

US008499874B2

(12) **United States Patent**  
**Dewis et al.**

(10) **Patent No.:** **US 8,499,874 B2**  
(45) **Date of Patent:** **Aug. 6, 2013**

(54) **GAS TURBINE ENERGY STORAGE AND CONVERSION SYSTEM**

(75) Inventors: **David William Dewis**, North Hampton, NH (US); **James Kesseli**, Greenland, NH (US); **Frank Wegner Donnelly**, North Vancouver (CA); **Thomas Wolf**, Winchester, MA (US); **Timothy Upton**, Exeter, NH (US); **John D. Watson**, Evergreen, CO (US)

(73) Assignee: **ICR Turbine Engine Corporation**, Vancouver, British Columbia (CA)

(\*) Notice: Subject to any disclaimer, the term of this patent is extended or adjusted under 35 U.S.C. 154(b) by 240 days.

(21) Appl. No.: **12/777,916**

(22) Filed: **May 11, 2010**

(65) **Prior Publication Data**  
US 2010/0288571 A1 Nov. 18, 2010

**Related U.S. Application Data**

(60) Provisional application No. 61/177,493, filed on May 12, 2009, provisional application No. 61/327,988, filed on Apr. 26, 2010.

(51) **Int. Cl.**  
**B60K 6/00** (2007.10)

(52) **U.S. Cl.**  
USPC ..... **180/165**; 180/65.265; 180/305

(58) **Field of Classification Search**  
USPC ..... 180/165, 65.2, 65.3, 303, 304, 305, 180/306; 60/668, 659, 645, 618  
See application file for complete search history.

(56) **References Cited**

U.S. PATENT DOCUMENTS

2,463,964 A	3/1949	Graf
2,543,677 A	2/1951	Traupel
2,696,711 A	12/1954	Marchant et al.
2,711,071 A	6/1955	Frankel
3,032,987 A	5/1962	Taylor
3,091,933 A	6/1963	Wagner et al.
3,166,902 A	1/1965	Maljanian et al.
3,209,536 A	10/1965	Howes et al.
3,518,472 A	6/1970	O'Callaghan
3,639,076 A	2/1972	Rowen
3,646,753 A	3/1972	Stearns et al.

(Continued)

FOREIGN PATENT DOCUMENTS

AT	311027	12/2005
AU	582981	4/1989

(Continued)

OTHER PUBLICATIONS

International Search Report for International (PCT) Application No. PCT/US2010/034375, mailed Jul. 15, 2010.

(Continued)

*Primary Examiner* — J. Allen Shriver, II

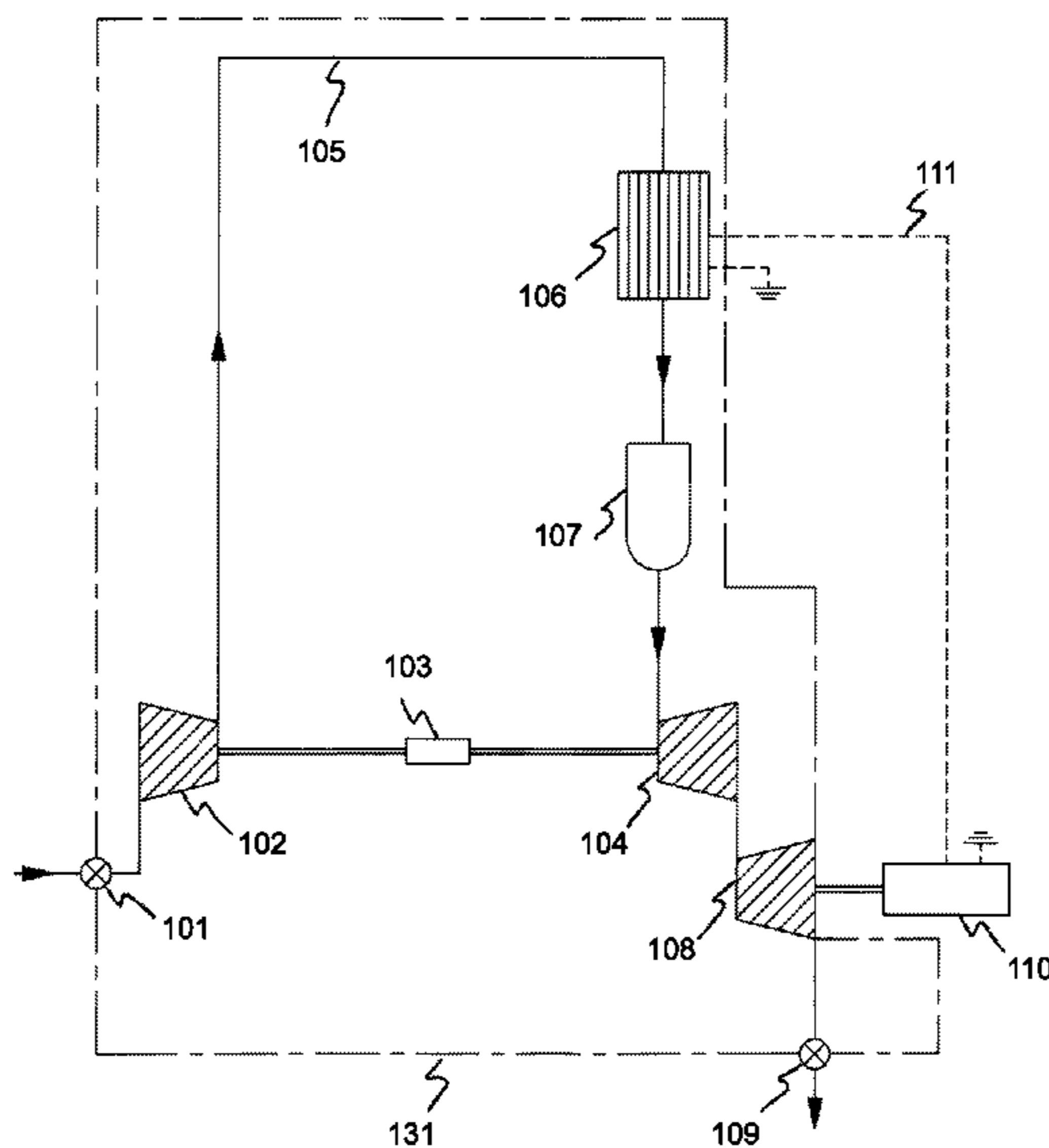
*Assistant Examiner* — James Triggs

(74) *Attorney, Agent, or Firm* — Sheridan Ross P.C.

(57) **ABSTRACT**

The present invention combines the principles of a gas turbine engine with an electric transmission system. A method and apparatus are disclosed for utilizing metallic and ceramic elements to store heat energy derived from a regenerative braking system. The subject invention uses this regenerated electrical energy to provide additional energy storage over conventional electrical storage methods suitable for a gas turbine engine. The subject invention provides engine braking for a gas turbine engine as well as reducing fuel consumption.

**15 Claims, 16 Drawing Sheets**



US 8,499,874 B2

U.S. PATENT DOCUMENTS							
3,660,977	A	5/1972	Reynolds	5,610,962	A	3/1997	Solorzano et al.
3,706,203	A	12/1972	Goldberg et al.	5,685,156	A	11/1997	Willis et al.
3,729,928	A	5/1973	Rowen	5,697,848	A	12/1997	Bosley
3,748,491	A	7/1973	Barrigher et al.	5,722,259	A	3/1998	Sorensen et al.
3,764,814	A	10/1973	Griffith	5,742,515	A	4/1998	Runkle et al.
3,766,732	A	10/1973	Woodcock	5,752,380	A	5/1998	Bosley et al.
3,817,343	A	6/1974	Albrecht	5,784,268	A	7/1998	Steffek et al.
3,848,636	A	11/1974	McCombs	5,791,868	A	8/1998	Bosley et al.
3,888,337	A	6/1975	Worthen et al.	5,819,524	A	10/1998	Bosley et al.
3,893,293	A	7/1975	Moore	5,820,074	A	10/1998	Trommer et al.
3,937,588	A	2/1976	Kisslan	5,827,040	A	10/1998	Bosley et al.
3,939,653	A	2/1976	Schirmer	5,850,732	A	12/1998	Willis et al.
3,945,199	A	3/1976	Bradley et al.	5,850,733	A	12/1998	Bosley et al.
3,953,967	A	5/1976	Smith	5,873,235	A	2/1999	Bosley et al.
3,964,253	A	6/1976	Paduch et al.	5,894,720	A	4/1999	Willis et al.
3,977,183	A	8/1976	Stearns	5,899,673	A	5/1999	Bosley et al.
3,986,364	A	10/1976	Cronin et al.	5,903,116	A	5/1999	Geis et al.
3,986,575	A *	10/1976	Eggmann ..... 180/302	5,915,841	A	6/1999	Weissert
3,999,373	A	12/1976	Bell et al.	5,918,985	A	7/1999	Bosley
3,999,375	A	12/1976	Smith et al.	5,928,301	A	7/1999	Soga et al.
4,002,058	A	1/1977	Wolfinger	5,929,538	A	7/1999	O'Sullivan et al.
4,005,946	A	2/1977	Brown et al.	5,954,174	A	9/1999	Costin
4,027,472	A	6/1977	Stearns	5,964,663	A	10/1999	Stewart et al.
4,027,473	A	6/1977	Baker	5,966,926	A	10/1999	Shekleton et al.
4,056,019	A	11/1977	Ahlen	6,002,603	A	12/1999	Carver
4,059,770	A	11/1977	Mackay	6,011,377	A	1/2000	Heglund et al.
4,082,115	A	4/1978	Gibb et al.	6,016,658	A	1/2000	Willis et al.
4,122,668	A	10/1978	Chou et al.	6,020,713	A	2/2000	Geis et al.
4,242,042	A	12/1980	Schwarz	6,023,135	A	2/2000	Gilbreth et al.
4,242,871	A	1/1981	Breton	6,031,294	A	2/2000	Geis et al.
4,248,040	A	2/1981	Kast	6,037,687	A	3/2000	Stewart et al.
4,270,357	A	6/1981	Rossi et al.	6,049,195	A	4/2000	Geis et al.
4,276,744	A	7/1981	Pisano	6,062,016	A	5/2000	Edelman
4,277,938	A	7/1981	Belke et al.	6,065,281	A	5/2000	Shekleton et al.
4,280,327	A	7/1981	Mackay	6,070,404	A	6/2000	Bosley et al.
4,282,948	A *	8/1981	Jerome ..... 180/165	6,082,112	A	7/2000	Shekleton
4,312,191	A	1/1982	Biagini	6,085,524	A	7/2000	Persson
4,336,856	A	6/1982	Gamell	6,093,975	A	7/2000	Peticolas
4,399,651	A	8/1983	Geary et al.	6,094,799	A	8/2000	Stewart et al.
4,411,595	A	10/1983	Pisano	6,107,693	A	8/2000	Mongia et al.
4,449,359	A	5/1984	Cole et al.	6,138,781	A *	10/2000	Hakala ..... 180/2.2
4,467,607	A	8/1984	Rydquist et al.	D433,997	S	11/2000	Laituri et al.
4,470,261	A	9/1984	Kronogard et al.	6,141,953	A	11/2000	Mongia et al.
4,474,007	A	10/1984	Kronogard et al.	6,155,076	A	12/2000	Cullen et al.
4,492,874	A	1/1985	Near	6,155,780	A	12/2000	Rouse
4,494,372	A	1/1985	Cronin	6,158,892	A	12/2000	Stewart et al.
4,499,756	A	2/1985	Medeiros et al.	6,169,334	B1	1/2001	Edelman
4,754,607	A	7/1988	Mackay	6,170,251	B1	1/2001	Skowronski et al.
4,783,957	A	11/1988	Harris	6,178,751	B1	1/2001	Shekleton et al.
4,815,278	A	3/1989	White	6,190,048	B1	2/2001	Weissert
4,819,436	A	4/1989	Ahner et al.	6,192,668	B1	2/2001	Mackay
4,858,428	A	8/1989	Paul	6,194,794	B1	2/2001	Lampe et al.
4,864,811	A	9/1989	Pfefferle	6,205,765	B1	3/2001	Iasillo et al.
5,010,729	A	4/1991	Adamson et al.	6,205,768	B1	3/2001	Dibble et al.
5,036,267	A	7/1991	Markunas et al.	6,213,234	B1	4/2001	Rosen et al.
5,069,032	A	12/1991	White	6,239,520	B1	5/2001	Stahl et al.
5,081,832	A	1/1992	Mowill	6,265,786	B1	7/2001	Bosley et al.
5,083,039	A	1/1992	Richardson et al.	6,274,945	B1	8/2001	Gilbreth et al.
5,090,193	A	2/1992	Schwarz et al.	6,281,596	B1	8/2001	Gilbreth et al.
5,097,658	A	3/1992	Klaass et al.	6,281,601	B1	8/2001	Edelman et al.
5,113,669	A	5/1992	Coffinberry	6,314,717	B1	11/2001	Teets et al.
5,129,222	A	7/1992	Lampe et al.	6,316,841	B1	11/2001	Weber
5,144,299	A	9/1992	Smith	6,324,828	B1	12/2001	Willis et al.
5,214,910	A	6/1993	Adair	6,324,846	B1	12/2001	Clarke
5,231,822	A	8/1993	Shekleton	6,325,142	B1	12/2001	Bosley et al.
5,253,470	A	10/1993	Newton	6,349,787	B1 *	2/2002	Dakhil ..... 180/302
5,276,353	A	1/1994	Kobayashi et al.	6,355,987	B1	3/2002	Bixel
5,301,500	A	4/1994	Hines	6,361,271	B1	3/2002	Bosley
5,333,989	A	8/1994	Missana et al.	6,381,944	B2	5/2002	Mackay
5,343,692	A	9/1994	Thomson et al.	6,405,522	B1	6/2002	Pont et al.
5,349,814	A	9/1994	Ciokajlo et al.	6,410,992	B1	6/2002	Wall et al.
5,427,455	A	6/1995	Bosley	6,425,732	B1	7/2002	Rouse et al.
5,448,889	A	9/1995	Bronicki	6,437,468	B2	8/2002	Stahl et al.
5,497,615	A	3/1996	Noe et al.	6,438,936	B1	8/2002	Ryan
5,529,398	A	6/1996	Bosley	6,438,937	B1	8/2002	Pont et al.
5,549,174	A *	8/1996	Reis ..... 180/165	6,453,658	B1	9/2002	Willis et al.
5,555,719	A	9/1996	Rowen et al.	6,468,051	B2	10/2002	Lampe et al.
5,586,429	A	12/1996	Kesseli et al.	6,487,096	B1	11/2002	Gilbreth et al.
				6,489,692	B1	12/2002	Gilbreth et al.

# US 8,499,874 B2

6,495,929 B2	12/2002	Bosley et al.	6,989,610 B2	1/2006	Gupta et al.
6,499,949 B2	12/2002	Schafrik et al.	6,998,728 B2	2/2006	Gupta et al.
6,522,030 B1	2/2003	Wall et al.	7,053,590 B2	5/2006	Wang
6,526,757 B2	3/2003	MacKay	7,065,873 B2	6/2006	Kang et al.
6,539,720 B2	4/2003	Rouse et al.	RE39,190 E	7/2006	Weissert
6,543,232 B1	4/2003	Anderson et al.	7,092,262 B2	8/2006	Ryan et al.
6,552,440 B2	4/2003	Gilbreth et al.	7,093,443 B2	8/2006	McKelvey et al.
6,605,928 B2	8/2003	Gupta et al.	7,112,036 B2	9/2006	Lubell et al.
6,606,864 B2	8/2003	Mackay	7,117,683 B2	10/2006	Thompson
6,612,112 B2	9/2003	Gilbreth et al.	7,147,050 B2	12/2006	Kang et al.
6,629,064 B1	9/2003	Wall	7,166,928 B2	1/2007	Larsen
6,634,176 B2	10/2003	Rouse et al.	7,181,337 B2	2/2007	Kosaka
6,639,328 B2	10/2003	Wacknov	7,185,496 B2	3/2007	Herlihy
6,644,916 B1	11/2003	Beacom	7,186,200 B1	3/2007	Hauser
RE38,373 E	12/2003	Bosley	7,211,906 B2	5/2007	Teets et al.
6,657,332 B2	12/2003	Balas	7,224,081 B2	5/2007	Larsen
6,657,348 B2	12/2003	Qin et al.	7,244,524 B2	7/2007	McCluskey et al.
6,663,044 B1	12/2003	Squier et al.	7,266,429 B2	9/2007	Travalay et al.
6,664,653 B1	12/2003	Edelman	7,285,871 B2	10/2007	Derouineau
6,664,654 B2	12/2003	Wall et al.	7,299,638 B2	11/2007	Mackay
6,675,583 B2	1/2004	Willis et al.	7,318,154 B2	1/2008	Tehee
6,683,389 B2	1/2004	Geis	7,325,401 B1 *	2/2008	Kesseli et al. .... 60/677
6,684,642 B2	2/2004	Willis et al.	7,343,744 B2	3/2008	Abelson et al.
6,698,208 B2	3/2004	Teets	7,393,179 B1	7/2008	Kesseli et al.
6,698,554 B2	3/2004	Dest a et al.	7,398,642 B2	7/2008	McQuiggan
6,702,463 B1	3/2004	Brockett et al.	7,404,294 B2	7/2008	Sundin
6,709,243 B1	3/2004	Tan et al.	7,415,764 B2	8/2008	Kang et al.
6,713,892 B2	3/2004	Gilbreth et al.	7,423,412 B2	9/2008	Weng et al.
6,720,685 B2	4/2004	Balas	7,464,533 B2	12/2008	Wollenweber
6,729,141 B2	5/2004	Ingram	7,513,120 B2	4/2009	Kupratis
6,732,531 B2	5/2004	Dickey	RE40,713 E	5/2009	Geis et al.
6,735,951 B2	5/2004	Thompson	7,572,531 B2	8/2009	Forte
6,745,574 B1	6/2004	Dettmer	7,574,853 B2	8/2009	Teets et al.
6,747,372 B2	6/2004	Gilbreth et al.	7,574,867 B2	8/2009	Teets et al.
6,748,742 B2	6/2004	Rouse et al.	7,595,124 B2	9/2009	Varatharajan et al.
6,751,941 B2	6/2004	Edelman et al.	7,605,487 B2	10/2009	Barton et al.
6,766,647 B2	7/2004	Hartzheim	7,605,498 B2	10/2009	Ledenev et al.
6,784,565 B2	8/2004	Wall et al.	7,607,318 B2	10/2009	Lui et al.
6,787,933 B2	9/2004	Claude et al.	7,608,937 B1	10/2009	Altenschulte
6,794,766 B2	9/2004	Wickert et al.	7,614,792 B2	11/2009	Wade et al.
6,796,527 B1	9/2004	Munoz et al.	7,615,881 B2	11/2009	Halsey et al.
6,804,946 B2	10/2004	Willis et al.	7,617,687 B2	11/2009	West et al.
6,810,677 B2	11/2004	Dewis	7,656,135 B2	2/2010	Schram et al.
6,812,586 B2	11/2004	Wacknov et al.	7,671,481 B2	3/2010	Miller et al.
6,812,587 B2	11/2004	Gilbreth et al.	7,766,790 B2	8/2010	Stevenson et al.
6,815,932 B2	11/2004	Wall	7,770,376 B1	8/2010	Brostmeyer
6,817,575 B1	11/2004	Munoz et al.	7,777,358 B2	8/2010	Halsey et al.
6,819,999 B2	11/2004	Hartzheim	7,804,184 B2	9/2010	Yuan et al.
6,823,675 B2	11/2004	Brunell et al.	7,841,185 B2	11/2010	Richards et al.
6,829,899 B2	12/2004	Benham, Jr. et al.	7,861,696 B2	1/2011	Lund
6,832,470 B2	12/2004	Dewis	7,866,532 B1	1/2011	Potter et al.
6,834,226 B2	12/2004	Hartzheim	7,921,944 B2 *	4/2011	Russell et al. .... 180/65.265
6,836,720 B2	12/2004	Hartzheim	7,926,274 B2 *	4/2011	Farkaly ..... 60/670
6,837,419 B2	1/2005	Ryan	7,957,846 B2	6/2011	Hakim et al.
6,845,558 B2	1/2005	Beacom	8,046,990 B2 *	11/2011	Bollinger et al. .... 60/410
6,845,621 B2	1/2005	Teets	8,188,693 B2 *	5/2012	Wei et al. .... 318/400.03
6,847,129 B2	1/2005	McKelvey et al.	2001/0030425 A1	10/2001	Gilbreth et al.
6,847,194 B2	1/2005	Sarlioglu et al.	2001/0052704 A1	12/2001	Bosley et al.
6,848,249 B2 *	2/2005	Coleman et al. .... 60/39.17	2002/0054718 A1	5/2002	Weissert
6,863,509 B2	3/2005	Dewis	2002/0063479 A1	5/2002	Mitchell et al.
6,864,595 B2	3/2005	Wall	2002/0067872 A1	6/2002	Weissert
6,870,279 B2	3/2005	Gilbreth et al.	2002/0073688 A1	6/2002	Bosley et al.
6,877,323 B2	4/2005	Dewis	2002/0073713 A1	6/2002	Mackay
6,883,331 B2	4/2005	Jonsson et al.	2002/0079760 A1	6/2002	Vessa
6,888,263 B2	5/2005	Satoh et al.	2002/0083714 A1	7/2002	Bakholdin
6,891,282 B2	5/2005	Gupta et al.	2002/0096393 A1	7/2002	Rouse
6,895,760 B2	5/2005	Kesseli	2002/0096959 A1	7/2002	Qin et al.
6,909,199 B2	6/2005	Gupta et al.	2002/0097928 A1	7/2002	Swinton et al.
6,911,742 B2	6/2005	Gupta et al.	2002/0099476 A1	7/2002	Hamrin et al.
6,931,856 B2	8/2005	Belokon et al.	2002/0103745 A1	8/2002	Lof et al.
6,951,110 B2	10/2005	Kang	2002/0104316 A1	8/2002	Dickey et al.
6,956,301 B2	10/2005	Gupta et al.	2002/0110450 A1	8/2002	Swinton
6,958,550 B2	10/2005	Gilbreth et al.	2002/0119040 A1	8/2002	Bosley
6,960,840 B2	11/2005	Willis et al.	2002/0120368 A1	8/2002	Edelman et al.
6,964,168 B1	11/2005	Pierson et al.	2002/0124569 A1	9/2002	Treece et al.
6,966,173 B2	11/2005	Dewis	2002/0128076 A1	9/2002	Lubell
6,973,880 B2 *	12/2005	Kumar ..... 105/35	2002/0148229 A1	10/2002	Pont et al.
6,977,446 B2	12/2005	Mackay	2002/0149205 A1	10/2002	Gilbreth et al.
6,979,914 B2	12/2005	McKelvey et al.	2002/0149206 A1	10/2002	Gilbreth et al.

