WO-2011014431		2/2011

OTHER PUBLICATIONS

“Eurocontrol, Pegasus Technical Notes on SBAS”, report [online] Dec. 7, 2004 [retrieved on May 18, 2010], Retrieved from the Internet: <<http://www.icao.int/icao/en/ro/nacc/meetings/2004/gnss/documentation/Pegasus/tn.pdg>> (Dec. 7, 2004), p. 89, paras [0001]-[0004].

“ISO”, 11783 Part 7 Draft Amendment 1 Annex, Paragraphs B.6 and B.7 ISO 11783-7 2004 DAM1, ISO; Mar. 8, 2004.

“Orthman Manufacturing Co., www.orthman.com/htm/guidance.htm regarding the “Tracer Quick-Hitch” <http://www.patentstorm.us/class/342/357.11-Attitude_determination.html>.

Bevly, David M., “Comparison of INS v. Carrier-Phase DGPS for Attitude Determination in the Control of Off-Road Vehicles”; ION 55th Annual Meeting; Jun. 28-30, 1999; Cambridge, Massachusetts; pp. 497-504. x Jun. 28, 1999.

Han, Shaowel, et al., “Single-Epoch Ambiguity Resolution for Real-Time GPS Attitude Determination with the Aid of One-Dimensional Optical Fiber Gyro”, GPS Solutions, vol. 3, No. 1, pp. 5-12 (1999) John Wiley & Sons, Inc.

International Search Report and Written Opinion for PCT/IB2008/003796 dated Jul. 15, 2009.

International Search Report and Written Opinion for PCT/US08/88070 dated Feb. 9, 2009.

International Search Report and Written Opinion for PCT/US09/63594 dated Jan. 11, 2010.

International Search Report and Written Opinion for PCT/US10/21334 dated Mar. 12, 2010.

International Search Report and Written Opinion for PCT/US10/26509 dated Apr. 20, 2010; 7 pages.

International Search Report and Written Opinion for PCT/US2004/015678 dated Jun. 21, 2005.

International Search Report and Written Opinion for PCT/US2008/081727 dated Dec. 23, 2008.

International Search Report and Written Opinion for PCT/US2010/043094 dated Sep. 17, 2010.

International Search Report for PCT/AU2008/000002 dated Feb. 28, 2008.

International Search Report for PCT/US09/039686 dated May 26, 2009.

International Search Report for PCT/US09/067693 dated Jan. 26, 2010.

International Search Report for PCT/US09/33567 dated Feb. 9, 2009.

International Search Report for PCT/US09/33693 dated Mar. 30, 2009.

International Search Report for PCT/US09/34376 dated Nov. 2, 2009.

International Search Report for PCT/US09/49776 dated Aug. 11, 2009.

International Search Report for PCT/US09/60668 dated Dec. 9, 2009.

Irsigler, M. et al., “PPL Tracking Performance in the Presence of Oscillator Phase Noise” GPS Solutions, vol. 5, No. 4; pp. 45-57 (2002).

Kaplan, E.D., “Understanding GPS: Principles and Applications”, Artech House, MA, 1996.

Keicher, R. et al., “Automatic Guidance for Agricultural Vehicles in Europe”; Computers and Electronics in Agriculture, vol. 25; Jan. 2000; pp. 169-194.

Last, J.D., et al., “Effect of skywave interference on coverage of radio beacon DGPS Stations”, IEEE Proc.-Radar. Sonar Navig., vol. 144, No. 3, Jun. 1997; pp. 163-168.

Lin, Dai et al., “Real-Time Attitude Determination for Microsatellite by Lamda Method Combined with Kalman Filtering”, A Collection of the 22nd AIAA International Communications Satellite Systems Conference and Exhibit Technical Papers vol. 1, Monterey, CA, America Institute of Aeronautics and Astronautics, Inc. (May 2004); pp. 136-143.

Park, Chansik, et al., “Integer Ambiguity Resolution for GPS Based Attitude Determination System”, SICE 2009, Jul. 29-31, 2009; Chiba; pp. 1115-1120.

Parkinson, Bradford W., et al., “Global Positioning System: Theory and Applications, vol. II”, Bradford W. Parkinson and James J. Spiker, Jr., eds., Global Positioning System: Theory and Applications vol. II, 1995, AIAA, Reston, VA, USA, pp. 3-50 (1995), 3-50.

Rho, Hyundho et al., “Dual-Frequency GPS Precise Point Positioning with WADGPS Corrections” [retrieved on May 18, 2010] Retrieved from the Internet: <<http://gauss.gge.unb.ca/papers.pdf/iongnss2005.rho.wadgps.pdf>> (Jul. 12, 2006).

Schaer, et al, “Determination and Use of GPS Differential Code Bias Values”; Presentation [online]. Retrieved May 18, 2010. Retrieved from the Internet: <<http://nng.esoc.esa.de/ws206/REPR2.pdf>> (May 8, 2006).

Takac, Frank, et al., “Smark RTK: A Novel Method of Processing Standardized RTCM Network RTK Information for High Precision Positioning”; Proceedings of ENC GNSS 2008; Toulouse, France; Apr. 22, 2008.

Ward, Phillip W., “Performance Comparisons Between FLL, PLL and a Novel FLL-Assisted-PLL carrier Tracking Loop Under RF Interference Conditions” 11th Int. Tech Meeting of the Satellite Division of the US Inst. Of Navigation, Nashville, TN; Sep. 15-18, 1998; pp. 783-795.

Xu, Jiangning, et al., “An EHW Architecture for Real-Time GPS Attitude Determination Based on Parallel Genetic Algorithm”, The Computer Society Preceedings of the 2002 NASA/DOD Conference on Evolvable Hardware (EH’02) (2002).

Noh, Kwang-Mo, Self-tuning controller for farm tractor guidance, Iowa State University Retrospective Theses and Dissertations, Paper 9874, (1990).

Van Zuydam, R.P., Centimeter-Precision Guidance of Agricultural Implements in the Open Field by Means of Real Time Kinematic DGPS, ASA-CSSA-SSSA, pp. 1023-1034 (1999).