# US 8,499,874 B2

2002/0157881	A1	10/2002	Bakholdin et al.	CA	1202099	3/1986
2002/0158517	A1	10/2002	Rouse et al.	CA	1244661	11/1988
2002/0166324	A1	11/2002	Willis et al.	CA	1275719	10/1990
2003/0110773	A1	6/2003	Rouse et al.	CA	2066258	3/1991
2004/0008010	A1	1/2004	Ebrahim et al.	CA	1286882	7/1991
2004/0011038	A1	1/2004	Stinger et al.	CA	2220172	5/1998
2004/0035656	A1	2/2004	Anwar et al.	CA	2234318	10/1998
2004/0080165	A1	4/2004	Geis et al.	CA	2238356	3/1999
2004/0090204	A1	5/2004	McGinley	CA	2242947	3/1999
2004/0103669	A1	6/2004	Willis et al.	CA	2246769	3/1999
2004/0106486	A1	6/2004	Jonsson	CA	2279320	4/2000
2004/0119291	A1	6/2004	Hamrin et al.	CA	2677758	4/2000
2004/0148942	A1	8/2004	Pont et al.	CA	2317855	5/2001
2004/0160061	A1	8/2004	Rouse et al.	CA	2254034	6/2007
2005/0000224	A1	1/2005	Jonsson	CA	2638648	2/2009
2005/0103931	A1	5/2005	Morris et al.	CA	2689188	7/2010
2005/0206331	A1	9/2005	Donnelly	CH	595552	2/1978
2005/0228553	A1 *	10/2005	Tryon ..... 701/22	CH	679235	1/1992
2006/0076171	A1	4/2006	Donnelly et al.	CN	1052170	6/1991
2006/0090109	A1 *	4/2006	Bonnet ..... 714/724	CN	1060270	4/1992
2007/0012129	A1	1/2007	Maty et al.	CN	1306603	8/2001
2007/0068712	A1 *	3/2007	Carnahan ..... 180/65.2	CN	1317634	10/2001
2007/0178340	A1	8/2007	Eickhoff	CN	1902389	1/2007
2007/0181294	A1	8/2007	Soldner et al.	CN	101098079	1/2008
2007/0239325	A1	10/2007	Regunath	CN	100564811	12/2009
2007/0290039	A1	12/2007	Pfleging et al.	CN	101635449	1/2010
2008/0080682	A1	4/2008	Ogunwale et al.	CN	101672252	3/2010
2008/0148708	A1	6/2008	Chou et al.	CS	9101996	1/1992
2008/0197705	A1	8/2008	Dewis et al.	CZ	20014556	4/2003
2009/0045292	A1	2/2009	Maddali et al.	DE	1272306	7/1968
2009/0071478	A1	3/2009	Kalfon	DE	2753673	6/1978
2009/0090109	A1	4/2009	Mills et al.	DE	2853919	6/1979
2009/0106978	A1	4/2009	Wollenweber	DE	3140694	7/1982
2009/0109022	A1	4/2009	Gangopadhyay et al.	DE	3736984	5/1988
2009/0158739	A1	6/2009	Messmer	DE	69519684	8/2001
2009/0211260	A1	8/2009	Kesseli et al.	DE	10305352	9/2004
2009/0211739	A1	8/2009	Nash et al.	DE	69828916	3/2006
2009/0211740	A1	8/2009	Kesseli et al.	DE	60125441	2/2007
2009/0249786	A1	10/2009	Garrett et al.	DE	60125583	2/2007
2009/0271086	A1	10/2009	Morris et al.	DK	331889	7/1989
2009/0292436	A1	11/2009	D'Amato et al.	EP	0092551	11/1983
2009/0313990	A1	12/2009	Mustafa	EP	0093118	11/1983
2010/0021284	A1	1/2010	Watson et al.	EP	0104921	4/1984
2010/0052425	A1	3/2010	Moore et al.	EP	0157794	10/1985
2010/0154380	A1	6/2010	Tangirala et al.	EP	0377292	7/1990
2010/0229525	A1	9/2010	Mackay et al.	EP	0319246	10/1990
2010/0288571	A1 *	11/2010	Dewis et al. .... 180/165	EP	0432753	6/1991
2010/0293946	A1	11/2010	Vick	EP	0455640	11/1991
2010/0319355	A1	12/2010	Prabhu	EP	0472294	2/1992
2011/0020108	A1	1/2011	Axelsson et al.	EP	0478713	4/1992
2011/0100777	A1	5/2011	Wilton et al.	EP	0493481	7/1992
2011/0215640	A1	9/2011	Donnelly	EP	0522832	1/1993
2011/0288738	A1	11/2011	Donnelly et al.	EP	0620906	10/1994
2011/0295453	A1 *	12/2011	Betz et al. .... 701/22	EP	0691511	1/1996
2012/0000204	A1	1/2012	Kesseli et al.	EP	0754142	1/1997
2012/0017598	A1	1/2012	Kesseli et al.	EP	0784156	12/1997
2012/0042656	A1	2/2012	Donnelly et al.	EP	0837224	4/1998
2012/0096869	A1	4/2012	Kesseli et al.	EP	0837231	4/1998
2012/0102911	A1	5/2012	Dewis et al.	EP	0901218	3/1999
2012/0175886	A1	7/2012	Donnelly et al.	EP	0698178	6/1999
2012/0201657	A1	8/2012	Donnelly et al.	EP	0963035	12/1999
2012/0260662	A1	10/2012	Nash et al.	EP	1055809	11/2000
2012/0324903	A1	12/2012	Dewis et al.	EP	1075724	2/2001
				EP	1046786	1/2002
				EP	1071185	1/2002
				EP	1215393	6/2002
				EP	0739087	8/2002
				EP	1240713	9/2002
				EP	1277267	1/2003
				EP	1283166	2/2003
				EP	1305210	5/2003
				EP	1340301	9/2003
				EP	1340304	9/2003
				EP	1341990	9/2003
				EP	1342044	9/2003
				EP	1346139	9/2003
				EP	1436504	7/2004
				EP	1203866	8/2004
				EP	0800616	12/2004
FOREIGN PATENT DOCUMENTS						
AU	587266	8/1989				
AU	8517301	3/2002				
AU	2025002	5/2002				
AU	2589802	5/2002				
AU	2004203836	3/2005				
AU	2004208656	2/2009				
AU	2004318142	6/2009				
CA	1050637	3/1979				
CA	1068492	12/1979				
CA	1098997	4/1981				
CA	1099373	4/1981				
CA	1133263	10/1982				
CA	1171671	7/1984				
CA	1190050	7/1985				

# US 8,499,874 B2

EP	1519011	3/2005	GB	2074254	10/1981
EP	1132614	1/2007	GB	2089433	6/1982
EP	1790568	5/2007	GB	2123154	1/1984
EP	1813807	8/2007	GB	2174824	11/1986
EP	1825115	8/2007	GB	2184609	6/1987
EP	1860750	11/2007	GB	2199083	6/1988
EP	1939396	7/2008	GB	2211285	6/1989
EP	2028104	2/2009	GB	2218255	11/1989
EP	1638184	3/2009	GB	2232207	12/1990
EP	1648096	7/2009	GB	2341897	3/2000
EP	2108828	10/2009	GB	2355286	4/2001
EP	1728990	11/2009	GB	2420615	5/2006
EP	2161444	3/2010	GB	2426043	11/2006
EP	2169800	3/2010	GB	2435529	8/2007
EP	1713141	5/2010	GB	2436708	10/2007
EP	1728304	6/2010	GB	2441924	3/2008
EP	1468180	7/2010	GB	2442585	4/2008
FR	2467286	11/1985	GB	2456336	7/2009
FR	2637942	4/1990	GB	2456672	7/2009
FR	2645908	10/1990	GB	2447514	12/2009
FR	2755319	4/1998	IN	4946DELNP2006	8/2007
FR	2848647	6/2004	IN	4341DELNP2005	10/2007
GB	612817	11/1948	IN	5879DELNP2008	9/2008
GB	671379	5/1952	IN	2502DEL2005	10/2009
GB	673961	6/1952	IN	1913DEL2009	6/2010
GB	706743	4/1954	IN	55DEL2010	7/2010
GB	731735	6/1955	IN	2013DEL2009	7/2010
GB	761955	11/1956	IT	1173399	6/1987
GB	768047	2/1957	IT	1194590	9/1988
GB	784119	10/1957	IT	MI911564	1/1992
GB	786001	11/1957	JP	51-065252	6/1976
GB	789589	1/1958	JP	56-088920	7/1981
GB	807267	1/1959	JP	56-148624	11/1981
GB	817507	7/1959	JP	56-148625	11/1981
GB	834550	5/1960	JP	60-184973	9/1985
GB	864712	4/1961	JP	61-182489	8/1986
GB	874251	8/1961	JP	3182638	8/1991
GB	877838	9/1961	JP	6201891	7/1994
GB	878552	10/1961	JP	2519620	7/1996
GB	885184	12/1961	JP	10-054561	2/1998
GB	917392	2/1963	JP	10-061660	3/1998
GB	919540	2/1963	JP	10-115229	5/1998
GB	920408	3/1963	JP	10-122180	5/1998
GB	924078	4/1963	JP	11-324727	11/1999
GB	931926	7/1963	JP	2000-054855	2/2000
GB	937278	9/1963	JP	2000-130319	5/2000
GB	937681	9/1963	JP	2000-329096	11/2000
GB	950015	2/1964	JP	2002-030942	1/2002
GB	950506	2/1964	JP	2002-115565	4/2002
GB	977402	12/1964	JP	2003-009593	1/2003
GB	993039	5/1965	JP	2003-013744	1/2003
GB	1004953	9/1965	JP	2003-041906	2/2003
GB	1008310	10/1965	JP	2004-163087	6/2004
GB	1009115	11/1965	JP	2005-345095	12/2005
GB	1012909	12/1965	JP	2006-022811	1/2006
GB	1043271	9/1966	JP	2006-170208	6/2006
GB	1083943	9/1967	JP	2006-174694	6/2006
GB	1097623	1/1968	JP	2006-200438	8/2006
GB	1103032	2/1968	JP	2007-231949	9/2007
GB	1127856	9/1968	JP	2008-111438	5/2008
GB	1137691	12/1968	JP	2008-132973	6/2008
GB	1138807	1/1969	JP	2009-108756	5/2009
GB	1141019	1/1969	JP	2009-108860	5/2009
GB	1148179	4/1969	JP	2009-209931	9/2009
GB	1158271	7/1969	JP	2009-216085	9/2009
GB	1172126	11/1969	JP	2009-250040	10/2009
GB	1174207	12/1969	JP	2010-014114	1/2010
GB	1211607	11/1970	JP	2010-106835	5/2010
GB	1270011	4/1972	KR	19840002483	12/1984
GB	1275753	5/1972	KR	880002362	10/1988
GB	1275754	5/1972	KR	890001170	4/1989
GB	1275755	5/1972	KR	1020010007189	1/2001
GB	1301104	12/1972	KR	1020020024545	3/2002
GB	1348797	3/1974	KR	1020030032864	4/2003
GB	1392271	4/1975	KR	1020060096320	9/2006
GB	1454766	11/1976	KR	1020070078978	8/2007
GB	1460590	1/1977	KR	1020070113990	11/2007
GB	1516664	7/1978	KR	1020080033866	4/2008
GB	2019494	10/1979	KR	1020090121248	11/2009

NL 7903120 10/1979  
 SE 437543 3/1985  
 SE 9901718 5/1999  
 SE 0103180 3/2003  
 WO WO 8501326 3/1985  
 WO WO 9207221 4/1992  
 WO WO 9524072 9/1995  
 WO WO 9722176 6/1997  
 WO WO 9722789 6/1997  
 WO WO 9726491 7/1997  
 WO WO 9825014 6/1998  
 WO WO 9854448 12/1998  
 WO WO 9919161 4/1999  
 WO WO 0140644 6/2001  
 WO WO 0182448 11/2001  
 WO WO 0202920 1/2002  
 WO WO 0240844 5/2002  
 WO WO 0242611 5/2002  
 WO WO 0244574 6/2002  
 WO WO 0250618 6/2002  
 WO WO 0237638 9/2002  
 WO WO 0229225 2/2003  
 WO WO 0239045 2/2003  
 WO WO 03093652 6/2004  
 WO WO 2004077637 9/2004  
 WO WO 2005045345 5/2005  
 WO WO 2005099063 10/2005  
 WO WO 2008044972 4/2008  
 WO WO 2008044973 4/2008  
 WO WO 2008082334 7/2008  
 WO WO 2008082335 7/2008  
 WO WO 2008082336 7/2008  
 WO WO 2009067048 5/2009

WO WO 2010050856 5/2010  
 WO WO 2010082893 7/2010  
 ZA 8608745 7/1987

OTHER PUBLICATIONS

Written Opinion for International (PCT) Application No. PCT/US2010/034375, mailed Jul. 15, 2010.  
 "A High-Efficiency ICR Microturbine for Commercial Vehicle Propulsion," PACCAR, date unknown, 11 pages.  
 "Analysis of Technology Options to Reduce the Fuel Consumption of Idling Trucks," Stodolsky, F., L. Gaines, and A. Vyas, Argonne National Laboratory, ANL/ESD-43, Jun. 2000, 40 pages.  
 "Benefits of the Microturbine to Power the Next Generation of Trucks." Kenworth Truck Company, date unknown, 9 pages.  
 "Why Gas Turbines have a Future in Heavy Duty Trucks." Capstone Turbine Corporation, Brayton Energy, LLC, Kenworth Truck Company, a PACCAR Company, Peterbilt Truck Company, a PACCAR Company, Apr. 2009, 10 pages.  
 Balogh et al. "DC Link Floating for Grid Connected PV Converters," World Academy of Science, Engineering and Technology Apr. 2008, Iss. 40, pp. 115-120.  
 Mackay et al. "High Efficiency Vehicular Gas Turbines," SAE International, 2005, 10 pages.  
 Nemeth et al. "Life Predicted in a Probabilistic Design Space for Brittle Materials With Transient Loads," NASA, last updated Jul. 21, 2005, found at <http://www.grc.nasa.gov/WWW/RT/2004/RS/RS06L-nemeth.html>, 5 pages.  
 Wolf et al. "Preliminary Design and Projected Performance for Intercooled-Recuperated Microturbine," Proceedings of the ASME TurboExpo 2008 Microturbine and Small Turbomachinery Systems, Jun. 9-13, 2008, Berlin, Germany, 10 pages.

\* cited by examiner

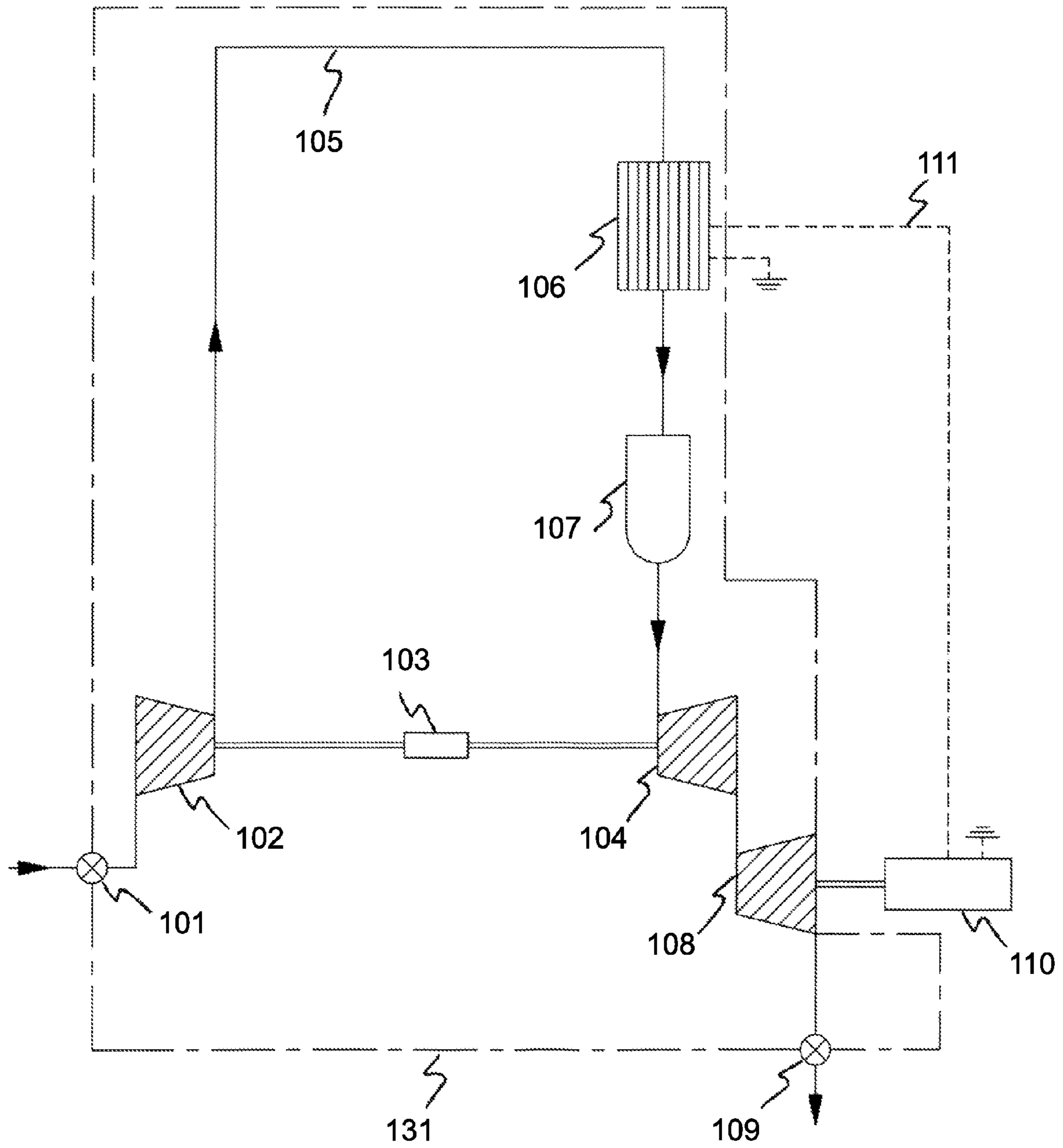


Figure 1





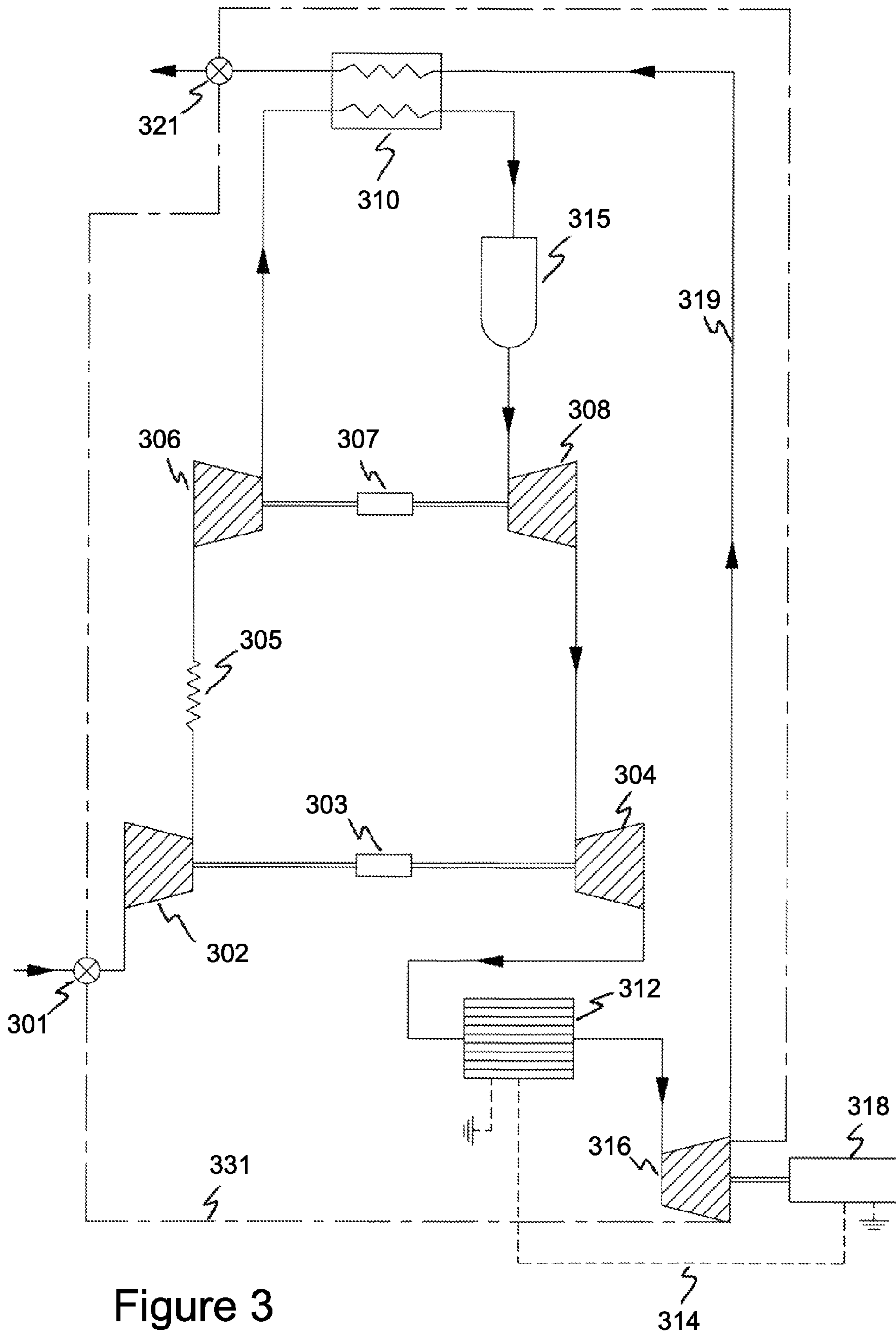


Figure 3

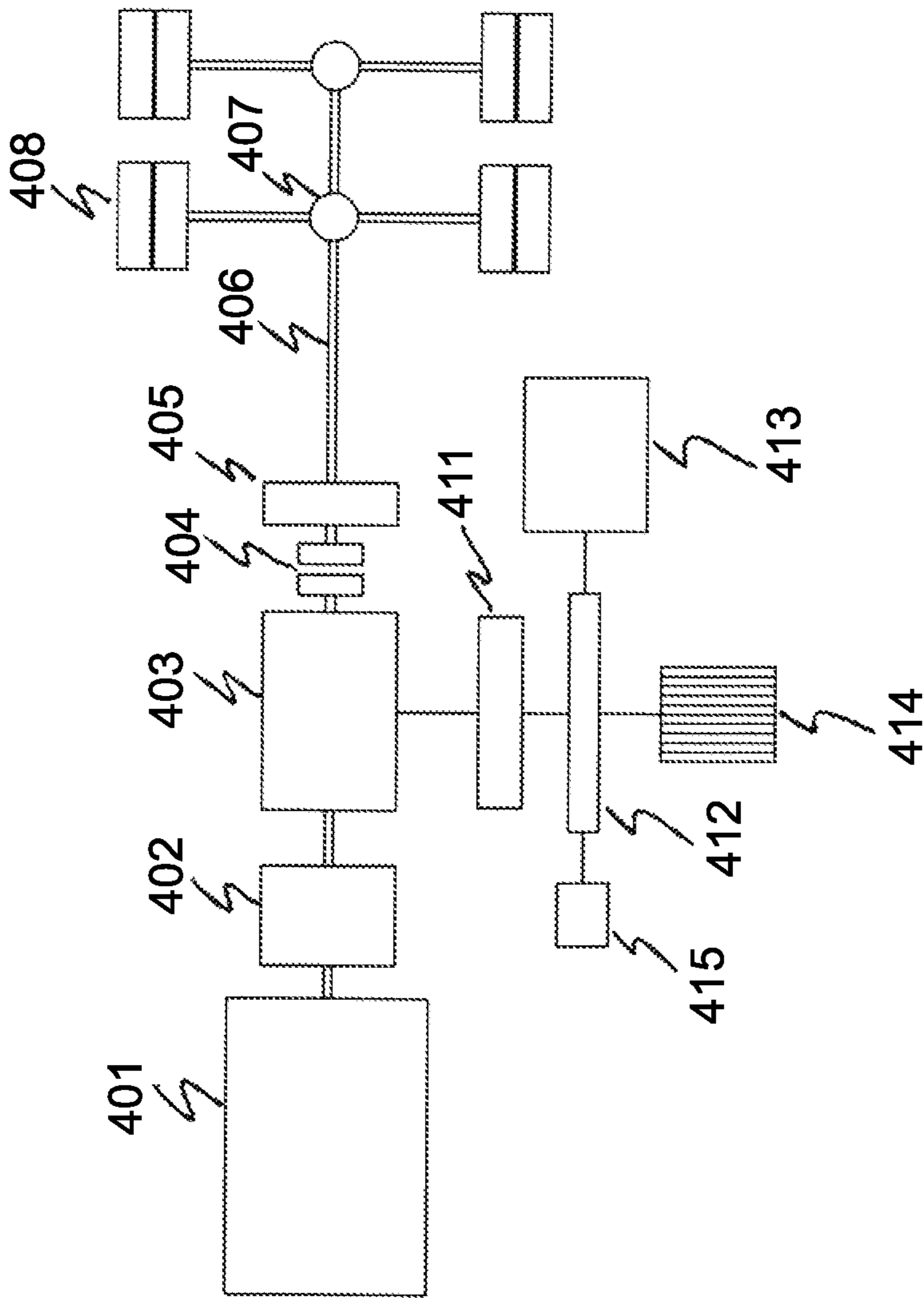


Figure 4

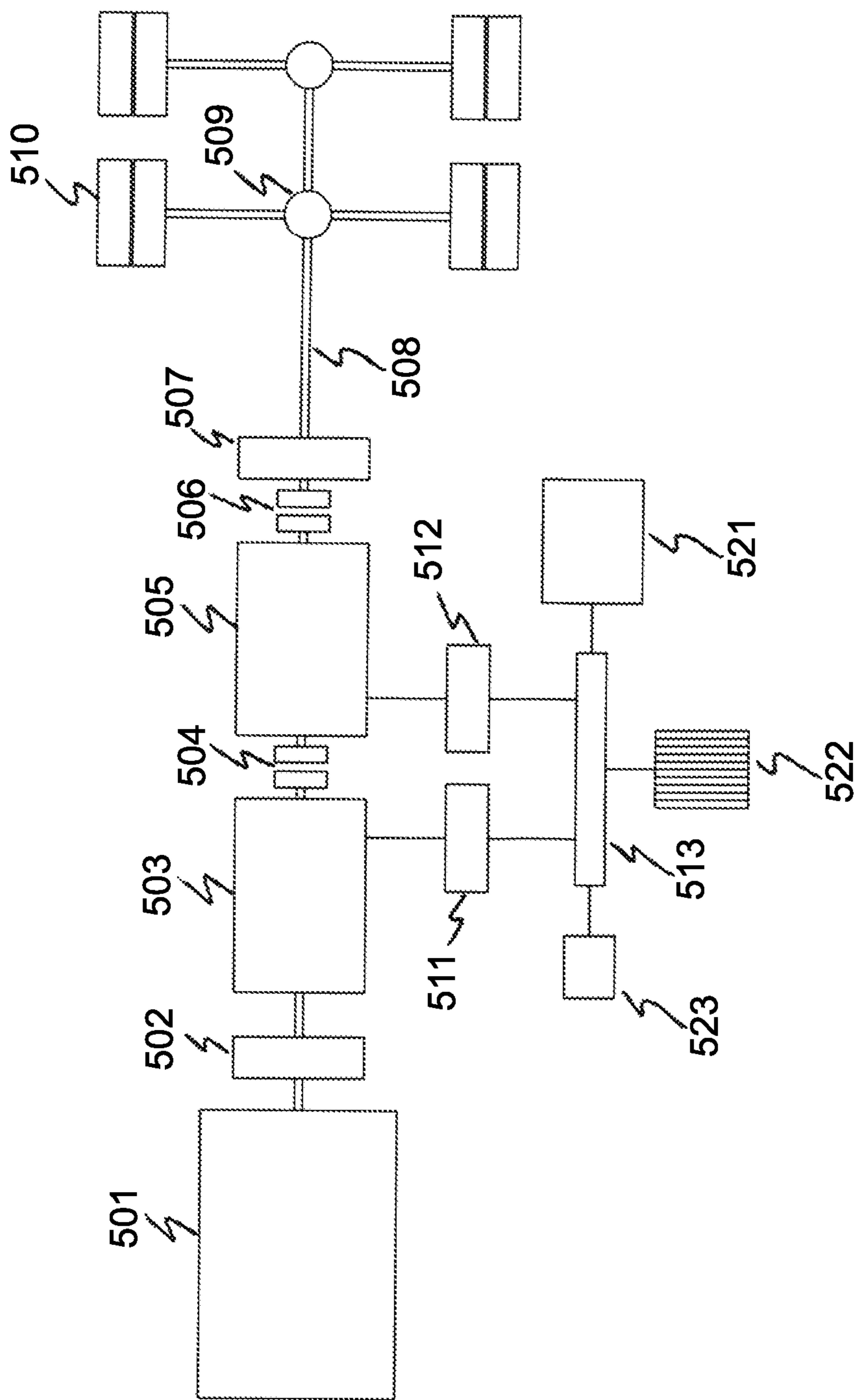


Figure 5

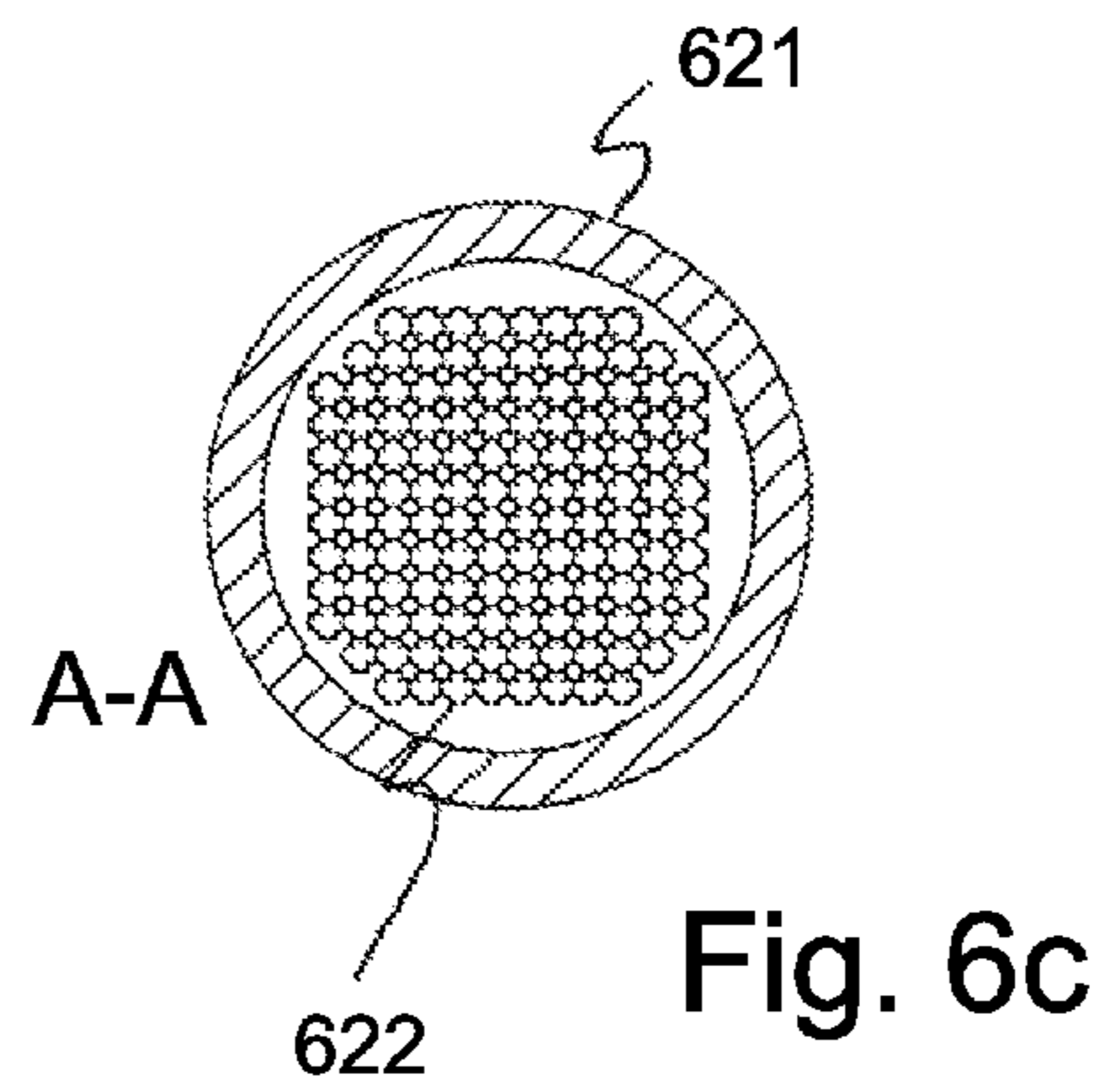
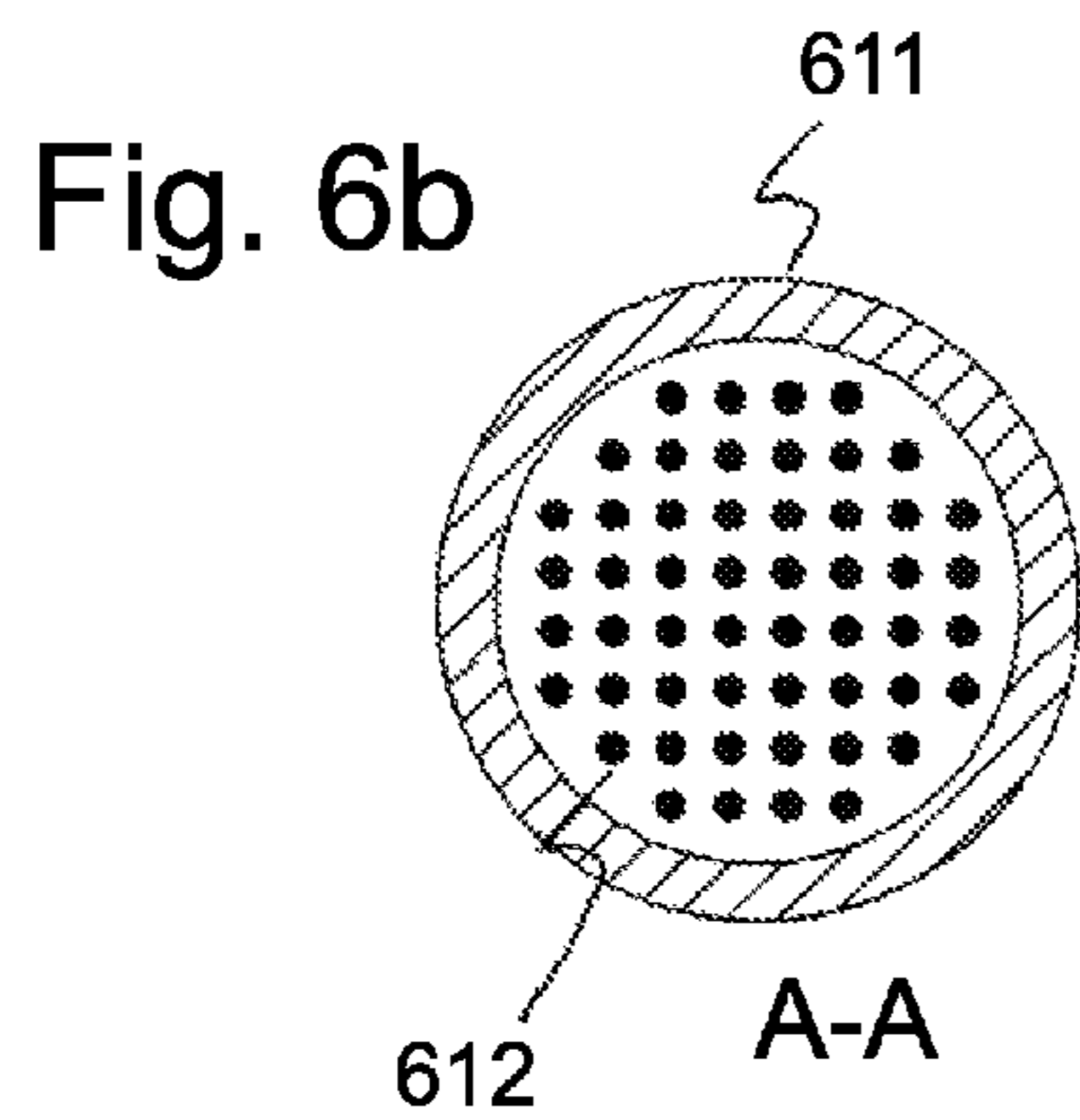
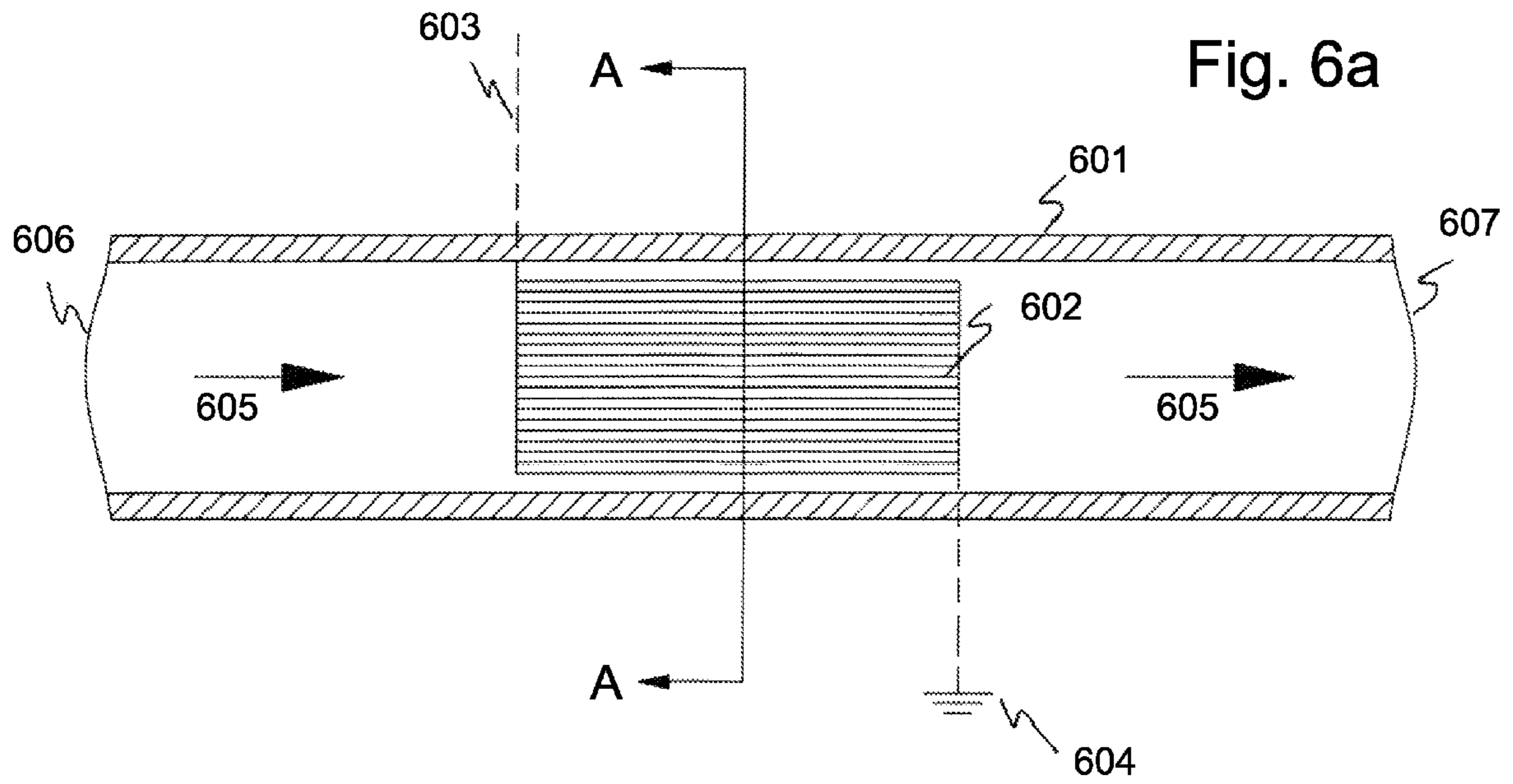