* cited by examiner

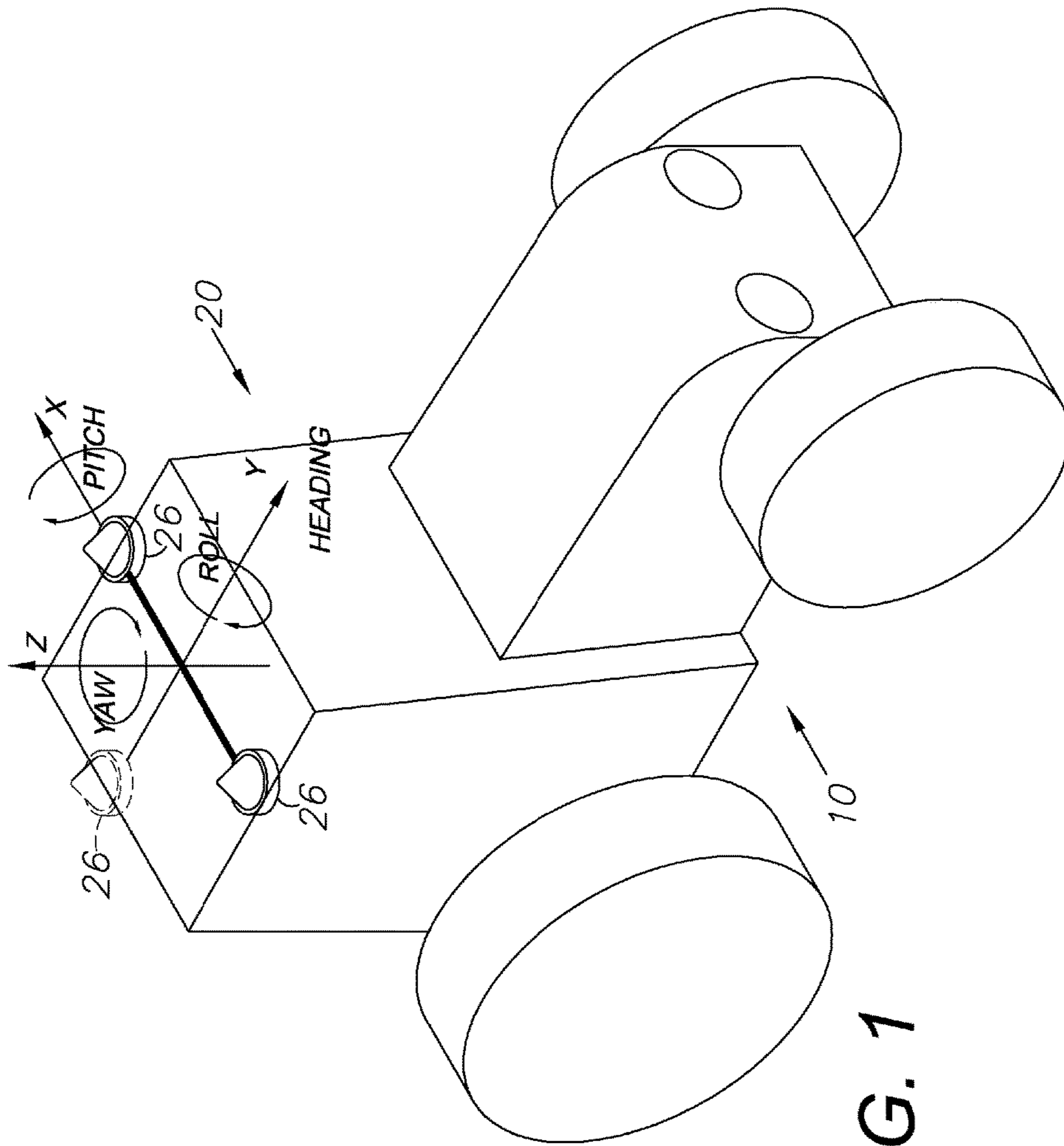


FIG. 1

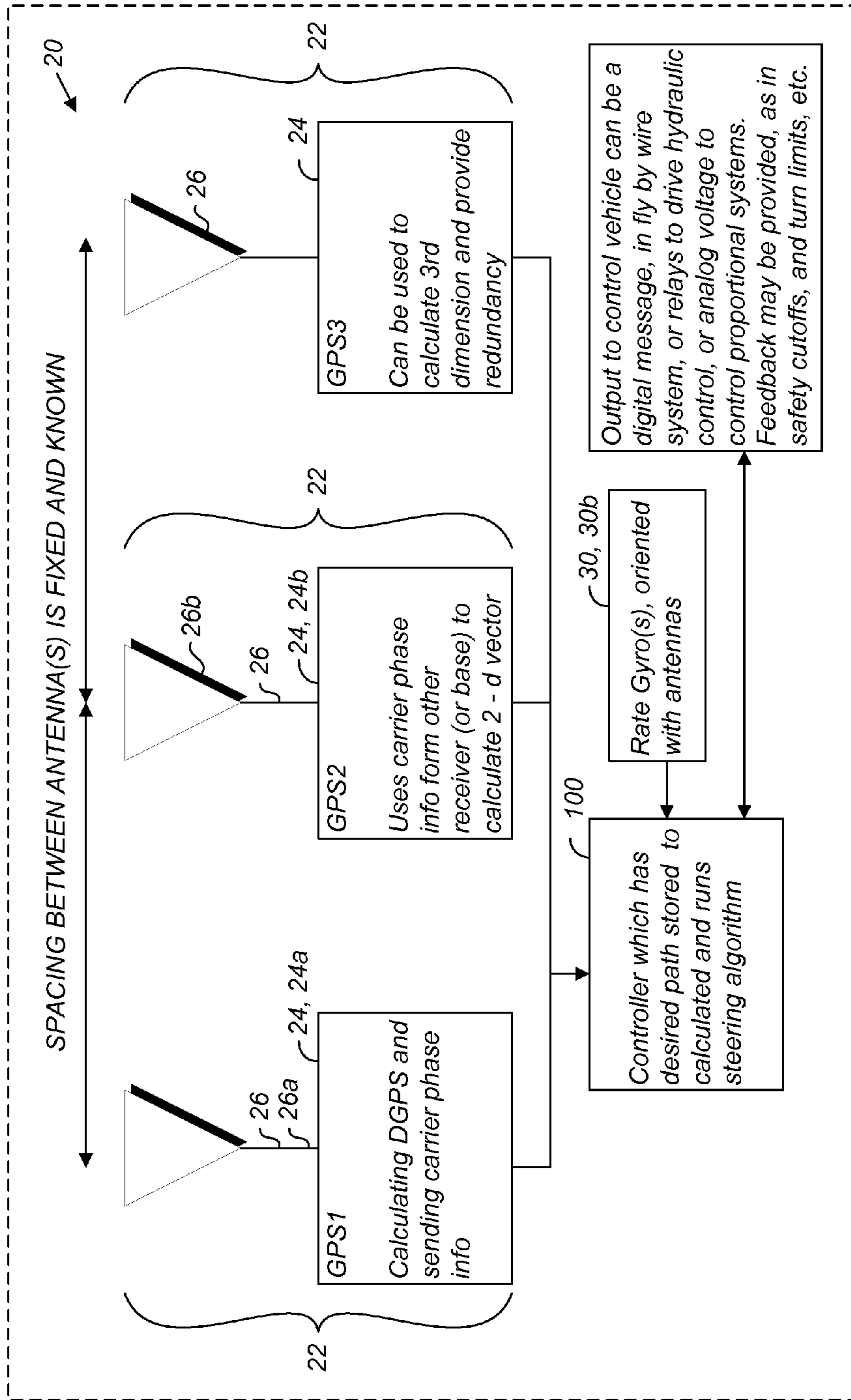


FIG. 2

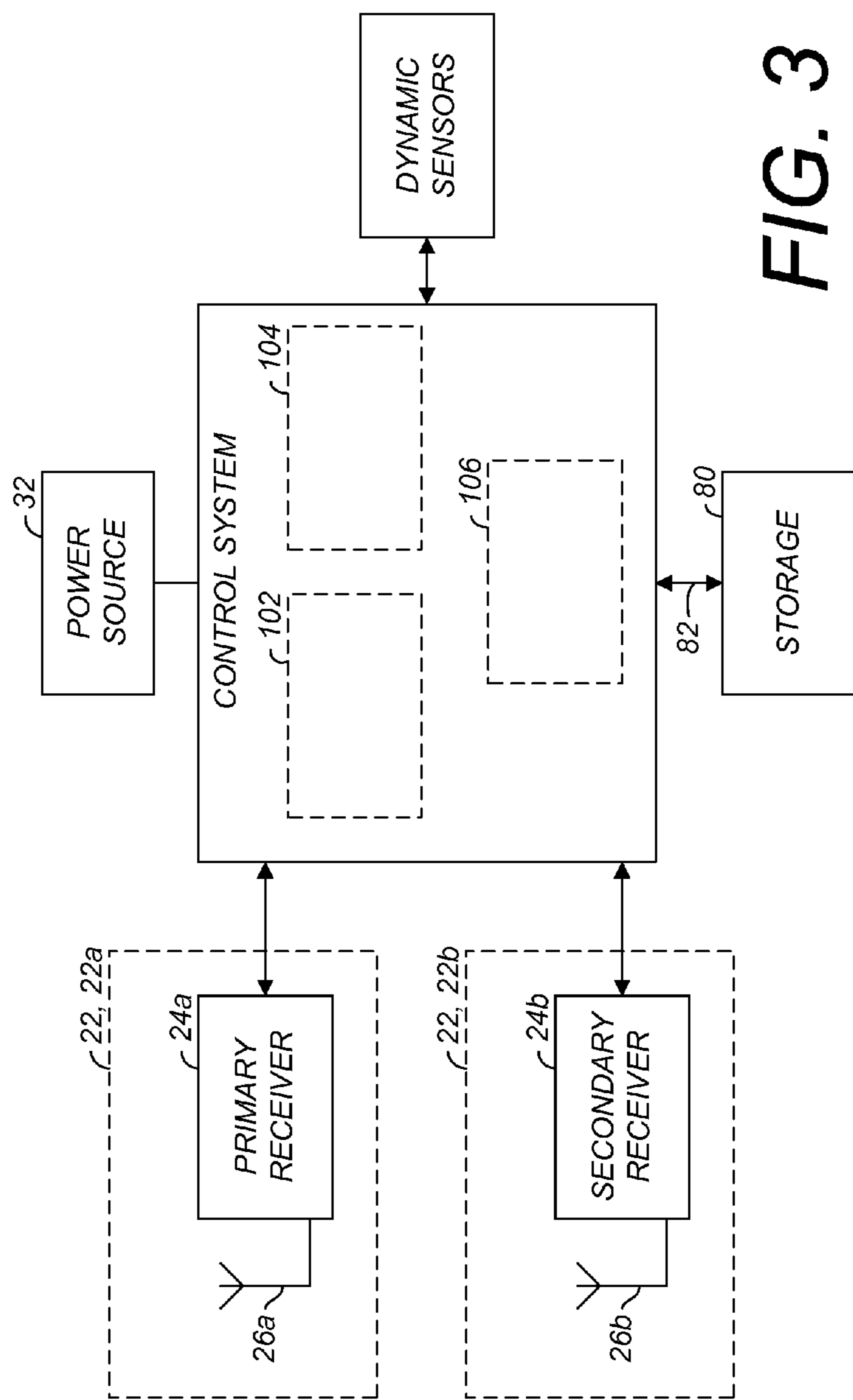


FIG. 3

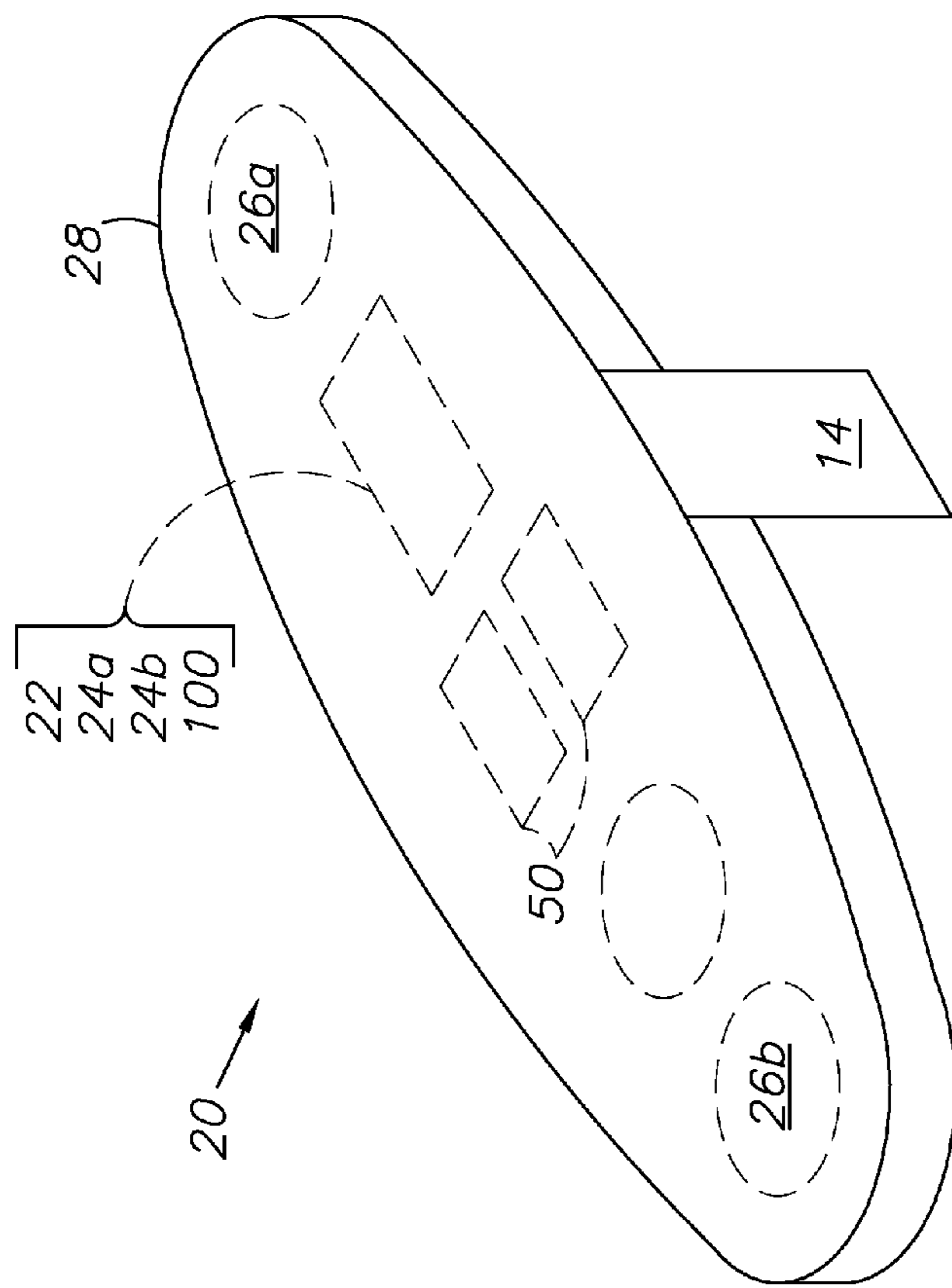


FIG. 4

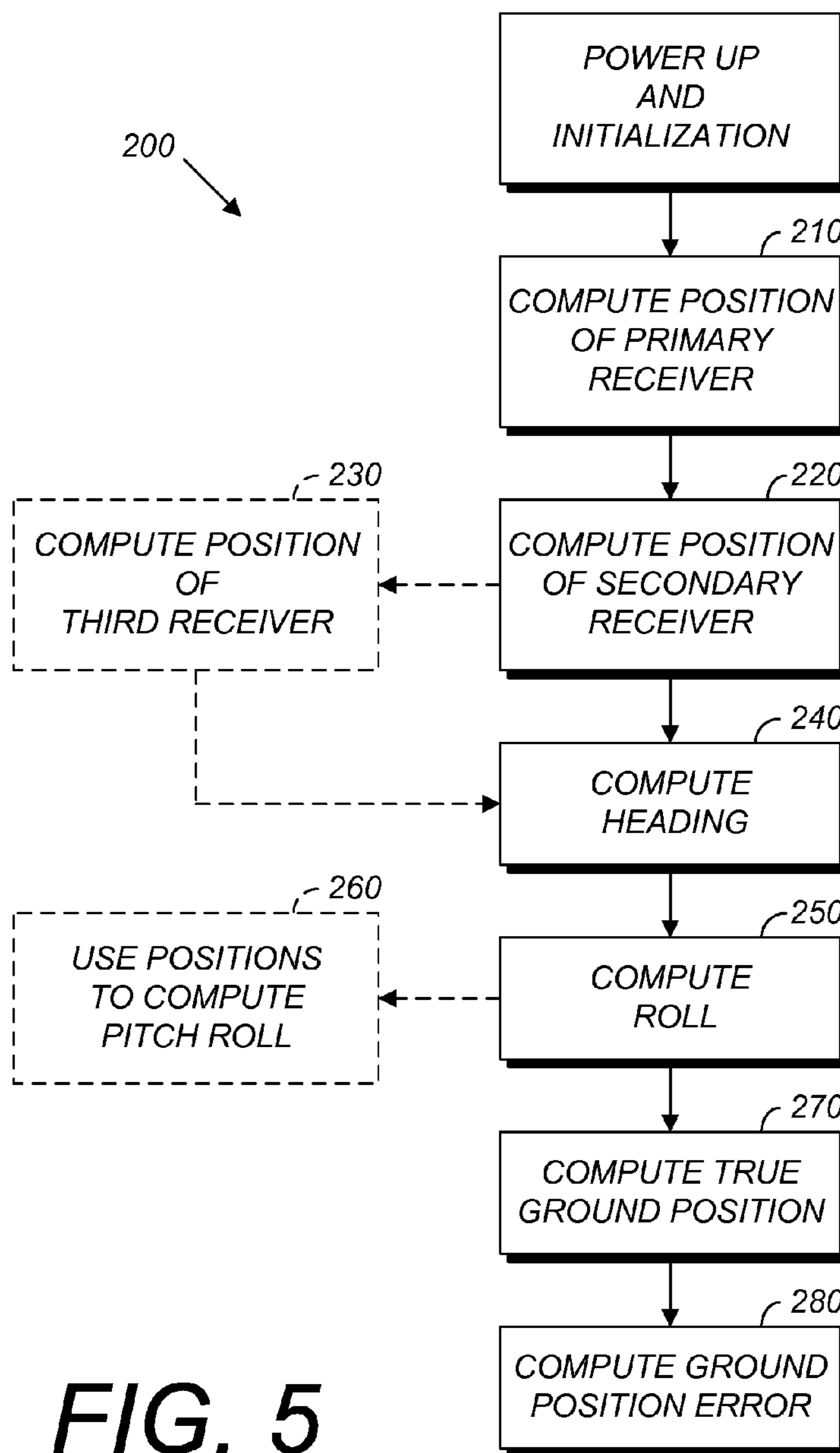


FIG. 5

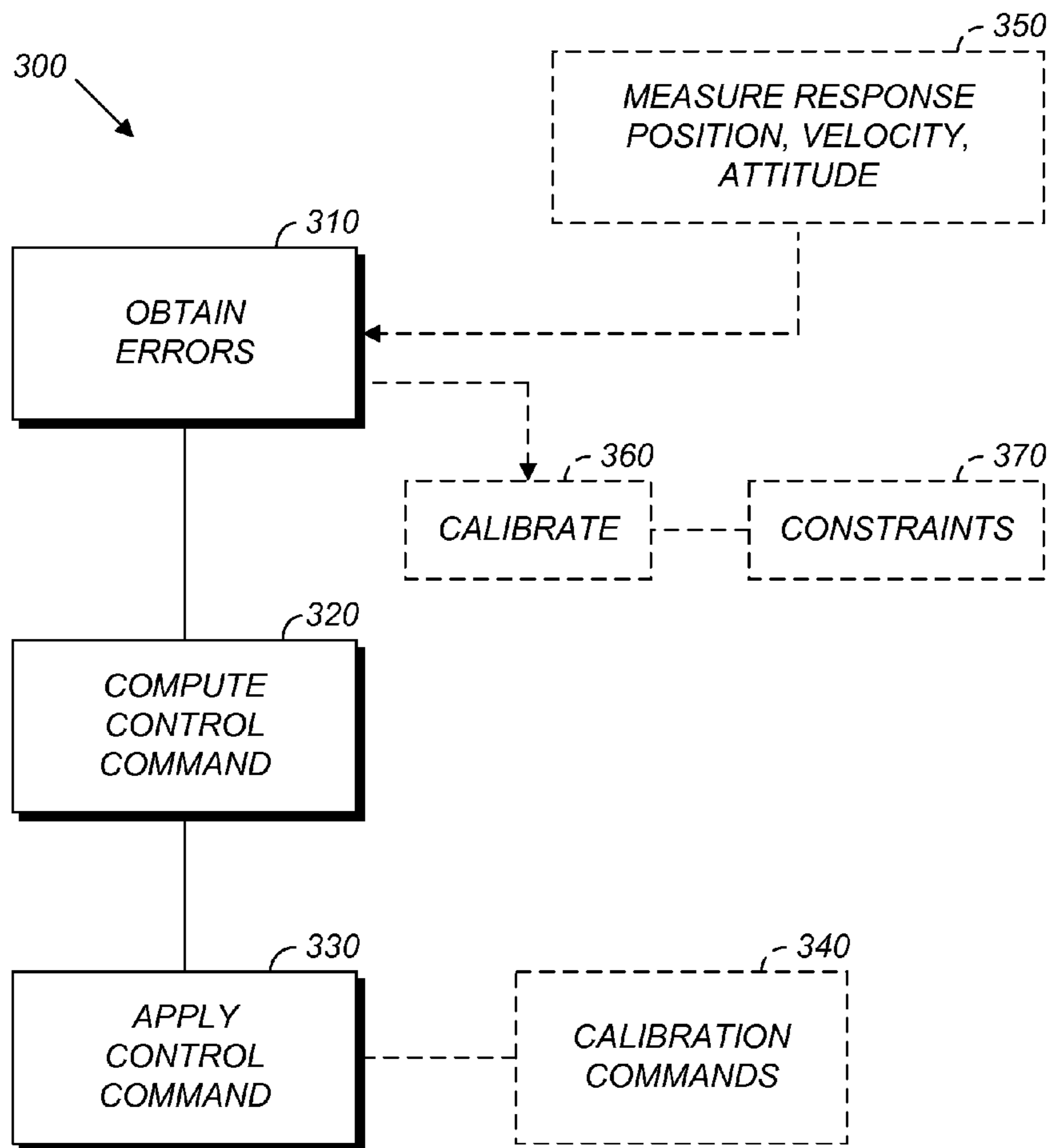


FIG. 6

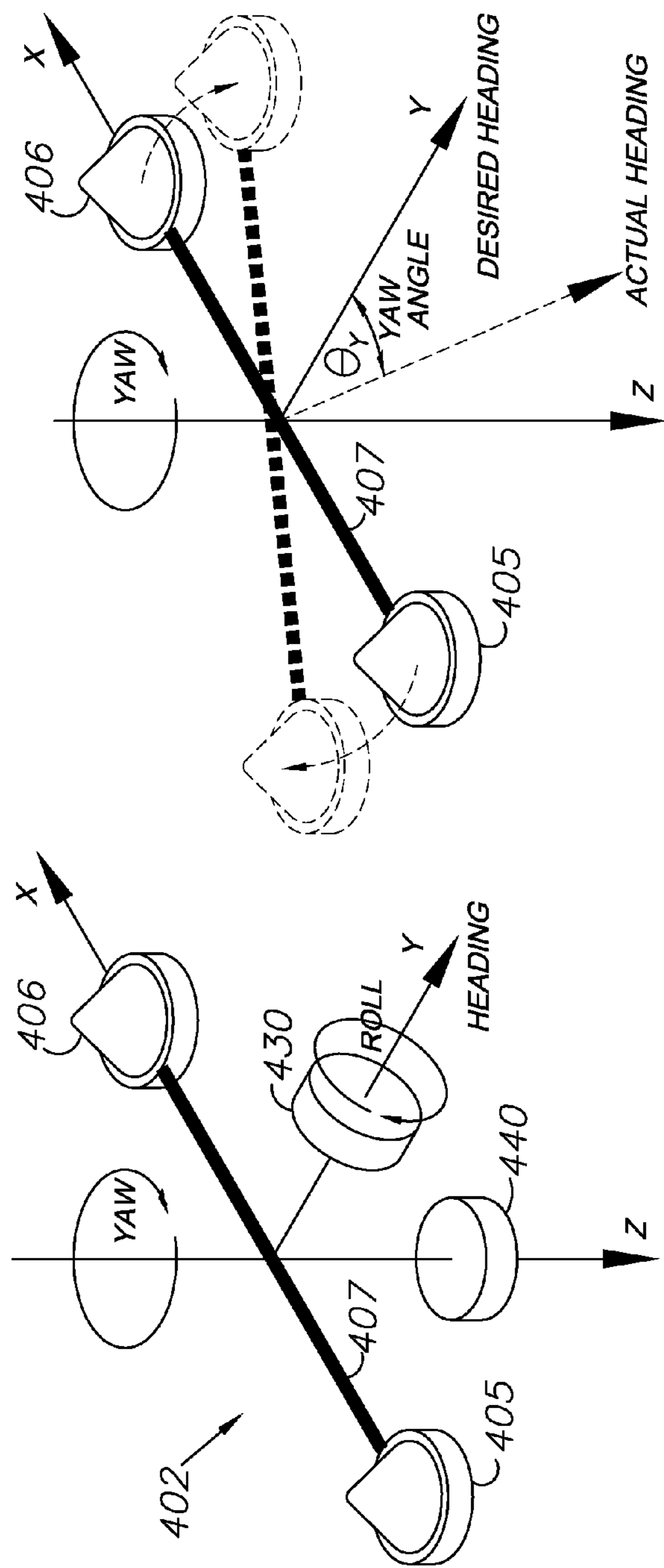


FIG. 7B
YAW ATTITUDE

FIG. 7A

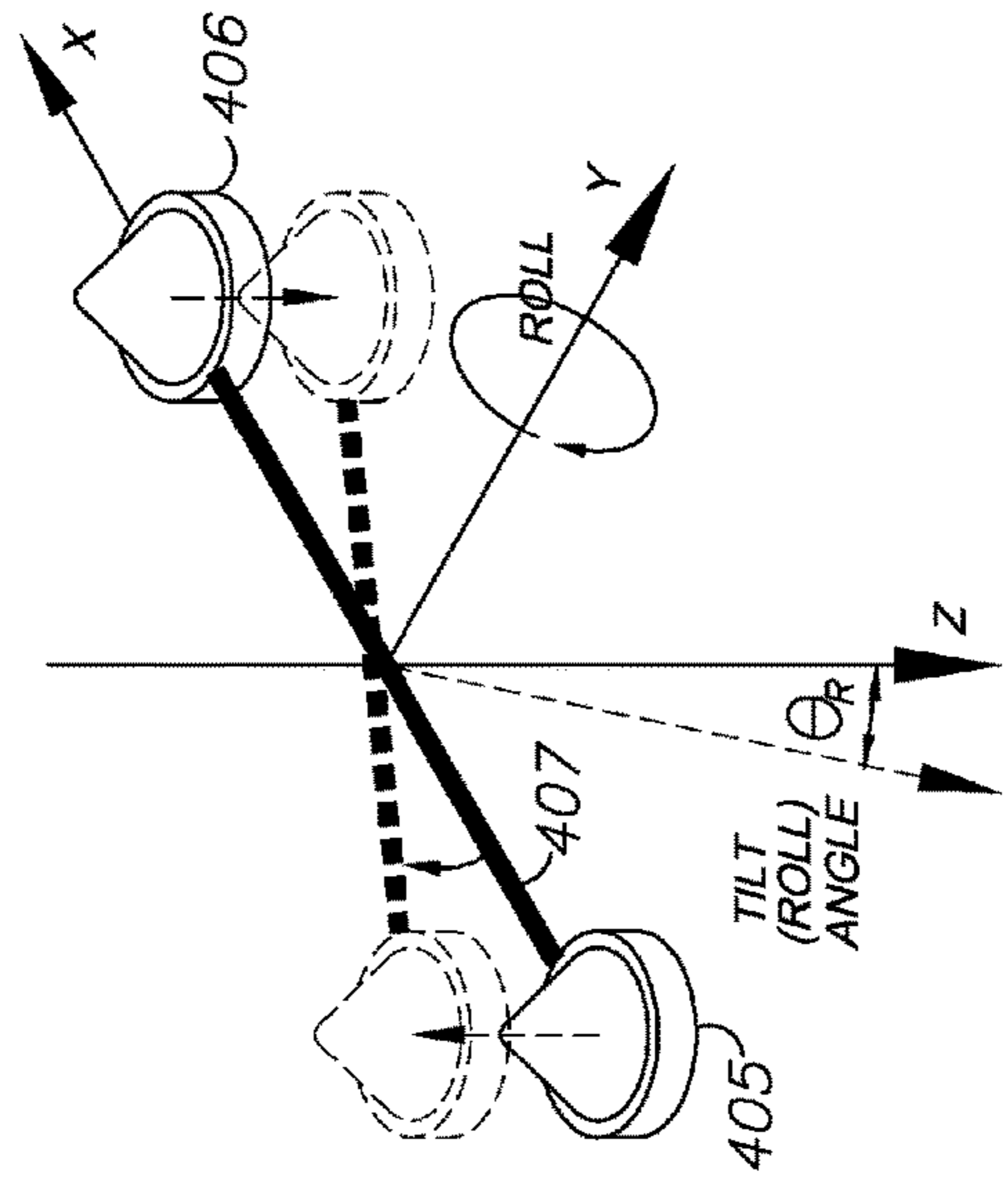


FIG. 7C
Tilt (Roll) Attitude

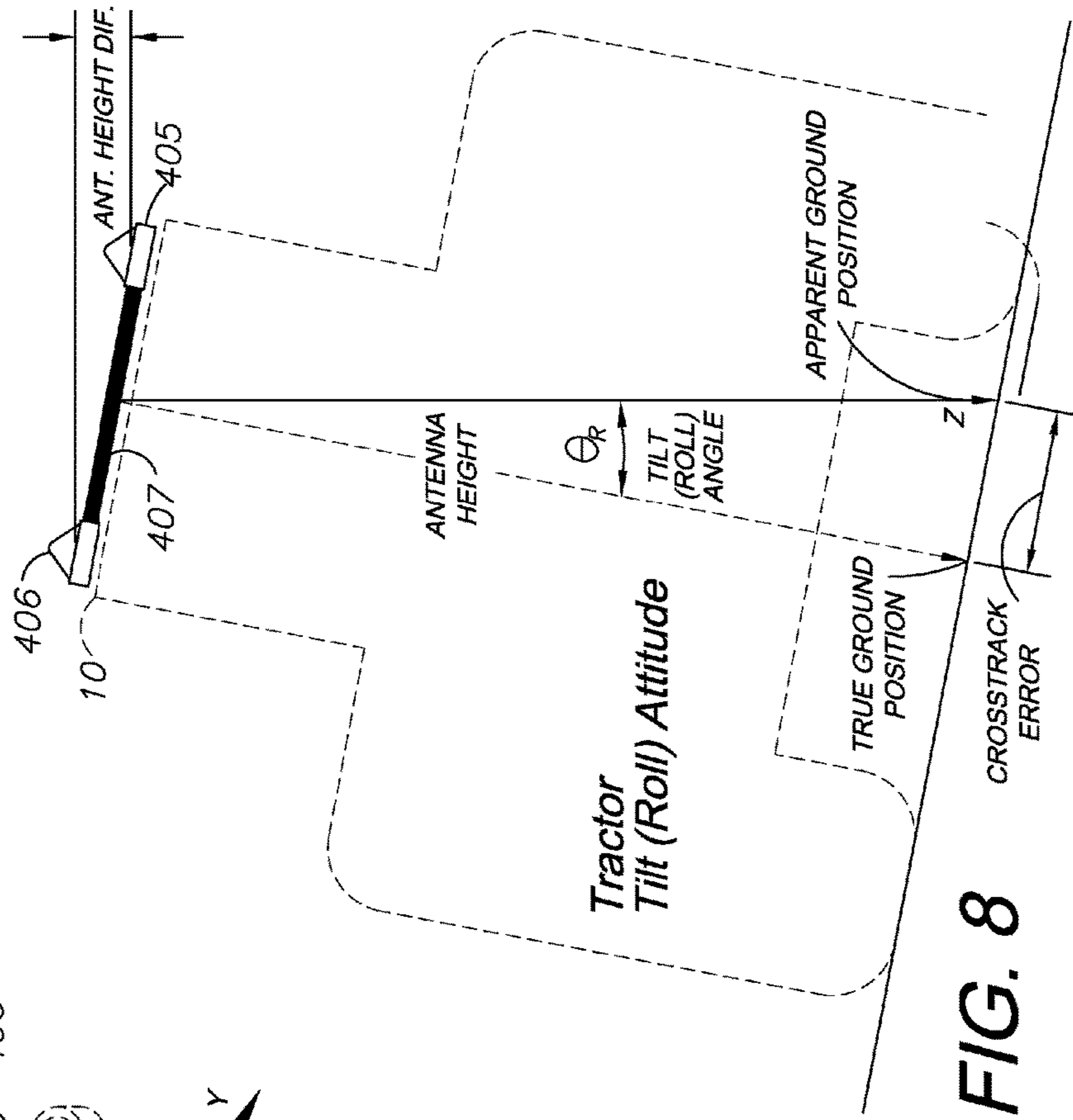


FIG. 8

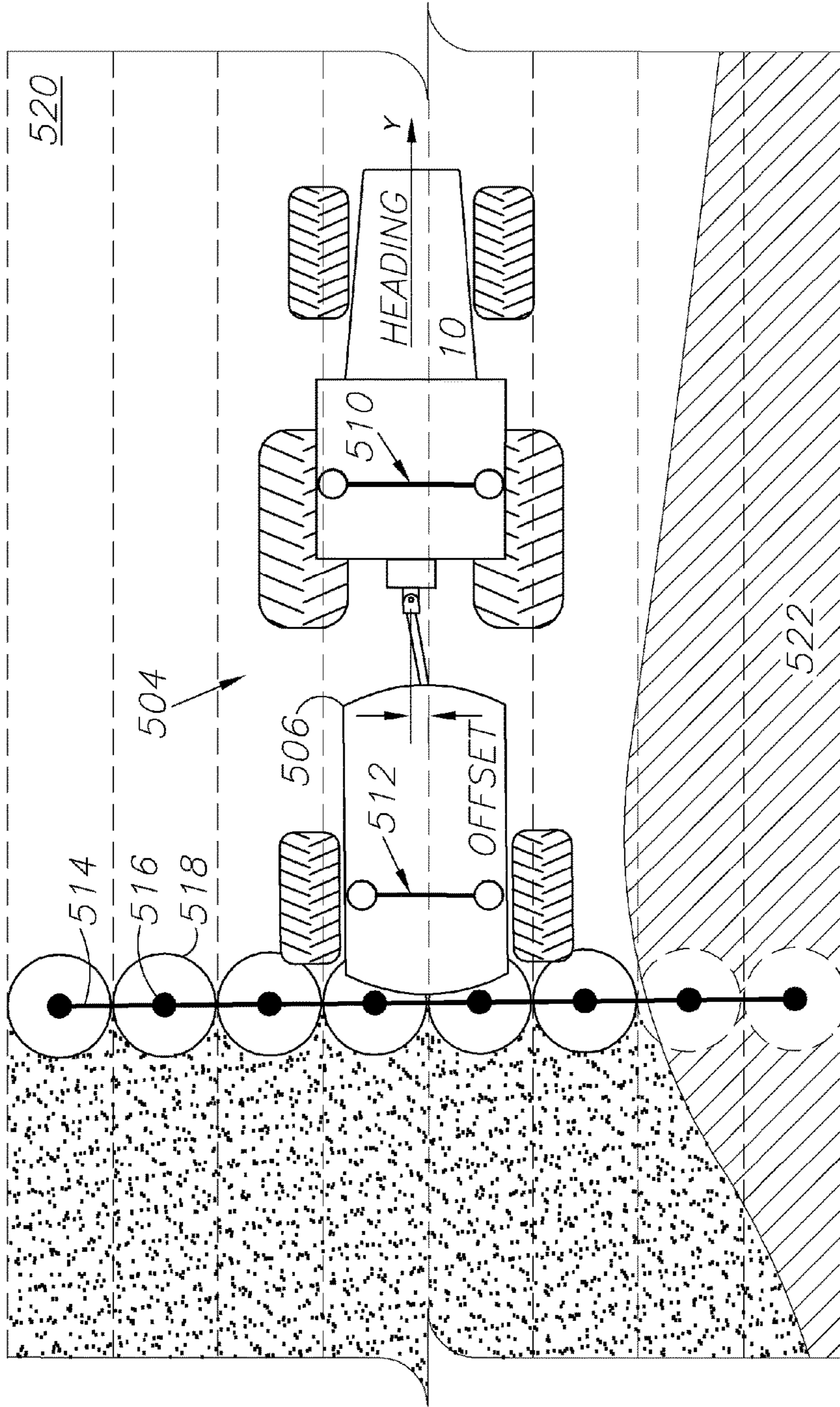


FIG. 9

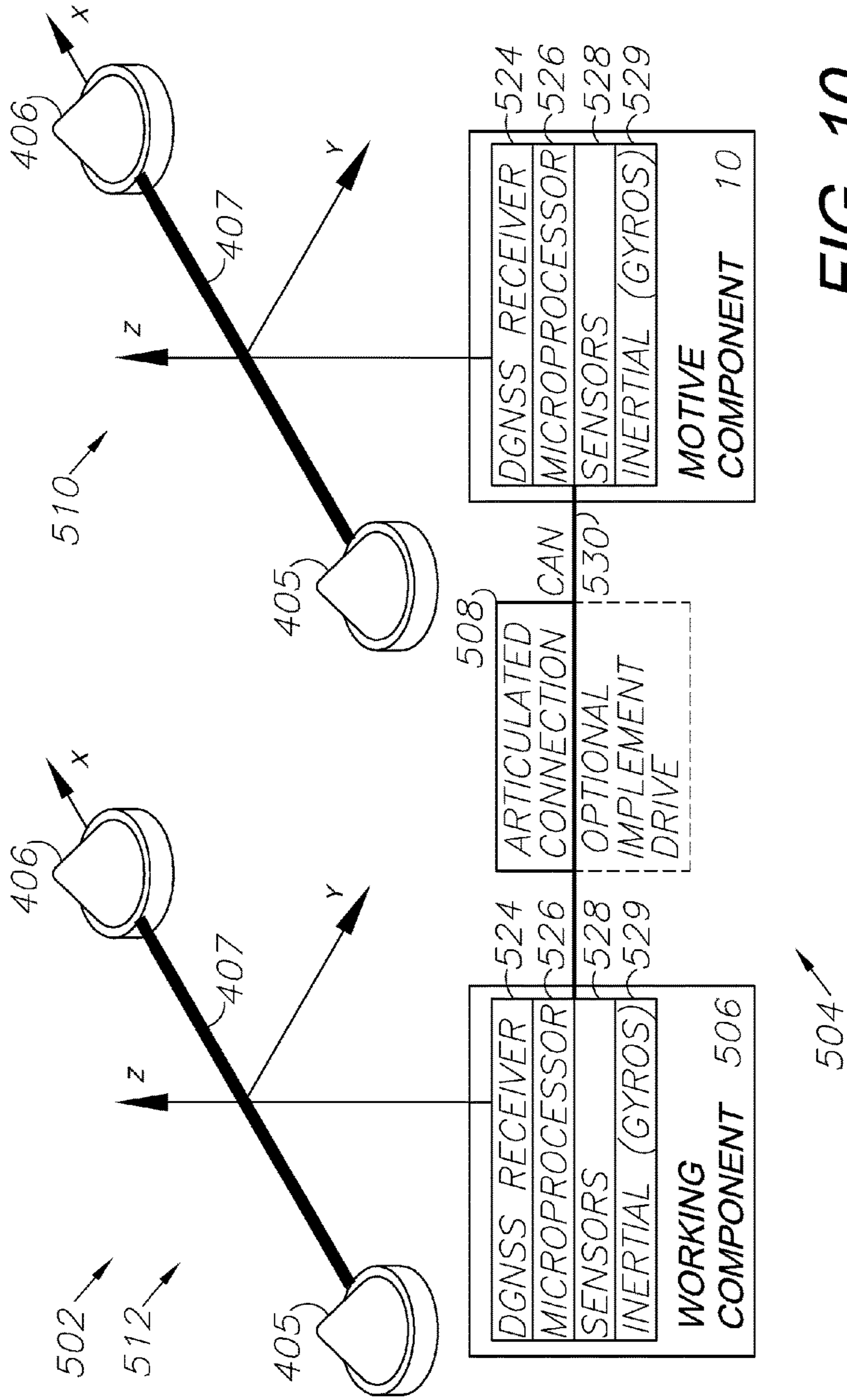


FIG. 10

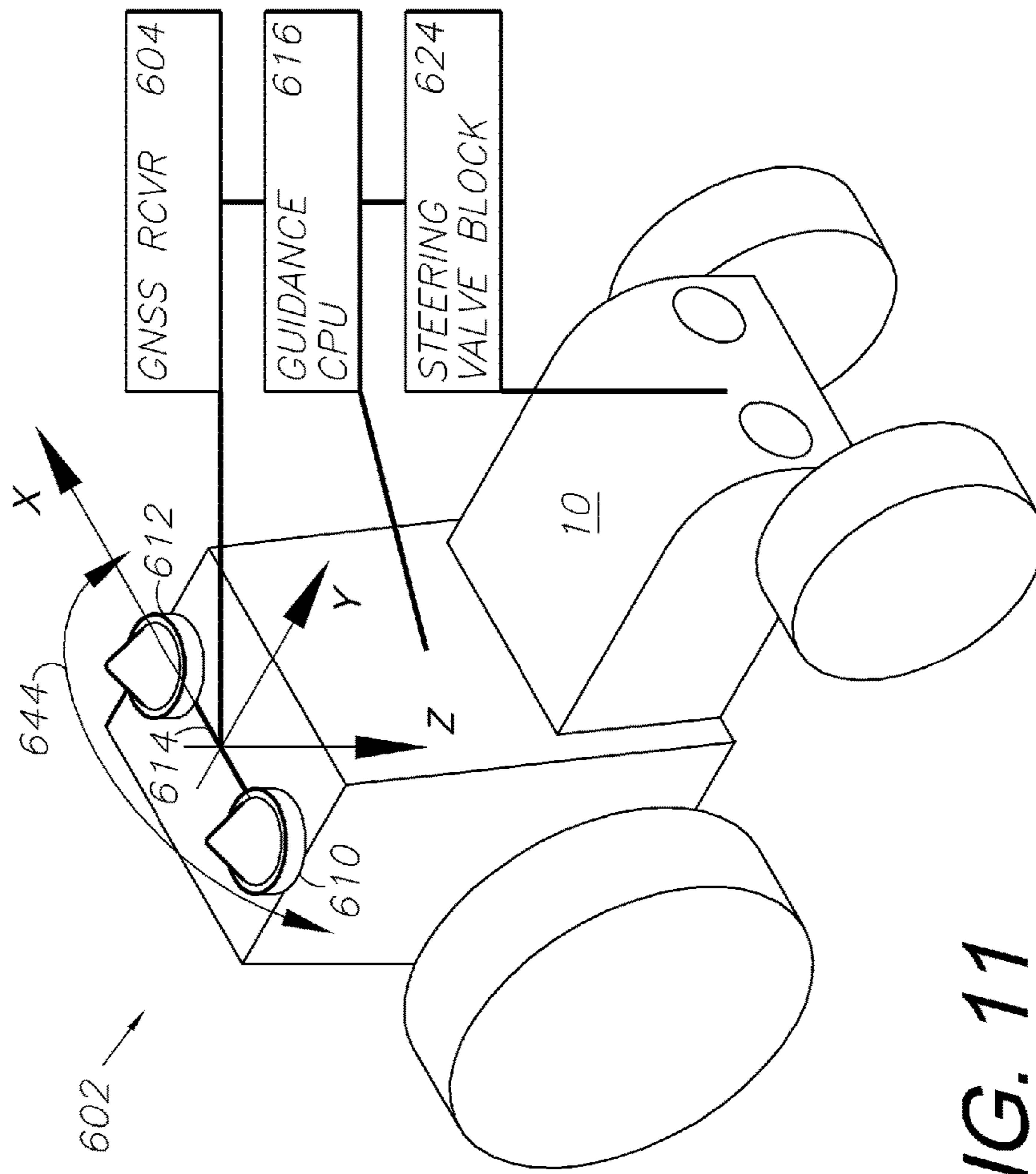
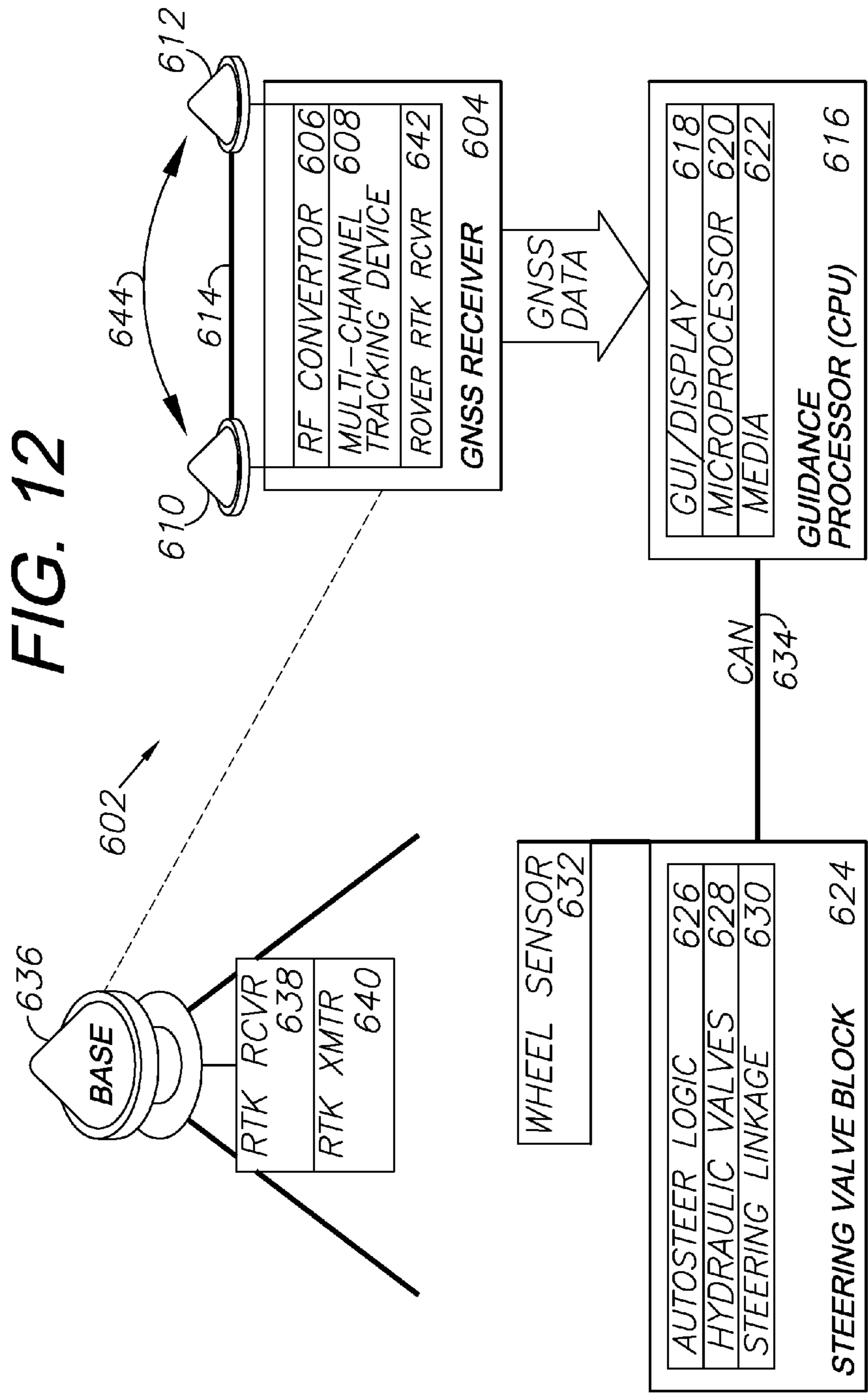