Figure 6

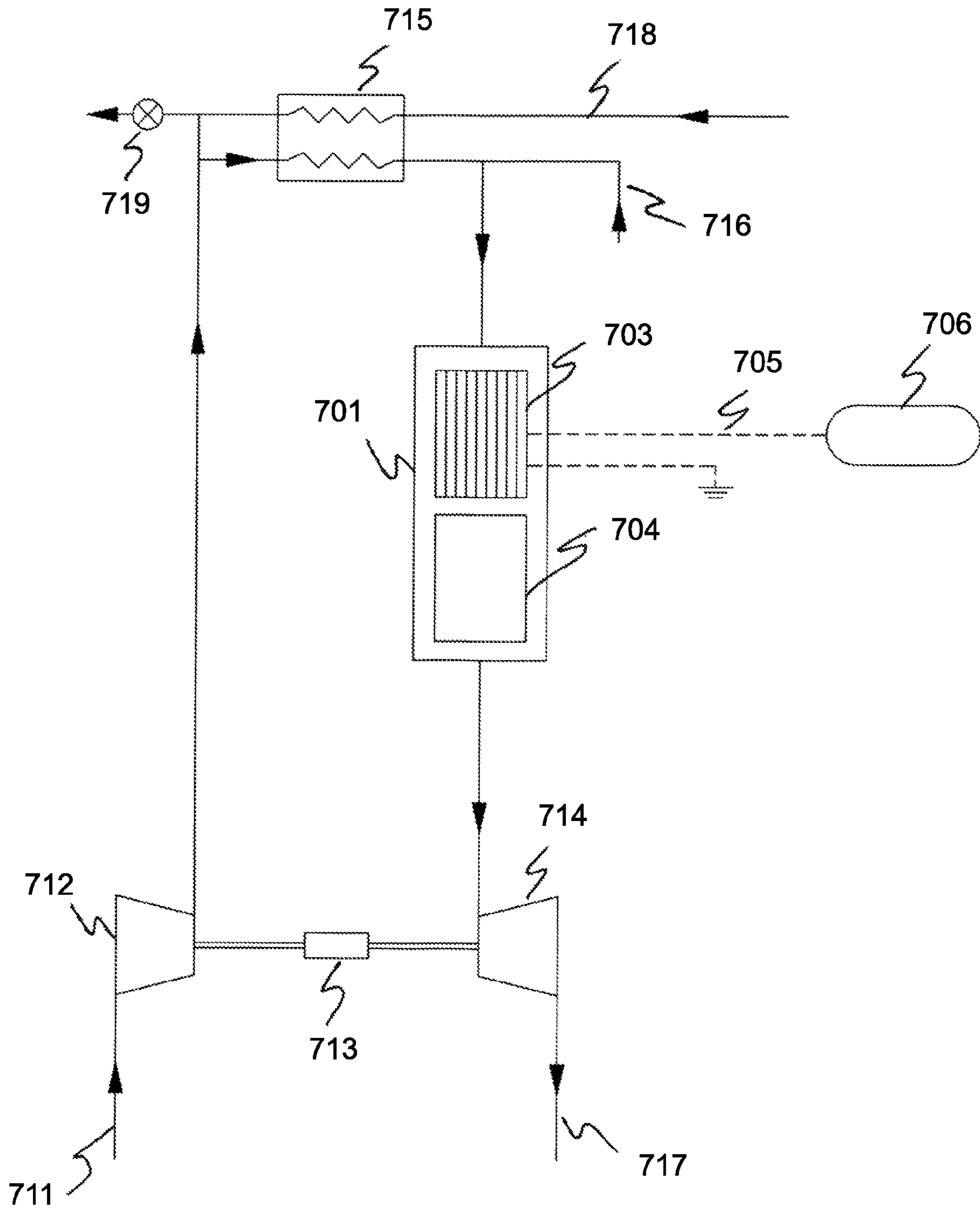


Figure 7

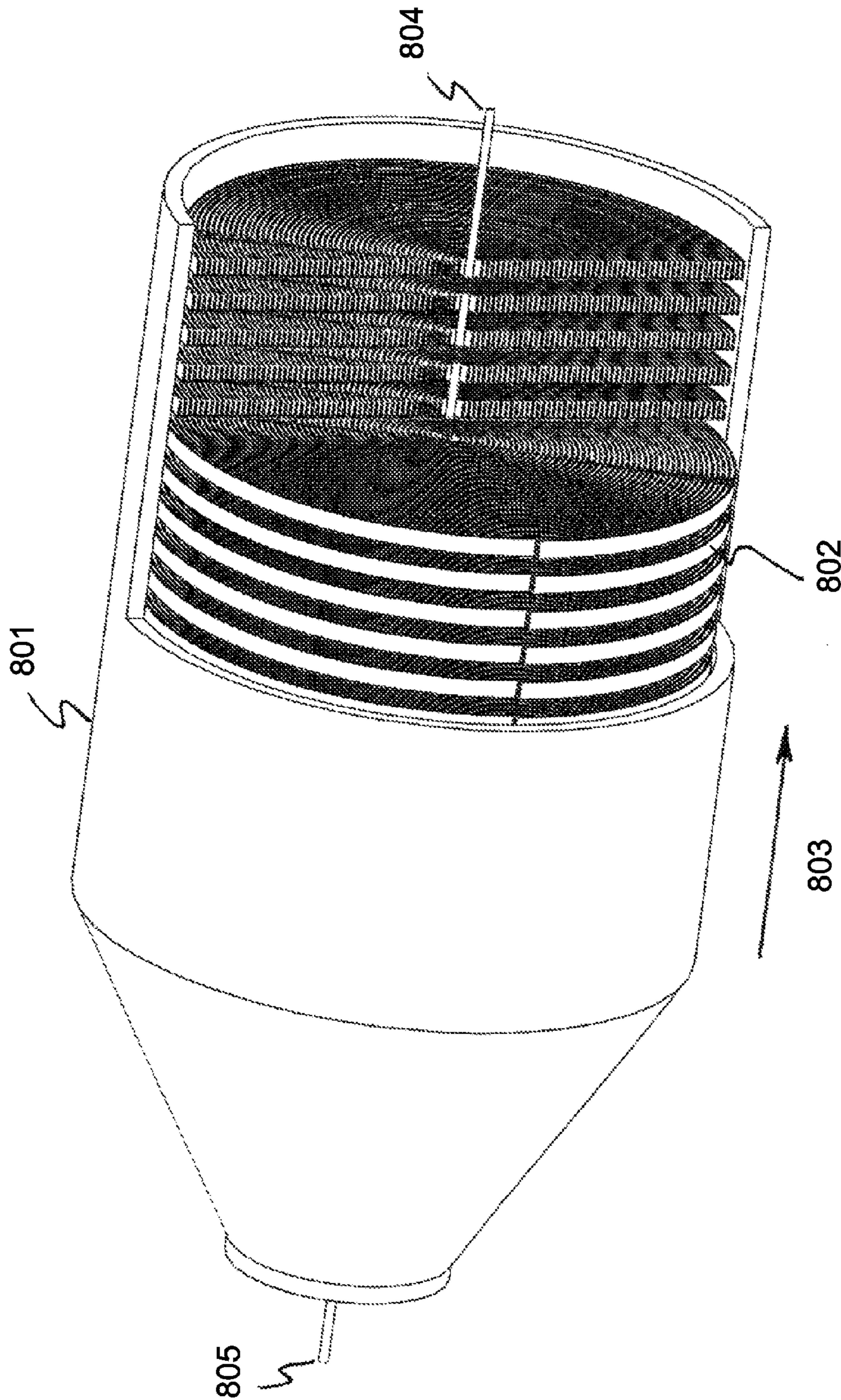


Figure 8

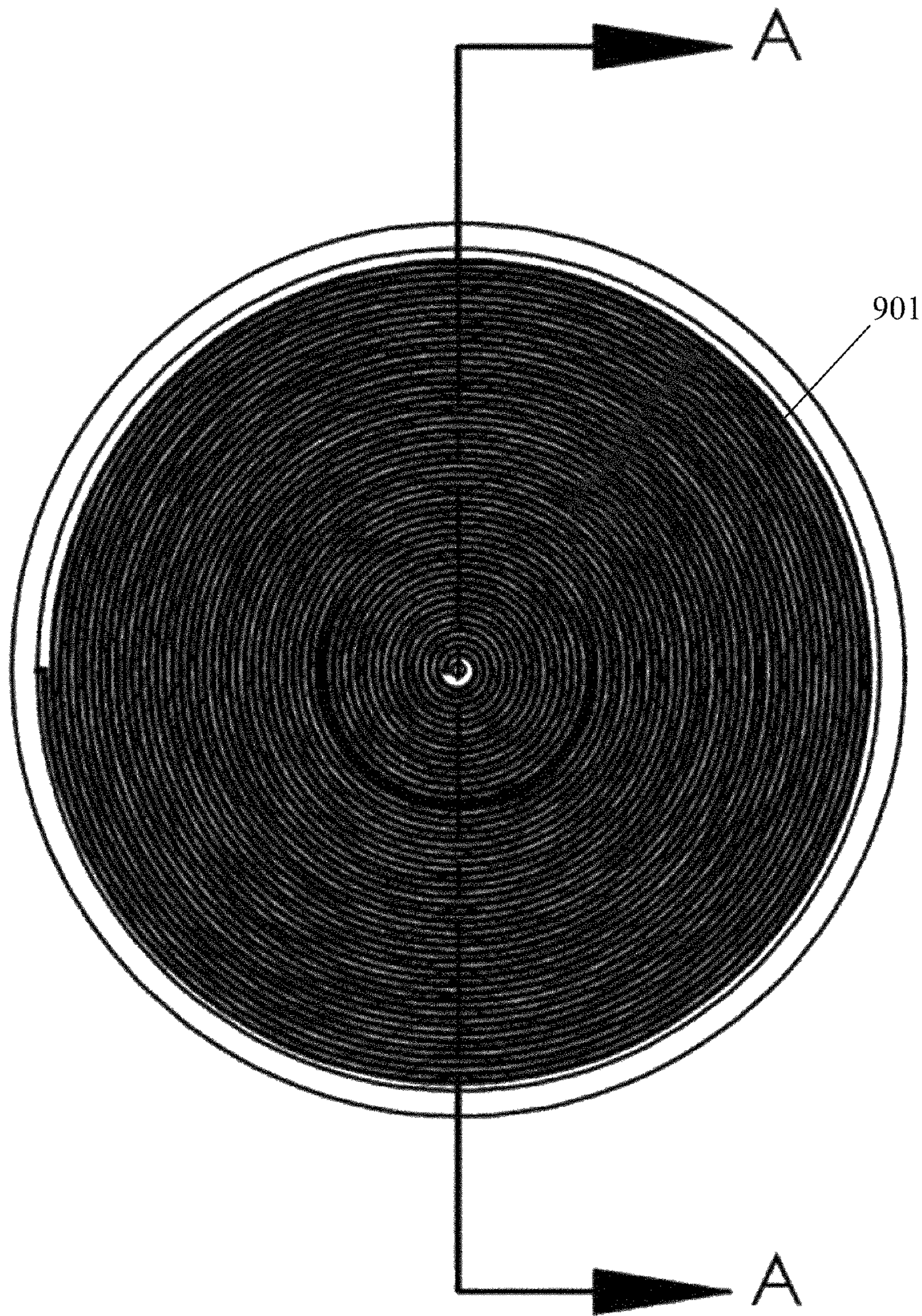


Fig. 9a

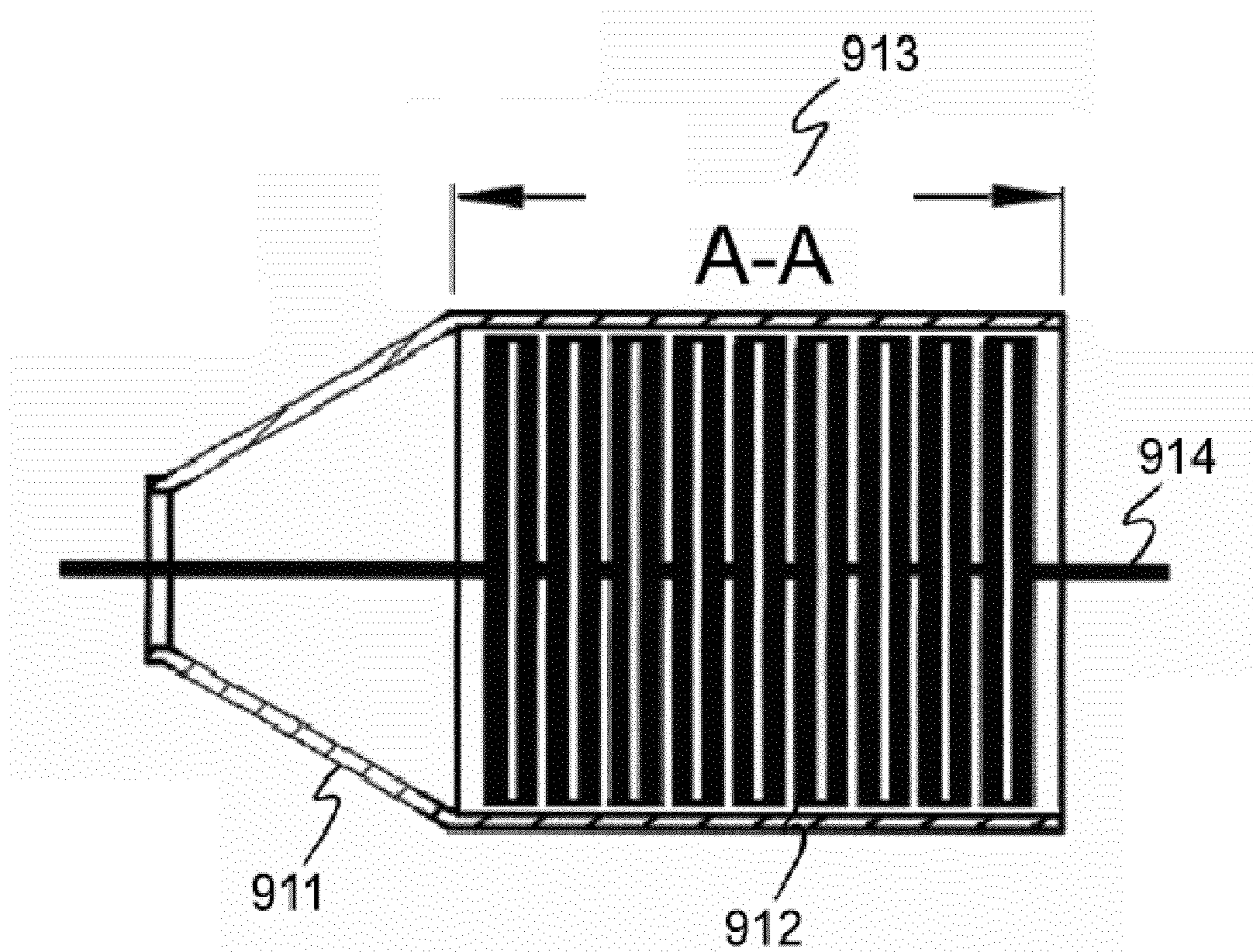


Fig. 9b



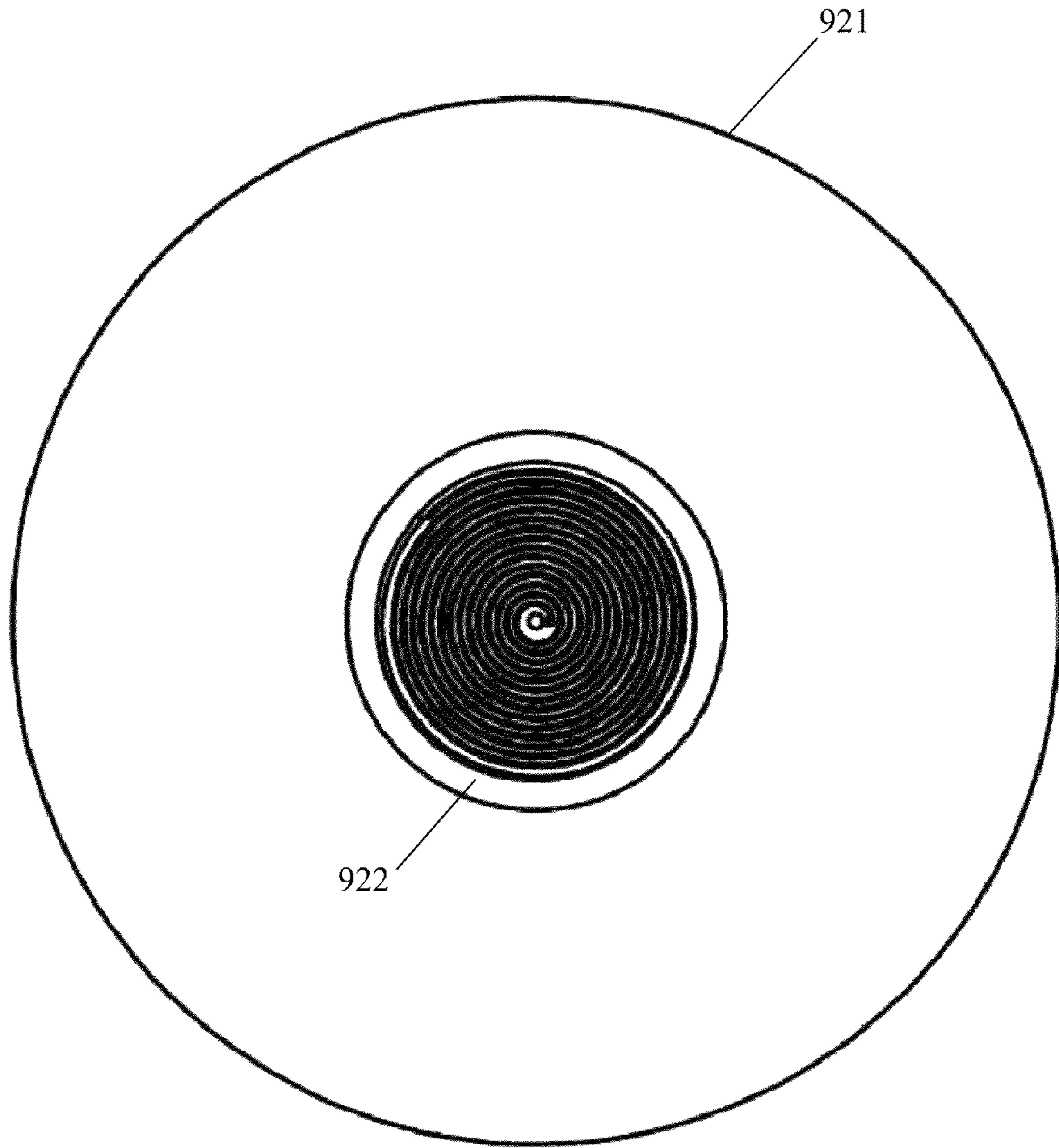


Fig. 9c

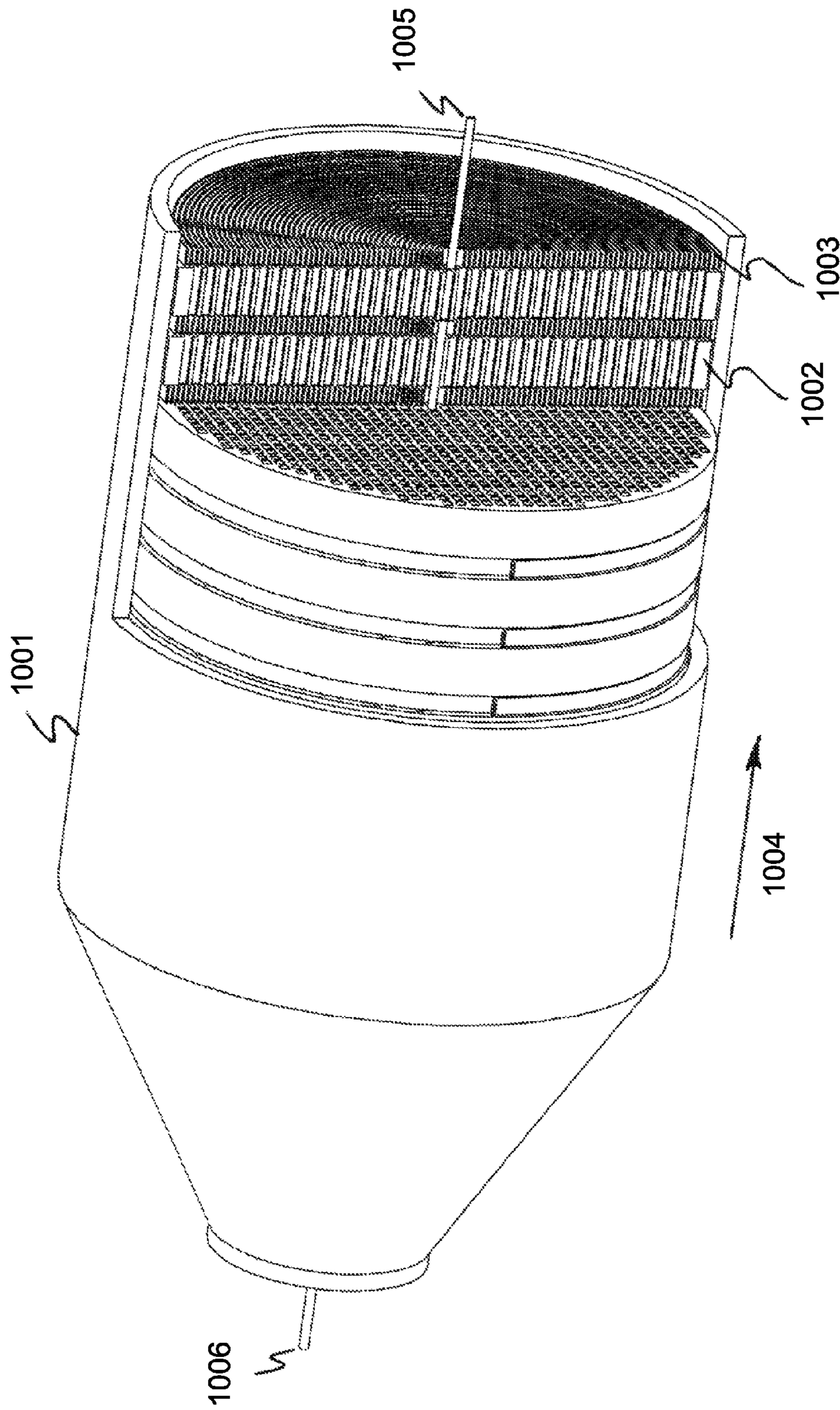


Figure 10

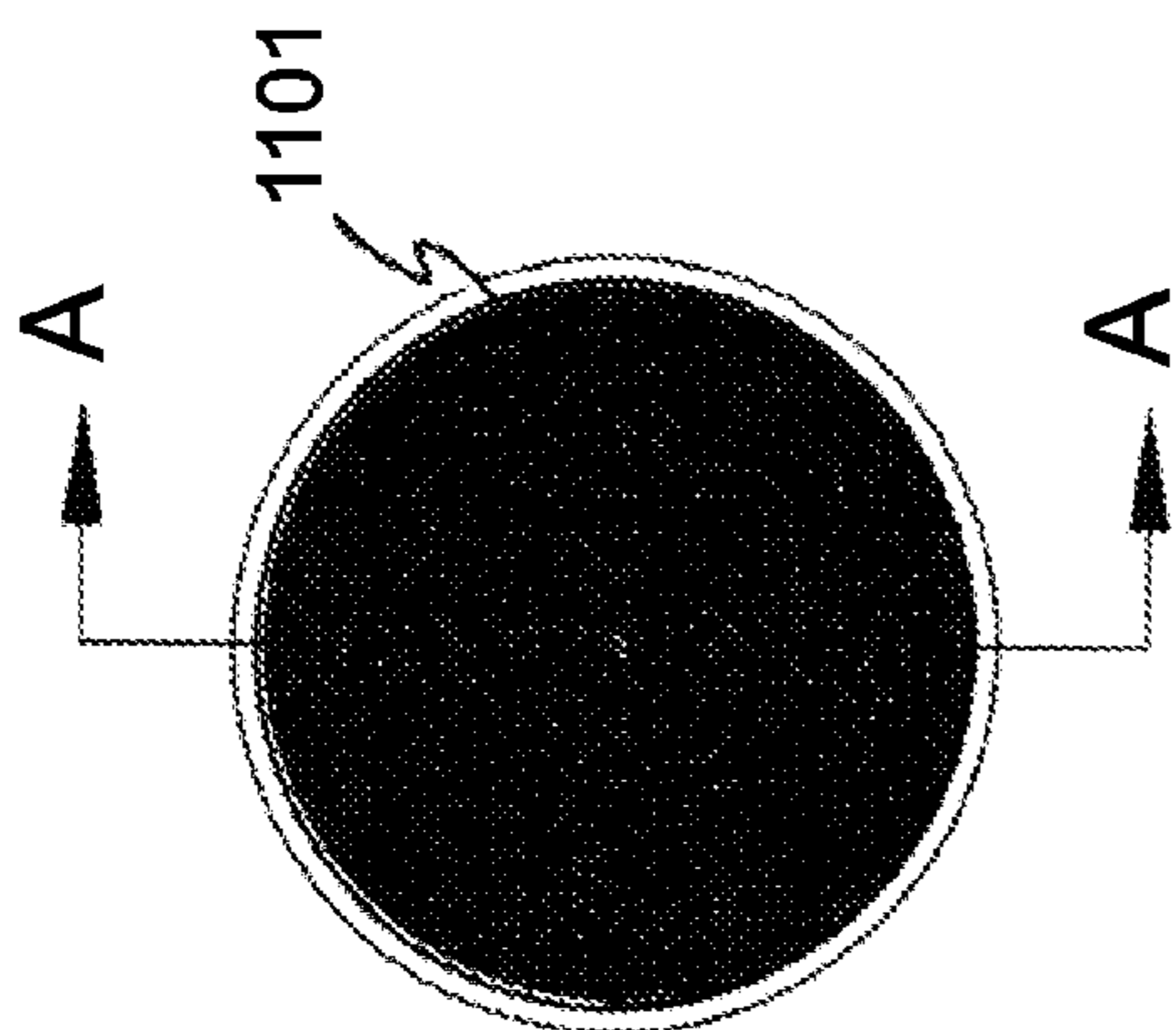


Fig. 11a

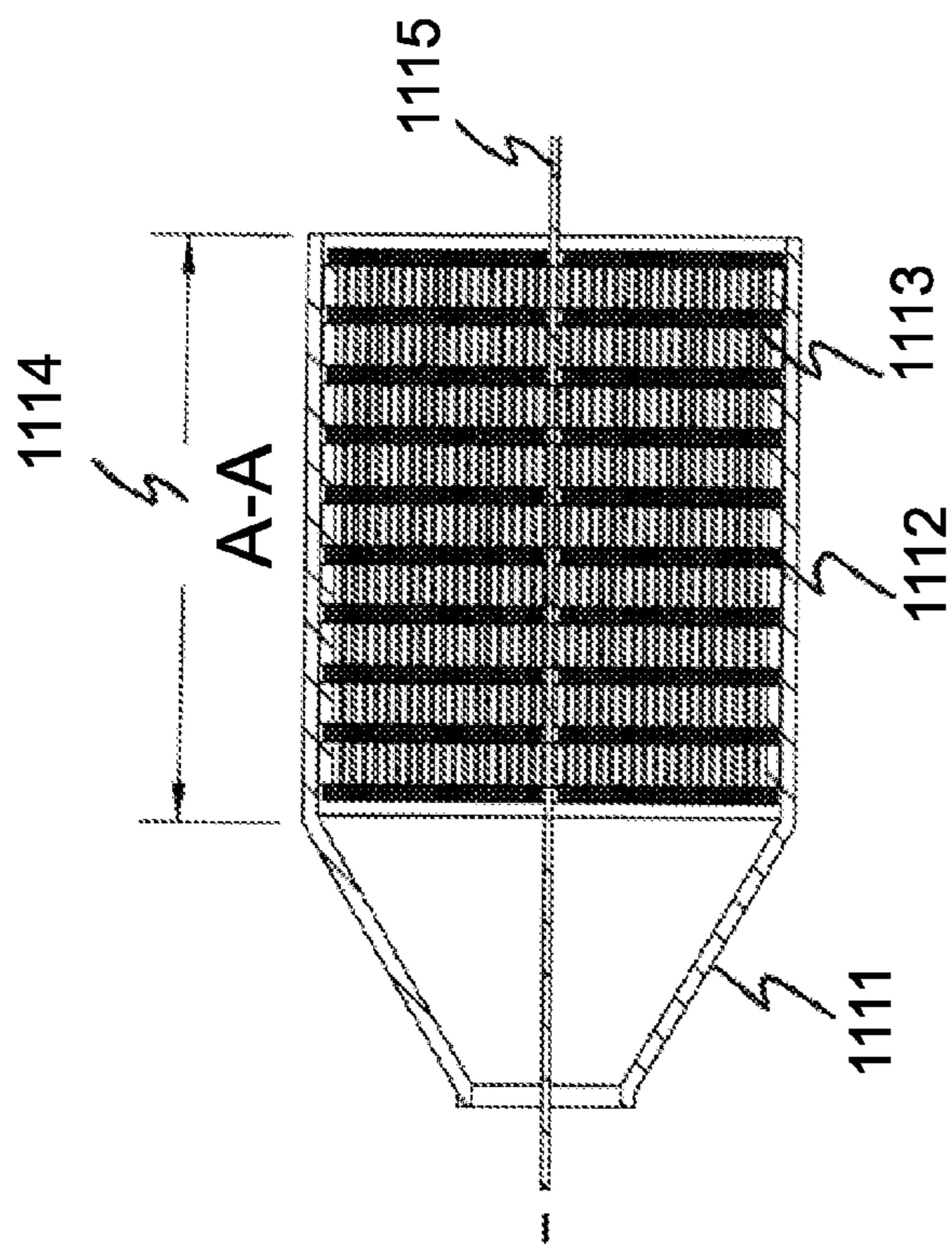


Fig. 11b

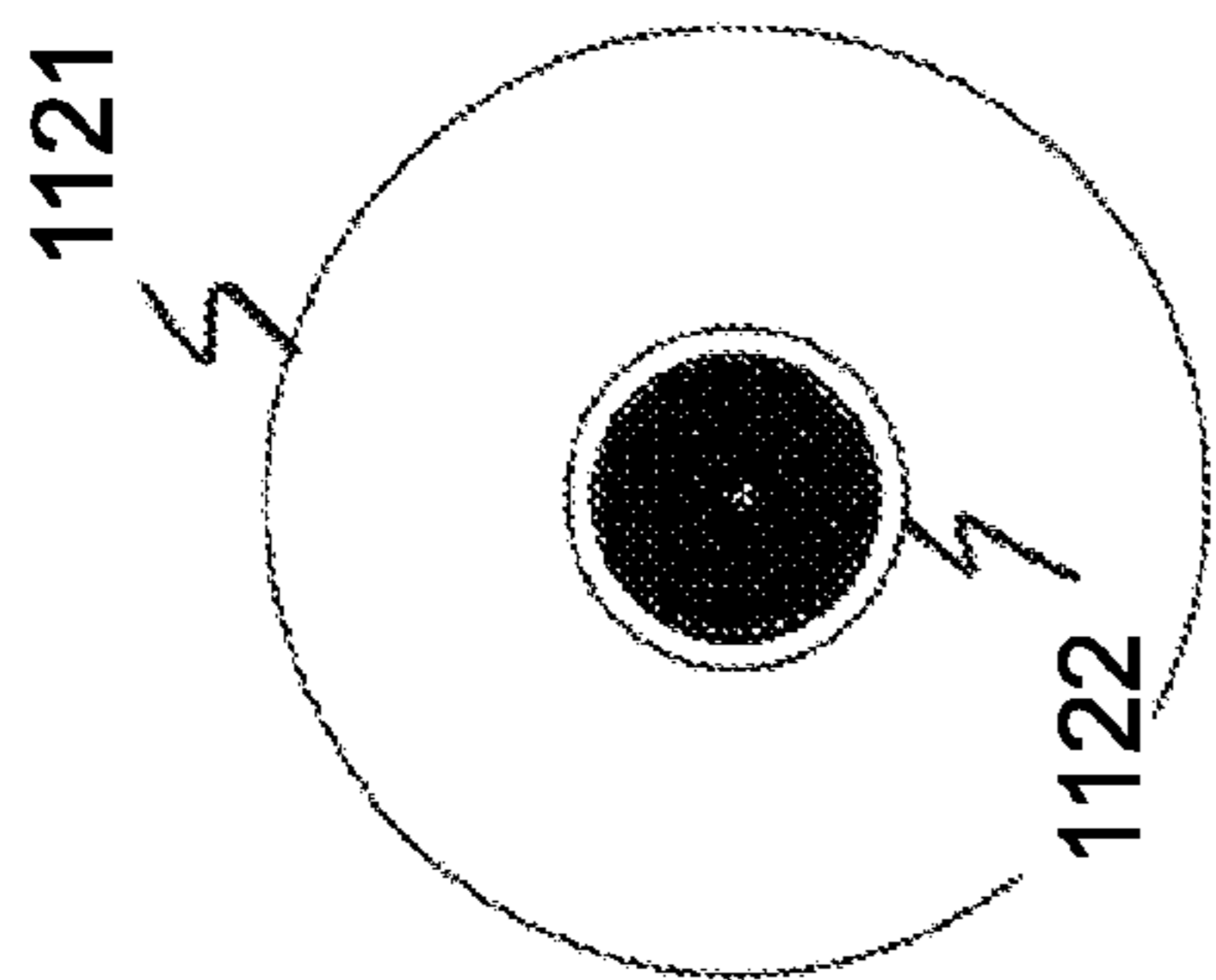


Fig. 11c

Figure 11

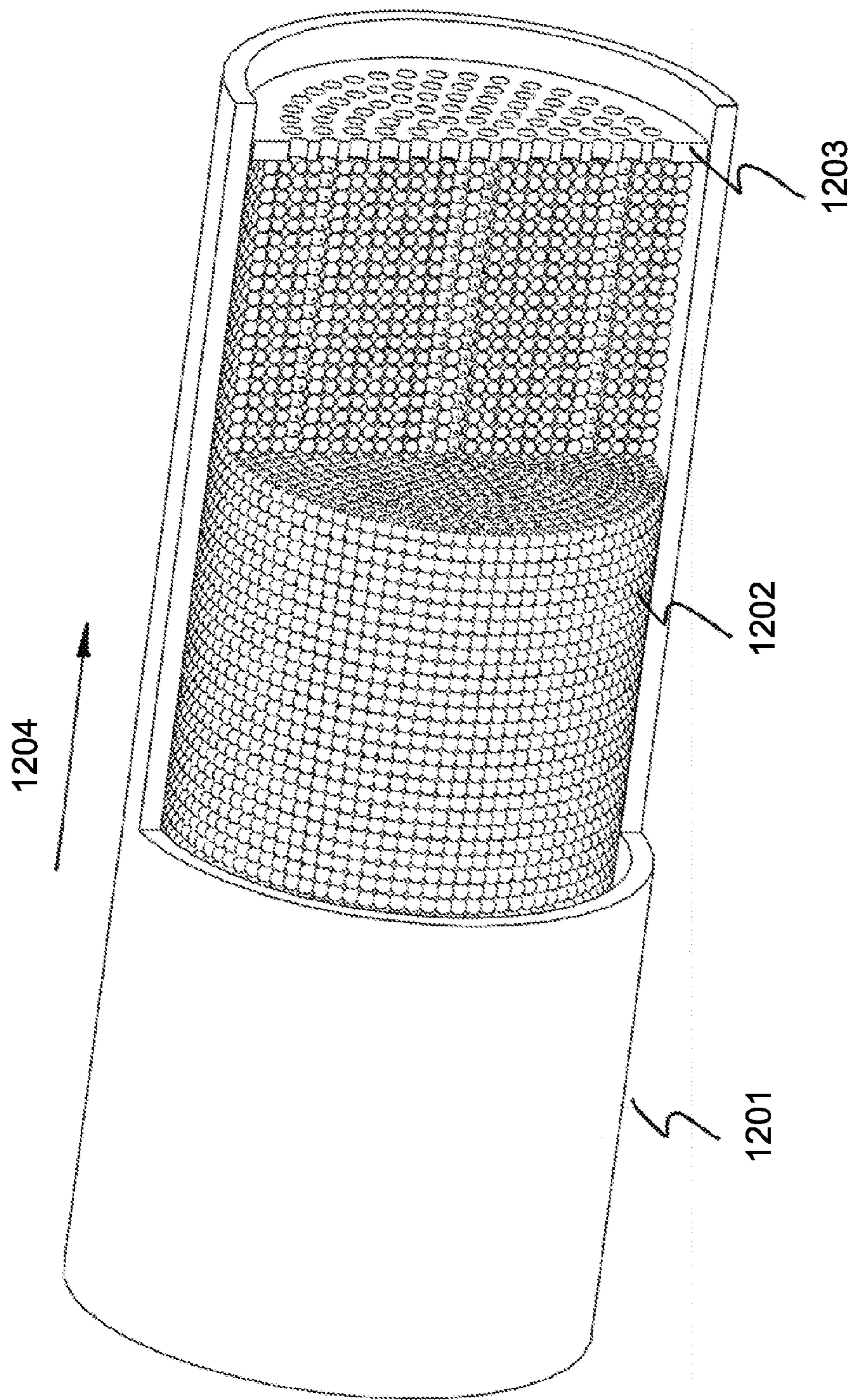


Figure 12

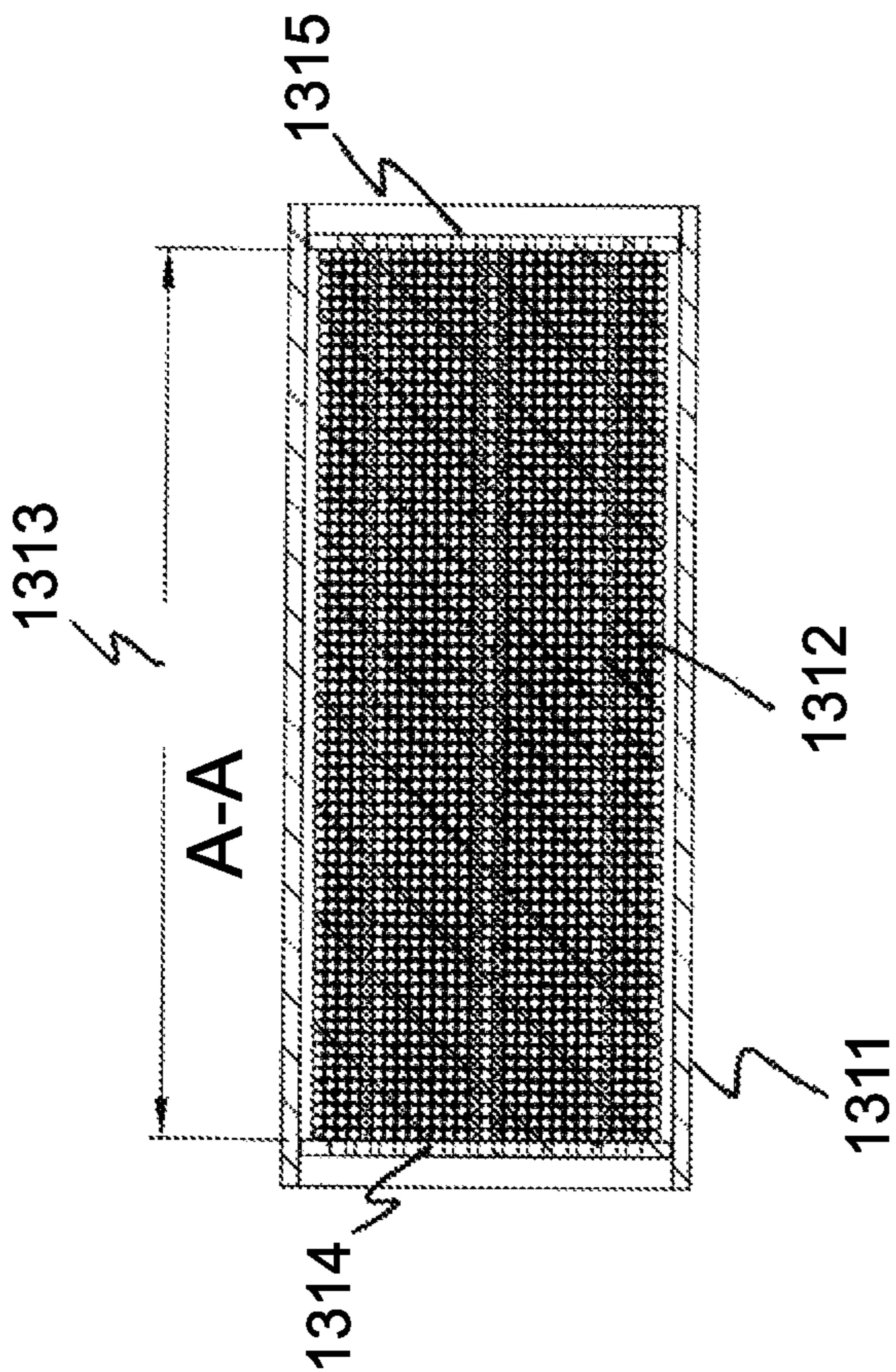


Fig. 13a

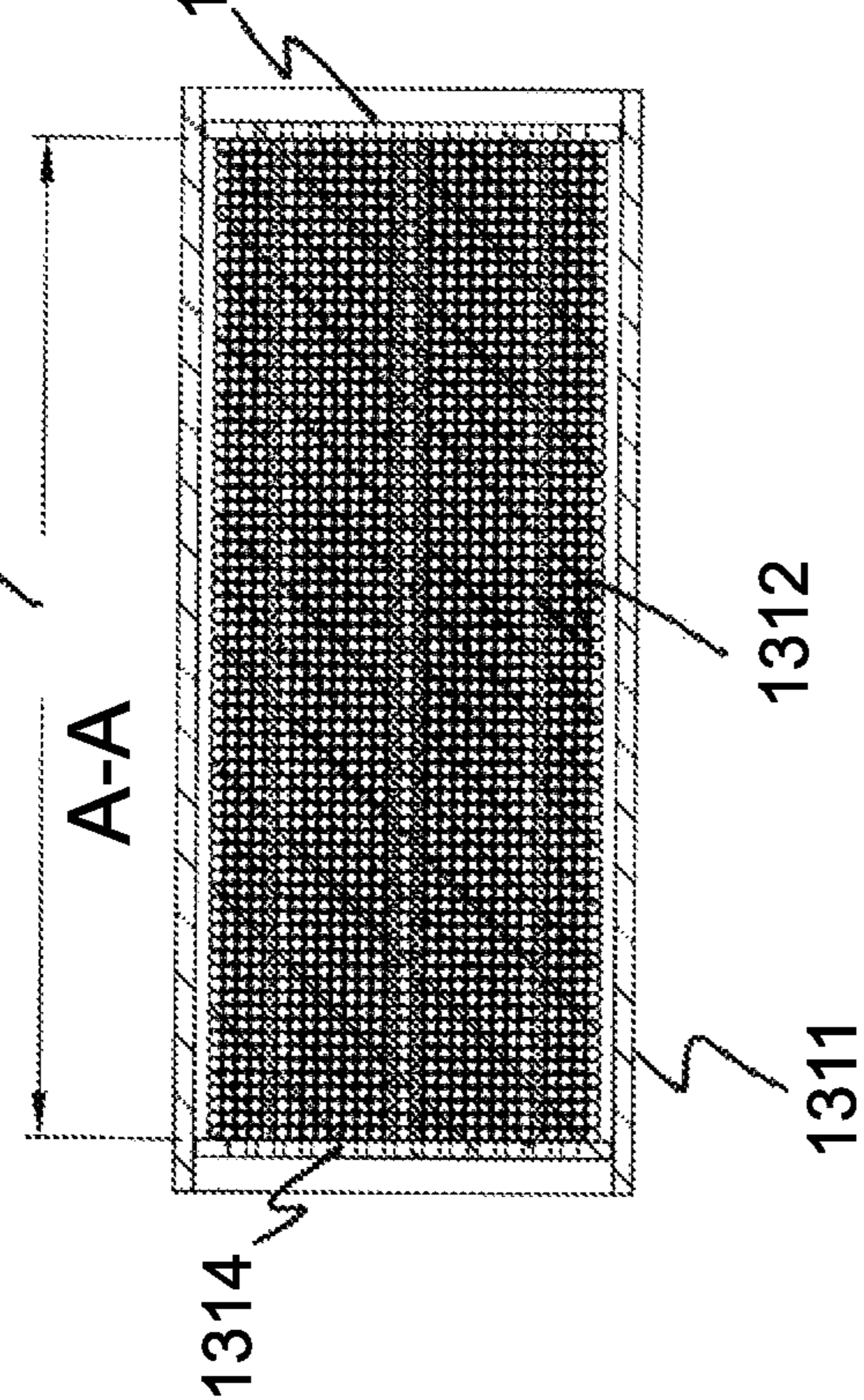


Fig. 13 b

Figure 13

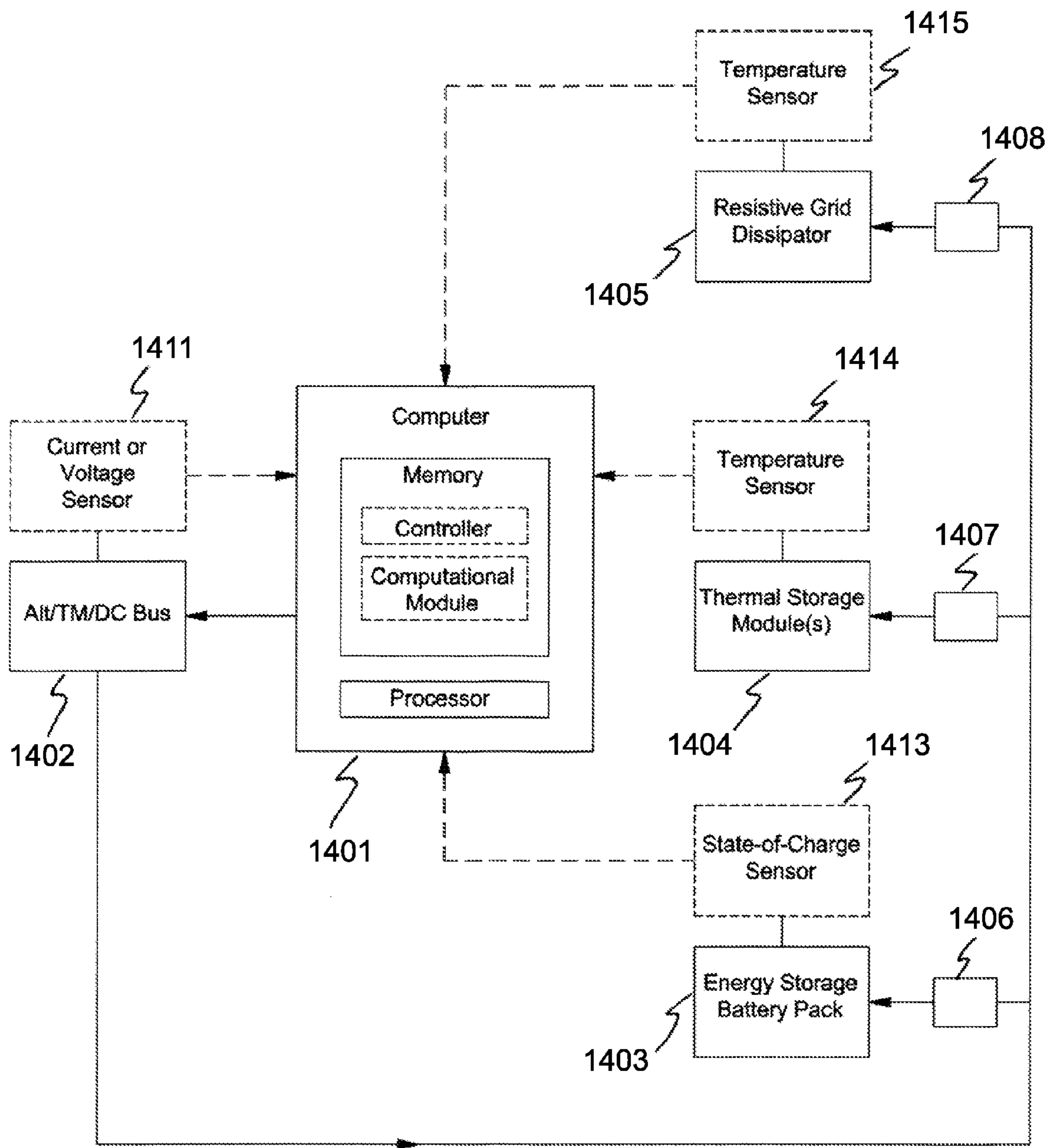


Figure 14

**1****GAS TURBINE ENERGY STORAGE AND  
CONVERSION SYSTEM****CROSS REFERENCE TO RELATED  
APPLICATION**

The present application claims the benefits, under 35 U.S.C. §119(e), of U.S. Provisional Application Ser. No. 61/177,493 entitled "Gas Turbine Energy Storage and Conversion System", filed May 12, 2009 and Provisional Application Ser. No. 61/327,988 entitled "Gas Turbine Energy Storage and Conversion System", filed Apr. 26, 2010, each of which is incorporated herein by this reference.

**FIELD**

The present invention relates generally to the field of regenerative braking and energy storage in gas turbine engines.

**BACKGROUND**

The world requires ever-increasing amounts of fuel for vehicle propulsion. Means of utilizing fuels needs to be accomplished more efficiently and with substantially lower carbon dioxide emissions and other air pollutants such as NO<sub>x</sub>s.

The gas turbine or Brayton cycle power plant has demonstrated many attractive features which make it a candidate for advanced vehicular propulsion. However, the gas turbine does not allow the normal "engine braking" or "compression braking" feature that is extensively used in piston-type engines. Further, many modern regenerative braking systems rely on batteries or other electrical storage subsystems to receive and absorb excess braking energy (others utilize pneumatic or hydraulic storage). In most cases, the cost of this energy storage is significant. Sizing a typical battery or ultra-capacitor energy storage system to absorb energy at high power associated with a long down-hill decent, for example, is prohibitively expensive.

Gas turbine engines have the additional advantage of being highly fuel flexible and fuel tolerant. For example, gas turbines can be operated on a variety of fuels such as diesel, gasoline, ethanol, methanol, natural gas, biofuels and hydrogen. The performance of gas turbine engines can be improved by making use of electrical energy recovered by a regenerative braking system. These improvements may include extending component lifetimes, pre-heating of fuels and providing an engine braking capability analogous to the Jacobs brake used by piston engines.

There remains a need for compact thermal energy storage devices to better enable gas turbine engines to recover energy from braking so as to improve both engine and braking performance of these engines applied to vehicular propulsion.

**SUMMARY**

These and other needs are addressed by the present invention. In one embodiment, the present invention is directed to a gas turbine engine that uses high temperature materials such as ceramic and/or metallic elements to store heat energy derived from a regenerative braking capability. The embodiment combines the principles of a gas turbine or Brayton cycle engine with an electric or hybrid transmission system. New techniques of thermal energy storage and thermal energy manipulation that can recover substantial amounts of energy normally discarded in braking are disclosed.

**2**

In one configuration of the embodiment, a method is provided that includes the steps of:

- (a) receiving electrical energy from a regenerative braking system;
- 5 (b) converting at least a portion of the received electrical energy into thermal energy;
- (c) transferring, directly and/or indirectly, the thermal energy to a pressurized working fluid to form a heated pressurized working fluid; and
- 10 (d) introducing the heated pressurized working fluid into at least one turbine to propel a vehicle.

In another configuration, a turbine power plant is provided that includes:

- (a) a source of compressed fluid;
- 15 (b) a turbine;
- (c) a mechanical linkage for extracting power from an output shaft of the turbine; and
- (d) a sensible thermal storage and/or thermal transfer medium contained within a pressure boundary of the turbine power plant, wherein the sensible thermal storage and/or thermal transfer medium transfers, by convection, thermal energy to the compressed fluid.

In another configuration, a turbine power plant is provided that includes:

- 25 (a) a turbine power plant;
- (b) a mechanical-to-electrical conversion device in mechanical communication with the turbine power plant to generate electrical energy from braking of the vehicle;
- (c) a direct current ("DC") bus in electrical communication with the mechanical-to-electrical conversion device to receive the electrical energy; and
- (d) at least one of a sensible thermal storage and/or thermal transfer medium

In another configuration, a vehicle is provided that includes:

- 35 (a) a mechanical-to-electrical conversion device in mechanical communication with an output shaft plant to generate electrical energy from braking of the vehicle;
- (b) a thermal energy storage medium to convert at least a portion of the electrical energy into thermal energy and store the thermal energy for use by a vehicle operation; an electrical energy storage system for storing at least a portion of the electrical energy; and
- (c) a controller to regulate, based on at least one of a state-of-charge of the electrical energy storage system and a temperature of the thermal energy storage and transfer medium, an amount of electrical energy sent to each of the thermal energy storage and transfer medium and electrical energy storage medium.

In another configuration, a vehicle is provided that includes:

- 55 (a) a conduit for transporting a gas-fuel mixture;
- (b) at least one of a combustor and re-heater for receiving and combusting the gas-fuel mixture to form a heated gas;
- (c) a turbine to receive the heated gas from the at least one of a combustor and re-heater;
- (d) an electrical storage system to store electrical energy;
- (e) at least one of a sensible thermal storage and/or thermal transfer medium contained within a pressure boundary of the turbine power plant to receive the electrical energy from the electrical energy storage system to heat at least one of the gas and gas-fuel mixture; and
- 65 (f) a controller operable to direct transfer of electrical energy from the electrical energy storage system to the at least one of a sensible thermal storage and/or thermal transfer medium.

The application of gas turbines to vehicular propulsion demands a wide range of power production from the engine. Further, improved driving economies are derived from recovering energy normally lost in braking. In some cases, associated with long down-hill descent, the engine can be configured to absorb considerable energy, so as to prevent the excessive load on other braking systems.

An electric drive-train uses electric traction motors to drive the wheels. During braking the flow of power reverses as the wheels drive the traction motors, thereby generating electricity derived from the energy of braking (referred to commonly as regenerative braking). This principal is prior art and may be incorporated into a vehicle with either an Otto cycle, Diesel cycle, Brayton (gas turbine) cycle, or any propulsion power plant. Embodiments of the subject invention use this regenerated electricity in a manner that provides economic advantages over normal battery or other electrical storage methods, which are typically charge-rate and capacity limited.

Embodiments of the present invention are referred to herein as thermal energy storage modules and incorporate one or more electric resistor/heat storage elements located within the engine's fluid conduit and pressure boundary to absorb over-flow braking energy from an electrical generator that is typically part of an electric or hybrid transmission. The electric resistor element converts electrical energy by resistive or Joule heating and delivers thermal energy to gas turbine's air flow during normal driving. These electric resistor elements may be located, for example, upstream of combustor, in the combustor, upstream of the free power turbine, upstream of the hot side of a recuperator, or any combination of these locations. Residual thermal energy remaining after braking or stopping may be used to assist combustor relight or ignition.

The following definitions are used herein:

The terms "at least one", "one or more", and "and/or" are open-ended expressions that are both conjunctive and disjunctive in operation. For example, each of the expressions "at least one of A, B and C", "at least one of A, B, or C", "one or more of A, B, and C", "one or more of A, B, or C" and "A, B, and/or C" means A alone, B alone, C alone, A and B together, A and C together, B and C together, or A, B and C together.

Dynamic braking is implemented when the electric propulsion motors are switched to generator mode during braking to augment the braking force. The electrical energy generated is typically dissipated in a resistance grid system. If the electrical energy generated is recaptured and stored in an electrical energy storage system, dynamic braking is then typically referred to as regenerative braking.

An energy storage system refers to any apparatus that acquires, stores and distributes mechanical, electrical or thermal energy which is produced from another energy source such as a prime energy source, a regenerative braking system, a third rail and a catenary and any external source of electrical energy. Examples are a heat block, a battery pack, a bank of capacitors, a compressed air storage system and a bank of flywheels or a combination of storage systems.

An engine refers to any device that uses energy to develop mechanical power, such as motion in some other machine. Examples are diesel engines, gas turbine engines, microturbines, Stirling engines and spark ignition engines.

A gear box as used herein is a housing that includes at least one gear set.

A gear set as used herein is a single ratio gear assembly.

A heat block is a solid volume of material with a high heat capacity and high melting temperature to which heat can be

added by electrical resistive or inductive heating and from which heat can be extracted by heat transfer to a fluid.

A hybrid vehicle combines an energy storage system, a prime power unit, and a vehicle propulsion system. A parallel hybrid vehicle is configured so that propulsive power can be provided by the prime power source only, the energy storage source only, or both. In a series hybrid vehicle, propulsive power is provided by the energy storage unit only and the prime power source is used to supply energy to the energy storage unit. When the energy storage capacity is small and the prime power source is large, the hybrid may be referred to as a power-assist hybrid. For example, an electric drive may be used primarily for starting and power assist while an internal combustion engine used primarily for propulsion. These vehicles are typically parallel hybrids. In a dual-mode hybrid, the energy storage and prime power are approximately balanced. For example, a dual-mode hybrid can operate on electric drive only, on engine power only, or on a combination of both. These vehicles are typically parallel hybrids.

Jake brake or Jacobs brake describes a particular brand of engine braking system. It is used generically to refer to engine brakes or compression release engine brakes in general, especially on large vehicles or heavy equipment. An engine brake is a braking system used primarily on semi-trucks or other large vehicles that modifies engine valve operation to use engine compression to slow the vehicle. They are also known as compression release engine brakes.