FIG. 11



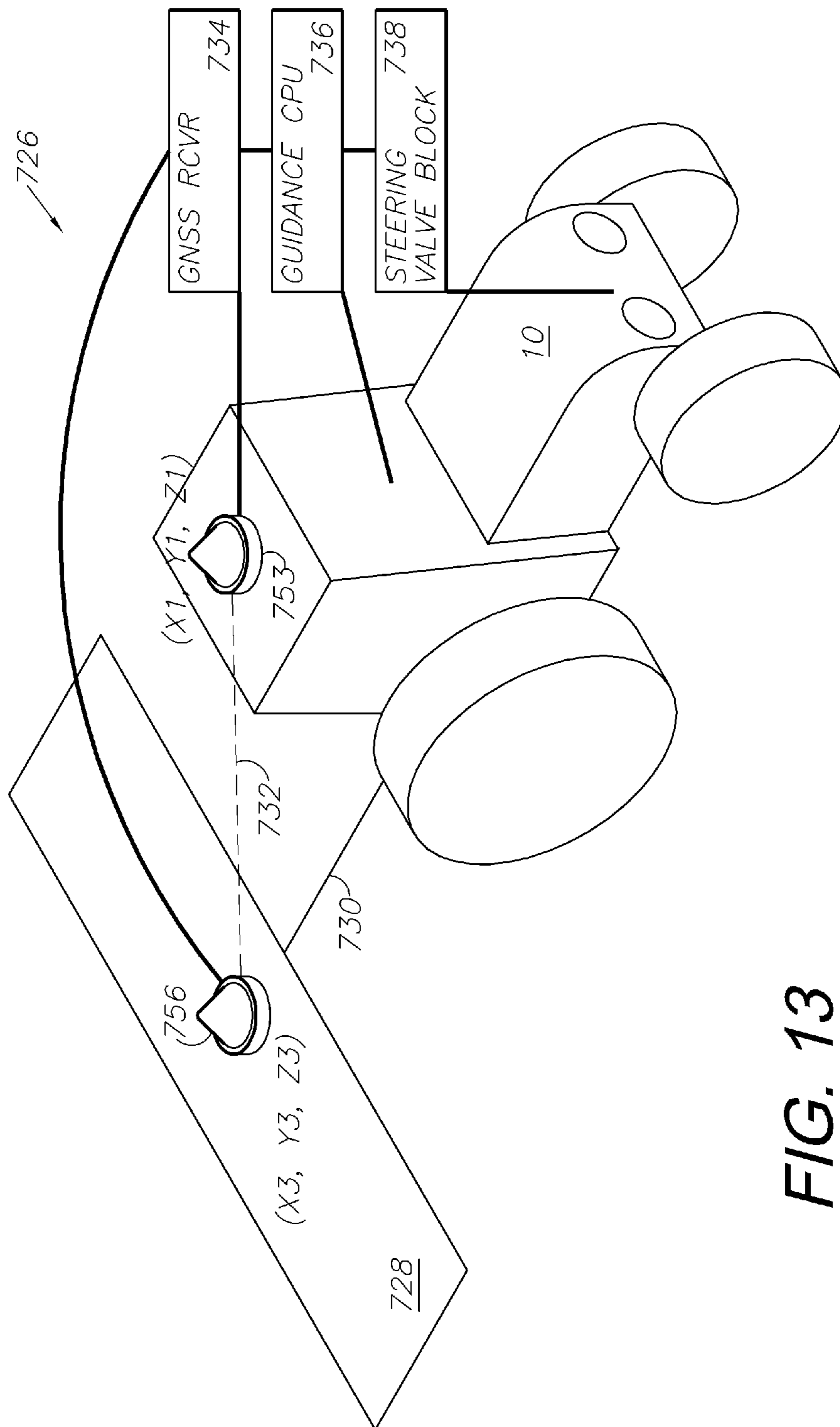


FIG. 13

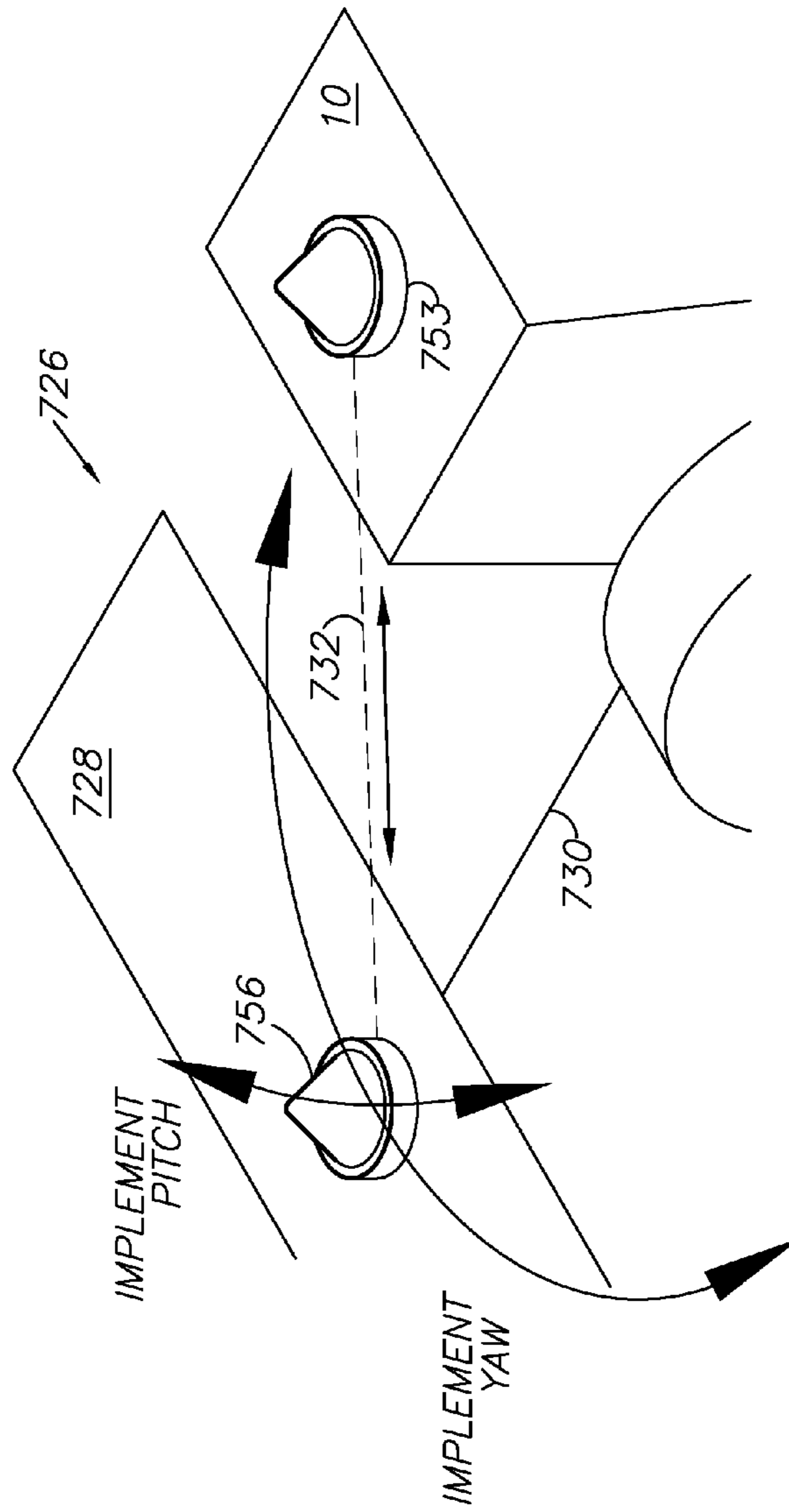


FIG. 14

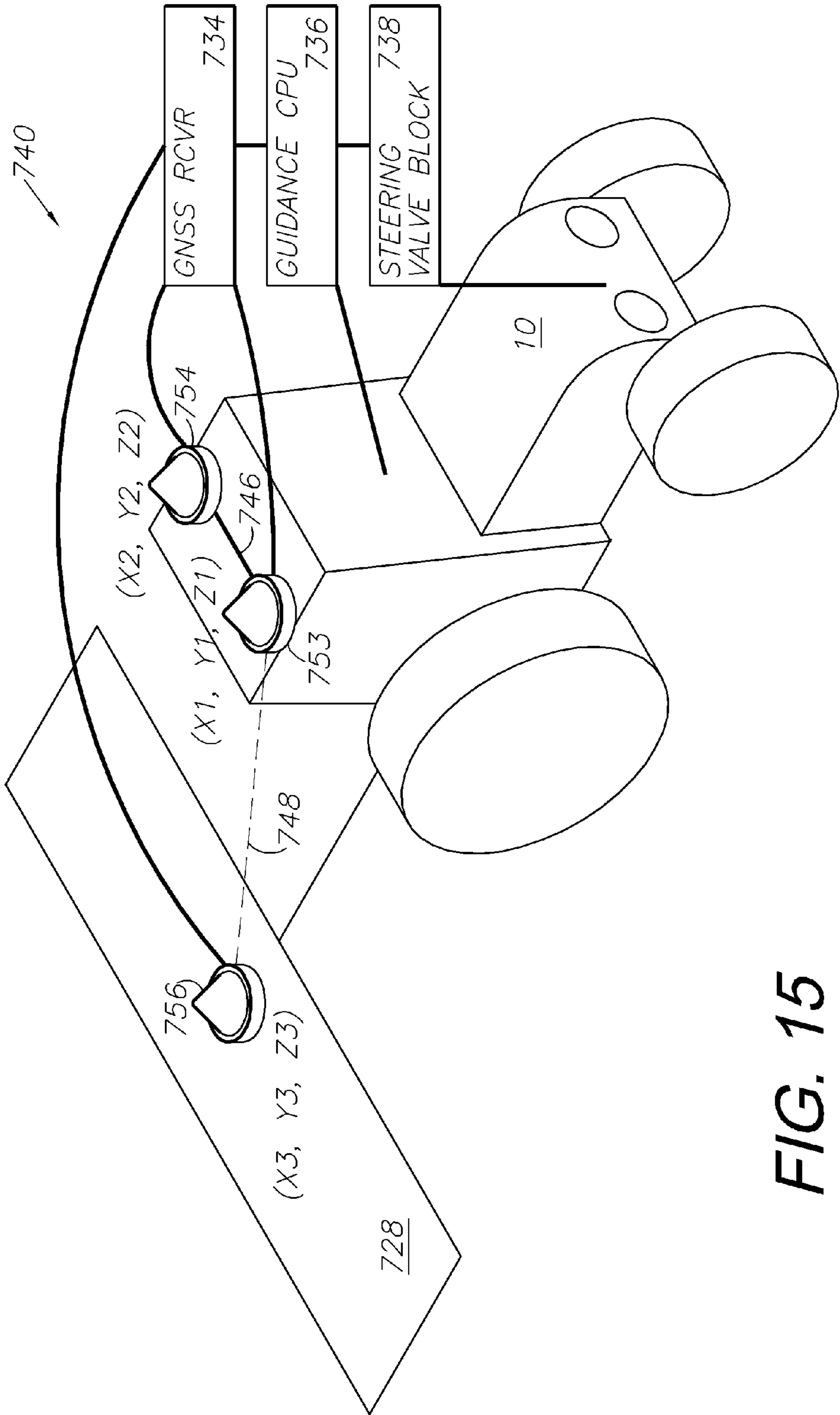


FIG. 15

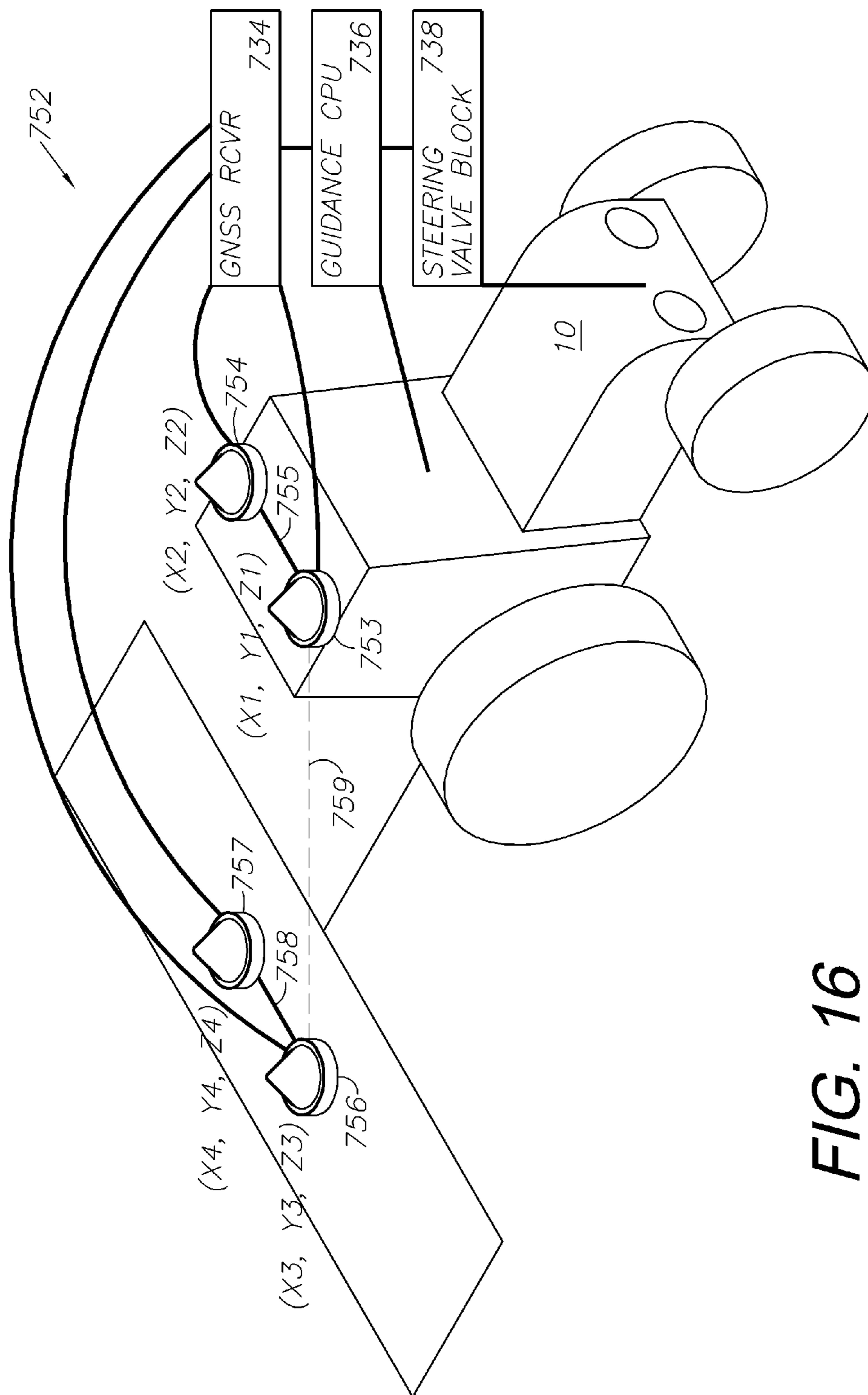


FIG. 16

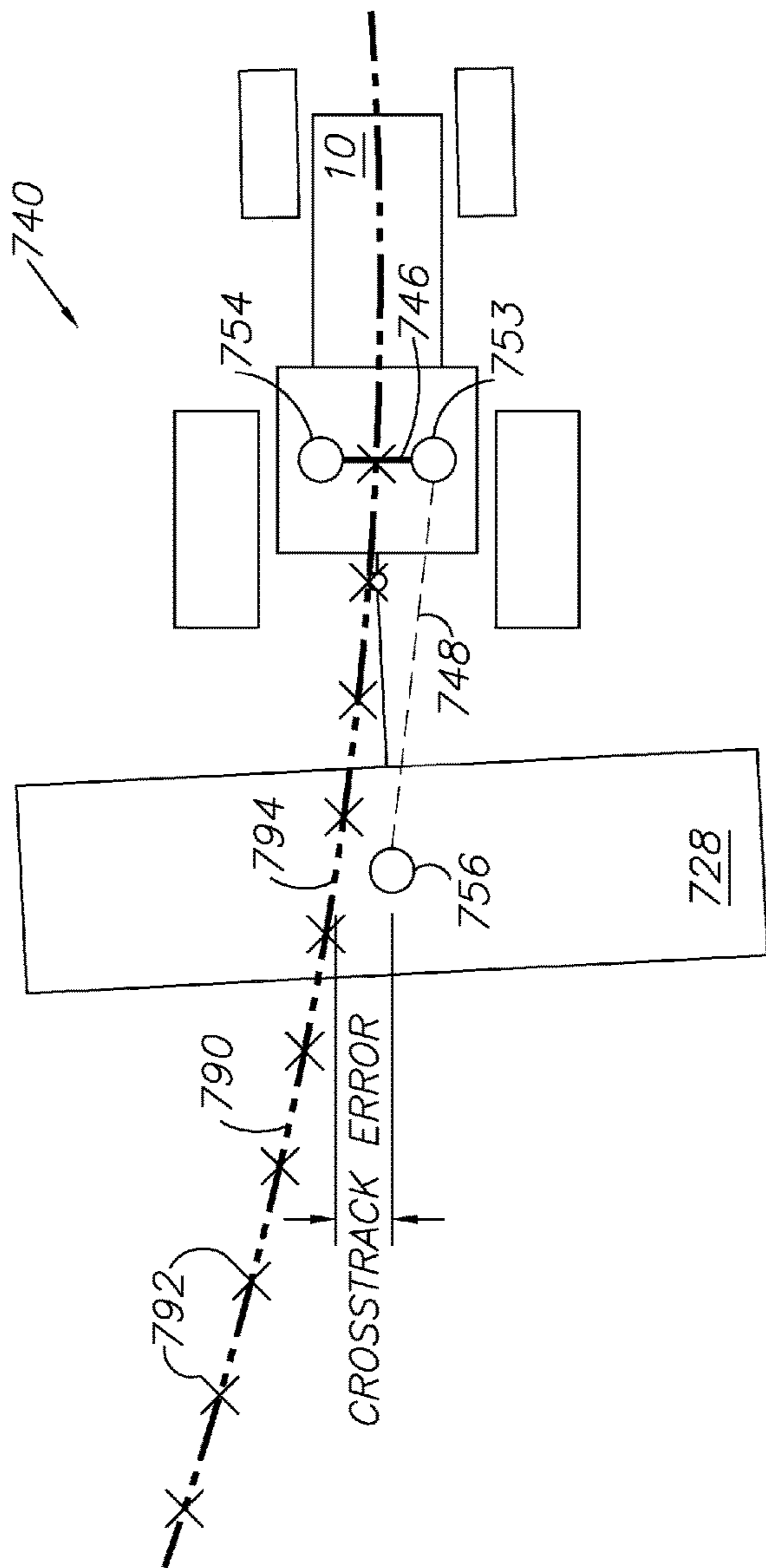
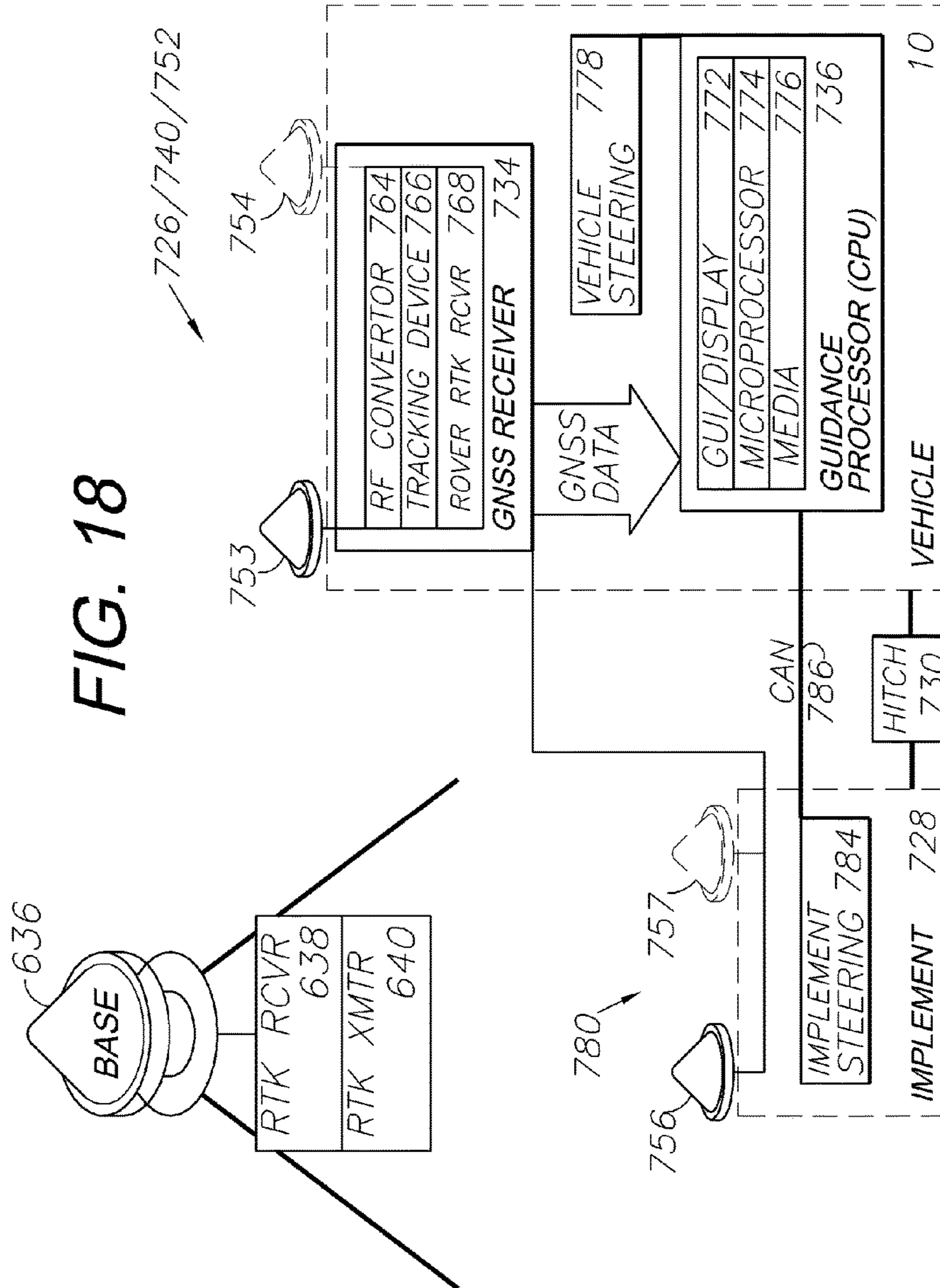
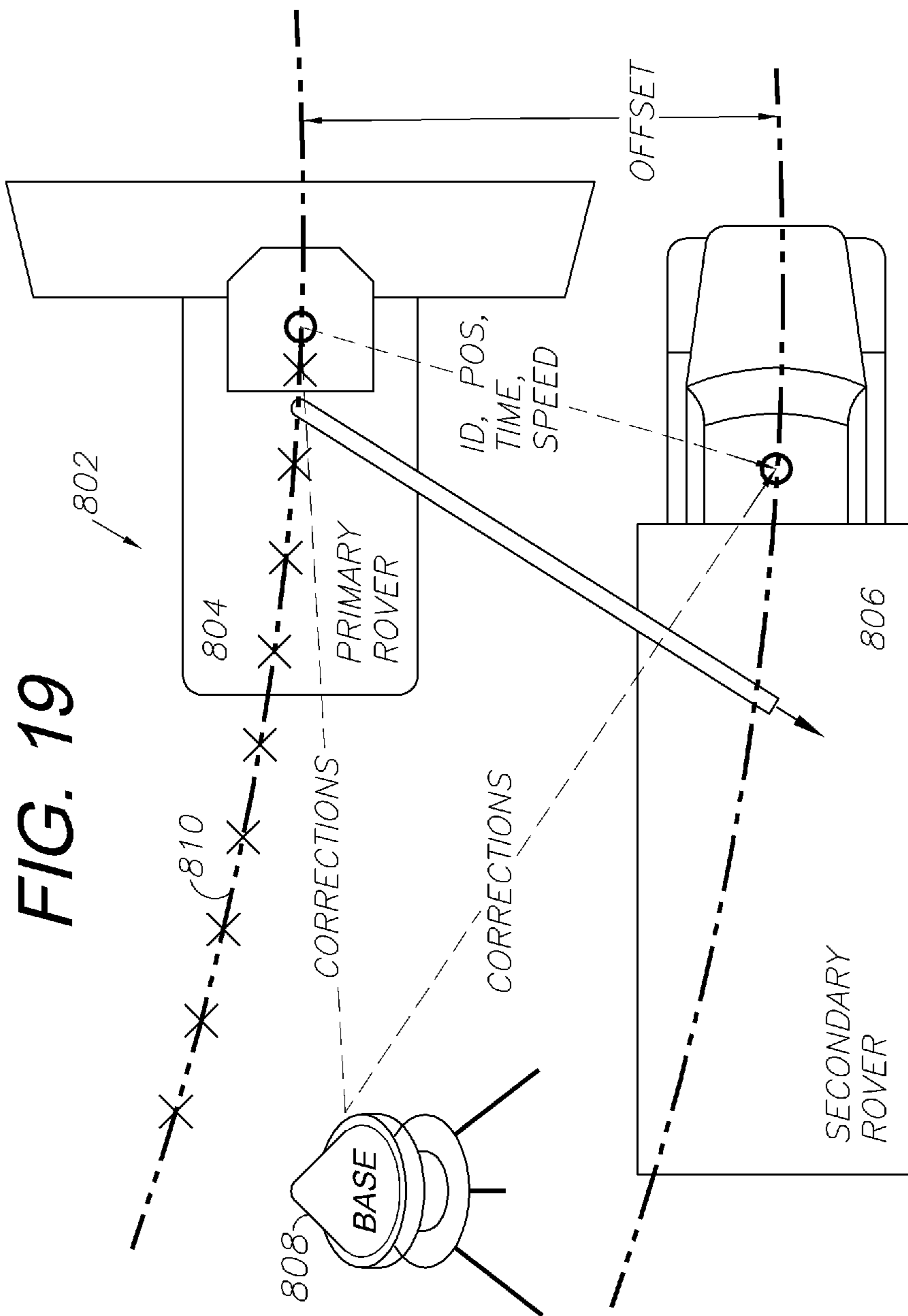
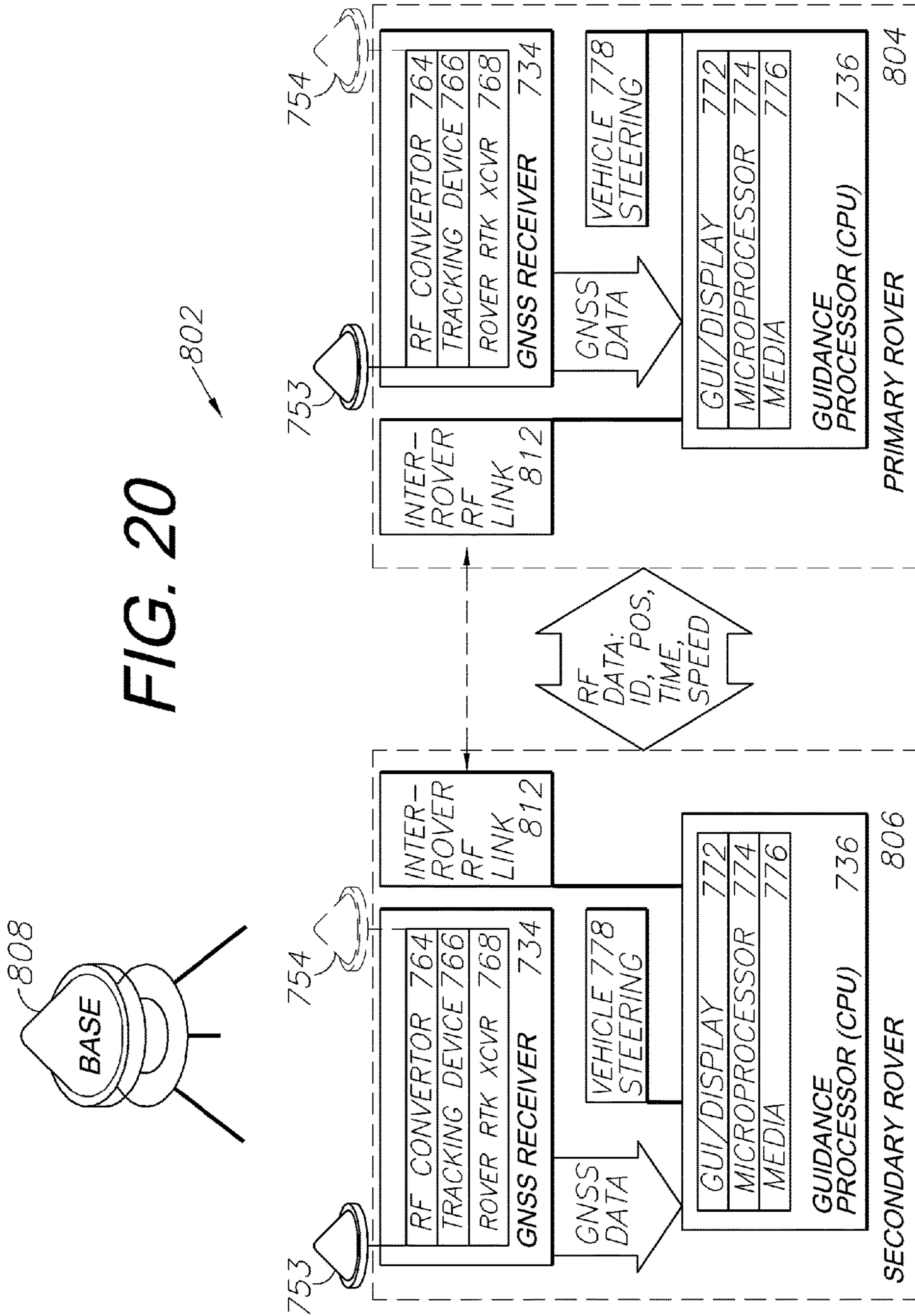