A mechanical-to-electrical energy conversion device refers to an apparatus that converts mechanical energy to electrical energy or electrical energy to mechanical energy. Examples include but are not limited to a synchronous alternator such as a wound rotor alternator or a permanent magnet machine, an asynchronous alternator such as an induction alternator, a DC generator, and a switched reluctance generator. A traction motor is a mechanical-to-electrical energy conversion device used primarily for propulsion.

Module as used herein in conjunction with a computer refers to any known or later developed hardware, software, firmware, artificial intelligence, fuzzy logic, or combination of hardware and software that is capable of performing the functionality associated with that element. Also, while the invention is described in terms of exemplary embodiments, it should be appreciated that individual aspects of the invention can be separately claimed.

A permanent magnet motor is a synchronous rotating electric machine where the stator is a three phase stator like that of an induction motor and the rotor has surface-mounted permanent magnets. In this respect, the permanent magnet synchronous motor is equivalent to an induction motor where the air gap magnetic field is produced by a permanent magnet. The use of a permanent magnet to generate a substantial air gap magnetic flux makes it possible to design highly efficient motors. For a common 3-phase permanent magnet synchronous motor, a standard 3-phase power stage is used. The power stage utilizes six power transistors with independent switching. The power transistors are switched in ways to allow the motor to generate power, to be free-wheeling or to act as a generator by controlling pulse frequency or pulse width.

A prime power source refers to any device that uses energy to develop mechanical or electrical power, such as motion in some other machine. Examples are diesel engines, gas turbine engines, microturbines, Stirling engines, spark ignition engines and fuel cells.



## 5

Power density as used herein is power per unit volume (watts per cubic meter).

A range-extended hybrid has a large energy storage capacity and a small prime power source. An example would be an electric drive vehicle with a small engine used for charging an electrical energy storage unit. These vehicles are typically series hybrids.

A recuperator is a heat exchanger that transfers heat through a network of tubes, a network of ducts or walls of a matrix wherein the flow on the hot side of the heat exchanger is typically exhaust gas and the flow on cold side of the heat exchanger is typically a gas such as air entering the combustion chamber. The flow of heat is from the hot side of the recuperator to the cold side of the recuperator.

Regenerative braking is the same as dynamic braking except the electrical energy generated is recaptured and stored in an energy storage system for future use.

Specific power as used herein is power per unit mass (watts per kilogram).

Spool means a group of turbo machinery components on a common shaft.

A thermal energy storage module is a device that includes either a metallic heat storage element or a ceramic heat storage element with embedded electrically conductive wires. A thermal energy storage module is similar to a heat storage block but is typically smaller in size and energy storage capacity.

A traction motor is a motor used primarily for propulsion such as commonly used in a locomotive. Examples are an AC or DC induction motor, a permanent magnet motor and a switched reluctance motor.

A turbine is any machine in which mechanical work is extracted from a moving fluid by expanding the fluid from a higher pressure to a lower pressure.

A vehicle is any device, apparatus or system for carrying, conveying, or otherwise transporting animate and/or inanimate objects, such as persons, including without limitation land conveyances (such as cars, trucks, buses, trains, and the like), maritime and other types of water vessels (such as ships, boats, and other watercraft), and aircraft.

## BRIEF DESCRIPTION OF THE DRAWINGS

FIG. 1 is a schematic of a simple cycle gas turbine with electrically heated resistor bank integrated within the engine conduit.

FIG. 2 is a schematic of an intercooled recuperated gas turbine with bypass around recuperator to increase thermal capacity of resistively heated thermal storage.

FIG. 3 is a schematic of an intercooled recuperated gas turbine cycle with reheat.

FIG. 4 is a schematic of an electric transmission suitable for regenerative braking with a gas turbine engine.

FIG. 5 is a schematic of an alternate electric transmission suitable for regenerative braking with a gas turbine engine.

FIG. 6a is a sectional view of an electrically heated thermal energy storage module representation.

FIG. 6b is a first sectional view of the energy storage module representation taken along line A-A of FIG. 6a.

FIG. 6c is a second sectional view of the energy storage module representation taken along line A-A of FIG. 6a.

FIG. 7 is a schematic of an electrically heated thermal storage module representation with surface combustion thermal reactor.

FIG. 8 is an isometric view of a first configuration of a thermal energy storage module.

## 6

FIG. 9a shows an end view of the thermal energy storage module of FIG. 8.

FIG. 9b is a sectional view along line A-A of FIG. 9a of the thermal energy storage module representation of FIG. 8.

FIG. 9c is an end view of the energy storage module representation of FIG. 8.

FIG. 10 is an isometric view of a second configuration of a thermal energy storage module.

FIG. 11a shows an end view of the thermal energy storage module of FIG. 10.

FIG. 11b is a sectional view along A-A of FIG. 11a of the thermal energy storage module representation of FIG. 10.

FIG. 11c is an end view of the energy storage module representation of FIG. 10.

FIG. 12 is an isometric view of a third configuration of a thermal energy storage module.

FIG. 13 shows an end view of the thermal energy storage module of FIG. 12.