FIG. 17







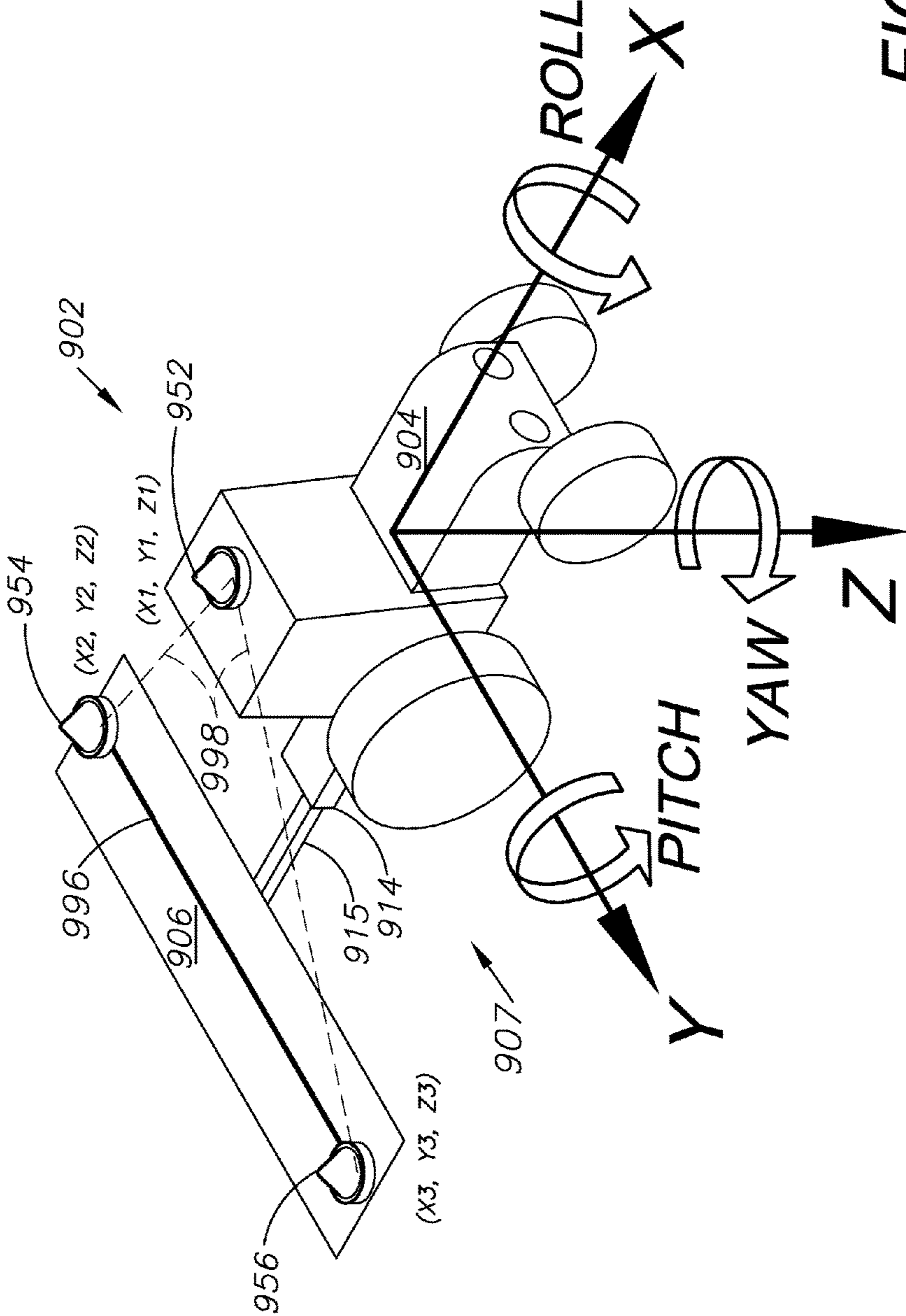


FIG. 21

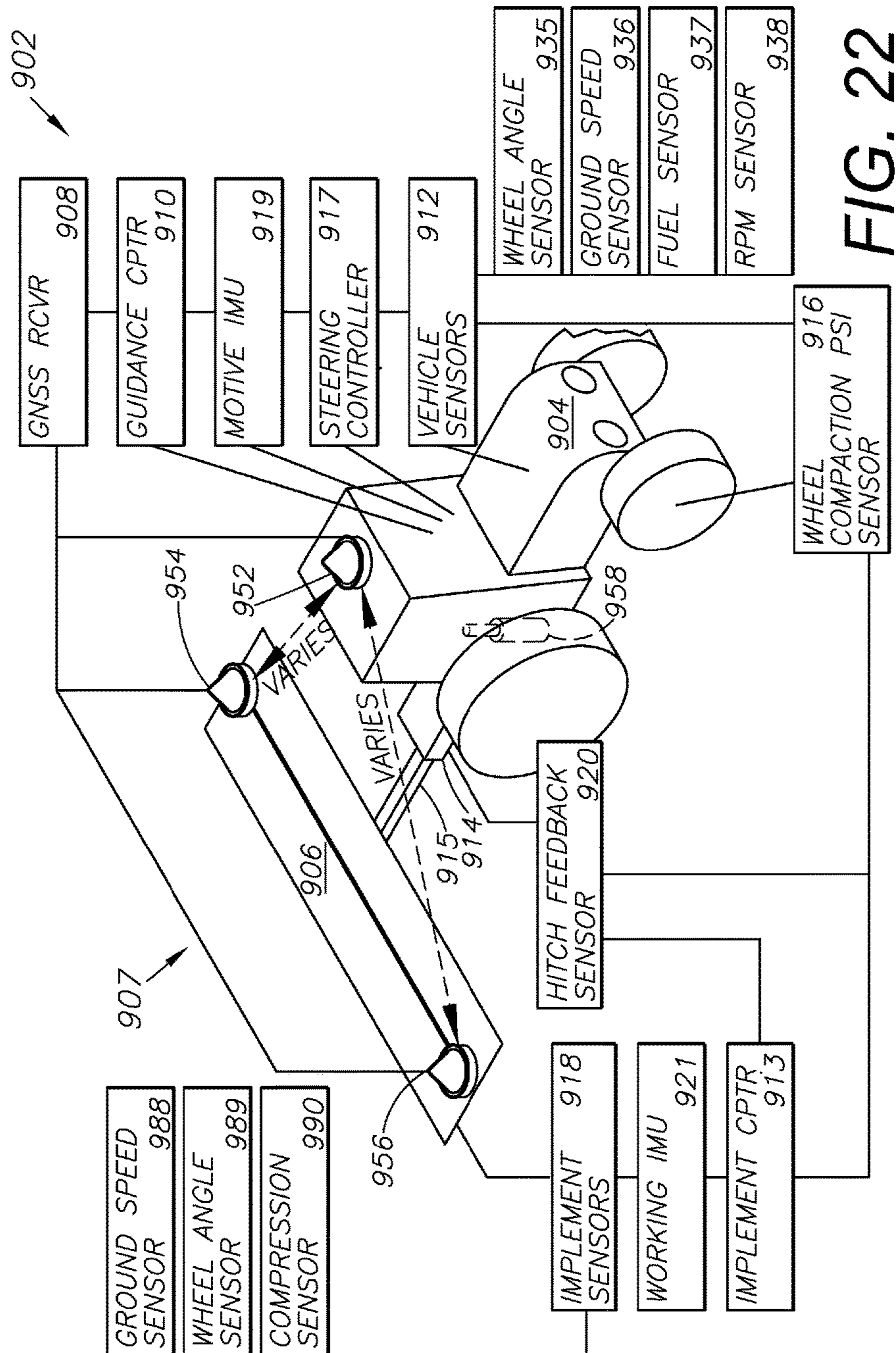
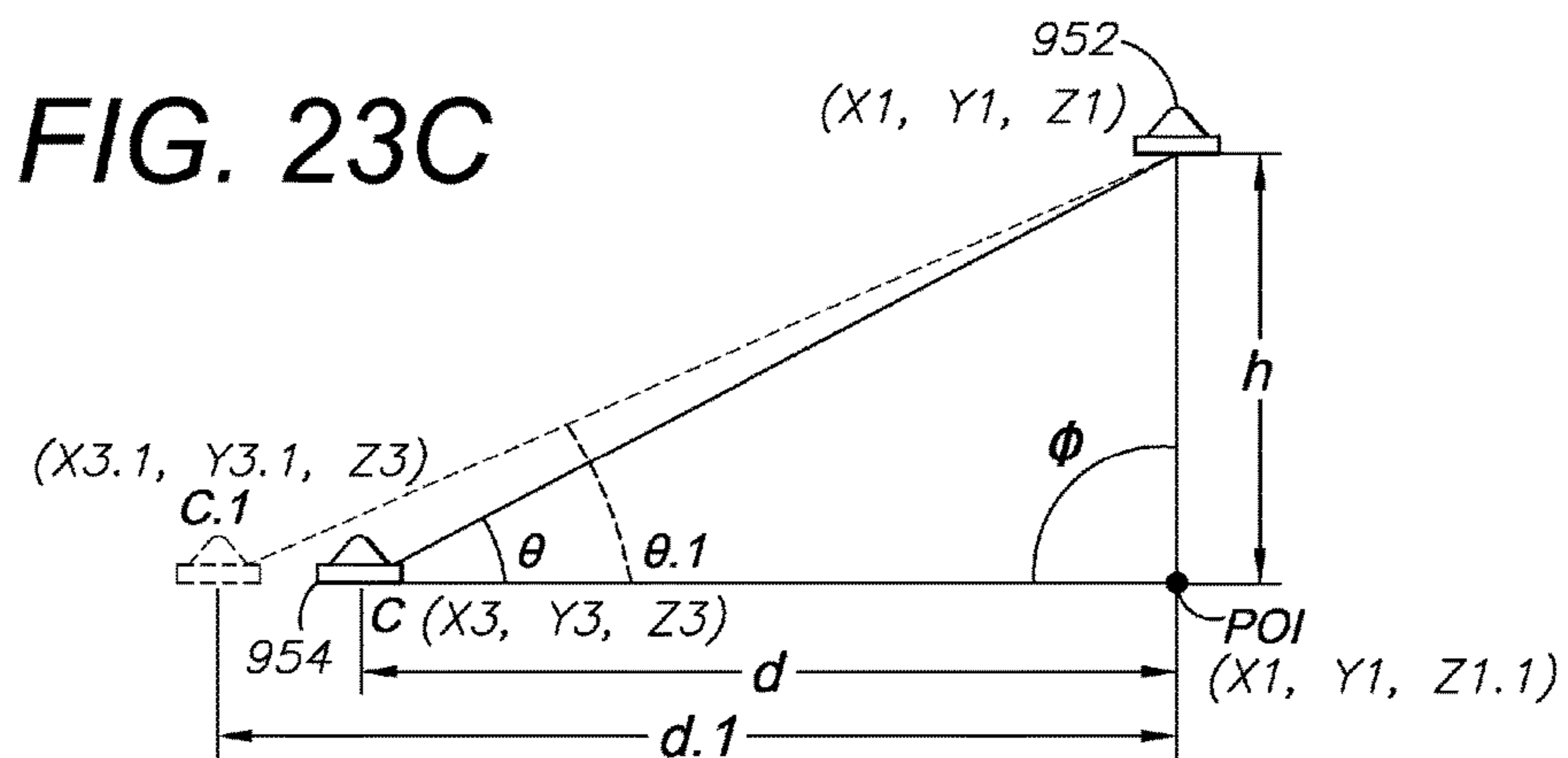
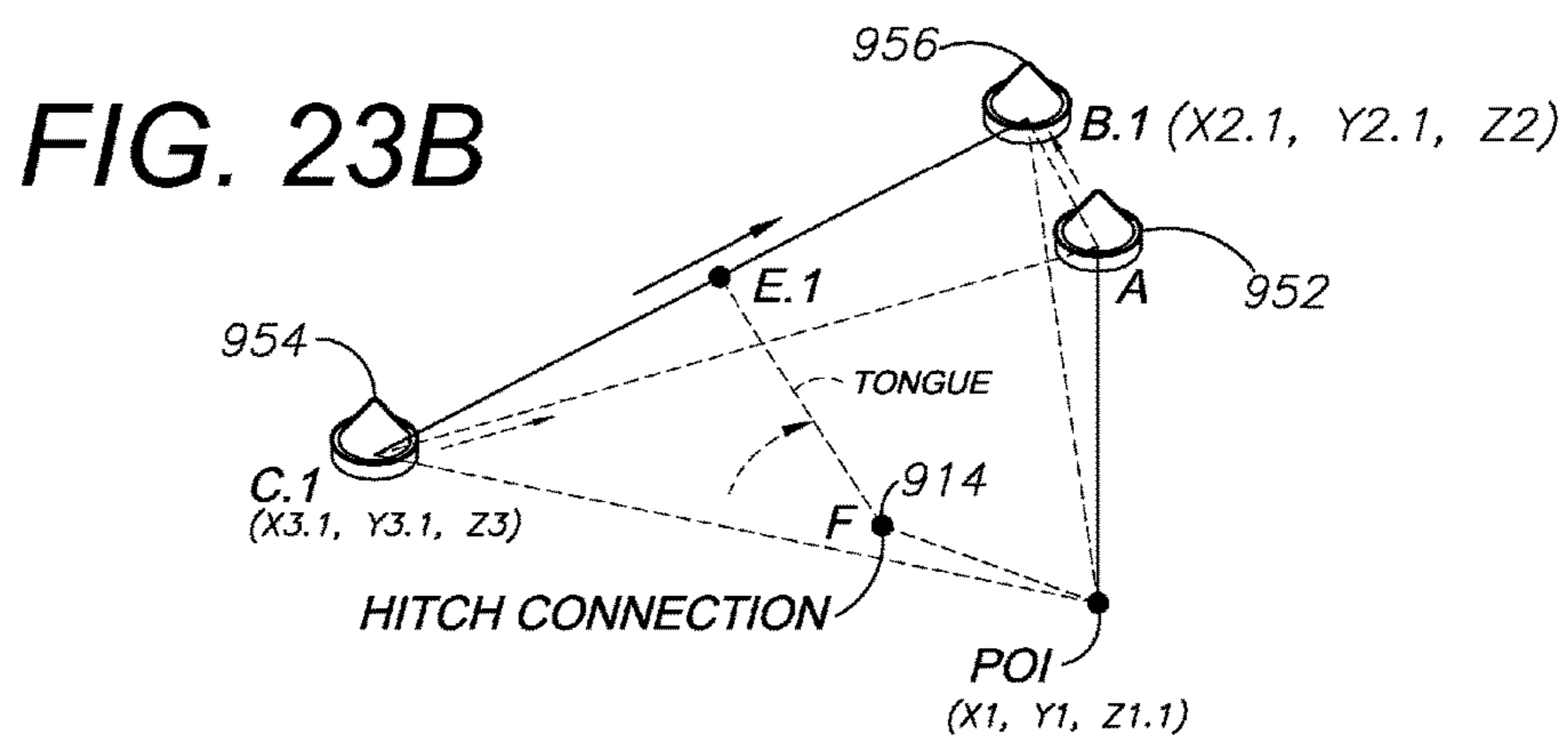
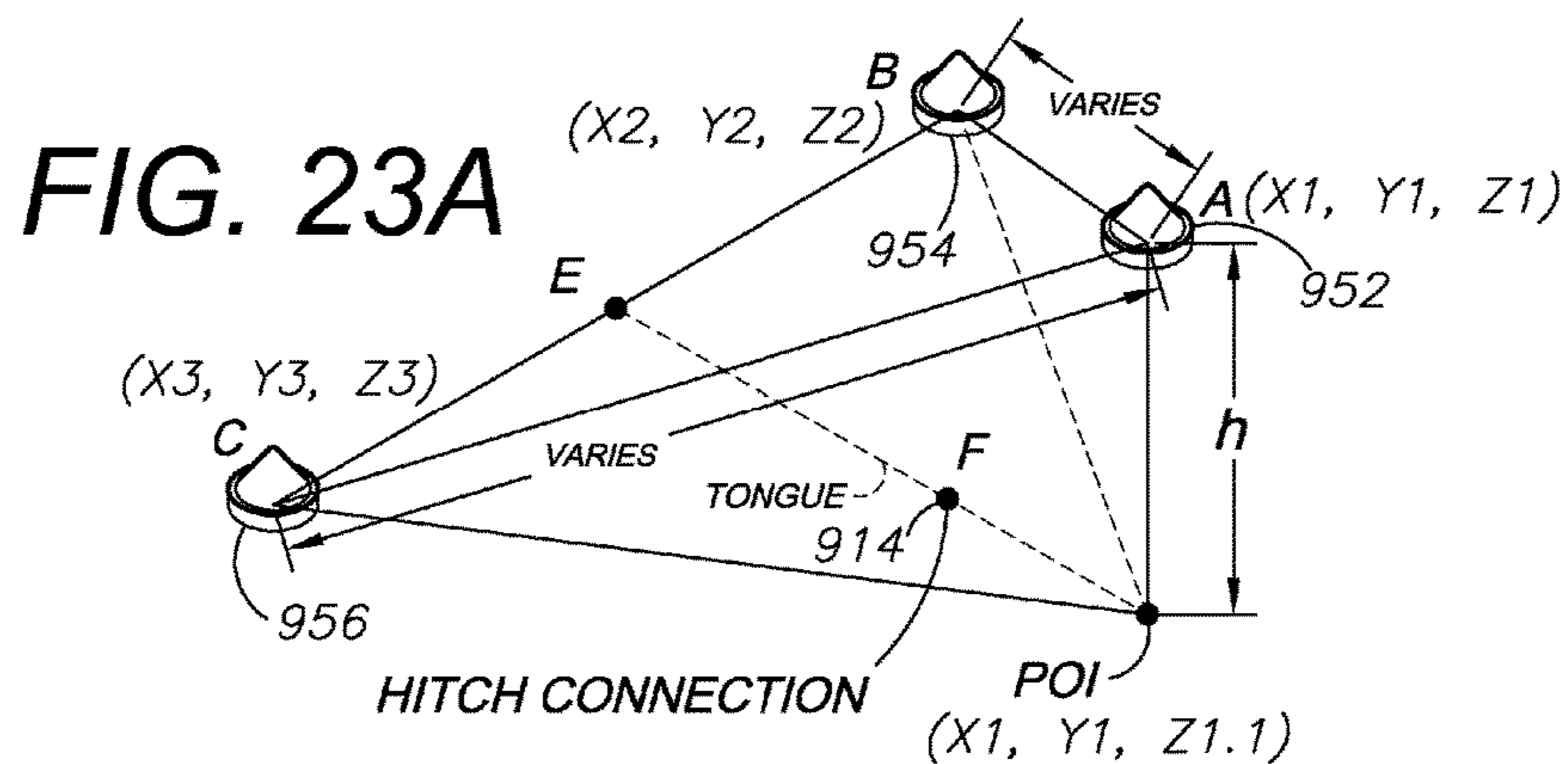
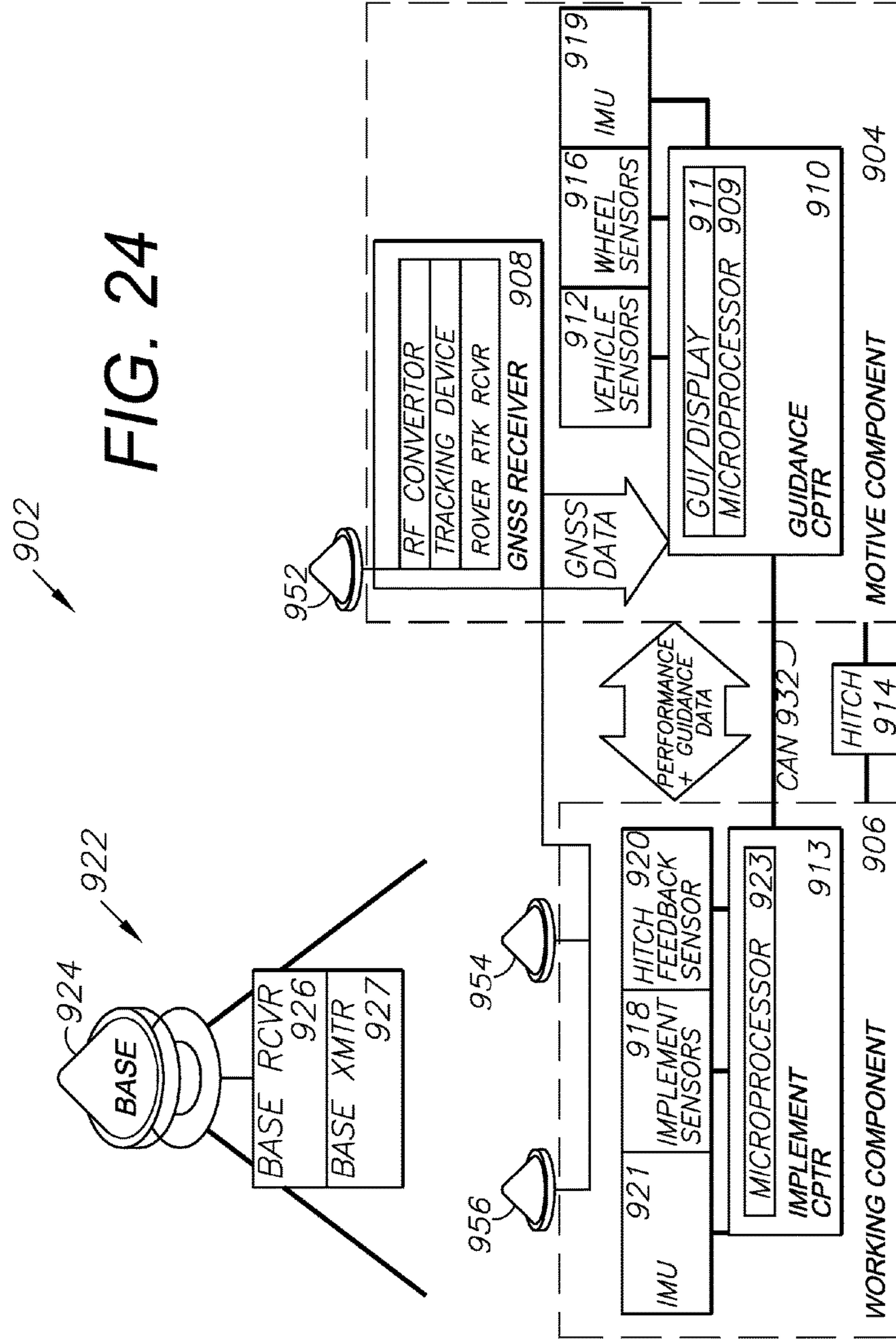


FIG. 22





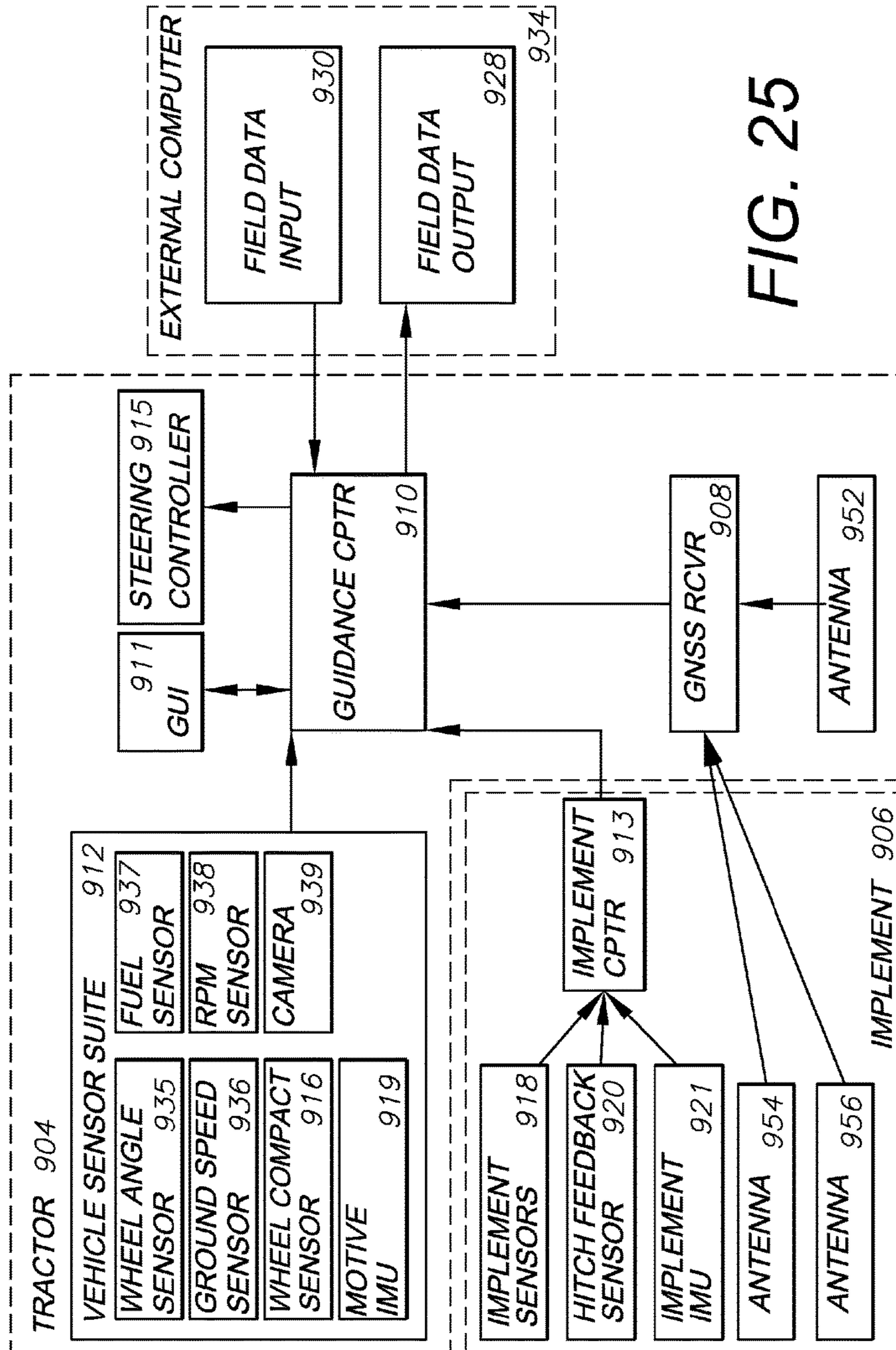


FIG. 25

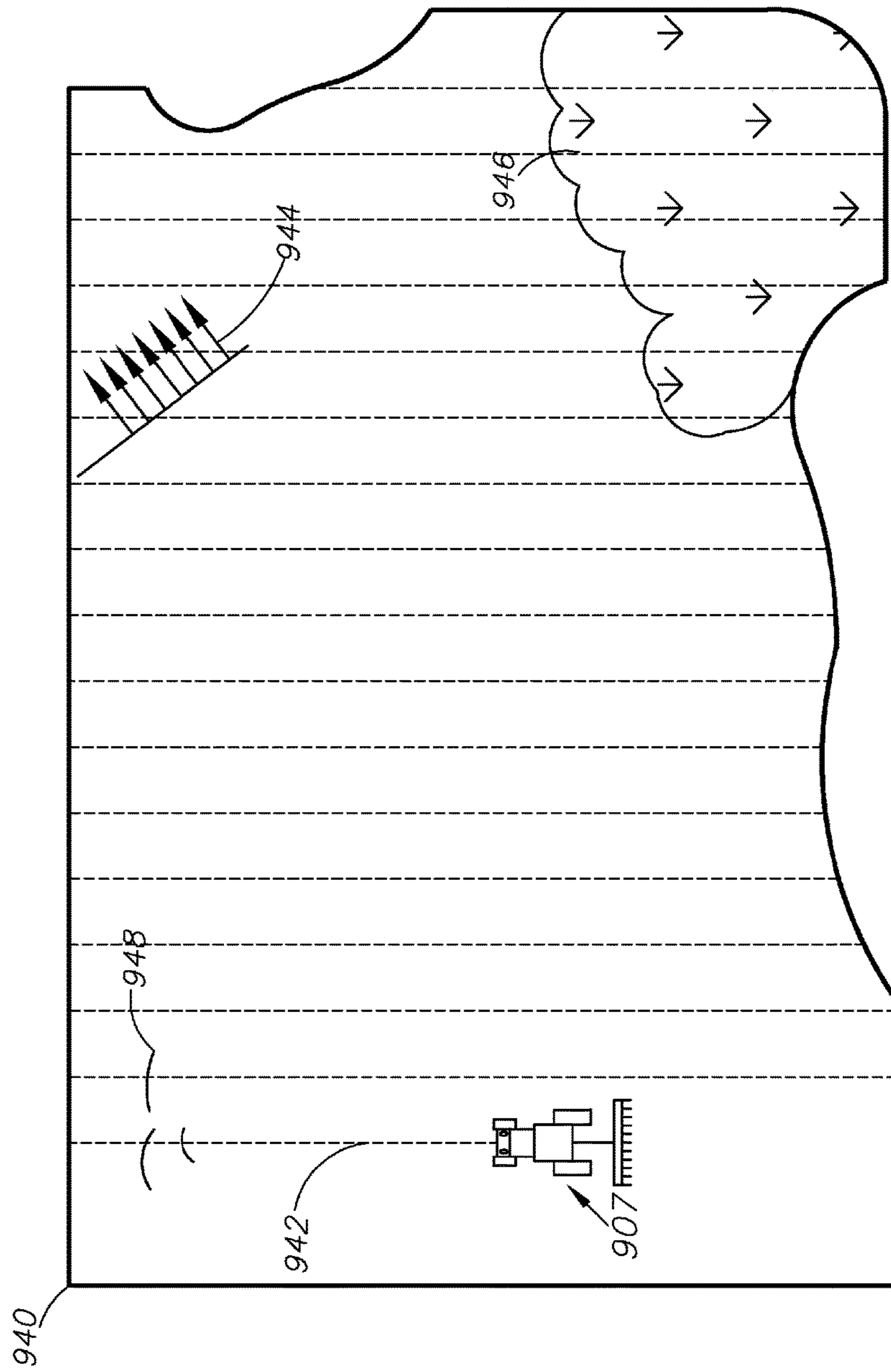
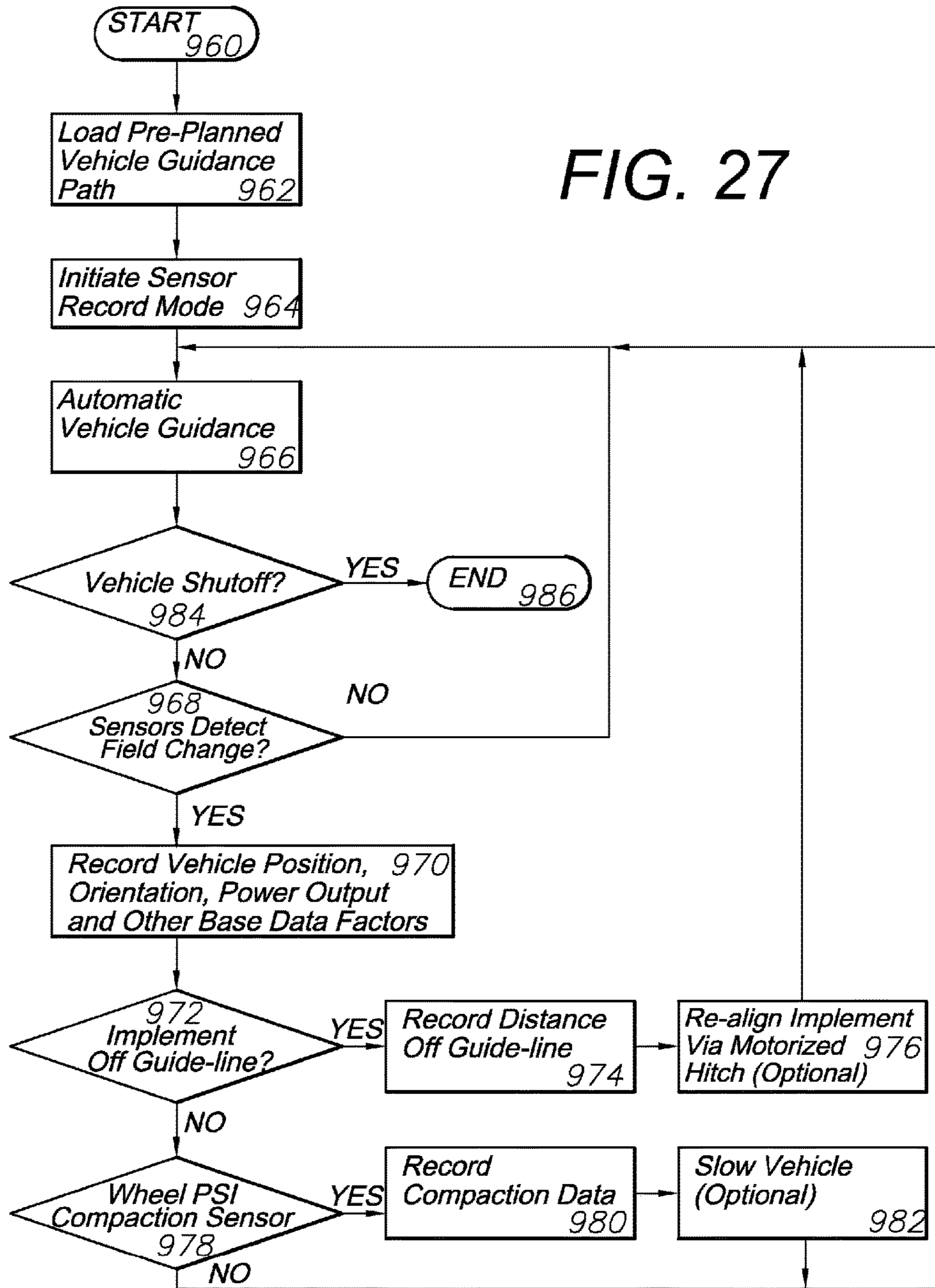
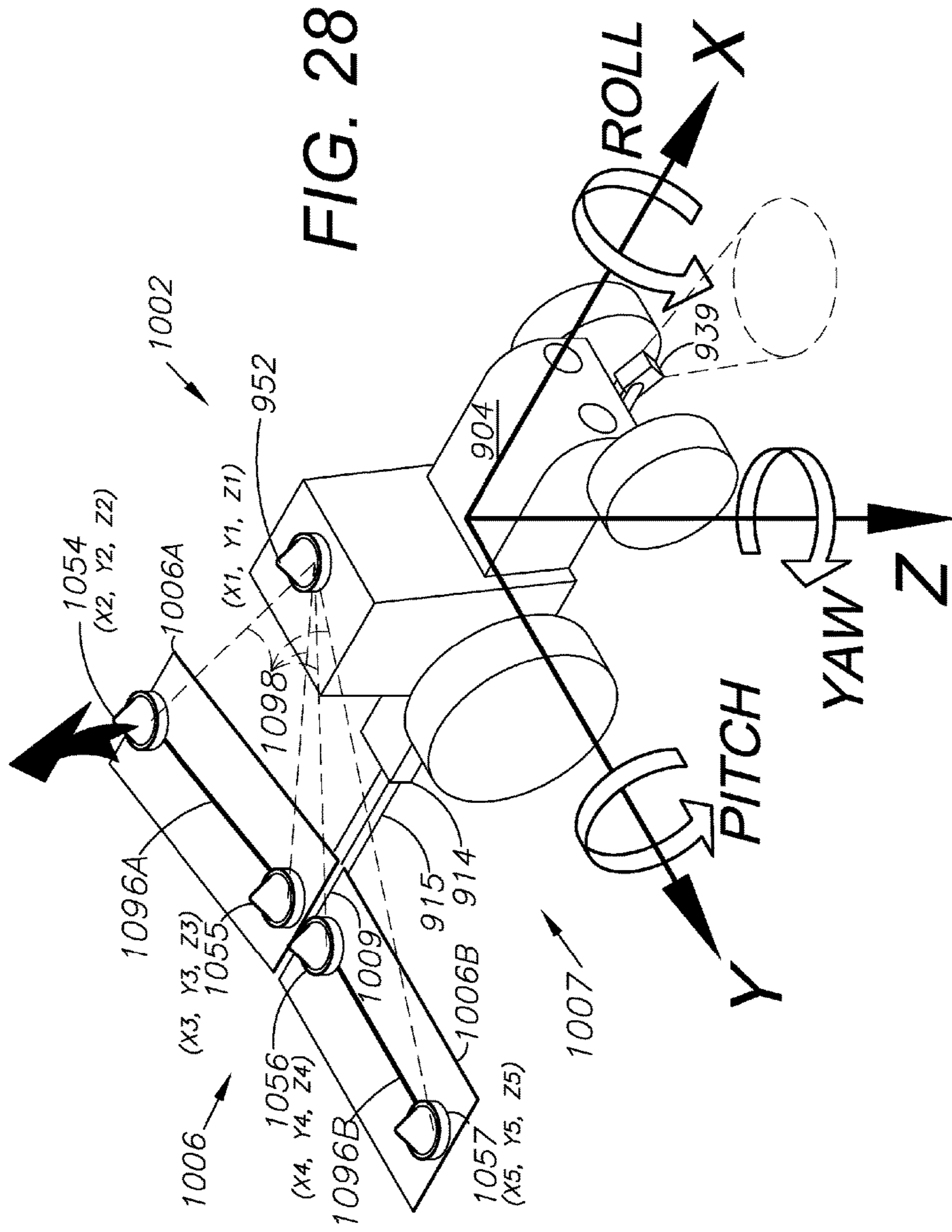


FIG. 26

FIG. 27





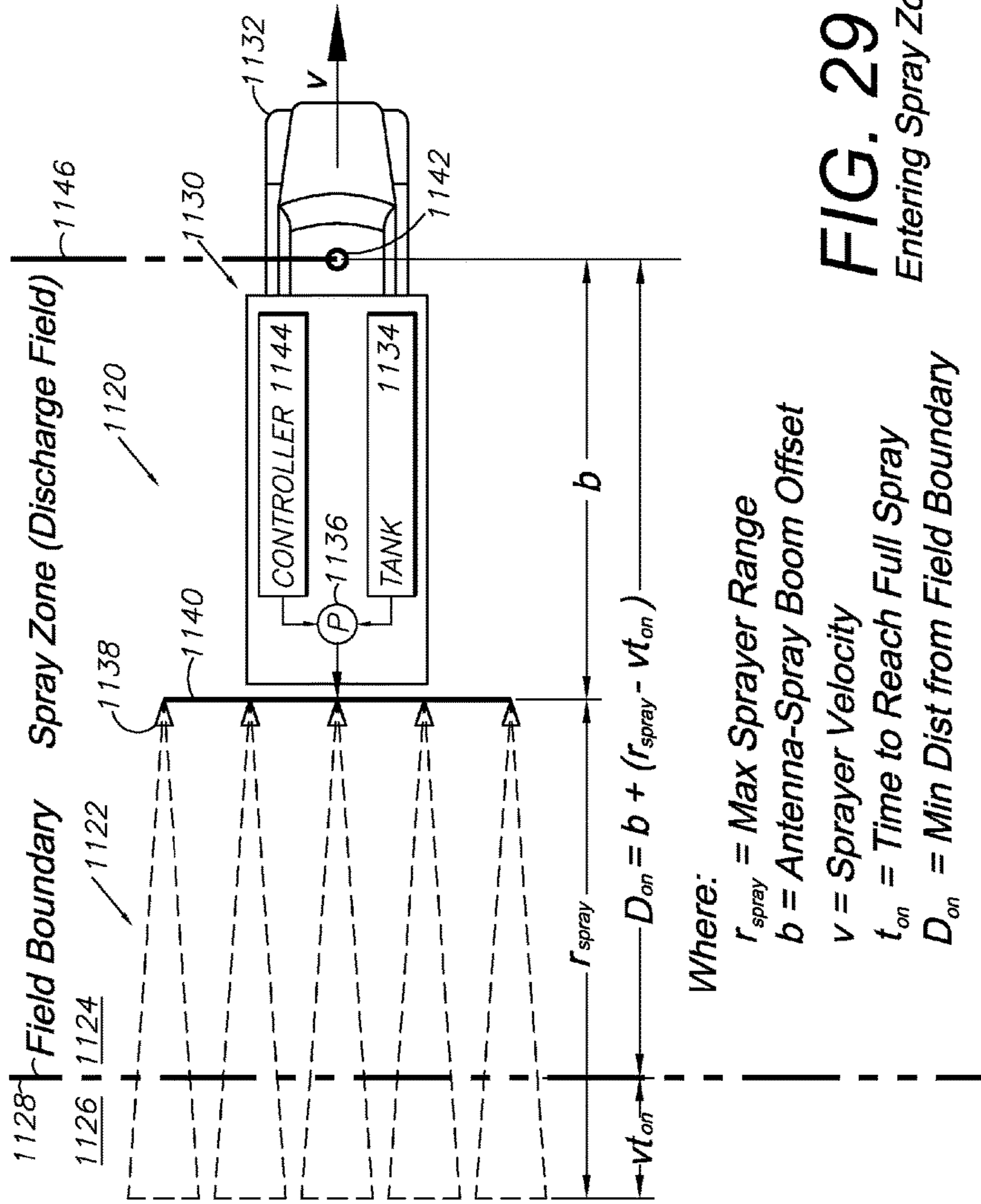


FIG. 29
Entering Spray Zone

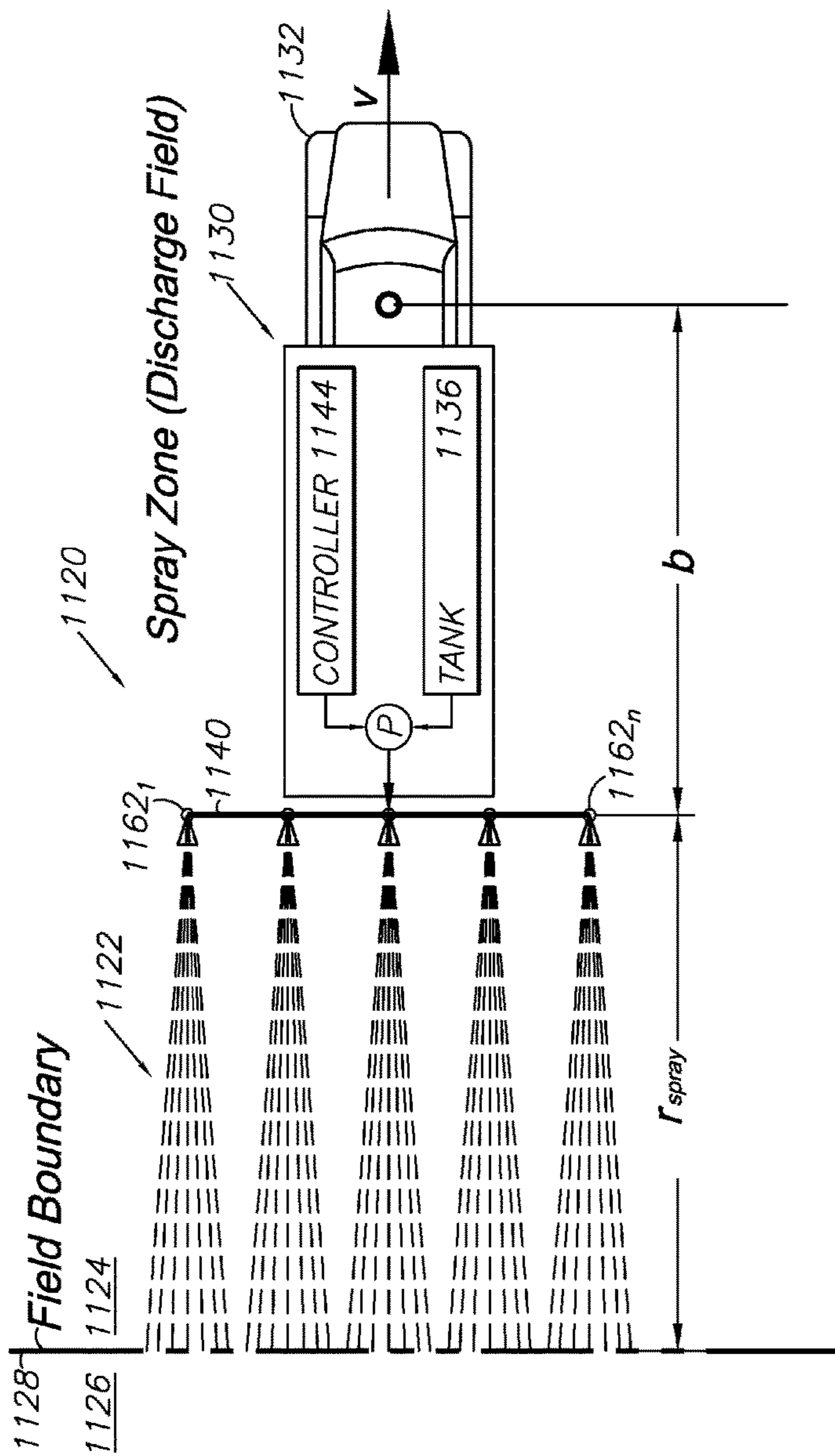
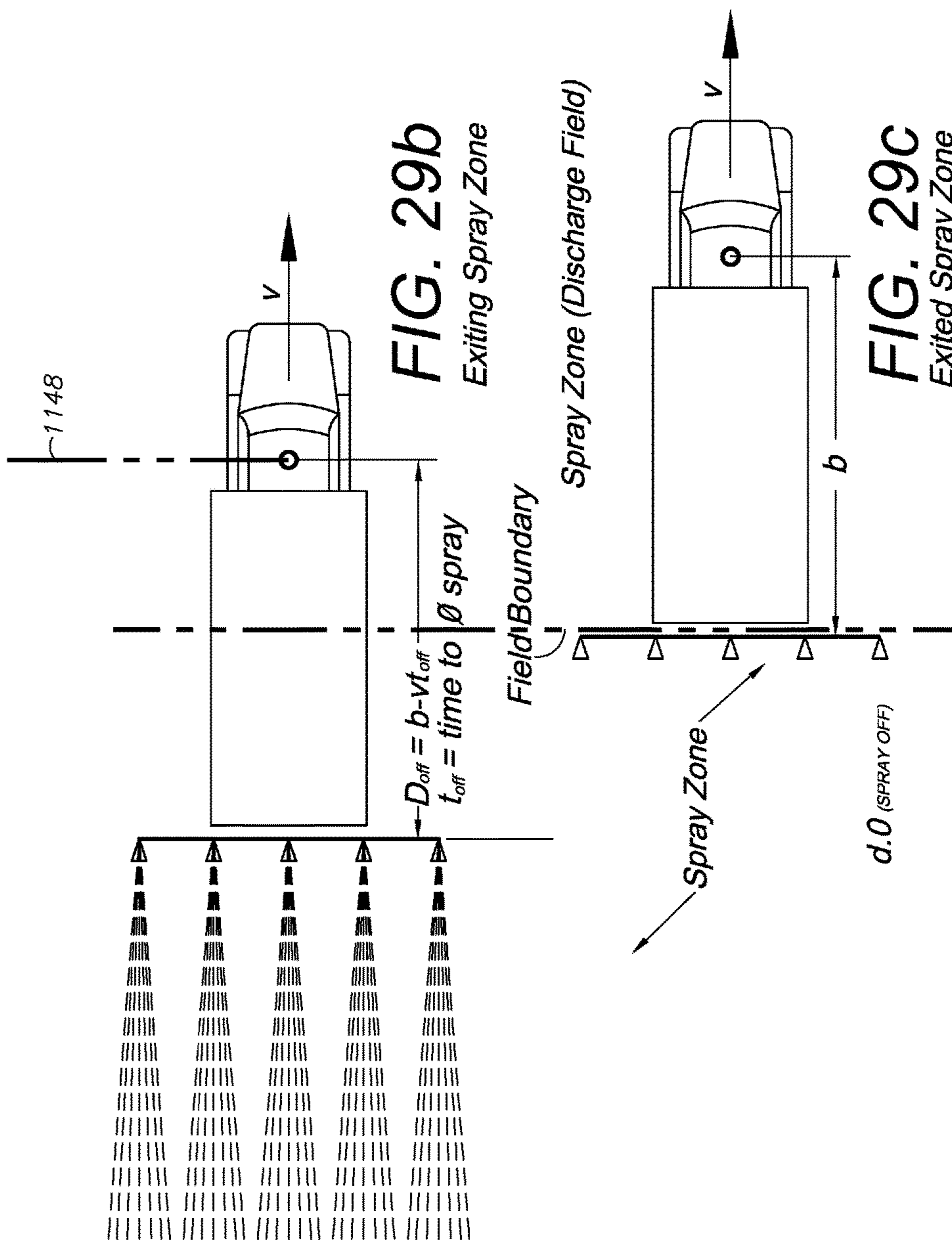


FIG. 29a
Entered Spray Zone



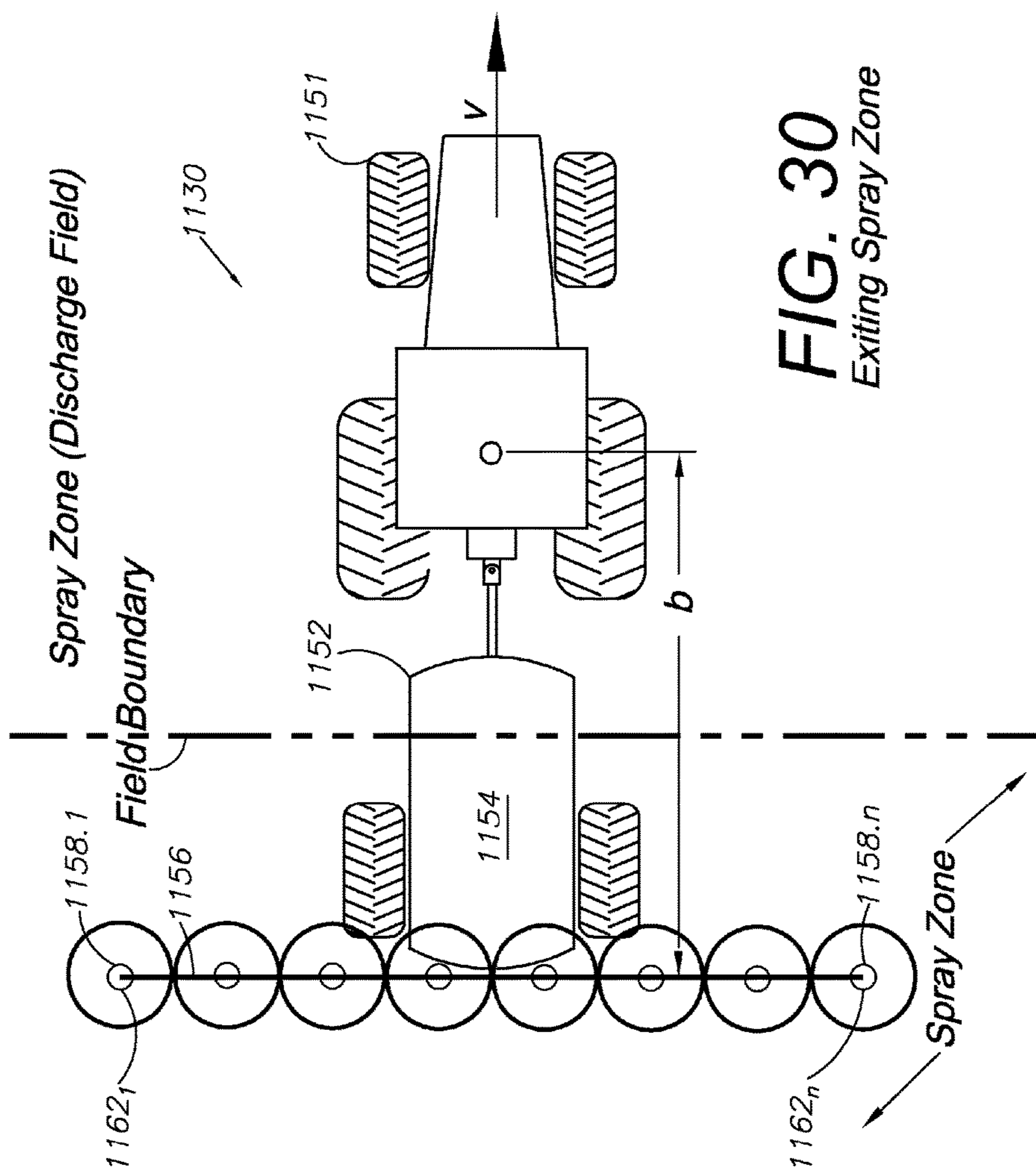


FIG. 30
Exiting Spray Zone

**CONTROL FOR DISPENSING MATERIAL
FROM VEHICLE**

Matter enclosed in heavy brackets [] appears in the original patent but forms no part of this reissue specification; matter printed in italics indicates the additions made by reissue; a claim printed with strikethrough indicates that the claim was canceled, disclaimed, or held invalid by a prior post-patent action or proceeding.

CROSS-REFERENCE TO RELATED
APPLICATIONS

[This application is a continuation-in-part of and claims the benefit of: U.S. patent application Ser. No. 12/355,776, filed Jan. 17, 2009 now U.S. Pat. No. 8,140,223; which is a continuation-in-part of Ser. No. 12/171,399, filed Jul. 11, 2008 now U.S. Pat. No. 8,265,826; which is a continuation-in-part of Ser. No. 10/804,758, filed Mar. 19, 2004, now U.S. Pat. No. 7,400,956 and a continuation-in-part of Ser. No. 10/828,745, filed Apr. 21, 2004, now abandoned which benefited from U.S. Provisional Patent Application Ser. No. 60/456,146, filed Mar. 20, 2003 and Ser. No. 60/464,756, filed Apr. 23, 2003. This application is also a continuation-in-part of U.S. patent application Ser. No. 12/750,429 filed Mar. 30, 2010 now U.S. Pat. No. 8,214,111; which is a continuation of U.S. patent application Ser. No. 11/184,657, filed Jul. 19, 2005, now U.S. Pat. No. 7,689,354. The contents of all of the aforementioned applications and patents are incorporated by reference herein in their entireties.] *The present application is a reissue application of U.S. Pat. No. 8,634,993, issued Jan. 21, 2014, entitled: GNSS BASED CONTROL FOR DISPENSING MATERIAL FROM VEHICLE; which is a continuation-in-part of and claims the benefit of: U.S. patent application Ser. No. 12/355,776, filed Jan. 17, 2009 now U.S. Pat. No. 8,140,223; which is a continuation-in-part of Ser. No. 12/171,399, filed Jul. 11, 2008 now U.S. Pat. No. 8,265,826; which is a continuation-in-part of Ser. No. 10/804,758, filed Mar. 19, 2004, now U.S. Pat. No. 7,400,956 and a continuation-in-part of Ser. No. 10/828,745, filed Apr. 21, 2004, now abandoned which benefited from U.S. Provisional Patent Application Ser. No. 60/456,146, filed Mar. 20, 2003 and Ser. No. 60/464,756, filed Apr. 23, 2003. This application is also a continuation-in-part of U.S. patent application Ser. No. 12/750,429 filed Mar. 30, 2010 now U.S. Pat. No. 8,214,111; which is a continuation of U.S. patent application Ser. No. 11/184,657, filed Jul. 19, 2005, now U.S. Pat. No. 7,689,354. The contents of all of the aforementioned applications and patents are incorporated by reference herein in their entireties.*

BACKGROUND OF THE INVENTION

Movable machinery, such as agricultural equipment, open-pit mining machines, airplane crop dusters, and the like all benefit from accurate global navigation satellite system (GNSS) high precision survey products, and others. However, in existing satellite positioning systems (SATPS) for guided parallel and contour swathing for precision farming, mining, and the like, the actual curvature of terrain may not be taken into account. This results in a less than precise production because of the less than precise parallel or contour swathing. Indeed, in order to provide swaths through a field (in farming, for example), the guidance system collects positions of the vehicle as it moves across

the field. When the vehicle commences the next pass through the field, the guidance system offsets the collected positions for the previous pass by the width of the equipment (i.e. swath width). The next set of swath positions is used to provide guidance to the operator as he or she drives the vehicle through the field.

The current vehicle location, as compared to the desired swath location, is provided to the vehicle's operator or to a vehicle's steering system. The SATPS provides the 3-D location of signal reception (for instance, the 3-D location of the antenna). If only 3-D coordinates are collected, the next swath computations assume a flat terrain offset. However, the position of interest is often not the same as where the satellite receiver (SR) is located since the SR is placed in the location for good signal reception, for example, for a tractor towing an implement, an optimal location for the SR may be on top of the cab. However, the position of interest (POI) for providing guidance to the tractor operator may be the position on the ground below the operator. If the tractor is on flat terrain, determining this POI is a simple adjustment to account for the antenna height.

However, if the tractor is on an inclined terrain with a variable tilt, which is often the case, the SATPS alone cannot determine the terrain tilt so the POI also cannot be determined. This results in a guidance error because the POI is approximated by the point of reception (POR), and this approximation worsens as the terrain inclination increases. This results in cross track position excursions relative to the vehicle ground track which would contaminate any attempt to guide to a defined field line or swath. On inclined terrain, this error can be minimized by collecting the vehicle tilt configuration along each current pass or the previous pass. The swath offset thus becomes a vector taking the terrain inclination into account with the assumption that from the first swath to the next one the terrain inclination does not change too much. It can therefore be seen that there is a need for a better navigation/guidance system for use with a ground-based vehicle that measures and takes into account vehicle tilt.

Various navigation systems for ground-based vehicles have been employed but each includes particular disadvantages. Systems using Doppler radar will encounter errors with the radar and latency. Similarly, gyroscopes, which may provide heading, roll, or pitch measurements, may be deployed as part of an inertial navigation package, but tend to encounter drift errors and biases and still require some external attitude measurements for gyroscope initialization and drift compensation. Gyroscopes have good short-term characteristics but undesirable long-term characteristics, especially those gyroscopes of lower cost such as those based on a vibrating resonator. Similarly, inertial systems employing gyroscopes and accelerometers have good short-term characteristics but also suffer from drift. Various systems include navigating utilizing GNSS; however, these systems also exhibit disadvantages. Existing GNSS position computations may include lag times, which may be especially troublesome when, for example, GNSS velocity is used to derive vehicle heading. As a result, the position (or heading) solution provided by a GNSS receiver tells a user where the vehicle was a moment ago, but not in real time. Existing GNSS systems do not provide high quality heading information at slower vehicle speeds. Therefore, what is needed is a low cost sensor system to facilitate vehicle swath navigation that makes use of the desirable behavior of both GNSS and inertial units while eliminating or reducing non-desirable behavior. Specifically, what is needed is a means to employ low-cost gyroscopes (e.g., micro electro-

mechanical (MEM) gyroscopes) which exhibit very good short-term low noise and high accuracy while removing their inherent long-term drift.

Providing multiple antennas on a vehicle can provide additional benefits by determining an attitude of the vehicle from the GNSS ranging signals received by its antennas, which are constrained on the vehicle at a predetermined spacing. For example, high dynamic roll compensation signals can be output directly to the vehicle steering using GNSS-derived attitude information. Components such as gyroscopes and accelerometers can be eliminated using such techniques. Real-time kinematic (RTK) navigation can be accomplished using relatively economical single frequency L1-only receivers with inputs from at least two antennas mounted in fixed relation on a rover vehicle. Still further, moving baselines can be provided for positioning solutions involving tractors and implements and multi-vehicle GNSS control can be provided.

Providing additional antennas in combination with standard SATPS and GNSS guidance, as mentioned above, along with optional gyroscopes is a great method to increase GNSS positioning precision and accuracy, such as is described in U.S. Patent Publication No. 2009/0164067 which is assigned to a common assignee and is incorporated herein. However, accuracy and precision can only improve the efficiency of working vehicles, such as those in the agricultural field, to a limited extent. Although such systems are able to track and guide vehicles in three dimensions, including along ridges and sloped-regions, errors may appear in other aspects of a working vehicle. For example, in an agricultural field-working situation where a tractor is towing an implement, the implement may slide on a sloped-region, or the tractor may list to one side or another when entering softer soil or rocky areas. This can happen repeatedly when a vehicle is guided around the same field, regardless of the precision of the guidance system in pre-planning a path. Thus, a system that can detect such changes in uniformity of a field as the vehicle traverses a path and remember those changes can predict and re-route a more accurate and more economical path than a guidance system alone. Heretofore there has not been available a system and method with the advantages and features of the present invention.

Conventional agricultural spraying operations are carried out over an entire field, everywhere the crop is planted. In contrast, environmental spraying allows the spraying of certain materials which require restrictions in the area of deposition due to potential toxicity or strength. The restrictions can include the distance from waterways and slope of the ground which can affect run-off and concentrations of deposits.

Drilling waste water is one such restricted application. Typically, large tanker trucks are used to disperse waste water from oil, gas, and water drilling operations. This water is mixed with specialized drilling mud with a large bentonite clay concentration along with stabilizing agents to cool the drill bit, to extract drill cuttings, and to maintain the hydrostatic pressure to seal and stabilize the well. The waste water is not particularly toxic and can act as a source of soil nutrients. However, if it is dumped into a surface water shed in large quantities and concentrations, it can have a detrimental effect on local fish populations. Environmental agencies now restrict dumping close to water channels. Current recycling methods involve preplanning of areas in open fields or spray zones that can be sprayed so that minimal run-off occurs, plus additional areas or exclusion zones that are not to be sprayed.

When a tanker truck turns on the spray, the area of deposition starts well behind the nozzles, depending on the speed and "windup" time of the spray machinery and plumbing. When the spray is turned off, the fluid continues to exit the nozzles for a certain amount of time. Even with a visually marked boundary of the spray zone, it is difficult for a tanker truck driver to accurately turn the spray on and off to fully cover the spray zone and to prevent some spray from getting in the exclusion zone.

SUMMARY OF THE INVENTION

Disclosed herein in an exemplary embodiment is a sensor system for vehicle steering control comprising: a plurality of global navigation satellite systems (GNSS) including receivers and antennas at a fixed spacing to determine a vehicle position, velocity and at least one of a heading angle, a pitch angle, and a roll angle based on carrier phase corrected real time kinematic (RTK) position differences. The roll angle facilitates correction of the lateral motion induced position errors resultant from motion of the antennas as the vehicle moves based on an offset to ground and the roll angle. The system also includes a control system configured to receive the vehicle position, heading, and at least one of roll, pitch and yaw, and configured to generate a steering command to a vehicle steering system.

Also disclosed herein in another exemplary embodiment is a method for computing a position of a vehicle comprising: initializing GNSS; computing a first position of a first GNSS antenna on the vehicle; computing a second position of a second GNSS antenna; and calculating a heading as a vector perpendicular to a vector joining the first position and the second position, in a horizontal plane aligned with the vehicle. The method also includes computing a roll angle of the vehicle as an arc-tangent of a ratio of differences in heights of the first GNSS antenna and the second GNSS antenna divided by a spacing between their respective phase centers and calculating an actual position at the center of the vehicle projected to the ground using the computed roll angle and a known height from the ground of at least one of the first GNSS antenna and the second GNSS antenna.

Further disclosed herein in yet another exemplary embodiment is a method of controlling a vehicle comprising: computing a position and a heading for the vehicle; computing a steering control command based on a proportionality factor multiplied by a difference in a desired position versus an actual position, plus a second proportionality factor multiplied by a difference in a desired heading versus an actual heading, the second proportionality factor ensuring that, when the vehicle attains the desired position, the vehicle is also directed to the desired heading, and thereby avoiding crossing a desired track. The method also includes a recursive adaptive algorithm employed to characterize the vehicle response and selected dynamic characteristics.

The method further includes applying selected control values to a vehicle steering control mechanism and measuring responses of the vehicle thereto; calculating response times and characteristics for the vehicle based on the responses; and calibrating the control commands by applying a modified control command based on the responses to achieve a desired response. Various alternative aspects and applications of the present invention are disclosed herein.

Additional alternative aspects include selective sprayer nozzle control, high dynamic roll compensation using GNSS attitude solutions from multiple antennas, moving baseline implement positioning and multiple vehicle control.

An additional exemplary embodiment is a sensor system for vehicle guidance using one or more global navigation satellite systems (GNSSs) according to the above-mentioned embodiments, in combination with a plurality of various sensors located throughout a vehicle and a towed implement. These sensors detect additional parameters from those calculated by the GNSS positioning system, such as vehicle and implement stress levels, fuel levels, power levels, optical guide path observations via an onboard camera, multi-section (articulated) implement position and attitude sensing via multiple antennas and other characteristics of the working vehicle. The combination of the two systems results in a much more accurate and economical preplanned path generated for use in later field work.

An additional embodiment of the present invention provides for precision control of spraying, particularly environmental spraying operations in which spraying of certain materials is only allowed in designated areas. In general, a spray vehicle includes a tanker truck with a spray boom having a plurality of transversely spaced nozzles, a tank holding a quantity of the material to be sprayed, a pump communicating with the tank and the nozzles, a GNSS receiver tracking its own position, and a controller interfaced with the GNSS receiver and the pump. The spray vehicle can also be a tractor towing a sprayer implement which is hitched to the tractor. The material tank, pump, and spray boom can be mounted on the sprayer implement.

Areas to be sprayed are designated as spray zones while areas not to be sprayed are designated as exclusion zones. The spray zones and exclusion zones are separated by linear boundaries, which are surveyed, and coordinates of the boundaries are stored in the controller. The controller is a computer having data storage, data inputs and outputs, and controlled outputs connected to suitable drivers and/or relays to control particular processes or actions. When the spray vehicle is within an exclusion zone and approaching a spray zone, the boundary is termed a spray turn-on boundary. When the spray vehicle is within a spray zone and approaching an exclusion zone, the boundary is termed a spray turn-off boundary. In general, the controller is programmed to control activation of the pump in relation to positions detected by the GNSS receiver to retain the spray of the material within spray zones and to prevent spray of the material within the exclusion zones.

The spray boom is mounted at an offset distance from the GNSS receiver in the direction of vehicle travel. The pump in cooperation with the nozzles and plumbing therebetween have operational characteristics such that a spray turn-on lag time is required for the spray to reach a maximum spray range distance from the nozzles after the pump is activated. Similarly, a spray turn-off lag time is required between the deactivation of the pump and cessation of material being sprayed from the nozzles. The nozzle/receiver offset distance, the spray range distance, the spray turn-on lag time, and the spray turn-off lag time are all stored in the controller.

In an embodiment of a spray control method, when the spray vehicle is moving in an exclusion zone toward a turn-on boundary at a particular vehicle velocity, the controller causes activation of the pump when the GNSS receiver detects its position within a spray zone beyond the turn-on boundary equal to the sum of the nozzle/receiver offset distance plus the spray range minus the product of the vehicle velocity times the turn-on lag time. The pump remains activated to cause the material to be sprayed from the nozzles as the spray vehicle approaches an exclusion zone. The controller causes deactivation of the pump when the GNSS receiver detects its position within the exclusion

zone beyond the turn-off boundary equal to the nozzle/receiver offset distance minus the product of the vehicle velocity times the turn-off lag time.

It is foreseen that the spray vehicle will not always be approaching a straight-line boundary or a boundary which is perpendicular to the direction of its travel. In an alternative embodiment of the spray control method, the nozzles are controlled individually by corresponding spray valves interfaced to the controller. The valves are opened and closed by the controller based, on the position of the GNSS receiver relative to a portion of an approached zone boundary which is aligned with a particular nozzle and its spray pattern in the direction of travel of the spray vehicle. Generally, when the spray vehicle is approaching a turn-on boundary, each valve is controlled to open when the GNSS receiver detects a position of the vehicle within the spray zone at which the spray of material from the nozzle associated with that valve is retained within the spray zone. Similarly, when the spray vehicle is approaching a turn-off boundary, each valve is controlled to close when the GNSS receiver detects a position of the vehicle within the exclusion zone at which the spray of material from that valve within the exclusion zone is prevented.

More particularly, an interval of time is required between the opening of a spray valve and the material reaching the spray range of the associated nozzle, which is referred to as a valve open lag time. Similarly, there is a valve close lag time required between the closing of a valve and the complete cessation of spray material exiting the nozzle. The valve open and close lag times may be different respectively from the spray turn-on and turn-off lag times described above. In environmental spraying, the principal criteria used in selecting valve open and close lag times is avoidance of spraying material on exclusion zones. The valve open and close lag times are entered into the controller.

When the spray vehicle is approaching a turn-on boundary, each spray valve is controlled to open when the GNSS receiver detects a position of the receiver within the spray zone at a valve open distance beyond a portion of the turn-on boundary aligned with the nozzle associated with that valve equal to the sum of nozzle/receiver offset distance plus the spray range minus the product of the spray vehicle velocity times the valve open lag time. Similarly, when the spray vehicle is approaching a turn-off boundary, each valve is closed when the GNSS receiver detects a position of the receiver within the exclusion zone at a valve close distance equal to the nozzle/receiver offset distance minus the product of the vehicle velocity times the valve close lag time.

Various objects and advantages of the present invention will become apparent from the following description taken in conjunction with the accompanying drawings wherein are set forth, by way of illustration and example, certain embodiments of this invention.

The drawings constitute a part of this specification, include exemplary embodiments of the present invention, and illustrate various objects and features thereof.

BRIEF DESCRIPTION OF THE DRAWINGS

FIG. 1 is a diagrammatic perspective view of a vehicle including an exemplary embodiment incorporating aspects of the present invention.

FIG. 2 is a block diagram of components mounted on the vehicle including an exemplary embodiment of a sensor system.

FIG. 3 is a block diagram of a sensor system in accordance with an exemplary embodiment.

FIG. 4 is a diagrammatic perspective view of an illustrative sensor system in accordance with an exemplary embodiment.

FIG. 5 is a flow diagram of an exemplary process for determining a steering command for a vehicle in accordance with an exemplary embodiment.

FIG. 6 is a flow diagram of an exemplary process for determining a steering command with an exemplary sensor system in accordance with an alternative embodiment.

FIGS. 7A-7C are diagrammatic perspective views depicting a multi-axis antenna and gyroscope system embodying an aspect of the present invention and including two antennas connected by a rigid link and yaw and roll gyroscopes and illustrating a determination of yaw attitude in FIG. 7B and roll attitude in FIG. 7C.

FIG. 8 is a diagrammatic end view of an agricultural vehicle and illustrates a tilt (roll) angle and cross track error measuring application of the invention.

FIG. 9 is a diagrammatic plan view of a tractor and towed implement and depicts an alternative aspect of the system with antenna and gyroscope subsystems mounted on both the tractor and the implement, e.g. a sprayer with selectively controllable spray nozzles.

FIG. 10 is a block diagram illustrating components the system shown in FIG. 9.

FIG. 11 is a diagrammatic perspective view which depicts a high dynamic roll compensation GNSS guidance system comprising an alternative aspect of the present invention.

FIG. 12 is a block diagram illustrating components of the system shown in FIG. 11.

FIG. 13 is a diagrammatic perspective view of a tractor and implement and depicts an alternative aspect of the present invention comprising a moving baseline GNSS system with the tractor and the implement each mounting a respective antenna for a 1+1 antenna configuration.

FIG. 14 is an enlarged fragmentary perspective view of the system shown in FIG. 13, particularly showing implement yaw and pitch movements in connection with the moving antenna-to-antenna baseline.

FIG. 15 is a view similar to FIG. 13 and depicts an alternative moving baseline aspect of the invention in a 2+1 antenna configuration.

FIG. 16 is a view similar to FIG. 13 and depicts another alternative moving baseline aspect of the invention in a 2+2 antenna configuration.

FIG. 17 is a diagrammatic plan view of a tractor and implement and depicts the 2+1 moving baseline system in a contour mode of operation with a multi-position tail.

FIG. 18 is a block diagram illustrating components of the moving baseline system(s).

FIG. 19 is a diagrammatic plan view and depicts a multi-vehicle GNSS relative guidance system including primary and secondary rovers.

FIG. 20 is a block diagram and illustrates components of the system shown in FIG. 19.

FIG. 21 is a diagrammatic perspective view of a tractor and implement equipped with a GNSS-based control system comprising another alternative embodiment of the present invention, and also depicts X, Y and Z axes corresponding to roll, pitch and yaw rotation respectively of the tractor.

FIG. 22 is a diagrammatic perspective view of a tractor and implement in combination with a block diagram illustrating components of the control system of FIG. 21.

FIG. 23A is a diagrammatic perspective view showing relative locations of a tractor-mounted antenna and two implement-mounted antennas thereof, with the implement directly in line behind the tractor.

FIG. 23B is a view similar to FIG. 23A showing the relative locations of the antennas with the implement swung to the left behind the tractor.

FIG. 23C is a right side elevational view of the antennas shown in FIGS. 23A and 23B and depicts the relative locations of the tractor antenna and an implement antenna.

FIG. 24 is a block diagram of the guidance system of FIGS. 23A-23C.

FIG. 25 flow diagram illustrating the flow of data among the various vehicle control system components of the system of FIG. 24.

FIG. 26 is a diagrammatic plan view and depicts the system controlling a tractor towing an implement in a cultivated field.

FIG. 27 is a flow diagram of a method of the present invention.

FIG. 28 is a diagrammatic perspective view of a guidance system on an agricultural vehicle comprising another alternative aspect of the present invention.

FIG. 29 is a plan view of another alternative embodiment or aspect of the present invention comprising a GNSS-based control for dispensing material from vehicles.

FIGS. 29a-c are diagrammatic plan views thereof illustrating the application of a material to selected areas and preventing application of the material to other areas.

FIG. 30 is a diagrammatic plan view of an embodiment or aspect of an environmental sprayer vehicle including a tractor and a towed spray implement.

DETAILED DESCRIPTION OF THE PREFERRED ASPECTS

As required, detailed embodiments of the present invention are disclosed herein; however, it is to be understood that the disclosed embodiments are merely exemplary of the invention, which may be embodied in various forms. Therefore, specific structural and functional details disclosed herein are not to be interpreted as limiting, but merely as a basis for the claims and as a representative basis for teaching one skilled in the art to variously employ the present invention in virtually any appropriately detailed structure.