FIG. 13b is a sectional view along line A-A of FIG. 13a of the thermal energy storage module representation of FIG. 13a.

FIG. 14 is a schematic of a regenerative braking energy distribution control system for a gas turbine with at least one thermal energy storage module.

## DETAILED DESCRIPTION

## Heat Energy Storage

Heat energy storage technology is known. For example, this technology has been proposed for non-nuclear submarines allowing them to operate for several hours to days while underwater. One of these technologies is the use of a graphite heat block as a heat source for a closed-cycle gas turbine power plant. In particular, the use of a graphite block heated to 2,750 K in an induction furnace to provide energy in place of a combustor has been disclosed. An inert gas flows through the block, picks up heat, spins the turbine and returns to complete the loop.

The energy storage possible with this technology is substantially higher than other forms of energy storage and, in particular, is compatible with gas turbines as a source of supplementary energy derived, for example, from regenerative braking.

Capacitors, inductors, some batteries and flywheels can release their energy at very high rates but typically at the expense of energy storage capacity. Graphite at high temperatures has a specific energy capacity comparable to chemical explosives and is a very compact form of energy storage compared to capacitors, inductors, flywheels and batteries commonly used in regenerative braking energy storage systems. Typical specific energy capacities associated with several energy storage technologies are shown in Table 1 below.

TABLE 1

Specific Energy Capacities of Some Storage Technologies	
Energy Storage Technology	Maximum Specific Energy Capacity (MJ/kg)
Capacitors	0.0004 to 0.001
Inductors - Room Temperature	0.001
Inductors - Cryogenic	0.003
Homopolar Generator (flywheel)	0.0085
Energy Storage Batteries	0.2
Kanthal Heat Storage at 1,700 K	0.4
Graphite Heat Block at 1,500 K	2.0
Graphite Heat Block at 2,000 K	3.0
Chemical Explosive (Octol)	4.8

## Some Thermal Properties of Heat Storage Materials

The properties of carbon and other materials such as ceramics make them useful for the collection and storage of thermal energy. These properties include: (1) a high heat capacity, especially at elevated temperatures; (2) a high melting point; and (3) a high thermal conductivity.

Preferred heat storage materials, for example, have a density of at least about 1,800 kg/m<sup>3</sup>, even more preferably of at least about 3,500 kg/m<sup>3</sup>, and even more preferably ranging from about 1,800 to about 7,500 kg/m<sup>3</sup>; and a heat capacity of at least about 400 J/kg-K, even more preferably of at least about 700 J/kg-K, and even more preferably ranging from about 400 to about 1,700 J/kg-K. The material should provide a high heat transfer efficiency. Preferably, the ratio of the thermal power transferred to the working fluid to the electrical power generated by regenerative braking is at least about 0.20 and even more preferably ranges from about 0.30 to about 0.70. This ratio is a function of working fluid flow velocity and density, surface area of the material, its thermal conductivity and its electrical resistivity. The material preferably has a thermal conductivity of at least about 5 W/m-K, more preferably of at least about 10 W/m-K, and even more preferably of at least about 20 W/m-K. Additionally, a preferred heat storage material also has a melting temperature in excess of the maximum temperature in the combustor (usually the combustor outlet temperature), even more preferably at least about 120% of the maximum temperature in the combustor, and even more preferably at least about 150% of the maximum temperature in the combustor.

A number of suitable materials, such as graphite, boron nitride, boron carbide, silicon carbide, silicon dioxide, magnesium oxide, tungsten carbide and alumina can be used for heat storage. Some important properties of ceramics and other high heat capacity materials that are typically used for gas turbine components are shown in Table 2 below.

TABLE 2

Some Properties of Heat Energy Storage Materials								
	Alumina	Cordierite	Silicon Carbide	Silicon Nitride	Graphite	Boron Nitride	Kanthal	Mullite
Density (kg/m <sup>3</sup> )	3,700-3,970	2,600	3,210	3,310	2,250	1,900	7,100	2,800
Specific Heat (J/kg-K)	670	1,465	628	712	712	1,610	460	963
Thermal Conductivity (W/m-K)	24	3	41	27	24	30	11	3.5
Coefficient Thermal Expansion (μm/m/K)	8.39	1.7	5.12	3.14	—	—	—	5.3
Thermal Shock Resistance ΔT (K)	200-250	500	350-500	750	—	—	—	300
Maximum Use Temperature (K)	2,060	1,640	1,670	1,770	2,270	2,100	1,670	2,000

High working temperature metals are required for electrical conduits embedded in ceramics and other non-conducting heat storage materials. Such electrical conductors may be formed from tungsten wire, for example.

Other high working temperature metals may be used both for electrical conduits and for heat storage. The material used is required to be a high temperature, oxidation resistant, electrically conductive alloy. Currently available candidate materials include the Kanthal alloys (specifically Kanthal A1, APM and Kanthal AF). It would also be possible to use one of the Inconel (nickel-chromium) alloys, but their high-temperature/oxidation resistance is lower than that of the Kanthal alloys. Due to lower cost per unit weight and possible lower thermal cycle degradation leading to longer component life, it appears Kanthal A1 is a preferable choice.

Additionally, when the heat storage material also converts electrical energy into thermal energy (for example, Kanthal or Inconel alloys), the material should have an appropriate electrical resistivity. Preferably, the electrical resistivity is at least about  $0.1 \times 10^{-6}$  ohm-meters, even more preferably at least about  $0.5 \times 10^{-6}$  ohm-meters.

The properties of Kanthal A1 (and Kanthal APM) are density of about 7,100 kg/cu m, heat capacity of about 460 J/kg-K, thermal conductivity of about 13 W/m-K. The practical maximum continuous operating temperature for Kanthal A1 (and Kanthal AF) is about 1,670 K. In operation, a Kanthal heat storage module of the present invention would be cycled between a low temperature of about 780 K (discharged) up to a fully charged temperature for short periods of about 1,700 K.

## Present Invention

The present invention integrates one or more electrical resistance heater elements into the gas turbine engine, receiving the regenerated electrical power and converting it to thermal storage and then to energy of the gas turbine's working fluid. The electrical heater is located within the engine's fluid conduit and pressure boundary, thereby eliminating the need for secondary transport fluids and facilitating the transport of the power through the engine's structural casing. FIG. 1 is a schematic representation of this basic principal for a single stage gas turbine engine. Compressor **102** pressurizes the engine working fluid, typically air or a lean air-fuel mixture. Conduit **105** is employed to transport the fluid to an electrically heated thermal storage module **106**, combustor **109** and turbine **104**. The conduits, such as conduit **105**, connecting the various components are denoted by solid lines. A free power turbine **108** is connected to a drive train **110** which includes gear assemblies, electrical generator/motors, drive shafts, differentials and axles. Drive train **110** transmits mechanical or electrical propulsion power while motoring

and generates electrical power while braking. Examples of these drive trains configured as electrical transmissions are described in FIGS. 4 and 5. In this embodiment, the electrical power generated within drive train **110** while braking is carried by conductor **111** (denoted by dashed lines) and passes through pressure boundary **131** (denoted by dot-dash lines) using a low resistance connector, well-known as an electrical feed-through. Within the high-pressure gas stream of the gas turbine, the electrical current causes Joule heating to occur in resistive element **106**. This resistive heating element **106** may be fabricated from metallic wire or ceramic materials in which conductive wires are embedded. The temperature of heating element **106** will rise as electrical energy is delivered and absorbed. During periods of power demand from the engine, the absorbed thermal energy is convected to the gas

turbine working fluid to offset energy that would otherwise be required from the fuel burned in the down-stream combustor **109**. In the example of FIG. 1, the electrical power is delivered to the resistively heated thermal storage module **106** from an electrical generator, located somewhere within drive train **110**, which derives its power from regenerative braking.

FIG. 2 shows a schematic of an intercooled recuperated Brayton cycle gas turbine. As in FIG. 1, conduits, such as conduit **219**, connecting the various components are denoted by solid lines, the electrical conductors, such as conductor **214**, are denoted by dashed lines and the pressure boundary **231** is denoted by dot-dash lines. This figure shows an intercooled, recuperated gas turbine engine which is comprised of a low pressure compressor **202**, an intercooler **205**, a high pressure compressor **206**, a recuperator **210**, a thermal storage module **212**, a combustor **215**, a high pressure turbine **208**, a low pressure turbine **204**, a free power turbine **216** which is connected to drive train **218** which includes gear assemblies, electrical generator/motors, drive shafts, differentials and axles. Drive train **110** transmits mechanical or electrical propulsion power while motoring and electrical power generation while braking. Examples of these drive trains configured as electrical transmissions are described in FIGS. 4 and 5. A regenerative braking system within drive train **218** delivers electrical to a heat energy storage module **212**. In typical operating mode, inlet air, which may be controlled by a valve such as **201**, is compressed by low pressure compressor **202**, then cooled at approximately constant pressure in intercooler **205**, compressed by high pressure compressor **206** to approximately maximum working pressure. The inlet air is heated by passing through recuperator **210** and heat storage module **212** and then heated to full working temperature by fuel energy added in combustor **215**. The hot, high pressure working fluid then expands in high pressure turbine **208** powering high pressure compressor **206** via mechanical coupling **207**, further expands in low pressure turbine **204** powering low pressure compressor **202** via mechanical coupling **203** and finally expanding in free power turbine **216** delivering mechanical shaft power to drive train **218**. The exhaust gases are then passed through the hot side of recuperator **210** giving up heat energy to the inlet air passing through the cool side of recuperator **210** before being vented to the atmosphere possibly by a valve **221**. Fuel is added to the air flow just upstream of or in combustor **215**. In certain types of ceramic matrix combustors, gaseous or vaporized fuels may be injected with the inlet air.

When the vehicle brakes, transmission **218** is disengaged and a mechanical to electrical conversion device within drive train **218** is engaged to generate electrical energy via conductors **213** where it is converted to heat energy by Joule heating within the resistive elements in thermal storage module **212**. A portion or all of the compressed inlet air heated by recuperator **210** can now be passed through thermal storage module **212** to gain further energy and temperature at approximately constant pressure before being delivered to combustor **215**. If the air entering combustor **215** is at the desired temperature or temperature set point for the combustor exit, no fuel need be added. If the injected air is at a lower temperature than the desired temperature or temperature set point for the combustor exit, an appropriate amount of fuel is added. As can be appreciated, when heat is added to the combustor inlet air via thermal storage module **212**, less fuel is required by combustor **215** than without the regenerative braking capability.

Depending on the duty cycle of the vehicle, the regenerative braking system described herein can have a modest or a large effect on the overall efficiency of the gas turbine. For

example, a delivery van or bus normally has a duty cycle with many stops and starts and so a regenerative braking system could substantially increase overall fuel efficiency. On the other hand, a long-haul Class 8 semi-trailer truck may have a duty cycle with few stops and starts. However, a regenerative braking system would provide some increase overall fuel efficiency by capturing energy from downhill travel or the occasional stop and go traffic conditions. Additionally, as discussed below, this system of regenerative braking can also assist the truck's normal braking system as serve the function of a Jacob's brake for a gas turbine engine.

As can be appreciated, when no energy is being added to the thermal energy storage module, the temperature of the thermal element will tend to follow the flow temperature and so may have an effect, for example, on combustor outlet temperature. A temperature sensor located just upstream of the combustor can be used to affect small adjustments in fuel-air ration to compensate for this effect. It should also be noted that a battery or other electrical energy storage device may be used to heat the thermal storage element to assist in engine start-up. That is, a thermal energy storage element, located for example just upstream of the combustor, can be used to add heat to the working fluid to assist an engine starter device for a gas turbine engine used in a vehicle.

As explained in FIG. 1, an electrically heated resistor bank thermal storage module **212** is integrated into the engine circuit upstream of combustor **215**, configured to receive regenerated electricity and pre-heat gas on route to combustor **215**. A recuperator **210** significantly improves the engine conversion efficiency, relative to the simple gas turbine cycle shown in FIG. 1, by recovering thermal energy from the free power (last stage) turbine duct to pre-heat the combustion gas. When employing a recuperator, the thermodynamic availability of energy from the electrically heated resistor and thermal storage module is reduced in proportion to the increased gas inlet temperature. To increase the energy absorbing capacity of the thermal storage module **212** during extended periods of regenerative braking (such as for example, a long decent), a simple by-pass duct **211** controlled by a solenoid valve **220** may be activated to introduce cool air over the resistor elements. This recuperator bypass allows for increased power dissipation from the thermal storage module by rapidly dropping combustor inlet temp. Although less preferred, a thermal storage module may be located in by-pass duct **211**.

It should be obvious to one skilled in the art of gas turbine architecture that the subject invention applies to gas turbines with and without intercooling, single shaft mechanical configurations, free-power turbine configurations, and varying numbers of compressor and turbine stages.

A further embodiment of the integrated resistance-heated thermal storage system is shown in FIG. 3. As in FIG. 1, conduits, such as conduit **319**, connecting the various components are denoted by solid lines, the electrical conductors, such as conductor **314**, are denoted by dashed lines and the pressure boundary **331** is denoted by dot-dash lines. This Brayton cycle gas turbine is essentially the same as that of FIG. 2, except that the thermal storage element **312** is located between low pressure turbine **304** and free power turbine **316**. By locating thermal storage element **312** as a re-heater between turbine stages, it is possible to derive thermodynamic benefits which improve overall efficiency and specific power (power/mass flow rate).

As can be appreciated, two electrically-heated thermal storage modules can be utilized in the gas turbine cycle. As an example of this configuration, one electrically-heated thermal storage module can be located upstream of the combustor (such as in FIG. 2) and a second between the low pressure and

free power turbines (such as in FIG. 3). It should be obvious to one skilled in the art of gas turbine design that the principal embodied herein may be extended to include multiple electrically heated thermal storage modules, each re-heating the engine's working fluid prior to entering each of a multiplicity of turbine stages. For example, electrically-heated thermal storage modules can be located in the bypass duct (duct 211 in FIG. 2) or even the recuperator (item 210 in FIG. 2) hot side inlet manifolds. These last two locations would not require any growth in size of the engines.

#### Compatible Transmissions

In a gas turbine engine in the power range of up to about 700 kW, the free power turbine typically rotates in the range from about 70,000 to about 120,000 rpm. The transmission must couple the output shaft of the free power turbine to the wheels of the vehicle which rotate in the range from about zero to about 500 rpm. This is preferably accomplished by one of a number of possible electric transmissions, although a purely mechanical transmission is feasible. However, an electric transmission offers the possibility of recovering some of the energy of braking by regenerative braking methods.

FIG. 4 is a schematic of a possible electric transmission suitable for regenerative braking with a gas turbine engine. A free power turbine 401 is shown with its output shaft connected to a reduction gearset 402 which might have a reducing gear ratio in the range of about 6:1 to about 10:1. In this example, gearset 402 is connected to traction motor 403 which can transmit mechanical shaft power via a clutch assembly 404 to a second gearset 405. Gearset 405 reduces the rpms of the transmission and might have a reducing gear ratio in the range of about 4:1 to about 10:1. Gearset 405 is connected to drive shaft 406, which is turn connected to differentials 407 which drive wheels 408. Traction motor 403 is electrically connected to a DC bus 412 by inverter 411. Vehicle auxiliary power 415, an electrical energy storage system 413 and a resistive heating element 414 are shown connected to DC bus 412. The electrical energy storage system 413 may be a battery pack, a capacitor bank or a flywheel apparatus, for example. The resistive heating element 414 may be a dissipative resistive grid (in which heat energy is removed by convection and discarded to ambient air) or a resistance-heated thermal storage system (in which heat energy is utilized such as described in FIGS. 1 through 3).

In motoring mode, electrical energy from electrical energy storage system 413 may be used to provide some or all of the propulsive power for the vehicle via traction motor 403. In braking mode, traction motor 403 becomes an electrical generator and can charge the energy storage system 413 or be dissipated in resistive dissipative grid 414 or both. For example, during braking, electrical energy derived from regenerative braking could be first directed to charge a battery pack. Once the battery pack is fully charged, electrical energy may be re-directed to a resistance-heated thermal storage system such as described in FIGS. 6 through 12. If additional dynamic braking is required and the battery pack is fully charged and the resistance-heated thermal storage system is at peak temperature, then additional electrical energy may be re-directed to the dissipative resistive grid in which heat energy is removed by convection and discarded to ambient air. Clutch assembly 404 allows the rotor of the traction motor to be disengaged during high speed motoring to reduce windage losses while engaging a separate shaft that continues to transmit mechanical power through the traction motor.

A traction motor is a mechanical-to-electrical energy conversion device used primarily for propulsion. Examples of traction motors include but are not limited to AC or DC

induction motors, permanent magnet machines and a switched reluctance generators.

FIG. 5 is a schematic of an alternate electric transmission suitable for regenerative braking with a gas turbine engine. This configuration is similar to that of FIG. 4 except there is a high speed alternator and a traction motor which can be operated electrically or mechanically depending on vehicle speed. A free power turbine 501 is shown with its output shaft connected to a reduction gearset 502 which might have a reducing gear ratio in the range of about 6:1 to about 10:1. In this example, gearset 502 is connected to alternator 503 which can output mechanical shaft power to a clutch assembly 504. When engaged, clutch assembly 504 connects alternator 503 to traction motor 505. Traction motor 505 can output mechanical shaft power via a clutch assembly 506 to a second gearset 507. Gearset 507 reduces the rpms of the transmission and might have a reducing gear ratio in the range of about 4:1 to about 10:1. Gearset 507 is connected to drive shaft 508, which is turn connected to differentials 509 which drive wheels 510. Alternator 503 and traction motor 505 are both electrically connected to a DC bus 513 by their respective inverters 511 and 512. Vehicle auxiliary power 523, an electrical energy storage system 521 and a resistive heating element 522 are shown connected to DC bus 513. The electrical energy storage system 521 may be a battery pack, a capacitor bank or a flywheel apparatus, for example. The resistive heating element 522 may be a dissipative resistive grid (in which heat energy is removed by convection and discarded to ambient air) or a resistance-heated thermal storage system (in which heat energy is utilized such as described in FIGS. 6 through 12).

In low speed motoring mode with clutch assembly 504 disengaged, electrical energy from alternator 503 and/or electrical energy storage system 521 may be used to provide propulsive power electrically for the vehicle via traction motor 505. In high speed motoring mode with clutch assembly 504 engaged, propulsive power may be provided mechanically via the shafts of alternator 503 and traction motor 505 which are locked together. Clutch assemblies 504 and 506 also allow the rotors of alternator 503 and traction motor 505 to be disengaged during high speed motoring to reduce windage losses while engaging a separate shaft that continues to transmit mechanical power through the alternator and traction motor which are locked together mechanically. The efficiency of the transmission in high speed motoring mode is typically higher (about 96% to about 98%) than the efficiency of the transmission in low speed motoring mode (about 92% to about 96%). High speed motoring mode is typically utilized for long distance driving where a higher transmission efficiency has its maximum efficiency benefit.

In braking mode with clutch assembly 504 may remain disengaged while clutch assembly 506 re-engages the rotor of traction motor 505. Traction motor 505 becomes an electrical generator and can charge the energy storage system 521 or be dissipated in resistive dissipative grid 522 or both. For example, during braking, electrical energy derived from regenerative braking could be directed first to charge a battery pack. Once the battery pack is fully charged, electrical energy may be re-directed to a resistance-heated thermal storage system such as described in FIGS. 6 through 12. If additional dynamic braking is required and the battery pack is fully charged and the resistance-heated thermal storage system is at peak temperature, then additional electrical energy may be re-directed to the dissipative resistive grid in which heat energy is removed by convection and discarded to ambient air. As can be appreciated, during braking, clutch assembly 504 may also be re-engaged to allow the rotor of alternator

**503** to allow it become an electrical generator and can charge the energy storage system **521** or be dissipated in resistive dissipative grid **522** or both.

An alternator is a mechanical-to-electrical energy conversion device. Examples include but are not limited to a synchronous alternator such as a wound rotor alternator or a permanent magnet machine, an asynchronous alternator such as an induction alternator, a DC generator, and a switched reluctance generator.

The drive trains shown in FIGS. **4** and **5** are known. These drive trains are examples of electric or hybrid transmissions which may be used in gas turbine engines to provide electrical power during motoring and braking and therefore provide the gas turbine engine a dynamic braking capability. The electrical energy generated during dynamic braking may be dissipated or it may be used to return heat energy to the engine as described in FIGS. **1** through **3** and FIGS. **6** through **12**.

Various Embodiments of Thermal Energy Storage Modules

The principle of the electrically heated thermal storage module is described in FIG. **6**. In FIG. **6a**, conduit **601** confines the pressurized working fluid to flow between two engine components such as in FIG. **1**, **2** or **3**. Electrode feed-through **603** permits the electrical connection to be made through the pressure boundary and communicate with the resistively heated elements **602**. An electrical ground **604** is required to complete the electrical circuit through the heating elements. As shown in the cross-section of FIG. **6b**, the resistive elements within conduit **611** may be a wire grid **612**. Alternately, as shown in the cross-section of FIG. **6c**, the resistive elements within conduit **612** may be a wire or ceramic matrix **622**. If a ceramic matrix is used, resistive conductors such as Kanthal or tungsten would be embedded in the ceramic elements. As shown in FIG. **6a**, working fluid **605** enters the conduit at end **606** and exits the conduit at end **607**. This working fluid is commonly air, but may be another gas, such as helium, nitrogen, argon, or xenon, or other gas or gas combination employed in open or closed cycle gas turbines (for example, a fuel-air mixture). The resistor wire or matrix elements are positioned within the gas turbine conduit to achieve high convective heat exchange between the fluid and the heating element while leaving sufficient flow cross-sectional area to maintain a selected pressure drop through the thermal energy storage module.