I. GNSS Introduction

Global navigation satellite systems (GNSS) are broadly defined to include GPS (U.S.), Galileo (proposed), GLONASS (Russia), Beidou/Compass (China, proposed), IRNSS (India, proposed), QZSS (Japan, proposed) and other current and future positioning technology using signals from satellites, with or without augmentation from terrestrial sources. Inertial navigation systems (INS) include gyroscopic (gyro) sensors, accelerometers and similar technologies for providing output corresponding to the inertia of moving components in all axes, i.e. through six degrees of freedom (positive and negative directions along transverse X, longitudinal Y and vertical Z axes). Yaw, pitch and roll refer to moving component rotation about the Z, X and Y axes respectively. Said terminology will include the words specifically mentioned, derivatives thereof and words of similar meaning.

Disclosed herein in an exemplary embodiment is a sensor system for vehicle guidance. The sensor system utilizes a plurality of GNSS carrier phase differenced antennas to derive attitude information, herein referred to as a GNSS attitude system. Moreover, the GNSS attitude system may optionally be combined with one or more rate gyro(s) used to measure turn, roll or pitch rates and to further calibrate bias and scale factor errors within these gyros. In an exemplary embodiment, the rate gyros and GNSS receiver/an-

tenna are integrated together within the same unit, to provide multiple mechanisms to characterize a vehicle's motion and position to make a robust vehicle steering control mechanism.

It is known in the art that by using a GNSS satellite's carrier phase, and possibly carrier phases from other satellites, such as WAAS satellites, a position may readily be determined to within millimeters. When accomplished with two antennas at a fixed spacing, an angular rotation may be computed using the position differences. In an exemplary embodiment, two antennas placed in the horizontal plane may be employed to compute a heading (rotation about a vertical Z axis) from a position displacement. It will be appreciated that an exemplary embodiment may be utilized to compute not only heading, but either roll (rotation about a longitudinal Y axis) or pitch (rotation about a lateral X axis) depending on the orientation of the antennas relative to the vehicle. Heading information, combined with position, either differentially corrected (DGPS or DGNSS) or carrier phase corrected real time kinematic (RTK) provides the feedback information desired for a proper control of the vehicle direction. Addition of one or more rate gyros further provides independent measurements of the vehicle's dynamics and facilitates vehicle steering control. The combination of GNSS attitude obtained from multiple antennas with gyroscopes facilitates calibration of gyroscope scale factor and bias errors which are present in low cost gyroscopes. When these errors are removed, gyro rates are more accurate and provide better inputs for guidance and control. Furthermore, gyroscopes can now effectively be integrated to obtain roll, pitch and heading angles with occasional adjustment from the GNSS-derived attitude.

Existing systems for vehicle guidance may employ separate gyros, and separate GNSS positioning or attitude systems. However, such systems do not provide an integrated heading sensor based on GNSS as disclosed herein. Moreover, separate systems exhibit the limitations of their respective technologies as mentioned earlier. The exemplary embodiments as described herein eliminate the requirements of existing systems for other means to correct for vehicle roll. Moreover, an implementation of an exemplary embodiment also provides a relatively precise means of calculating heading and heading rate of change (turn rate), in both the short-term and the long-term.

Another benefit achieved by incorporating a GNSS-based heading sensor is the elimination or reduction of drift and biases resultant from a gyro-only or other inertial sensor approach. Yet another advantage is that heading may be computed while the vehicle is stopped or moving slowly, which is not possible in a single-antenna GNSS based approach that requires a vehicle velocity vector to derive heading. This can be very important in applications where a vehicle has to turn slowly to align with another path. During these slow turns the gyro can drift away, but by adding the use of a dual antenna GNSS solution the orientation of the gyro can be continuously corrected. This also permits immediate operation of a slow moving vehicle after being at rest, rather than requiring an initialization from motion. Yet another advantage of an exemplary embodiment is that a combination of the aforementioned sensors provides sufficient information for a feedback control system to be developed, which is standalone and independent of a vehicle's sensors or additional external sensors. Thus, such a system is readily maintained as vehicle-independent and may be moved from one vehicle to another with minimal effort. Yet another exemplary embodiment of the sensor employs global navigation satellite system (GNSS) sensors

and measurements to provide accurate, reliable positioning information. GNSS sensors include, but are not limited to GNSS, Global Navigation System (GLONAS), Wide Area Augmentation System (WAAS), and the like, as well as combinations including at least one of the foregoing.

An example of a GNSS is the Global Positioning System (GPS) established by the United States government that employs a constellation of 24 or more satellites in well-defined orbits at an altitude of approximately 26,500 km. These satellites continually transmit microwave L-band radio signals in two frequency bands, centered at 1575.42 MHz and 1227.6 MHz, denoted as L1 and L2 respectively. These signals include timing patterns relative to the satellite's onboard precision clock (which is kept synchronized by a ground station) as well as a navigation message giving the precise orbital positions of the satellites, an ionosphere model and other useful information. GNSS receivers process the radio signals, computing ranges to the GNSS satellites, and by triangulating these ranges, the GNSS receiver determines its position and its internal clock error.

In standalone GNSS systems that determine a receiver's antenna position coordinates without reference to a nearby reference receiver, the process of position determination is subject to errors from a number of sources. These include errors in the GNSS satellite's clock reference, the location of the orbiting satellite, ionosphere induced propagation delay errors, and troposphere refraction errors.

To overcome the errors of the standalone GNSS system, many positioning applications have made use of data from multiple GNSS receivers. Typically, in such applications, a reference receiver, located at a reference site having known coordinates, receives the GNSS satellite signals simultaneously with the receipt of signals by a remote receiver. Depending on the separation distance between the two GNSS receivers, many of the errors mentioned above will affect the satellite signals equally for the two receivers. By taking the difference between signals received both at the reference site and the remote location, the errors are effectively eliminated. This facilitates an accurate determination of the remote receiver's coordinates relative to the reference receiver's coordinates.

The technique of differencing signals from two or more GNSS receivers to improve accuracy is known as differential GNSS (DGNSS or DGPS). Differential GNSS is well known and exhibits many forms. In all forms of DGNSS, the positions obtained by the end user's remote receiver are relative to the position(s) of the reference receiver(s). GNSS applications have been improved and enhanced by employing a broader array of satellites such as GNSS and WAAS. For example, see commonly assigned U.S. Pat. No. 6,469, 663 B1 to Whitehead et al. titled Method and System for GNSS and WAAS Carrier Phase Measurements for Relative Positioning, dated Oct. 22, 2002, the disclosures of which are incorporated by reference herein in their entirety. Additionally, multiple receiver DGNSS has been enhanced by utilizing a single receiver to perform differential corrections. For example, see commonly assigned U.S. Pat. No. 6,397, 147 B1 to Whitehead titled Relative GNSS Positioning Using a Single GNSS Receiver with Internally Generated Differential Correction Terms, dated May 28, 2002, the disclosures of which are incorporated by reference herein in their entirety.

II. GNSS and Gyro Control System and Method

Referring now to FIGS. 1 through 4, an illustrative vehicle **10** is depicted including a sensor system **20** in accordance with an exemplary embodiment. Referring also to FIGS. 2 and 3, block diagrams of the sensor system **20** are

depicted. The sensor system **20** includes, but is not limited to a GNSS attitude system **22**, comprising at least a GNSS receiver **24** and an antenna **26**. The GNSS receiver/antenna systems comprising GNSS attitude system **22** cooperate as a primary receiver system **22a** and a secondary receiver system **22b**, with their respective antennas **26a** and **26b** mounted with a known separation. The primary receiver system **22a** may also be denoted as a reference or master receiver system, while the secondary receiver system **22b** may also be denoted as a remote or slave receiver system. It will also be appreciated that the selection of one receiver as primary versus secondary need not be of significance; it merely provides a means for distinguishing between systems, partitioning of functionality, and defining measurement references to facilitate description. It should be appreciated that the nomenclature could readily be transposed or modified without impacting the scope of the disclosure or the claims.

The sensor system **20** is optionally configured to be mounted within a single enclosure **28** to facilitate transportability. In an exemplary embodiment, the enclosure **28** can be any rigid assembly, fixture, or structure that causes the antennas **26** to be maintained in a substantially fixed relative position with respect to one another. In an exemplary embodiment, the enclosure **28** may be a lightweight bracket or structure to facilitate mounting of other components and transportability. Although the enclosure **28** that constrains the relative location of the two antennas **26a** and **26b** may have virtually any position and orientation in space, the two respective receivers **24** (reference receiver **24a** and remote receiver **24b**) are configured to facilitate communication with one another and resolve the attitude information from the phase center of the reference antenna **26a** to the phase center of the remote antenna **26b** with a high degree of accuracy.

Yet another embodiment employs a GNSS sensor **20** in the embodiments above augmented with supplementary inertial sensors **30** such as accelerometers, gyroscopes, or an attitude heading reference system. More particularly, in an implementation of an exemplary embodiment, one or more rate gyro(s) are integrated with the GNSS sensor **20**.

In yet another exemplary embodiment, a gyro that measures roll-rate may also be combined with this system's GNSS-based roll determination. A roll rate gyro denoted **30b** would provide improved short-term dynamic rate information to gain additional improvements when computing the sway of the vehicle **10**, particularly when traveling over uneven terrain.

It will be appreciated that to supplement the embodiments disclosed herein, the data used by each GNSS receiver **24** may be coupled with data from supplementary sensors **50**, including, but not limited to, accelerometers, gyroscopic sensors, compasses, magnetic sensors, inclinometers, and the like, as well as combinations including at least one of the foregoing. Coupling GNSS data with measurement information from supplementary sensors **30**, and/or correction data for differential correction improves positioning accuracy, improves initialization durations and enhances the ability to recover for data outages. Moreover, such coupling may further improve, e.g., reduce, the length of time required to solve for accurate attitude data.

It will be appreciated that although not a requirement, the location of the reference antenna **26a** can be considered a fixed distance from the remote antenna **26b**. This constraint may be applied to the azimuth determination processes in order to reduce the time required to solve for accurate azimuth, even though both antennas **26a** and **26b** may be

moving in space or not at a known location. The technique of resolving the attitude information and position information for the vehicle **10** may employ carrier phase DGNSS techniques with a moving reference station. Additionally, the use of data from auxiliary dynamic sensors aids the development of a heading solution by applying other constraints, including a rough indication of antenna orientation relative to the Earth's gravity field and/or alignment to the Earth's magnetic field.

Producing an accurate attitude from the use of two or more GNSS receiver and antenna systems **22** has been established in the art and therefore will not be expounded upon herein. The processing is utilized herein as part of the process required to implement an exemplary embodiment.

Referring also to FIG. **4**, a mechanism for ensuring an accurate orientation of the sensor system **20** to the vehicle **10** may be provided for by an optional mounting base **14** accurately attached to the enclosure **28**. An accurate installation ensures that substantially no misalignment error is present that may otherwise cause the sensor system **20** to provide erroneous heading information. The mounting base **14** is configured such that it fits securely with a determinable orientation relative to the vehicle **10**. In an exemplary embodiment, for example, the mounting base **14** is configured to fit flatly against the top surfaces of the vehicle **10** to facilitate an unimpeded view to the GNSS satellites.

With the sensor system **20** affixed and secured in the vehicle **10** power up and initialization of the sensor system **20** is thereafter executed. Such an initialization may include, but not be limited to, using the control system **100** to perform any initialization or configuration that may be necessary for a particular installation, including the configuration of an internal log file within the memory of the sensor system **20**.

The sensor system **20** may further include additional associated electronics and hardware. For example, the sensor system **20** may also include a power source, e.g., battery, or other power generation means, e.g., photovoltaic cells, and ultrahigh capacity capacitors and the like. Moreover, the sensor system **20** may further include a control system **100**. The control system **100** may include, without limitation, a controller/computer **102**, a display **104** and an input device **106**, such as a keypad or keyboard for operation of the control system **100**. The controller **102** may include, without limitation, a computer or processor, logic, memory, storage, registers, timing, interrupts, input/output signal interfaces, and communication interfaces as required to perform the processing and operations prescribed herein. The controller preferably receives inputs from various systems and sensor elements of the sensor system **20** (GNSS, inertial, etc.), and generates output signals to control the same and direct the vehicle **10**. For example, the controller **102** may receive such inputs as the GNSS satellite and receiver data and status, inertial system data, and the like from various sensors. In an exemplary embodiment, the control system **100** computes and outputs a cross-track and/or a direction error relating to the current orientation, attitude, and velocity of the vehicle **10** as well as computing a desired swath on the ground. The control system **100** will also allow the operator to configure the various settings of the sensor system **20** and monitor GNSS signal reception and any other sensors of the sensor system **20**. In an exemplary embodiment, the sensor system **20** is self-contained. The control system **100**, electronics, receivers **24**, antennas **26**, and any other sensors, including an optional power source, are contained within the enclosure **28** to facilitate ease of manipulation, transportability, and operation.

Referring now to FIG. 5, a flowchart diagrammatically depicting an exemplary methodology for executing a control process 200 is provided. An exemplary control process 200, such as may be executed by an operator in conjunction with a control system 100, acts upon information from the sensor system 20 to output cross-track and/or direction error based upon corrected 3-D position, velocity, heading, tilt, heading rate (degrees per second), radius of curvature and the like.

System 22a computes its position, denoted $p_1 (x_1, y_1, z_1)$. Referring now to block 220, the secondary receiver and antenna system 22b computes its position, denoted $p_2 (x_2, y_2, z_2)$. Referring now to block 230, optionally additional receiver and antenna system(s) 22 compute their respective positions, denoted $p_3 (x_3, y_3, z_3), \dots p_n (x_n, y_n, z_n)$.

At process block 240, employing a geometric calculation the heading is computed as the vector perpendicular to the vector joining the two positions, in the horizontal plane (assuming they are aligned with the vehicle 10). Furthermore, at block 250 the roll of the vehicle 10 may readily be computed as the arc-tangent of the ratio of the difference in heights of the two antennas 26a and 26b divided by the spacing between their phase centers (a selected distance within the enclosure 28). It will be appreciated that optionally, if additional receiver and antenna systems are utilized and configured for additional measurements, the pitch and roll angles may also be computed using differential positioning similar to the manner for computing heading. Therefore, in FIG. 5, optionally at process block 260, the pitch and/or roll may be computed.

Continuing with FIG. 5, at process block 270, using the computed roll angle and a known antenna height (based on the installation in a given vehicle 10), the actual position at the center of the vehicle 10 projected to the ground may be calculated. This position represents a true ground position of the vehicle 10. Once the ground position is known, the error value representing the difference between where the vehicle should be based on a computed swath or track, and where it actually is, can be readily calculated as shown at block 280.

Optionally, the vector velocities of the vehicle 10 are also known or readily computed based on an existing course and heading of the vehicle 10. These vector velocities may readily be utilized for control and instrumentation tasks.

Turning now to FIG. 6, in another exemplary embodiment a steering control process 300 can utilize the abovementioned information from the sensor system 20 to direct the vehicle motion. At process block 310 the steering control may be initiated by obtaining the computed errors from process 200. Turning to block 320, the steering control process 300 may be facilitated by computing a steering control command based on a proportionality factor times the difference in desired position versus actual position (computed position error), plus a second proportionality factor times the difference in desired heading versus actual heading (heading error). The second proportionality factor ensures that when the vehicle attains the desired position it is actually directed to the correct heading, rather than crossing the track. Such an approach will dramatically improve steering response and stability. At process block 330, a steering command is generated and directed to the vehicle 10.

Moreover, continuing with FIG. 6, optionally a recursive adaptive algorithm may also be employed to characterize the vehicle response and selected dynamic characteristics. In an exemplary embodiment, the sensor system 20 applies selected control values to the vehicle steering control mechanism as depicted at optional block 340 and block 330. The sensor system 20 measures the response of the vehicle

10 as depicted at process block 350 and calculates the response times and characteristics for the vehicle. For example, a selected command is applied and the proportionality of the turn is measured given the selected change in steering. Turning to process block 360, the responses of the vehicle 10 are then utilized to calibrate the control commands applying a modified control command to achieve a desired response. It will be appreciated that such an auto-calibration feature would possibly be limited by constraints of the vehicle to avoid excess stress or damage as depicted at 370.

It will be appreciated that while a particular series of steps or procedures is described as part of the abovementioned alignment process, no order of steps should necessarily be inferred from the order of presentation. For example, the process 200 includes installation and power up or initialization. It should be evident that power-up and initialization could potentially be performed and executed in advance without impacting the methodology disclosed herein or the scope of the claims.

It should further be appreciated that while an exemplary partitioning functionality has been provided, it should be apparent to one skilled in the art that the partitioning could be different. For example, the control of the primary receiver 24a and the secondary receiver 24b, as well as the functions of the controller 102, could be integrated in other units. The processes for determining the alignment may, for ease of implementation, be integrated into a single receiver. Such configuration variances should be considered equivalent and within the scope of the disclosure and claims herein.

The disclosed invention may be embodied in the form of computer-implemented processes and apparatus for practicing those processes. The present invention can also be embodied in the form of computer program code containing instructions embodied in tangible media, such as floppy diskettes, CD-ROMs, DVD's, flash drives, hard drives, or any other computer-readable storage medium 80 wherein the computer becomes an apparatus for practicing the invention when the computer program code is loaded into and executed by the computer. The present invention can also be embodied in the form of computer program code stored in a storage medium or loaded into and/or executed by a computer, for example. The present invention can also be embodied in the form of a data signal 82 transmitted by a modulated or unmodulated carrier wave, over a transmission medium, such as electrical wiring or cabling, through fiber optics, or via electromagnetic radiation. When the computer program code is loaded into and executed by a computer, the computer becomes an apparatus for practicing the invention. When implemented on a general-purpose microprocessor, the computer program code segments configure the microprocessor to create specific logic circuits.

III. Alternative Aspect GNSS Control Systems and Methods

FIG. 7A shows another alternative aspect of the invention including a GNSS antenna and gyroscope attitude system 402 with antennas 405, 406 separated by a rigid link 407. In a typical application, the rigid link 407 is attached to the vehicle 10 and extends along the X (transverse) axis or transversely with respect to the vehicle's direction of travel, which generally corresponds to the Y (heading) axis. Alternatively, the vehicle 10 itself can provide the rigid link between the antennas 405, 406, for example, by mounting the antennas 405, 406 at predetermined, fixed locations on the roof of the vehicle cab with a predetermined, fixed distance therebetween. Another alternative is to provide a GNSS attitude device with antennas, receivers and sensors (e.g., gyroscopes (gyros), accelerometers, and other sensors)

in a self-contained, unitary enclosure, such as the device **20** shown in enclosure **28** in FIG. **4**. Regardless of the antenna-mounting structure, the orientation of the antenna pair and the rigid link **407** (or vehicle **10**) is determined with respect to an Earth-fixed coordinate system. The XYZ axes shown in FIG. **7A** provide an example for defining this relation. Roll and yaw gyros **430**, **440** are generally aligned with the Y and Z axes respectively for detecting and measuring vehicle **10** attitude changes with respect to these axes.

With the system **402** installed on a vehicle **10** (FIG. **8**), the two antennas **405**, **406** can provide angular orientations with respect to two axes. In the example shown, angular orientation with respect to the Y (heading) axis corresponds to vehicle roll and with respect to the Z (vertical) axis corresponds to vehicle yaw. These orientations are commonly of interest in agricultural vehicles whereby this is the preferred mounting and orientation arrangement for such applications. The vehicle's roll most adversely affects GNSS-measured vehicle cross-track error. By measuring the vehicle's roll, such cross-track errors can be compensated for or eliminated. Such roll-induced cross-track errors include variable roll errors due to uneven terrain and constant roll errors due to hill slopes. It will be appreciated that adding a third antenna provides three-axis (XYZ) attitude solutions corresponding to pitch, roll, and yaw. Of course, reorienting the two-antenna system **402** can provide other attitude solutions. For example, locating the antennas' baseline (aligned with the rigid link **407**) fore-and-aft along the vehicle's Y axis will provide pitch and yaw attitudes.

FIG. **7B** shows the system **402** in a yaw attitude or condition whereby the vehicle **10** has deviated from a desired heading along the Y axis to an actual heading by a yaw angle θ_y . In other words, the vehicle **10** has rotated (yawed) clockwise with respect to the Z axis. FIG. **7C** shows the system **402** in a roll attitude or condition whereby the vehicle **10** has deviated from level to a tilt or roll angle of θ_R . In other words, the vehicle **10** has rotated (rolled) counterclockwise with respect to the Y axis.

The system **402** includes roll and yaw gyros **430**, **440** mounted and oriented for detecting vehicle rotational movement with respect to the Y and Z axes. The system **402** represents a typical strap-down implementation with the vehicle **10**, antennas **405**, **406** and gyros **430**, **440** rigidly connected and moving together. A body-fixed coordinate system is thus defined with the three perpendicular axes XYZ.

In all but the most extreme farmlands, the vehicle **10** would normally deviate relatively little from level and horizontal, usually less than 30° in most agricultural operations. This simplifies the process of calibrating the gyros **430**, **440** using the GNSS attitude system **402** consisting of two or more antennas **405**, **406**. For simplicity, it is assumed that the body-fixed axes XYZ remain relatively close to level. Thus, the change in the heading (yaw) angle θ_y of FIG. **7B** is approximately measured by the body-fixed yaw gyro **440**, even though there may be some small discrepancy between the axes of rotation. Similar assumptions can be made for the roll angle θ_R (FIG. **7C**), which is approximately measured by the body-fixed roll gyro **430**. A similar assumption could be used for measuring pitch attitude or orientation angles with a pitch gyro.

This simplifying assumption allows the gyros to be decoupled from one another during integration and avoids the necessity of using a full strap-down quaternion implementation. For example, heading deviation is assigned only to the yaw gyro **440** (gyro axis perturbations from the assumed level axis alignment are ignored). Similarly,

vehicle roll is assumed to be measured completely by a single roll gyro **430**. GNSS attitude-measured heading and roll can then be used to calibrate the gyros **430**, **440**. Such simplifying assumptions tend to be relatively effective, particularly for agricultural operations on relatively flat, level terrain. Alternatively, a full six-degrees-of-freedom strap-down gyro implementation with quaternion integration could be employed, but such a solution would normally be excessive and represent an ineffective use of computing resources, unless an inertial navigation system (INS) was also being used to backup GNSS, for example, in the event of GNSS signal loss.

For the purpose of calibrating the gyroscopes **430**, **440**, the angles measured by the GNSS attitude system **402** are used as truth in a Kalman filter estimator of gyro bias and scale factor errors. Over a small interval of time, T, the following equation holds:

$$\vec{\theta}_T = A\theta_{true} + BT$$

Where

$\vec{\theta}$ = average gyro reading over

$$T = 1/n \sum_n \dot{\theta}_{gyro}$$

(with n readings taken over time T)

θ_{true} = truth angular change over interval T as measured by the GNSS attitude system.

A = gyro scale factor error

B = gyro rate bias error

A two state Kalman filter is defined to have the gyro rate basis and scale factor error as states. The Kalman process model is a first-order Markov:

$$X_{k+1} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} X_k + \begin{bmatrix} \sigma_A & 0 \\ 0 & \sigma_B \end{bmatrix} W_k$$

where the state vector $X = [A \ B]$

Here σ_A and σ_B are noise amplitudes and W is white noise. This dictates what is known as a random walk of the state [A B]. The designer of the Kalman filter chooses σ_A and σ_B according to how rapidly the bias and scale factor errors are expected to vary (usually variations due to temperature dependencies of scale and bias in a low cost gyro). Typical variations, especially of the scale factor, are quite small (A and B are nearly constant), and σ_A and σ_B are chosen accordingly. Typical values for a low-cost gyroscope, using a time interval T are:

$$\sigma_A = \frac{0.02T}{1200}, \sigma_B = \frac{T}{1200}$$

where T is expressed in seconds and 1200 means 1200 seconds. For example, here the random walk is chosen to cause a drift in scale factor of 0.02 in 1200 seconds. The Kalman measurement equation is:

$$y = Hx + v$$

Where

$y = \vec{\theta}_{gyro} T$, $H = [\theta_{true} \ T]$ and v is measurement noise. The Kalman covariance propagation and gain calculation is designed according to well-known techniques.

Similar Kalman filters are deployed in both yaw and roll (and/or pitch) channels. The GNSS attitude devices **20** provides a reference yaw and roll that act as the Kalman measurements enabling the calibration of gyro rate basis and scale factor errors. The GNSS device provides heading and roll, even when the vehicle is stationary or traveling in reverse. This provides a significant advantage over single-antenna systems which provide a vehicle direction only when moving (i.e., a velocity vector). The multi-antenna attitude device **20** enables continuous calibration regardless of whether or not and in what direction the vehicle **10** is moving.

The calibrated gyros **430**, **440** are highly advantageous in a vehicle steering control system. High precision heading and heading-rate produced by the calibrated yaw gyro is a very accurate and instantaneous feedback to the control of vehicle changes in direction. The angular rate produced by the gyro is at least an order of magnitude more accurate than the angular rate produced by pure GNSS systems, even those with multiple antennas. The system **402** is also very responsive. The feedback control needs such relatively high accuracy and responsiveness in heading and heading-rate to maintain control loop stability. It is well known that rate feedback in a control loop enhances stability. On a farm vehicle, where vehicle dynamics may not be fully known or modeled, this aspect is particularly important. The rate term allows a generic control system to be developed which is fairly insensitive to un-modeled vehicle dynamics. A relatively accurate heading and heading-rate-of-turn can be calculated for use in a vehicle automatic steering system.

Another advantage of the system **402** is that a gyro calibrated to measure tilt angle can provide the vehicle's tilt much more accurately than a system relying exclusively on GNSS positioning signals. This advantage is particularly important in high-precision autosteering, e.g., to the centimeter level. Errors in GNSS attitude are effectively increased by the ratio of the antenna spacing to the mounted height of the antennas above the ground, as illustrated in FIG. **8**, which shows an attitude system **402** comprising a pair of antennas **405**, **406** connected by a link **407**, as described above. The system **402** is shown tilted through a tilt (roll) angle θ_R . An imaginary antenna height line perpendicular to the rigid link **407** is projected to the "true" ground position of the vehicle **10** in FIG. **8** and forms the roll angle with respect to the Z axis. The relative antenna height differential can be projected along the vertical Z axis to a ground intercept point and establishes a cross-track error (distance between the vehicle true ground position and the Z axis ground intercept point), whereby errors in the antenna height differential are amplified by the ratio of the rigid link **407** length to the antenna height. The spacing of the antennas **405**, **406**, which corresponds to the length of the rigid link **407**, is typically limited by the width of the vehicle **10**, which can be relatively tall, thereby resulting in a relatively large antenna height-to-spacing ratio, e.g., five-to-one. Furthermore, noise-induced errors present in GNSS relative antenna height differentials (e.g., carrier phase noise, etc.) will be multiplied by this ratio, which can cause steering errors, including steering oscillations, etc.

The GNSS attitude system **402** utilizes a roll gyro (e.g., **430**) for measuring rate-of-change of the roll angle, rather than the absolute roll angle, which rate of change is integrated to compute absolute roll angle. The constant of integration can be initialized to the current GNSS-derived roll angle and then subsequently steered to the GNSS roll angle by filtering with a Hatch filter or similar filter used for smoothing the code phase against the carrier phase in the

GNSS receivers. Relatively smooth vehicle roll estimates can thus be achieved with a gyro.

More specifically, in an exemplary embodiment, the filtering is supplemented by the equation:

$$\theta_{filter}(k) = \Delta_{gyro}(k) + Gain * [\theta_{GNSS}(k) - \theta_{filter}(k-1) - \Delta_{gyro}(k)]$$

$$\Delta_{gyro}(k) = \theta_{gyro}(k) - \theta_{gyro}(k-1)$$

Where $\theta_{filter}(k)$ is the desired output roll angle (at time k) smoothed by gyro roll angle, but steered to GNSS roll angle. The GNSS roll (at time k) is $\theta_{GNSS}(k)$ while the raw gyro angular reading is $\theta_{gyro}(k)$ which is obtained by integrating gyro angular rate. The difference in gyro integrated rate over one time interval (k-1 to k) is denoted $\Delta_{gyro}(k)$. The filter bandwidth and weighting of the GNSS roll angle into the solution is set by the filter's gain (denoted Gain). One method to choose the gain is to assign $Gain = T/\tau$ where T is the time span from epoch to epoch and τ is a time-constant, typically much larger than T. The smaller the Gain, the less the GNSS roll angle is weighted into the solution. The gain is chosen to give a smooth filtered roll output, dominated by the low gyro noise characteristics, but also maintaining alignment with GNSS roll. Since the gyro is calibrated in terms of its scale and bias errors per the methods described earlier, the gain can be chosen to be very small (much less than 1) and still the filtered roll angle closely follows the GNSS roll angle, but without the noise of the GNSS derived roll angle. Similar schemes can be deployed for pitch and heading angles if needed, all with the benefit of improved steering if such angles are used in the steering control feedback.

FIGS. **9** and **10** show a GNSS and gyroscopic control system **502** comprising an alternative aspect of the present invention in a tractor and sprayer agricultural equipment application **504**. The vehicle (e.g., a motive component or tractor) **10** is connected to a working component (e.g., a sprayer) **506** by an articulated connection **508**, which can comprise a conventional tongue-and-hitch connection, or a powered, implement steering system or hitch, such as those shown in U.S. Pat. No. 6,865,465, No. 7,162,348 and No. 7,373,231, which are assigned to a common assignee herewith and are incorporated herein by reference.

The tractor **10** and the sprayer **506** mount respective tractor and sprayer GNSS antenna and gyroscope attitude subsystems **510**, **512**, which are similar to the system **402** described above and provide GNSS-derived position and attitude outputs, supplemented by gyro-derived rate of rotation outputs for integration by the control system **502**. The sprayer **506** includes a spray boom **514** with multiple nozzles **516** providing spray patterns **518** as shown, which effectively cover a swath **520**. The system **502** can be programmed for selectively controlling the nozzles **516**. For example, a no-spray area **522** is shown in FIG. **9** and can comprise, for example, an area previously sprayed or an area requiring spray. Based on the location of the no-spray area **522** in relation to the spray boom **514**, one or more of the nozzles **516** can be selectively turned on/off. Alternatively, selective controls can be provided for other equipment, such as agricultural planters wherein the seed boxes can be selectively turned on/off.

FIG. **10** shows some of the major components of the system **502**, including the GNSS antenna and gyroscope attitude subsystems **510**, **512** with antennas **405**, **406** separated by rigid links **407**, as described above, and inertial gyros **514**. The tractor and implement **10**, **506** can be equipped with comparable systems including DGNSS

receivers **524**, suitable microprocessors **526** and the inertial gyros **529**. Additional sensors **528** can include wheel counters, wheel turn sensors, accelerometers, etc. The system components can be interconnected by a controller-area network or CAN connection **530**. Alternatively, components

can be wirelessly interconnected, e.g., with various types of RF transmitters and receivers or transceivers. In operation, the functions described above can be implemented with the system **502**, which has the additional advantage of providing GNSS and gyro-derived positioning and attitude signals independently from the tractor **10** and the implement **506**. Such signals can be integrated by one or both of the microprocessors **526**. The tractor **10** can be automatically steered accordingly whereby the implement **506** is maintained on course, with the additional feature of selective, automatic control of the nozzles **516**. For example, FIG. **9** shows the course of the tractor **10** slightly offset to the course of the sprayer **516**, which condition could be caused by a downward left-to-right field slope. Such sloping field conditions generate roll attitudes, which could also be compensated for as described above. For example, the system **502** can adjust the output from the spray nozzles **516** to compensate for such variable operating conditions as sloping terrain, turning rates, tire slippage, system responsiveness and field irregularities whereby the material is uniformly applied to the entire surface area of the field. Moreover, the GNSS-derived positioning and heading information can be compared to actual positioning and heading information derived from other sensors, including gyros, for further calibration.

IV. Multi-Antenna High Dynamic Roll Compensation and Rover L1 RTK

Another alternative aspect GNSS guidance system **602** is shown in FIGS. **11** and **12** and provides high dynamic roll compensation, heading and rate-of-turn (ROT) in an RTK system including a GNSS receiver **604** including an RF converter **606** connected to a multi-channel tracking device **608** and first and second antennas **610**, **612**, which can be mounted on top of a vehicle **10** in fixed relation defining a transverse (X axis) fixed baseline **614**. The receiver **604** provides a GNSS data output to a guidance processor (CPU) **616**, which includes a GUI/display **618**, a microprocessor **620** and media (e.g., for data storage) **622**. A steering valve block **624** includes autosteer logic **626**, hydraulic valves **628** and steering linkage **630**. A wheel sensor **632** is connected to the steering valve block **624**, which in turn is connected to the guidance processor **616** by a suitable CAN bus **634**.

GNSS positioning signals are received from a constellation of GNSS satellites and an RTK base transceiver **636**, which includes a receiver **638** and a transmitter **640** for transmitting carrier phase signals to a rover RTK receiver, such as the receiver **604**. By using GNSS positioning signals from the satellites and correctional signals from the RTK base transceiver **636**, the guidance system **602** can calculate a relatively accurate position relative to the base transceiver **636**, which can be located at a predetermined position, such as a benchmark. The guidance system **602** described thus far is an RTK system utilizing a dual frequency receiver and is capable of achieving sub-centimeter accuracy using the carrier phase signals.

Roll compensation, heading, and rate of turn can all be calculated using vector-based heading (yaw and roll) information derived from the rover GNSS receiver **604**. High-dynamic vehicle roll is a problem with certain applications, such as agricultural vehicles, which traverse uneven terrain and tend to be relatively tall with antennas mounted three meters or more above ground level. Antenna arrays can

swing significant distances from side to side with vehicle roll, as indicated by a roll arrow **644**. Such deviations can be detrimental to precision farming, and require compensation. The fixed-baseline vehicle antennas **610**, **612** provide the necessary dynamic vector outputs for processing and compensation by the steering valve block **624**. For example, the microprocessor **620** can be preprogrammed to instantly respond to such roll errors by providing counteracting output signals via the CAN bus **634** to autosteer logic **626**, which controls the hydraulic valves **628** of the steering valve block **624**. A slight delay phase shift can be programmed into the microprocessor **620**, thus reflecting the inherent lag between vehicle roll and the steering system reaction. The delay phase shift can be adjustable and calibrated for accommodating different equipment configurations. The GNSS receiver **604** output provides relatively accurate guidance at slow speeds, through turns and in reverse without relying on sensing vehicle motion via an inertial navigation system (INS), utilizing gyroscopes and/or accelerometers. Moreover, the guidance system **602** can eliminate the calibration procedures normally needed for INS-corrected systems.

The system **602** can likewise provide high dynamic yaw compensation for oscillation about the vertical Z axis using the two-antenna fixed baseline configuration of the receiver **604**. Adding a third antenna would enable high dynamic compensation with respect to all three axes XYZ e.g., in a six-degrees-of-freedom mode of operation.

Providing multiple antennas **610**, **612** on a rover vehicle **10** can significantly improve the ability to resolve integer ambiguities by first obtaining an attitude solution by solving for the locations of the rover antennas **610**, **612** with respect each other. Then, using the non-relative locations and the known relative ambiguities, solving for the global ambiguities using observations taken at each antenna **610**, **612**. The number of observations is thus significantly increased over conventional RTK. Solving the global ambiguities enables locating the rover antennas **610**, **612** in a global sense relative to a base station **636**. Using multiple antennas in this manner enables using L1 single frequency receivers, which tend to be less expensive than dual frequency (L1 and L2) receivers, as in conventional RTK systems. An exemplary method consists of:

1. Transmitting code and carrier phase data from a base station **636** to a multiple antenna rover system (e.g., **602**).
2. At the rover **602** side, determining the relative locations and the relative ambiguities of the multiple antennas using an attitude solution taking advantage of known geometry constraints and/or a common clock. Such a method is disclosed in U.S. Pat. No. 7,388,539, which is assigned to a common assignee herewith and is incorporated herein by reference.
3. Optionally store off the attitude solution (locations and ambiguities) for later time-tag matching with the data from the base station **636**. Optionally, also store off the current GNSS observations (carrier phase) for the same purpose. Although this step is not necessary, time tag matching of base and rover data improves results by avoiding extrapolation errors.
4. Form single or double difference equations and solve for the global ambiguities using knowledge of the relative antenna locations and/or common clocks and/or the relative ambiguities.

Example using a two-antenna rover system (e.g., 602):

At antenna 1 (e.g., 610) of the rover, we can write the equation

$$R1=[A]x1-N1,$$

where R1 is a carrier phase observation vector (single or double difference) at antenna 1, A is a design matrix, X1 is the location vector of antenna 1 (may include clock if single differencing is used), and N1 is an ambiguity vector for antenna 1.

Similarly, at antenna 2 (e.g., 612) we can write

$$R2=[A]x2-N2$$

Where R2 is a carrier phase observation vector at antenna 1, A is a design matrix, X2 is the location vector of antenna 2, and N2 is an ambiguity vector for antenna 2.

Note, that in this example, the design matrix A is taken to be the same in both antenna equations. But, this is true only if both antennas see the same satellites. A more general example would use separate A1 and A2 for the two equations.

Solving an attitude solution (for example, see U.S. Pat. No. 7,388,539), we find the relative antenna displacement V, and the relative ambiguity M where

$$V=x2-x1$$

and

$$M=N2-N1$$

Thus, combining the above equations, we have

$$R1=[A]x1-N1$$

$$R2=[A](x1+V)-(N1+M)$$

Rearranging gives

$$R1=[A]x1-N1$$

$$R2-[A]V+M=[A]x1-N1$$

And, combining into a single vector equations gives

$$R=[A]x1-N$$

Where

$$R=[R1,R2-[A]V+M]^T \text{ and } N=[N1,N1]^T$$

Where 'T' denotes transpose

Referring to the above example, twice as many equations are obtained for the same number of unknowns (e.g. X1 and N1). Solving for the global integer ambiguity N1 is facilitated by the multiple available equations.

Multiple antennas can also be utilized at the base and would provide the advantage of canceling multipath signals. However, multiple antennas on the rover are generally preferred because they provide attitude for the rover 10, which is generally not of concern for the base 636.

V. Moving Baseline Vehicle/Implement Guidance Systems

Alternative embodiment multiple-antenna GNSS guidance systems are shown in FIGS. 13-18 and utilize a moving baseline between a vehicle-mounted antenna(s) and an implement-mounted antenna. Independent implement steering can be accomplished with a powered, implement steering system or hitch, such as those shown in U.S. Pat. No. 6,865,465, No. 7,162,348 and No. 7,373,231, which are assigned to a common assignee herewith and are incorporated herein by reference.

FIGS. 13-14 show a GNSS guidance system 726 comprising another modified embodiment of the present inven-

tion and including a vehicle 10 connected to an implement 728 by a hitch 730. The hitch 730 permits the implement 728 to move through three axes of movement relative to the vehicle 10 as the system 726 maneuvers and traverses ground with irregularities causing the vehicle 10 and the implement 728 to yaw, pitch, and roll somewhat independently of each other. A moving baseline 732 is defined between points on each, e.g., between a vehicle antenna 753 and an implement antenna 756. The moving baseline 732 is generally a 3D vector with variable length and direction, which can be derived from the differences between the vehicle antenna 753 location (X1, Y1, Z1) and the implement antenna location (X3, Y3, Z3), or other predetermined point locations on the vehicle 10 and the implement 728. The guidance system 726 includes a single GNSS receiver 734 (e.g., a single printed circuit board (PCB) receiver) receiving ranging data streams from the antennas 753, 756, which can include the normal front end RF down-converter components. Using the geodetic-defined position solutions for the antennas 753, 756, the moving baseline 732 is defined and used by a guidance CPU 736 in real-time for computing guidance solutions, which include steering command outputs to the steering valve block 738. The varying separation of the antennas 753, 756 occurs both at the start of attitude acquisition and during operation.

FIG. 15 shows another alternative aspect vehicle/implement GNSS guidance system 740 with first and second vehicle antennas 753, 754, which can include front end down converter RF components providing ranging signal outputs, along with the implement antenna 756, to the single GNSS receiver 734 as described above. The vehicle antennas 753, 754 define a fixed baseline 754 by their respective positions (X1, Y1, Z1), (X2, Y2, Z2), which function to provide vector heading and rate-of-turn (ROT) output information. Such positioning data is input to the guidance CPU 736 by measuring yaw and roll attitudes whereby such guidance and performance information can be determined solely on GNSS-defined ranging data utilizing the fixed-relationship mounting of the vehicle antennas 753, 754 on the vehicle 10. Such information can be processed in connection with the implement antenna 756 position information in order to provide more complete GNSS positioning and guidance solutions, including travel paths for the vehicle 10 and the implement 728.

FIG. 16 shows another modified aspect GNSS positioning system 752, which includes first and second vehicle antennas 753, 754 at GNSS-defined positions (X1, Y1, Z1), (X2, Y2, Z2) respectively, which positions define a vehicle fixed baseline 755. The implement 728 includes first and second implement antennas 756, 757 at GNSS-defined positions (X3, Y3, Z3), (X4, Y4, Z4) respectively, which define an implement fixed baseline 758 and from which the guidance CPU 736 determines heading and ROT for the implement 728 using similar vector techniques to those described above. A movable baseline 759 can be defined between a vehicle antenna 753 and an implement antenna 756 as shown, or between other corresponding antenna pairs, or other predetermined locations on the vehicle 10 and the implement 728. The system 752 utilizes a single GNSS receiver 734 receiving input ranging information from the four antennas 753, 754, 756, 757 and providing a single output stream to the guidance CPU 736. It will be appreciated that various other antenna/receiver combinations can be utilized. For example, a third vehicle and/or implement antenna can be provided for 3-axis attitude computation. INS components, such as gyroscopes and/or accelerometers, can also be utilized for additional guidance correction,

although the systems described above can provide highly accurate guidance without such INS components, which have certain disadvantages.

FIG. 17 shows the 2+1 antenna system 740 operating in a guidance mode whereby a predetermined number of positions 790 at predetermined intervals are retained by the guidance CPU 736, thereby defining a multi-position “breadcrumb” tail 792 defining the most recent guidepath segment traversed by the vehicle 10 based on the locations of the vehicle antenna(s) 753 (754). Although the 2+1 antenna guidance system 740 is used as an example, the 1+1 antenna guidance system 726 and the 2+2 guidance system 752 can also be used in this mode and function in a similar manner, with more or less ranging signal sources. The guidance CPU 736 utilizes the retained tail “breadcrumb” positions 790 in conjunction with the GNSS-derived antenna locations for computing a crosstrack error representing implement 728 deviation from a desired guidepath 794, and the necessary steering signals for correcting the vehicle 10 course to maintain the implement 728 on track. Still further, in a multi-position tail 792 operating mode the high dynamic roll compensation function described above can be utilized to compensate for vehicle and/or implement roll using the fixed baseline(s) 746, 755, 758 for further guidance solution accuracy based solely on GNSS ranging information.

With the systems 726, 740 and 752, a single receiver can be used for achieving carrier phase relative accuracy, even without differential correction. A single clock associated with the receiver facilitates ambiguity resolution, as compared to dual receiver and dual clock systems. Direct connections among the components further enhance accuracy and facilitate high dynamic roll corrections, as described above. Continuous base and rover ranging data are available for positioning and control. With the 2+1 and the 2+2 configurations, the fixed baseline(s) provide heading and ROT guidance for the vehicle and/or the implement. Steering control for the vehicle is derived from crosstrack error computations utilizing the multi position tail 792.

FIG. 18 is a schematic block diagram showing the components of the GNSS guidance systems 726, 740 and 752. The vehicle 10 components include a GNSS receiver 734 including a first vehicle antenna 753, an optional second vehicle antenna 754, an RF down converter 764, a tracking device 766 and an optional rover RTK receiver 768. A guidance processor CPU 736 includes a GUI display 772, a microprocessor 774 and a media storage device 776. Vehicle steering 778 is connected to the guidance processor CPU 736 and receives steering commands therefrom. GNSS-derived data is transferred from the GNSS receiver 734 to the guidance processor CPU 736. The implement 728 mounts an implement positioning system 780 including a first implement antenna 756 and an optional second implement antenna 757, which are connected to the vehicle GNSS receiver 734 and provide GNSS data thereto. An implement steering subsystem 784 receives steering commands from the guidance processor CPU 736 via a CAN bus 786. The implement 728 is mechanically connected to the vehicle 10 by a hitch 788, which can be power-driven for active implement positioning in response to implement steering commands, or a conventional mechanical linkage. The hitch 788 can be provided with sensors for determining relative attitudes and orientations between the vehicle 10 and the implement 728.

VI. Multi-Vehicle GNSS Tracking Method

FIG. 19 shows a multi-vehicle GNSS tracking system 802 adapted for tracking primary and secondary rover vehicles

804, 806, which can comprise, for example, a combine and an offloading truck. Other exemplary multi-vehicle combinations include crop picking and harvesting equipment, snowplows, aircraft engaged in mid-air refueling, etc. Data transfer among the vehicles 804, 806 and a base transceiver 808 can be accomplished with short-range radio links, such as Bluetooth and Wi-Fi wireless technologies. For example, the base transceiver 808 can transmit corrections to the rovers 804, 806 at predetermined intervals of one second (i.e., 1 Hz).

Between the base transmissions the primary rover 804 can transmit its identifying information (ID) and GNSS-derived position and timing information to the secondary rover 806. The secondary rover 806 thus receives both differential corrections and the primary rover data over the same radio link, or through an additional radio link. Such data can comprise a multi-position tail 810 as described above and against which the secondary rover 806 can guide. For example, the secondary rover 806 can directly follow the primary rover 804 at a predetermined distance by aligning its travel path with the multi-position tail 810 at a predetermined following distance, or it can offset its own parallel travel path a predetermined offset distance, as shown in FIG. 19. The secondary rover 806 can position itself relative to the primary rover 804 based on either a predetermined time interval or a predetermined separation distance. As discussed above, the multi-position tail 810 can automatically update whereby only a predetermined number of detected positions are stored, which can correspond to a predetermined time duration or distance behind the primary rover 804.

FIG. 20 shows a schematic block diagram of components comprising the multi-vehicle tracking system 802. The onboard systems for the primary rover 804 and the secondary rover 806 can be similar to the vehicle-based GNSS guidance systems described above, with the addition of an inter-rover radio link 812.

VII. Alternative Embodiment Multi-Antenna System 902

FIG. 21 shows a multi-antenna, GNSS-based guidance system 902 installed on a motive component 904, herein exemplified by a tractor, towing a working component 906, herein exemplified by a towed implement, and collectively comprising a vehicle 907. Without limitation, the vehicle 907 is configured for agricultural operations. However, the system 902 could also be used for guiding and controlling a wide range of vehicles, equipment and machines. For example, the system 902 could be applied to earth-moving equipment, examples of which are shown in U.S. patent application Ser. No. 12/857, 298, which is assigned to a common assignee here with and is incorporated herein by reference. The motive and working components can be interconnected, articulated components of a piece of equipment, such as the base vehicle and boom assembly components of an excavator. Also shown are the three axes X, Y, and Z, and the positive directions of rotation about those axes, i.e., roll, pitch, and yaw respectively. Using three antennas 952, 954, 956, the GNSS guidance system 902 can track the motive component 904 and working component 906 in all six degrees of freedom and in relation to each other. The motive component 904 includes a motive component antenna 952, and the working component 906 includes first and second working component antennae 954, 956, i.e. a “1+2” configuration. Other tractor/implement antenna combinations could also be used, such as 2+2. This transfers not only positional information to a GNSS guidance computer 910, but also data on the slope of the earth below the vehicle 907 and whether the working component

906 is traveling laterally (“offset”) compared with the motive component 904, indicating a sliding motion and crosstrack displacement. In FIGS. 21 and 22 the antennas 952, 954, 956 are shown in “normal” positions with the working component 906 aligned with and positioned directly behind the motive component 904. The distances between the working component antennas and the motive component antenna can vary depending on the relative orientations of either the motive component 904 or the working component 906, or both.

FIG. 22 shows the motive component 904 towing the working component 906 with an articulated hitch 914 and a tongue 915, and the various attached sensors and systems which create an embodiment of the guidance path memory system 902. The GNSS system includes the antennas 952, 954, 956, a GNSS receiver 908, a guidance computer 910 including a microprocessor/CPU 909, a working component computer 913 including a microprocessor/CPU 923 and a graphical user interface (GUI) 911. This embodiment of the present invention uses differential GNSS (DGNSS) by using a base station 922 located generally in the vicinity of the work to be performed (FIG. 24). The base station 922 includes an antenna 924, a base receiver 926 and a base transmitter 927. The base and rover configuration is similar to other differential (DGNSS) guidance systems, such as the Outback S Series produced by Hemisphere GPS LLC of Calgary, Canada. The GNSS components are preferably configured to use carrier phase GNSS signals with a base-and-rover receiver combination, which is generally referred to as real-time kinematic (RTK). See U.S. Pat. No. 6,469,663, which is incorporated herein by reference. The guidance system 902 will track the three-dimensional position of the motive component 904 and the working component 906, along with the roll, pitch, and yaw (collectively attitude) of the motive component 904 and the working component 906, both independently and relative to each other. Additionally, the GNSS system 902 will determine the heading of the motive component 904, and will detect when the working component 906 is facing a different heading from the motive component 904 or if the working component 906 is moving laterally compared with the motive component 904, inferring that the working component 906 may have become misaligned due to a bump in the path or because the path is along a slope.

Also shown in FIGS. 22, 24 and 25 are several sensor devices for detecting other vehicle parameter values. These sensors include various vehicle sensors 912, a wheel compaction PSI sensor 916, a hitch feedback sensor 920, and various working component sensors 918. The various vehicle sensors 912 include a motive component wheel angle sensor 935, ground speed sensor 936, fuel sensor 937, RPM sensor 938, and various other optional sensors that detect variables of vehicle performance and may enhance the information received about the terrain being driven over. The various working component sensors 918 include a ground speed sensor 988, a working component wheel angle sensor 989, and compression sensors 990 for determining the amount of soil being compressed similar to the wheel compaction PSI sensor 916. The information harvested from these various sensors is taken and combined with the positional data received by the GNSS system 902, and finally computed by the guidance computer 910. The information is output to an external computer 934, as shown in FIG. 25, where it can be analyzed and future pre-planned paths can be designed based on the data gathered during the field pass.

Alternatively, the guidance computer 910 could calculate and modify its own stored, pre-planned path based on the

gathered data and programmed functions for dealing with different field conditions. The guidance computer 910 can be pre-programmed to adapt to field conditions in different ways depending on the circumstances. For example, the microprocessor 909 can be programmed to instruct an articulated hitch 914 that is included with an optional motorized component, such as the device covered by previously mentioned and incorporated U.S. Pat. No. 7,162,348, to adjust the position of the working component 906 relative to the motive component 904 depending on the severity of the slope as the vehicle 907 is traversing that slope. The computer 910 will update commands to the hitch 914 as data is reported by working component and motive component gyro sensors 921, 919 and other relevant sensors for detecting a change in pitch or roll. All of this can be performed in real time as data is reported to the guidance computer 910. The concept of real-time, pre-planned path modification for the present invention follows similar techniques as described in U.S. Patent Publication No. 2007/0021913, which is assigned to a common assignee herewith and incorporated herein by reference.

Also located on the motive component 904 is a steering controller 917 receiving steering commands from the guidance computer 910 and applying them to the motive component 904, steering it around the field. The guidance computer 910 also controls the power settings of the motive component 904, reducing or increasing speed, and optionally controls other vehicle 907 operations, e.g., adjusting the stiffness of shock absorbing components via adjustable hydraulic shock absorbers 958. A controller for controlling the amount of shock absorbed by the hydraulic shock absorbers 958 can be connected directly to and controlled by the guidance computer 910. This will allow the vehicle to increase the resistance of the shock absorbers 958 prior to the vehicle traversing a particularly rough terrain, or decrease their resistance for softer terrain, depending on performance desired from the vehicle 907. Similarly, other elements of the vehicle can be controlled in this way, which will lead to increased vehicle performance and control.

The use of a moving baseline 998 between at least three antennas 952, 954, 956, with two antennas located on the working component 906 and at least one on the motive component 904, allows the guidance system 902 to track the position of the working component relative to the motive component. The working component 906 may actually roll in one direction while the motive component 904 rolls in the opposite direction. Including additional data provided by a motive component inertial measurement unit (IMU) 919 and a working component IMU 921 allows the guidance computer 910 to distinguish yaw, pitch, and roll movement of the working component 906 relative to yaw, pitch, and roll movement of the motive component 904. Because the working component 906 is doing the actual work in a field, it is important to ensure that the working component 904 is being properly guided and aligned relative to the motive component 906. The use of an optional motorized hitch 914, as mentioned above, allows the guidance computer 910 to readjust and realign the working component 906 if the guidance system detects that it is no longer properly aligned. This optional aspect is further discussed in the previously mentioned and referenced U.S. Patent Publication No. 2009/0164067.

FIG. 23A demonstrates the relationship among the three antennas' 952, 954, 956 positions. Using basic trigonometric equations, unknown distances between antenna pairs can be solved and used by the guidance computer 910 to recalculate driving directions. The motive component antenna 952

location is denoted by A. The working component antennas B (954) and C (956) are located a fixed distance BC away from each other. The point where the hitch 914 pivots, allowing the working component 906 to rotate independent from the motive component 904, is at point F. The pivot arm is alternatively labeled the tongue 915. A point-of-interest (POI) directly below the motive component-mounted antenna. Point E is a point directly between the two working component-mounted antennas 954, 956.

The known distances include the distance between the working component-mounted antennas (BC) and the height (H) of the motive component-mounted antenna 952 above the working component-mounted antennas 954, 956. When the working component is directly behind the motive component, as depicted in FIG. 23A, and points B and C are at approximately the elevation of the point of interest (POI), several right-isosceles triangles are formed and the distances among the antennas can be computed.

FIGS. 23B and 23C show the trigonometric relationship changes when the working component 906 rotates about point F (hitch 914) via the tongue 915. The working component will shift in a direction along the X-Y plane, changing the moving baseline relationship AB and AC.

FIG. 23C demonstrates the positional relationship between the motive component-mounted antenna 952 at A and the working component-mounted antenna 956 at C as it moves from the starting position shown in FIG. 23A and moves to the ending position shown in FIG. 23B. The height 'h' is known, and the X, Y, and Z coordinates of both point A and point C are known. The coordinates of the Point of Interest (POI) are:

$$(X_1, Y_1, Z_{1.1}) = (X_1, Y_1, Z_1 - h)$$

Because point C and POI are at the same elevation, $Z_{1.1} = Z_3$. Thus, the distances d and d.1 can be calculated:

$$d = \sqrt{(X_3 - X_1)^2 + (Y_3 - Y_1)^2}$$

$$d.1 = \sqrt{(X_{3.1} - X_1)^2 + (Y_{3.1} - Y_1)^2}$$

And therefore:

$$\tan \theta = h/d$$

$$\tan \theta.1 = h/d.1$$

$$AC = h/\sin \theta$$

$$AC.1 = h/\sin \theta.1$$

Alternatively:

$$AC = \sqrt{(X_3 - X_1)^2 + (Y_3 - Y_1)^2 + (Z_3 - Z_1)^2}$$

$$AC.1 = \sqrt{(X_{3.1} - X_1)^2 + (Y_{3.1} - Y_1)^2 + (Z_3 - Z_1)^2}$$

$$\sin \theta = h/AC$$

$$\sin \theta.1 = h/AC.1$$

This alternative formula can be used because the three-dimensional points A and C can be determined by their actual GNSS positions as determined by GNSS satellite signals received by the various antennas 952, 954, 956.

This same method can be used as long as points B, C, and POI are at the same elevation; e.g. $Z_B = Z_C = Z_{POI}$, leaving Φ to equal 90° . The distances AB and AC will vary as the working component 906 is rotated about point F as shown in FIG. 22C. Using the formulas above, the distances AB and AC can always be determined as long as $Z_B = Z_C = Z_{POI}$.

The working component 906 and the motive component 904 can independently roll (X-axis), pitch (Y-axis) and yaw (Z-axis) relative to each other. For example, rolling and pitching will alter the elevation of points B, C, and POI relative to each other because the motive component 904 and the working component 906 will not be coplanar. The above-mentioned equations will not be able to solve the distances AB and AC. Also, the angle Φ has changed to Φ' , which is no longer a right angle. In such an instance, the height h will not change, however, and the distances between points can still be calculated using $AB = \sqrt{(X_B - X_A)^2 + (Y_B - Y_A)^2 + (Z_B - Z_A)^2}$ or $AC = \sqrt{(X_C - X_A)^2 + (Y_C - Y_A)^2 + (Z_C - Z_A)^2}$. The various angles can then be calculated using the law of cosines:

$$\frac{\cos \Phi' = h^2 - AC^2 - d^2}{-2(AC * d)}$$

Knowing the lengths of at least two sides and a known angle Φ allows calculation of the other side and angles. This will allow the guidance computer 910 to calculate the distance between the antennas 952, 954, 956 no matter what the three-dimensional orientation of the working component is with respect to the motive component. The roll, pitch, or yaw difference between the motive component 904 and the working component 906 can be determined by including IMUs 919, 921 and measuring the differences recorded by those IMUs. The IMU measurements will provide additional values for unknown distances necessary to solve the relative position of the working component 906 in relation to the motive component 904.

FIG. 25 is a block diagram showing the relationship between the various sensors and the GNSS guidance system. The working component 906 contains its own CPU 913, which collects data from both the working component sensors 918 and the hitch feedback sensor 920. These elements are separate to allow the working component 906 to move itself relative to the vehicle 4 by maneuvering the mechanical hitch 914, which will realign the working component being towed by the vehicle, as explained in further detail in U.S. Pat. No. 7,292,186, which is incorporated herein by reference. The guidance computer 910, on the other hand, is directly connected to the GNSS receiver 8, the vehicle sensors 912, and the wheel compaction sensor 916. A controller area network (CAN) cable 932 connects the working computer 913 with the guidance computer 910 located in the vehicle 904. Alternatively, the two computers may communicate over a local wireless network. The wireless network may be located somewhere on the vehicle 907 or may be located elsewhere in the vicinity. Such a network typically requires a wireless router and a wireless communication device connected to each computer.

Communication between the two computers 910, 913 compares data received from the various sensors and the GNSS guidance system and results in problem solving for future pre-planned paths. Problem solving can either be done in real-time, as mentioned above, or used in generating future, pre-planned paths off-site. This may be performed by uploading gathered data onto an external PC 934 or using the guidance computer 910 directly to calculate a new path. Field data that has been gathered by the various sensors can include, without limitation: the slope of the field at various point locations; the speed at which the vehicle previously navigated the field; and the GNSS positional data recorded as the vehicle traversed the field, including locations where

the working component **906** and/or the motive component **904** were no longer in line with the previous pre-planned path. The user may interpret the data and create a new pre-planned program based on it, or an optional computer program can take the data and generate a pre-planned path based on programmed configurations for dealing with different field conditions.

It should be noted that the components of the system **902** can be combined in various ways and will function in a similar manner. For example, a commonly used component is a combination receiver and antenna unit, sometimes referred to as a "smart antenna." Other components may also optionally be combined, such as the various base station components. A common example of such a combination antenna is the A-220 "Smart Antenna" manufactured by Hemisphere GPS LLC of Calgary, Canada, which are typically combined with Hemisphere GPS receivers and other components, as described in U.S. Patent Application Ser. No. 61/377,355, which is assigned to a common assignee herewith and incorporated herein by reference.

As mentioned above, a motorized hitch **914** connects the working component **6** to the motive component **904**. This motorized hitch contains a feedback sensor **920** which communicates with the working component computer **913**, which in turn communicates with the guidance computer **910**. This allows commands to be sent to the motorized hitch **914** from the guidance computer **910** regarding positioning of the working component **906**, and feedback data to then be reported to the guidance computer **910** for recording and additional guidance commands. Stresses on the hitch **914** from holding the working component **906** along a slope and relative position to the motive component **904** are among the variables reported to the guidance computer **910** by the hitch feedback sensor **920**.

FIG. **25** is a block diagram dividing the separate subsystems of the system **902**. FIG. **25** shows the flow of information from the sensors and GNSS positional system to a finished form of field output data **928** as it is gathered by the various sensors located on the motive component **904** and the working component **906** and communicated between the working component computer **913** and the guidance computer **910**. The various sensors including the vehicle sensors **912**, wheel compaction psi sensor **916**, working component sensors **918**, and hitch feedback sensor **920** feed into the guidance computer **910**. Additionally, the antennas **952**, **954**, **956** receive satellite positional information and transmit that information to the GNSS receiver **908** which is directly connected to the guidance computer **910**. The guidance computer **910** is connected to the GUI **911** which both displays information to the user and allows for user input via an interface device, such as a touchscreen display or other interface device. Finally, the field output data **928** is created by combining the entirety of the recorded data and relating it to the layout of the field or piece of land that has been worked. This will allow for a more efficient and accurate pre-planned path the next time a vehicle **904** is to work the field in question by combining the data and configuring an automatic steering program focused on guiding the vehicle while addressing the landscape concerns. Knowing where field irregularities are located is the easiest way to ensure the vehicle **907** correctly navigates these irregularities.

The guidance computer **910** can interface with an external computer (e.g., PC) **934** which can receive recorded field data, edit that data, and turn that data into a pre-planned guidance path. Input field data **930** is data includes pre-planned path and controller data. This data is installed in the guidance computer **910** and actively and automatically

guides and controls the vehicle through interaction with the steering controller **917**. The steering controller **917** will take guidance commands, steering commands, and other commands to control various vehicle functions and will physically perform those functions. Thus a preplanned path based on earlier field data will know to slow down when the motive component is approaching a particularly sharp curve or may instruct the motorized hitch **914** to adjust the position of the working component **906** prior to entering a sloped area.

Output to an external computer **934**, such as a personal computer (PC), can be performed in a number of ways. Field data output **928** can be delivered over a direct connection established between the onboard computer **910** and the external computer **934**, or field data output **928** can be offloaded onto a portable storage device and then connected to the external computer **934**. Similarly, input data **930** can be generated by an external (e.g., offsite) computer **934** and stored onto a portable storage device, and later uploaded to the CPU **910**. Such input data **930** may include a pre-planned driving path for an initial field test, or an updated planned path based on previous data collection.

The vehicle sensor suite **912** can also include a camera **939**, or other suitable optical device. For example, U.S. Patent Publication No. 2009/0204281, which is assigned to a common assignee herewith, shows a video input system for autosteering control of an agricultural vehicle and other machines. U.S. patent application Ser. No. 12/504,779, which is also assigned to a common assignee herewith, shows an optical tracking vehicle control system and method. Both of these applications are incorporated herein by reference. The camera **939** can be directed at the projected guide path of the tractor **904**, towards crop rows on either side, along vehicle tracks or towards any area of interest relative to the tractor **904** or the implement **906**. Optical input from the camera **939** can be used by the guidance computer **910** for guiding the vehicle **907** using video input. Alternatively, the camera **939** can be used for recording, observing and archiving the path of the vehicle **907** for purposes of record-keeping or future guidance. For example, in a "match tracks" mode, it may be desirable for the vehicle **907** to accurately retrace previous guide paths, which may be optically observable. Still further, such optical data can be useful for observing the crop plants (typically in rows) whereby the operator and/or the guidance computer **910** can avoid driving over crops and can monitor and record their growth. Still further, the camera **939** can be user-controlled and adjustable for visually observing the vehicle **907** guide path or the crops close-up, for example, on the GUI **911** in the cab.

FIG. **26** shows a plan view of a field with a border **940** and a vehicle **907** traversing a pre-planned path **942**. The field contains several irregularities, including a severely sloped section **944**, a section of soft earth **946** where water or soil type will cause the vehicle and working component wheels to slightly sink into the ground, an uneven area **948** which may be rocky or otherwise uneven. The various sensors attached to the motive component and working component will record data as the vehicle **907** traverses the areas of irregularity. For instance; as the motive component approaches the uneven area **948**, the wheel sensors **916** may detect compression psi differences if the ground contains rocks. Likewise, there may be a sensor attached to the shock absorbers of the motive component to determine the stress levels on said shocks when traversing such an uneven area. The GNSS guidance system **902** will detect whether the vehicle **904** or working component **906** rolls or pitches to a

side, or if the heading is altered due to a bump. If the effects of the uneven ground result in the vehicle 907 being deflected off course, the guidance CPU can record this information and instruct the vehicle to slow down in that location at a future date.

The measurement of the varying distance of the three GNSS antennas 952, 954, 956 from one another, a plurality of satellites, and the base station 922 along with heading, attitude, motive component speed, motive component gearing, power, fuel consumption, working component load, stress loads, and other factors which may affect vehicle progression through a field will result in providing knowledge to an extreme detail of the field or piece of land being driven. Once all measurements are taken, the end-user will be able to zoom in on any particular spot in a field map and view near topographic details of any location. Knowing where rocks, slopes, and obstacles are and controlling the vehicle according to this knowledge will result in greater efficiency, less wear on the vehicle and working component, and lower costs on vehicle fuel as well as seeds, chemicals, and other products being distributed.

Recording field conditions in a variety of weather types and a variety of soil types can also increase efficiency and safety. For instance; if the field needs to be worked while it is raining, preplanned path data can be fed to the guidance computer 910 from a previous field pass from when it was raining. This will present a completely customizable method of vehicle guidance and control which can be optimized depending on weather type, vehicle type, soil condition, and other factors.

A preferred embodiment of the present invention will result in better positioning of the motive component 904 for improved working component 906 position, attitude, and track. The hitch feedback sensor 920 will provide feedback regarding working component attitude and will aid in adjusting a skewed heading. Real time and post analysis of motive component and working component stress areas in the field will result in resolving those areas with additional field preparation or alternations to the motive component's tires, speed, or power. Generation of data based off of stressed field conditions will allow future passes to supply guidance changes to preempt working component track distortions in difficult field conditions. Applying the preferred embodiment to a system using adjustable variable rate controllers for applying chemical, seed, or other material to a field will result in a guidance system with unparalleled accuracy.

FIG. 27 is a flowchart demonstrating an embodiment of a method of practicing the present invention. This embodiment does not contain all possible sensor data, but instead represents an example of an embodiment of the present invention. The method of practicing the memory system 902 starts at 960. The guidance computer 910 is loaded with a pre-planned vehicle guidance path at 962. All vehicle sensors are initiated and record mode begins at 964. From there, the vehicle is automatically guided around the selected field at 966. This is either done using automatic steering or by directing a driver using a light track bar or other typical guidance method. Alternatively, the vehicle 907 may be driven manually around the field with the sensors recording data, without the need of a preplanned path or vehicle guidance process.

The various vehicle and working component sensors are constantly checking the various systems of the vehicle 907. Simultaneously, the GNSS guidance subsystem is recording position and orientation data of the vehicle as it travels upon the pre-planned path. When one of the various sensors detects a change in the field at 968, the system 902 stores

data to a storage device such as a hard drive connected to the guidance CPU 910 in the form of a reference point at 970. This reference point data includes vehicle speed, vehicle position, vehicle orientation, power output, and any other base system sensor desired to be recorded by the end user. This reference data is important for calculating what has occurred at the particular point in the field where a sensor has picked up a change in the field layout according to the pre-planned path.

At 972 is a check to determine whether the sensors have determined if the working component has moved off of the guide-line 942 by an amount pre-set by the user. If the response to this check is "yes," the system 902 records the distance the working component has moved off of the guideline at 974. From there, the system 902 can optionally re-align the working component via the connected motorized hitch 914 at 976. The user may wish not to re-align and determine the full effect of the field irregularity on the pre-planned guidance path, in which case the optional step at 976 can be ignored. The method will then loop back to the guidance step at 966, where sensor checks will continue.

If, at 972, the sensors do not determine the working component has drifted off of the guide-line 942, then the method proceeds to the next check-step. This step involves the wheel PSI compaction sensor at 978, wherein the wheel compaction sensors of the working component, the motive component, or both determine that the soil beneath the tire has changed in some fundamental way. If the answer to this check is a "yes," then the compaction data is recorded at 980 in reference to positional data and orientation data. From there, the system 902 can optionally slow the vehicle at 982 in order to compensate for the irregular soil type and ensure a smoother and more accurate ride by the vehicle 907. From here, the method loops back to the guidance step at 966, where sensor checks will continue.

A constant "vehicle shutoff" check is present in the loop at 984. If the vehicle or system is ever shutoff, it will result in the system ending at 986.

The guidance computer 910 of the present invention can use guidance algorithms in common with U.S. Patent Publication No. 2009/0164067 (incorporated herein by reference) for position determination in a multiple antenna moving-baseline guidance system. Position and guidance algorithms used by the processors of the present invention are well known and documented in the prior art.

VIII. Alternative Embodiment Multi-Antenna System 1002

A guidance system 1002 comprising an alternative embodiment of the present invention is shown in FIG. 28 and includes a motive component (e.g., tractor) 904 substantially as described above. Without limitation on the generality of articulated working components adapted for use with the present invention, an articulated implement 1006 is shown with first and second implement sections 1006A and 1006B, which are adapted for rotating relative to each other around a hinge line 1009 extending generally parallel to the X axis. Alternatively, various other working components, such as machines, earthworking equipment, articulated excavator booms, motor graders and agricultural implements can be utilized with the system 1002. For example, a wide range of tillage, cultivating, harvesting, seeding, and spraying implements can be controlled with the system 1002. Such implements include side-by-side and front-and-back components, which can be pivotably connected by hinges and other articulated connections, such as hitches.

As shown in FIG. 28, the implement sections 1006A, 1006B can accommodate field conditions requiring indepen-

dent rotation. For example, implement section **1006A** can be positioned on a sloping ground surface, such as a terrace, while the other implement section **1006B** can be relatively flat.

The guidance system **1002** can utilize multiple antennas for independently monitoring positional and attitude (orientation) data from the tractor **904** and the implement sections **1006A**, **1006B**. For example: the tractor **904** can be provided with an antenna **952**; the first implement section **1006A** can be provided with antennas **1054**, **1055**; and the second implement section **1006B** can be provided with antennas **1056**, **1057**. Respective XYZ GNSS-based coordinates can be read from each antenna for computing their respective positions, either on an absolute or relative basis. The first implement section antennas **1054**, **1055** define a first fixed baseline **1096A** and the second implement section antennas **1056**, **1057** define a second fixed baseline **1096B**. Variable baselines **1098** are defined between the tractor antenna **952** and the implement antennas **1054-57**. Although specific antenna configurations are shown, they are not limiting and the present invention generally contemplates the use of multiple antennas in various suitable multiples, distributions and configurations.

The position/attitude determining algorithms used by the guidance computer **910** can be programmed for the baseline constant and variable factors for use in computing position/attitude solutions. For example, the guidance computer **910** can include a switching function for switching among the antennas in order to optimize the available GNSS ranging information. See, U.S. Patent Publication No. 2004/0212533, which is incorporated herein by reference. The availability of ranging information from multiple antennas can be important in compensating for GNSS signal blockage, for example, when equipment or environment obstructions prevent individual antennas from “seeing” enough satellites. Interference, multipath and other error sources can lead to position dilution of precision (“PDOP”). These conditions can be compensated for by the multi-antenna configuration shown in FIG. **28**. Of course, the tractor **904** can also be equipped with multiple antennas defining additional fixed and variable baselines. Multiple antennas are also useful for computing “Vector” guidance solutions comprising object attitude or orientation, direction of travel (heading) and velocity.

XI. Spray Control System and Method **1120**

Referring to FIGS. **29**, **29a-c** and **30**, elements of a spray control system and method **1120** are illustrated. The spray control system **1120** can be applied to any kind of field spraying but has particularly advantageous application in environmental spraying in which the application of certain materials **1122** is restricted by regulation to designated areas to avoid or minimize adverse environmental impacts. Areas in which spray of the material **1122** is allowed are referred to as spray zones **1124**, while areas in which spray of the material **1122** is required to be avoided are referred to as exclusion zones **1126**. The spray zones **1124** are separated from the exclusion zones **1126** by field boundaries **1128**.

The method **1120** makes use of a spray vehicle **1130** which is illustrated in FIGS. **29** and **29a-c** as a tanker truck **1132** having a tank **1134** holding a quantity of the material **1122** to be sprayed, such as waste water from oil, gas, and water drilling operations, or the like. A pump **1136** is connected to the tank **1134** and is activated to pump the material **1122** to nozzles **1138** transversely spaced along a transverse spray boom **1140**. A GNSS antenna **1142** is mounted on the spray vehicle **1130** and is interfaced to a GNSS receiver (e.g., **24a**, **524**, **604**, **734** or **908** described

above, or any other suitable GNSS receiver), which is connected to a controller or control computer **1144**, which is interfaced by way of suitable drivers and/or relays with the pump **1136** to thereby activate and deactivate the pump **1136**. The controller **1144** may be a conventional type of computer, including one or more central processing units (CPUs), memory, mass storage, user interface devices, and input/output (I/O) ports (not detailed) which are widely available.

The field boundaries **1128** are surveyed and position coordinates of the boundaries, as determined by a GNSS receiver or other position coordinate detecting system, are recorded for entry into the controller **1144**. The boundaries may be surveyed by a vehicle (not shown) other than the spray vehicle **1130**. A longitudinal nozzle/receiver or boom offset distance “*b*” is measured between the GNSS receiver **1142** and the spray boom **1140**. The illustrated nozzles **1138** are aimed to spray the material **1122** generally to the rear of the spray vehicle **1130**. When the pump **1136** is at a steady state of operation spraying the material **1122** from the nozzles **1138**, the material **1122** is sprayed to an average spray range “*r*” behind the spray boom **1140**. However, there is a spray turn-on lag time “*t_{on}*” between activation of the pump **1136** and the material **1122** reaching the average spray range. Similarly, there is a spray turn-off lag time “*t_{off}*” that occurs between deactivation of the pump **1136** and the cessation of emission of the material **1122** from the nozzles **1138**.

In the spray control method **1120**, the GNSS receiver generally tracks the position of antenna **1142**, which it communicates to the controller **1144**. When the spray vehicle **1130** is traveling in an exclusion zone **1126** toward a spray zone **1124**, the **1120** turns on when the antenna **1142** reaches a spray turn-on boundary **1146**. Conversely, when the spray vehicle **1130** is traveling in a spray zone **1124** toward an exclusion zone **1126**, the system **1120** turns off when the antenna **1142** reaches a spray turn-off boundary **1148**. In general, when the spray vehicle **1130** is approaching a turn-on boundary **1146**, the controller **1144** causes the pump **1136** to activate when the GNSS receiver detects that it is at a distance beyond the turn-on boundary **1146** at which spray of the material **1122** will be retained within the spray zone **1124**. Similarly, when the spray vehicle **1130** is approaching a turn-off boundary **1146**, the controller **1144** causes the pump **1136** to deactivate when the vehicle **1130** is at a distance beyond the turn-off boundary **1146** at which deposition of the material **1122** within the exclusion zone **1126** is prevented.

In an embodiment of the method **1120**, when the spray vehicle **1130** is traveling within an exclusion zone **1126** approaching a spray zone **1124**, the controller **1144** causes the pump **1136** to activate within the spray zone **1124** when the system **1120** detects that the vehicle **1130** is at a turn-on distance “*D_{on}*” beyond the turn-on boundary **1146** equal to the sum of the nozzle/receiver offset distance plus the spray range minus the product of the spray vehicle velocity “*v*” times spray turn-on lag time. The turn-on distance can be expressed as:

$$D_{on}=(b+r)-v*t_{on}$$

When the spray vehicle **1130** is traveling in a spray zone **1124** approaching an exclusion zone **1126**, the controller **1144** causes the pump **1136** to deactivate when the system **1120** detects that it is at a turn-off distance “*D_{off}*” beyond the turn-off boundary **1148** equal to the nozzle/receiver offset

minus the product of the spray vehicle velocity times the spray turn-off lag time. The turn-off distance can be expressed as:

$$D_{off} = b - v * t_{off}$$

It is foreseen that the turn-on and turn-off distances may need to be adjusted to insure that the material 1122 is not sprayed onto the exclusion zone 1126. Thus, the turn-on distance may need to be increased somewhat and the turn-off distance decreased somewhat to avoid any spraying onto the exclusion zone 1126.

Referring to FIG. 30, the illustrated spray vehicle 1150 includes a towing vehicle or tractor 1150 and a towed vehicle or spray implement 1152, which is hitched to the tractor 1150, which are similar in many respects to the tractor 10 and the spray implement 506 shown in FIG. 9. The illustrated spray implement 1152 includes a tank 1154 holding the material 1122 to be sprayed and a transverse spray boom 156 having a plurality of nozzles 158 transversely spaced therealong and communicating with a pump 1136, which is controlled by a controller 1144. While the illustrated spray patterns of the nozzles 1158 suggest a downward spray direction, it is foreseen that the nozzles 1158 could also be arranged to have spray patterns similar to the spray nozzles 1138. The illustrated tractor 1150 has the GNSS antenna 1142 mounted thereon at a nozzle/receiver offset distance from the spray boom 1156 and interfaced to the controller. The spray control method 1120 may be operated with the spray vehicle 1130 formed by the tractor 1150 and spray implement 1152 in a manner substantially similar to operation using the spray truck 1132.

In surveying the field boundaries 1128 for environmental spraying, it is desirable to simplify the shape of the boundaries, with any error adjustment being in the direction of avoiding applying the material 1122 to an exclusion zone. It is foreseen that the field boundaries 1128 may not always be straight-lined and that the direction of travel of the spray vehicle 1130 may not always be perpendicular to a field boundary 1128.

X. Optional Spray Control Valves

The spray control system and method 1120 can optionally include valves 1162₁ through 1162_n by the controller 1144 to control the flow of material 1122 from the pump 1136 to the nozzles 1158₁ through 1158_n. There is a valve open lag time which occurs between opening of a valve 1162 and the material 1122 reaching the average spray range behind the spray boom 1156. The valve open lag time may be different from the spray lag time described above and it may vary depending on whether or not the pump 1136 is being activated simultaneously. Additionally, the valve open lag time may vary depending on the number of valves 1162 which are currently open or are being opened. Similarly, there is a valve close lag time between closure of a valve 1162 and the cessation of material 1122 being emitted from the associated nozzle 1156, which may vary for reasons similar to variation in the valve open lag time. The variations in the valve open and close lag times can be measured and entered into the controller 1144 along with the sets of conditions which are to be processed in selecting a given valve lag time. In the alternative spray control embodiment 1160, each valve 1162 is controlled in relation to a segment or portion of the upcoming field boundary 1128 that is aligned with the valve 1160 and its spray pattern. Although the spray truck 1132 is not illustrated with individual valves for its nozzles 1138, it is foreseen that the spray truck 1132 could also be provided with individual spray valves for use in the alternative spray control method 1160.

When the spray vehicle 1130 is traveling within an exclusion zone 1126 and approaching a spray zone 1124, the controller 1144 causes each valve 1162 to be opened within the spray zone 1124 when the system 1120 detects that it is at a valve open distance beyond the portion of the turn-on boundary 1146 aligned with that particular valve 1162 which is equal to the sum of the nozzle/receiver offset distance plus the spray range minus the product of the spray vehicle velocity times the valve turn-on lag time, which is selected according to the conditions described above. When the spray vehicle 1130 is traveling in a spray zone 1124 approaching an exclusion zone 1126, the controller 1144 causes each valve 1162 to be closed when the GNSS receiver 1142 detects that it is at a valve close distance beyond a portion of the turn-off boundary 1148 aligned with that particular valve 1162 which is equal to the nozzle/receiver offset minus the product of the spray vehicle velocity times the valve close lag time, also selected according to the conditions described above. It is foreseen that, in an environmental spraying operation, the valve open and close distances may need to be adjusted somewhat to insure that the material 1122 is not applied to any exclusion zones 1126.

The spray control method 1120, as described, generally assumes that the truck 1132 and the tractor 1150 are driven by human operators, with the method 1120 causing automatic turn-on and turn-off of spraying equipments based on the locations detected by the GNSS receivers. It is also foreseen that the truck 1132 or tractor 1150 could be operated in fully automatic navigation modes using apparatus and techniques described above, in cooperation with the spray control method 1120.

While the description has been made with reference to exemplary embodiments, it will be understood by those of ordinary skill in the pertinent art that various changes may be made and equivalents may be substituted for the elements thereof without departing from the scope of the disclosure. In addition, numerous modifications may be made to adapt the teachings of the disclosure to a particular object or situation without departing from the essential scope thereof. Therefore, it is intended that the claims not be limited to the particular embodiments disclosed as the currently preferred best modes contemplated for carrying out the teachings herein, but that the claims shall cover all embodiments falling within the true scope and spirit of the disclosure.

Having thus described the invention, what is claimed as new and desired to be secured by Letters Patent is:

1. A spray control method for controlling spraying of a material on a spray zone of a field also including an exclusion zone which is not to receive said material[, said spray zone being separated from said exclusion zone by a spray turn-on boundary when passing from said exclusion zone to said spray zone and by a spray turn-off boundary when passing from said spray zone to said exclusion zone]; said method employing a spray vehicle including a material tank *storing said material*, a pump communicating with said tank, [and a nozzle communicating with said pump, said pump when activated causing said material to be sprayed from said nozzle to a spray range from said nozzle and requiring a turn-on lag time between activation of said pump and said material reaching said spray range and a turn-off lag time between deactivation of said pump and cessation of said material being sprayed from said nozzle, said vehicle having] a [GNSS] receiver [mounted thereon at an offset distance from said nozzle and] outputting position data representing a position [of said] *thereof*, and a controller interfaced between said receiver[, said receiver being interfaced to a controller which is interfaced to] and said pump,

said controller selectively activating said pump and having data representing [said turn-on boundary and said turn-off boundary] *boundaries of said spray zone relative to said exclusion zone* stored therein, [and] said method comprising [the steps of]:

[(a)] moving said vehicle in said exclusion zone toward said [turn-on boundary at a vehicle velocity] *spray zone*;

[(b)] communicating position data from said receiver to said controller to track the position of said receiver;

[(c)] activating said pump by said controller when said receiver detects a position of said receiver within said spray zone at [a spray turn-on distance beyond said turn-on boundary equal to the sum of said offset distance plus said spray range minus the product of said vehicle velocity times said turn-on lag time] *which spray of said material is retained within said spray zone*;

[(d)] moving said vehicle in said spray zone toward said [turn-off boundary] *exclusion zone*; [and]

[(e)] deactivating said pump by said controller when said receiver detects a position of said receiver within said exclusion zone at [a] *which spray [turn-off distance beyond said turn-off boundary equal to said offset distance minus the product of said vehicle velocity times said turn-off lag time] of said material within said exclusion zone is prevented*;

storing data identifying a spray turn-on boundary separating said spray zone from said exclusion zone when passing from said exclusion zone to said spray zone;

activating said pump causing said material to spray to a spray range from a nozzle communicating with said pump and positioned at an offset distance from said receiver, a turn-on lag time being required between activation of said pump and said material reaching said spray range;

moving said vehicle toward said spray turn-on boundary at a vehicle velocity; and

activating said pump by said controller when said receiver detects a position of said receiver within said spray zone beyond said spray turn-on boundary equal to the sum of said offset distance plus said spray range minus the product of said vehicle velocity times said turn-on lag time.

2. A spray control method as set forth in claim 1 [and including the steps of] *further comprising*:

[(a)] mounting a transverse spray boom on said vehicle; and

[(b)] supporting said nozzle and a plurality thereof on said spray boom in transversely spaced relation and in communication with said pump.

3. A method as set forth in claim 1 and including a plurality of spray valves connecting [said] *a plurality of nozzles* respectively with said pump, said valves being interfaced to said controller, and *said* valves having a valve turn-on lag time and a valve turn-off lag time which are stored in said controller, and including [the steps of]:

[(a)] individually controlling each valve to open when said receiver detects a position of said receiver within said spray zone at a valve open distance beyond a portion of said turn-on boundary aligned with the nozzle associated with said valve equal to the sum of said offset distance plus said spray range minus the product of said vehicle velocity times said [valve open] *turn-on lag time*; and

[(b)] individually controlling each valve to close when said receiver detects a position of said receiver within

said exclusion zone at a valve close distance beyond a portion of said turn-off boundary aligned with the nozzle associated with said valve equal to said offset distance minus the product of said vehicle velocity times said [valve close] *turn-off lag time*.

4. A spray control method as set forth in claim 1 wherein: [(a)] said vehicle includes a tractor and a towed spray implement hitched to said tractor and having said tank and said pump mounted thereon.

5. A [spray control] method for controlling spraying of a material *from a vehicle* on a spray zone of a field *that also [including] has* an exclusion zone which is not to receive said material; [said method employing a spray vehicle including a material tank storing said material, a pump communicating with said tank, a GNSS receiver outputting position data representing a position thereof, and a controller interfaced between said receiver and said pump, said controller selectively activating said pump and having data representing boundaries of said spray zone relative to said exclusion zone stored therein, and], said method comprising [the steps of]:

[(a)] moving said vehicle in said exclusion zone toward said spray zone;

[(b)] communicating [receiving, by a controller, position data from [said] a receiver to [said controller to] track [the] a position of said receiver;

[(c)] activating [said] a pump by said controller when [said receiver detects a] *the position of said receiver is detected* within said spray zone at which spray of said material is retained within said spray zone;

[(d)] moving [detecting movement of] said vehicle in said spray zone toward said exclusion zone;

[(e)] deactivating said pump by said controller when [said receiver detects a] *the position of said receiver is* within said exclusion zone at which spray of said material within said exclusion zone is prevented;

[(f)] *storing data identifying a turn-on boundary* separating said spray zone from said exclusion zone [by a spray turn-on boundary] when passing from said exclusion zone to said spray zone;

[(g)] activating said pump causing said material to spray to a spray range from a nozzle communicating with said pump and positioned at an offset distance from said receiver[.];

storing a measurement of a turn-on lag time [being required] between activation of said pump and said material reaching said spray range;

[(h)] moving [detecting movement of] said vehicle toward said turn-on boundary at a vehicle velocity; and

[(i)] activating said pump [by] *when* said controller [when said receiver] detects a position of said receiver within said spray zone beyond said turn-on boundary *substantially* equal to [the] a sum of said offset distance plus said spray range minus [the] a product of said vehicle velocity times said turn-on lag time.

6. A method as set forth in claim 5 [wherein said spray zone is separated from said exclusion zone by a spray turn-on boundary when passing from said exclusion zone to said spray zone, and said method includes the steps of] *further comprising*:

[(a)] moving [detecting] said vehicle *moving* toward said turn-on boundary; and

[(b)] activating said pump [by] *when* said controller [when said receiver] detects a position of said receiver within said spray zone *is* beyond said turn-on boundary at which spray of said material is retained within said spray zone.

7. A method as set forth in claim 5 [wherein said spray zone is separated from said exclusion zone by a spray turn-off boundary when passing from said spray zone to said exclusion zone, and said method includes the steps of] *further comprising:*

[(a) moving] *detecting movement of* said vehicle in said spray zone toward [said] a turn-off boundary *separating the spray zone from the exclusion zone;* and

[(b)] deactivating said pump [by] *when* said controller [when said receiver] detects a position of said receiver within said exclusion zone beyond said turn-off boundary at which spray of said material within said exclusion zone is prevented.

8. A method as set forth in claim 5 [and including the steps of] *further comprising:*

[(a)] mounting a transverse spray boom on said vehicle; and

[(b)] supporting said nozzle and a plurality thereof on said spray boom in transversely spaced relation and in communication with said pump.

9. A method as set forth in claim 8 and including a plurality of spray valves connecting said nozzles respectively with said pump, said valves being interfaced to said controller, [and including the steps of] *further comprising:*

[(a)] individually controlling each valve to open when said [receiver] *controller* detects a position of said vehicle within said spray zone at which spray of said material from the nozzle associated with said valve is retained within said spray zone; and

[(b)] individually controlling each valve to close when said [receiver] *controller* detects a position of said receiver within said exclusion zone at which spray of said material within said exclusion zone from the nozzle associated with said valve is prevented.

10. A method as set forth in claim 5 wherein:

[(a)] said vehicle includes a tractor and a towed spray implement hitched to said tractor and having [said] a tank and said pump mounted thereon.

11. A spray control method for controlling spraying of a material on a spray zone of a field also including an exclusion zone which is not to receive said material; said method employing a spray vehicle including a material tank storing said material, a pump communicating with said tank, a GNSS receiver outputting position data representing a position thereof, and a controller interfaced between said receiver and said pump, said controller selectively activating said pump and having data representing boundaries of said spray zone relative to said exclusion zone stored therein, and said method comprising [the steps of]:

[(a)] moving said vehicle in said exclusion zone toward said spray zone;

[(b)] communicating position data from said receiver to said controller to track the position of said receiver;

[(c)] activating said pump by said controller when said receiver detects a position of said receiver within said spray zone at which spray of said material is retained within said spray zone;

[(d)] moving said vehicle in said spray zone toward said exclusion zone;

[(e)] deactivating said pump by said controller when said receiver detects a position of said receiver within said exclusion zone at which spray of said material within said exclusion zone is prevented;

[(f)] *storing data in the controller identifying a turn-off boundary* separating said spray zone from said exclusion zone [by a spray turn-off boundary] when passing from said spray zone to said exclusion zone;

[(g) said vehicle including a nozzle positioned] *storing data in the controller identifying* an offset distance of a nozzle from said receiver [and communicating with said pump];

[(h)] *storing data in the controller identifying* a turn-off lag time elapsing between deactivation of said pump and cessation of material being sprayed from said nozzle;

[(i) moving] *monitoring* said vehicle *moving* in said spray zone toward said turn-off boundary at a vehicle velocity; and

[(j)] deactivating said pump by said controller when [said receiver detects] a position of said receiver within said exclusion zone beyond said turn-off boundary *is at least equal to* said offset distance minus the product of said vehicle velocity times said turn-off lag time.

12. A spray control method for controlling spraying of a material on a spray zone of a field also including an exclusion zone which is not to receive said material, said spray zone being separated from said exclusion zone by a [spray] turn-on boundary when passing from said exclusion zone to said spray zone and by a [spray] turn-off boundary when passing from said spray zone to said exclusion zone; said method employing a spray vehicle including a material tank storing said material, a pump communicating with said tank, a GNSS receiver outputting position data representing a position thereof, and a controller interfaced between said receiver and said pump, said controller selectively activating said pump and having data representing said turn-on boundary and said turn-off boundary stored therein, and said method comprising [the steps of]:

[(a)] moving said vehicle in said exclusion zone toward said turn-on boundary;

[(b)] communicating position data from said receiver to said controller to track the position of said receiver;

[(c)] activating said pump by said controller when said receiver detects a position of said receiver within said spray zone beyond said turn-on boundary at which spray of said material is retained within said spray zone;

[(d)] moving said vehicle in said spray zone toward said turn-off boundary;

[(e)] deactivating said pump by said controller when said receiver detects a position of said receiver within said exclusion zone beyond said turn-off boundary at which spray of said material within said exclusion zone is prevented;

[(f) said vehicle including a nozzle positioned] *storing data identifying* an offset distance of a nozzle from said receiver [and], *said nozzle* communicating with said pump;

[(g)] *storing data identifying* a turn-off lag time elapsing between deactivation of said pump and cessation of material being sprayed from said nozzle;

[(h)] moving said vehicle in said spray zone toward said turn-off boundary at a vehicle velocity; and

[(i)] deactivating said pump by said controller when said receiver detects a position of said receiver within said exclusion zone beyond said turn-off boundary equal to said offset distance minus the product of said vehicle velocity times said turn-off lag time.

13. A method as set forth in claim 12 wherein [said spray zone is separated from said exclusion zone by a spray turn-on boundary when passing from said exclusion zone to said spray zone,] activation of said pump causes said material to spray to a spray range from [a] said nozzle [communicating with said pump and positioned at an offset distance

from said receiver,] and a turn-on lag time is required between activation of said pump and said material reaching said spray range, and said method [includes the steps of] *further comprising*:

5 [(a)] moving said vehicle toward said turn-on boundary at a vehicle velocity; and

[(b)] activating said pump by said controller when said [receiver] *controller* detects a position of said receiver within said spray zone beyond said turn-on boundary equal to the sum of said offset distance plus said spray range minus the product of said vehicle velocity times said turn-on lag time.

14. A method as set forth in claim 12 [and including the steps of] *further comprising*:

15 [(a)] mounting a transverse spray boom on said vehicle; and

[(b)] supporting a plurality of said nozzles on said spray boom in transversely spaced relation and in communication with said pump.

15. A method as set forth in claim 14 and including a plurality of spray valves connecting said nozzles respectively with said pump, said valves being interfaced to said controller, [and including the steps of] *further comprising*:

25 [(a)] individually controlling each valve to open when said receiver detects a position of said receiver within said spray zone beyond said turn-on boundary at which spray of said material from the nozzle associated with said valve is retained within said spray zone; and

[(b)] individually controlling each valve to close when said receiver detects a position of said receiver within said exclusion zone beyond said turn-off boundary at which spray of said material within said exclusion zone from the nozzle associated with said valve is prevented.

16. A method as set forth in claim 12 wherein:

35 [(a)] said vehicle includes a tractor and a towed spray implement hitched to said tractor and having said tank and said pump mounted thereon.

17. *A control system for controlling spraying of a material from a vehicle or from an implement hitched to the vehicle, the vehicle or implement including a tank storing the material, a pump communicating with the tank, and a receiver outputting position data, the control system comprising*:

a processor; and memory storing one or more stored sequences of instructions which, when executed by the processor, cause the processor to:

5 *activate the pump to cause the material to spray to a spray range distance from a nozzle communicating with the pump;*

store data identifying a spray zone for a field which is to receive the material;

10 *store data identifying an exclusion zone for the field which is not to receive the material;*

store data identifying a turn-on lag time between activation of the pump and said material reaching the spray range;

15 *store data identifying a turn-off lag time between deactivation of the pump and termination of emission of the material from the nozzle;*

store data identifying an offset distance between the receiver and the nozzle;

monitor a vehicle velocity;

20 *identify the vehicle entering the spray zone; and*

activate the pump when a position of the receiver in the spray zone is at least equal to a sum of the offset distance plus the spray range distance minus a product of the vehicle velocity times the turn-on lag time.

18. *The control system according to claim 17, including: identifying the vehicle leaving the spray zone and entering the exclusion zone;*

30 *deactivating the pump when a location of the receiver in the exclusion zone is at least equal to the offset distance minus a product of the vehicle velocity times the turn-off lag time.*

19. *The control system according to claim 17, wherein the nozzle is located on a transverse spray boom.*

20. *A control system according to claim 19 wherein the spray boom is attached to the vehicle through an articulated joint.*

21. *A control system according to claim 17, wherein the receiver is at least one of an inertial navigation system, an attitude heading reference system, or a global navigation satellite system (GNSS) receiver.*

22. *A control system according to claim 17, wherein the receiver uses one or more antennas.*

* * * * *