As shown in FIG. **7**, another configuration for utilizing an electrically activated thermal storage module combines the principals described above with a combustion system. In a typical application wherein the gas turbine working fluid is air and a compressor **712** delivers air to a combustor assembly **701**. Commonly a recuperator **715** may also be employed as an energy savings device, but un-recuperated variations are equally feasible. In the example of FIG. **7**, combustor vessel **701** contains an electrically heated thermal storage module **703**, arranged as described in FIG. **6** to serve as an effective heat exchanger, and a combustor unit **704**. The combustor **704** may be a conventional metallic combustor or it may be a ceramic matrix combustor. In the configuration illustrated in FIG. **7**, gaseous or vaporized fuel is introduced from conduit **716**, preferably upstream of the combustor assembly **701**. When thermal storage module **703** is not operating, the pre-mixed fuel and air passing through the combustor **704** is reacted. If combustor **704** is a ceramic matrix combustor, the pre-mixed fuel and air passing through matrix will react on the high temperature surfaces, releasing the heat of combustion. This ceramic matrix reactor has certain advantages in a gas turbine combustor, providing very low pressure drop, low levels of NOx emissions, an a homogenous temperature distribution to the flow entering the turbine section down-stream.

The use of an electrically heated thermal storage system provides a convenient means of controlling the conditions of the reaction of a lean fuel-air in a ceramic matrix combustor.

It was noted previously that a battery or other electrical energy storage device may be used to heat the thermal storage element to assist in engine start-up. That is, a thermal energy storage element, located for example just upstream of the combustor, can be used to add heat to the working fluid to assist an engine starter device for a gas turbine engine used in a vehicle. Such a starter device has been disclosed in U.S. patent application Ser. No. 12/115,134 entitled "Multi-Spool Intercooled Recuperated Gas Turbine".

Finally, it should be obvious to one skilled in the art of gas turbine design that the aforementioned invention would function equally well as an inter-turbine re-heater, as illustrated, for example, in FIG. **3**.

FIG. **8** is an isometric view with a cutaway section of a first configuration of a thermal energy storage module. The module casing **801** contains a metallic heating element **802**. Electrical energy flows in via conductor **804**, through each Kanthal spiral and out via conductor **805**. The flow direction of the gas turbine working gas is indicated by arrow **803**. Heating element **803** can be made of a material such as, for example, Kanthal A1 which is a material commonly used in automobile cigarette lighters. The module shown in FIG. **8** is about 0.33 meters in diameter with a cylindrical section about 0.4 meters long. The Kanthal heat storage element is formed by 18 spirals joined together, each about 10 mm wide by about 1.0 mm thick by about 17 meters long for a total length of Kanthal of about 306 meters. The Kanthal spirals are all connected in series to form a single long resistive element. The connections are shown as alternately at the center of each spiral then at the outside of adjacent spirals. At maximum working temperature of about 1,670 K to about 1,700 K, the storage module which weighs about 15 kg can store about 5 to 6 MJ in the form of useable heat energy. The spirals are separated by an air gap of about 3 to 10 mm.

FIGS. **9a-c** show various views of the thermal energy storage module of FIG. **8**. This thermal energy storage module is designed for a gas turbine engine with an approximate peak power of 375 kW. The 15 kg Kanthal thermal strip, configured as a series of spiral windings, is housed in an approximately 0.334 meter diameter steel housing **901** with a wall thickness in the range of about 9.5 to about 11 mm. The cylindrical portion of the housing **911** is about 0.395 meters long and tapers down from about a 0.334 meter diameter at about 30 degrees to about a 0.12 meter diameter **922**.

FIG. **10** is an isometric view with a cutaway section of a second configuration of a thermal energy storage module. The module casing **1001** contains a heating element **1003**. Heating element spirals **1003** contained within housing **1001** are interspersed with ceramic honeycomb discs **1002** to increase the thermal mass of the module while reducing the overall module mass. Ceramic honeycomb discs **1002** may be made of alumina or silicon carbide, for example. Electrical energy flows in via conductor **1005**, through each Kanthal spiral and out via conductor **1006**. The flow direction of the gas turbine working gas is indicated by arrow **1004**. For example, the heating element **1003** is made of a material such as Kanthal A1 which is a material commonly used in automobile cigarette lighters. The module shown in FIG. **10** is about 0.33 meters in diameter with a cylindrical section about 0.4 meters long. The Kanthal heat storage element is formed by a number of spirals joined together, each about 10 mm wide by about 1.0 mm thick by about 17 meters long for a total length of Kanthal of about 150 meters. As in FIG. **8**, the Kanthal spirals are all connected in series to form a single long

resistive element. The connections are shown as alternately at the center of each spiral then at the outside of adjacent spirals. At maximum working temperature of about 1,670 K to about 1,700 K, the storage module which weighs about 15 kg can store about 5 to 6 MJ in the form of useable heat energy. The metallic spirals are separated by ceramic layers which range from about 5 mm to about 30 mm wide. As can be appreciated, the ratio of metallic strip width to ceramic strip width can be varied to change the ratio of active heating element storage capacity to passive thermal storage capacity and to adjust the overall weight of the thermal storage module.

FIGS. 11a-c show various views of the thermal energy storage module of FIG. 10. This thermal energy storage module is designed for a gas turbine engine with an approximate peak power of 375 kW. A 7 kg Kanthal thermal strip is housed in a 0.334 meter diameter steel housing 1111 with a wall thickness in the range of about 9.5 to about 11 mm. The cylindrical portion of the housing 1111 is about 0.395 meters long and tapers down from about a 0.334 meter diameter at about 30 degrees to about a 0.12 meter diameter 1122.

FIG. 12 is an isometric view with a cutaway section of a third configuration of a thermal energy storage module. This thermal energy storage module is also designed for a gas turbine engine with an approximate peak power rating of about 375 kW. In this configuration, the thermal storage is formed by a porous metal pebble bed in a metallic pellet configuration. Higher resistance sintered Kanthal metal pellets 1202 are held between lower resistance bonded metal conductor end caps 1203 at both ends, all contained in cylindrical housing 1201. Metal conductor end caps 1203 are preferably made from lower resistance Kanthal. The void fraction (air cross-section to solid material cross-section) is adjusted for a selected, tolerable low pressure drop and compact size.

FIGS. 13a-b show various views of the thermal energy storage module of FIG. 12. As shown in FIG. 13a, the diameter of housing 1301 is about 0.1684 meters in diameter with the sintered Kanthal bed contained inside. As shown in FIG. 13b, the cylindrical housing 1311 is about 0.368 meters long with electrically positive end cap 1314 and electrically negative (ground) end cap 1315.

#### Sizing of Electrically Heated Thermal Storage Module

As an example, electrical energy recovered from braking for a 375 kW gas turbine engine is assumed to be recovered at a rate of about 200 kWe from the braking electrical generator system. If this system is operated for about 30 seconds, the total energy recovered is about 6 MJ, which is typical of a short descent down a modest hill. This performance is typical of a thermal energy storage system of about 15 kg of Kanthal coiled into a 1 mm thick by 10 mm wide by 306 meter long spiral structure located between the recuperator and combustor as shown in FIG. 2. Such a structure is shown, for example, in FIG. 8.

The maximum energy stored is the mass of the thermal storage element times its average heat capacity times the temperature change and is typically about 6 MJ for the size of engine assumed. The total energy input to the thermal storage element may be larger than 6 MJ as the working fluid flow through the thermal storage element simultaneously removes heat during the charging cycle. For materials such as Kanthal, the maximum useful energy storage capacity is about 0.3 to about 0.4 MJ/kg. Thus a 15 kg Kanthal heat storage module will store about 6 MJ. More energy than this (approximately 20% to about 50% depending on the design of the thermal storage element) can be received from the braking system as some energy is simultaneously transferred out of the thermal

storage element by convection to the air flow through the thermal storage module during charging.

If the mass of the element is 15 kg and the density of the thermal storage element material is about 7,100 kg per cubic meter, the approximate volume of Kanthal is about 0.002 cubic meters. The fill factor is about 15% to about 30% Kanthal with the remainder being volume available for the flow of working fluid. Thus, the overall volume of a thermal storage module is on the order of 0.01 cubic meters. The reduction in flow volume by the thermal storage element causes a pressure drop. It is desired to minimize this pressure drop while maximizing the convective heat transfer surface area. The design problems related to these considerations are discussed in FIG. 14.

At full engine power, the initial flow conditions over the Kanthal are typically about 1.2 kg/s at about 8 m/s and flow density of about 6.5 kg/m<sup>3</sup> (corresponding to inlet conditions of about 1,450 kPa and about 775 K).

At engine idle power of about 25% full power, the initial flow conditions over the Kanthal are typically about 0.3 kg/s at about 8 m/s and flow density of about 1.6 kg/m<sup>3</sup> (corresponding to inlet conditions of about 360 kPa and about 775 K).

The convection heat transfer coefficient for the "cigarette lighter" design as shown in FIG. 8, is that characteristic of a flat plate since the flow length through the channels is too short for the flow to develop and transition to fully turbulent. The flow over the Kanthal can be approximated as initially laminar flow transitioning to turbulent flow near the end of the flat plate where the length of the flat plate assumed in the analysis is approximately the width of the Kanthal strip or about 10 mm. The average convective heat transfer coefficient ranges from about 200 to about 400 W/sq m-K over operating flow conditions. In this example, the active convective heat transfer area is about 6 square meters (10 mm wide strips, about 306 meters long, both sides). The energy transfer out of the heat storage module to the flow occurs at about 75 kW or about 35% of the electrical power input to the thermal storage element.

In a calculation typical of a 375 kW gas turbine engine at idle power during braking and charging of the thermal storage module, flow temperature and pressure at the entry to the thermal storage module are about 360 kPa and about 775 K and the thermal storage element is at about 775 K. After about 36 seconds of 200 kWe braking, the temperature of the Kanthal from front to back of the thermal storage module is approximately constant at about 1,000 K while the air flow temperature ranges from about 780 K at the entry to about 920 K at the exit. After about 60 seconds of 200 kWe braking, the temperature of the Kanthal from front to back of the thermal storage module ranges from about 1,500 K at the entry to about 1,750 K at the exit while the corresponding air flow temperatures range from about 780 K to about 1,310 K.

After discharging of the thermal storage module for about 28 additional seconds (or about 88 seconds after the start of braking), the temperature of the Kanthal from front to back of the thermal storage module ranges from about 1,045 K at the entry to about 1,370 K at the exit while the corresponding air flow temperatures range from about 780 K to about 1,060 K. From these calculations, it can be seen that the thermal storage module raises inlet temperature to the combustor to nearly the desired combustor outlet temperature so that very little fuel is required to achieve the desired combustor outlet temperature, thus resulting in a reduction of fuel consumption.

## Control of Braking Energy

FIG. 14 is a schematic of a regenerative braking energy distribution control system for a gas turbine powered vehicle with at least one thermal energy storage module. This figure shows a computer 1401 in communication with electrical components 1402 of a drive train such as shown in FIGS. 4 and 5. Computer 1401 is also in communication with one or more energy storage systems 1403, one or more thermal storage modules 1404 and a resistive grid 1405 that can dissipate its thermal energy to ambient air. Examples of electrical energy storage systems 1403 are battery packs, capacitor banks or flywheels. Thermal storage modules 1404 are those shown, for example, in FIGS. 8 through 13 which convert electrical energy to thermal energy by Joule heating within the pressure boundary of a gas turbine engine. An example of a resistive grid 1405 is the roof-mounted resistive braking grid typically used for dynamic braking of locomotives.

As shown in FIG. 14, choppers 1406, 1407 and 1408 are included and may be used as on/off switches and current regulators to the energy storage pack 1403, thermal storage module(s) 1404 and resistive dissipater grid 1405 respectively. These choppers may be controlled by the controller in computer 1401. This is a known method of control for energy storage packs and resistive dissipater grids.

In the design of a thermal storage module, it is desired to prescribe both the volume and surface area of the resistive storage element. The volume must be adjusted to control the pressure drop of the flow through the element while the surface area must be adjusted to achieve the desired heat transfer rate from the element to the flow. For a given material, this design process cannot also control the electrical resistance of the storage element. Therefore, a chopper 1407 can be used to regulate the desired current flow to the storage element in module 1404 consistent with the resistance of the storage element and the voltage of the DC bus 1402. The chopper may be pulse width modulated or pulse frequency modulated. Alternatively or additionally, the voltage of the DC bus may be altered to regulate the desired current flow to the storage element.

When the vehicle is braking, the electric components 1402 of the drive train generate power to a DC bus such as described in FIGS. 4 and 5. A voltage and/or current sensor 1411 on the DC bus is monitored by computer 1401 and a controller in computer 1401 determines where to distribute the electrical power generated by dynamic braking. If energy storage system 1403 is a battery pack or capacitor bank, its state-of-charge is determined typically by a voltage sensor 1412 which is monitored by computer 1401. Any suitable voltage or current sensor may be employed. Examples of voltage sensors include voltmeters, other common voltage transducers or voltage sensing devices. Examples of current sensors include ammeters, current-sensing resistors, Hall current sensors, current-sensing transformers, current transducers, Rogowski coils or other common current measuring devices.

If the energy storage system can accept additional charge and if power is not needed by thermal energy storage module 1404, then the controller in computer 1401 directs electrical power to energy storage system 1403 until state-of-charge sensor 1412 communicates to computer 1401 that the state-of-charge of the battery pack or capacitor bank is at its selected operating (e.g., maximum) voltage. If energy storage system 1403 is a flywheel, then sensor 1413 may be an rpm indicator and addition of energy to the flywheel would be terminated by the controller of computer 1401 when rpm sensor 1412 communicates to computer 1401 that the rpm of

the flywheel is at its selected operating rpm. Examples of rotary speed sensors include tachometers such as axle alternators or reluctance pickups.

If braking power is still being generated and if temperature sensor 1413 indicates that the temperature of thermal storage module 1404 is below its operating (e.g., maximum acceptable) temperature, then the controller in computer 1401 directs electrical power to thermal storage module 1404 until temperature sensor 1413 communicates to computer 1401 that the temperature of the thermal energy storage module is at its selected operating temperature.

If braking power is still being generated then the controller in computer 1401 directs electrical power to the resistive grid dissipater 1405 until temperature sensor 1414 communicates to computer 1401 that the temperature is at its selected operating (e.g., maximum) temperature. Once energy storage system 1403 has reached its operating state-of-charge and once thermal storage modules 1404 have reached their operating temperatures and once resistive grid dissipater 1405 has reached its operating (e.g., maximum) temperature, then dynamic braking is terminated and further braking is accomplished by the vehicles mechanical braking system.

If the vehicle is not braking and if power is needed by thermal energy storage module 1404, for example for assisting in engine start-up, then the controller in computer 1401 may direct electrical power electrical power to thermal storage module 1404 from energy storage system 1403 until temperature sensor 1413 communicates to computer 1401 that the temperature of the thermal energy storage module is at its selected operating temperature.

A number of variations and modifications of the inventions can be used. As will be appreciated, it would be possible to provide for some features of the inventions without providing others.

The present invention, in various embodiments, includes components, methods, processes, systems and/or apparatus substantially as depicted and described herein, including various embodiments, sub-combinations, and subsets thereof. Those of skill in the art will understand how to make and use the present invention after understanding the present disclosure. The present invention, in various embodiments, includes providing devices and processes in the absence of items not depicted and/or described herein or in various embodiments hereof, including in the absence of such items as may have been used in previous devices or processes, for example for improving performance, achieving ease and/or reducing cost of implementation.

The foregoing discussion of the invention has been presented for purposes of illustration and description. The foregoing is not intended to limit the invention to the form or forms disclosed herein. In the foregoing Detailed Description for example, various features of the invention are grouped together in one or more embodiments for the purpose of streamlining the disclosure. This method of disclosure is not to be interpreted as reflecting an intention that the claimed invention requires more features than are expressly recited in each claim. Rather, as the following claims reflect, inventive aspects lie in less than all features of a single foregoing disclosed embodiment. Thus, the following claims are hereby incorporated into this Detailed Description, with each claim standing on its own as a separate preferred embodiment of the invention.

Moreover though the description of the invention has included description of one or more embodiments and certain variations and modifications, other variations and modifications are within the scope of the invention, e.g., as may be within the skill and knowledge of those in the art, after under-

standing the present disclosure. It is intended to obtain rights which include alternative embodiments to the extent permitted, including alternate, interchangeable and/or equivalent structures, functions, ranges or steps to those claimed, whether or not such alternate, interchangeable and/or equivalent structures, functions, ranges or steps are disclosed herein, and without intending to publicly dedicate any patentable subject matter.

What is claimed is:

1. A method, comprising:
  - receiving electrical energy from a regenerative braking system;
  - converting at least a portion of the received electrical energy into thermal energy;
  - transferring, directly and/or indirectly, the thermal energy to a pressurized working fluid to form a heated pressurized working fluid;
  - introducing the heated pressurized working fluid into at least one turbine to propel a vehicle; and
  - wherein the converting step is performed within a pressure boundary of a gas turbine engine.
2. The method of claim 1, wherein the regenerative braking system comprises a mechanical-to-electrical conversion device, at least one of which is a synchronous or asynchronous alternator, a generator, a permanent magnet machine, a direct current ("DC") generator, a switched reluctance machine and a traction motor, and a DC bus and further comprising:
  - when the vehicle brakes, engaging the mechanical-to-electrical conversion device to generate the electrical energy; and
  - storing the thermal energy in a heat storage element in thermal communication with the pressurized working fluid.
3. The method of claim 1, wherein the transferring step is performed between a cold side of a recuperator and a combustor.
4. The method of claim 1, wherein the transferring step is performed in a combustor.
5. The method of claim 1, wherein the transferring step is performed between a combustor and a high pressure turbine.
6. The method of claim 1, wherein the transferring step is performed between a high pressure turbine and a low pressure turbine.
7. The method of claim 1, wherein the transferring step is performed between a low pressure turbine and a free power turbine.
8. The method of claim 1, wherein the transferring step is performed upstream of a hot side of a recuperator.
9. The method of claim 2, wherein the heat storage element has the following characteristics:
  - a density of at least about 1,800 kg/m<sup>3</sup>;
  - a heat capacity of at least about 400 J/kg-K; and
  - a melting temperature in excess of a maximum temperature in the combustor.
10. The method of claim 9, wherein the heat storage element is at least one of graphite, boron nitride, boron carbide, silicon carbide, silicon dioxide, magnesium oxide, tungsten carbide, alumina, a Kanthal alloy, and an Inconel alloy.
11. The method of claim 1, wherein, during an extended period of regenerative braking, a first portion of the pressur-

ized working fluid is passed through a cold side of a recuperator and a second portion of the pressurized working fluid bypasses the recuperator.

12. A method, comprising:
  - receiving electrical energy from a regenerative braking system;
  - converting at least a portion of the received electrical energy into thermal energy;
  - transferring, directly and/or indirectly, the thermal energy to a pressurized working fluid to form a heated pressurized working fluid, wherein said transferring is performed between a cold side of a recuperator and a combustor; and
  - introducing the heated pressurized working fluid into at least one turbine to propel a vehicle.
13. A method, comprising:
  - receiving electrical energy from a regenerative braking system;
  - converting at least a portion of the received electrical energy into thermal energy;
  - transferring, directly and/or indirectly, the thermal energy to a pressurized working fluid to form a heated pressurized working fluid;
  - introducing the heated pressurized working fluid into at least one turbine to propel a vehicle;
  - wherein the regenerative braking system comprises a mechanical-to-electrical conversion device, which is at least one of a synchronous or asynchronous alternator, a generator, a permanent magnet machine, a direct current ("DC") generator, a switched reluctance machine, and a traction motor, and a DC bus and further comprising:
    - when the vehicle brakes, engaging the mechanical-to-electrical conversion device to generate the electrical energy; and
    - storing the thermal energy in a heat storage element in thermal communication with the pressurized working fluid; and
    - wherein the heat storage element has the following characteristics:
      - a density of at least about 1,800 kg/m<sup>3</sup>,
      - a heat capacity of at least about 400 J/kg-K, and
      - a melting temperature in excess of a maximum temperature in the combustor.
14. The method of claim 13, wherein the heat storage element is at least one of graphite, boron nitride, boron carbide, silicon carbide, silicon dioxide, magnesium oxide, tungsten carbide, alumina, a Kanthal alloy, and an Inconel alloy.
15. A method, comprising:
  - receiving electrical energy from a regenerative braking system;
  - converting at least a portion of the received electrical energy into thermal energy;
  - transferring, directly and/or indirectly, the thermal energy to a pressurized working fluid to form a heated pressurized working fluid;
  - introducing the heated pressurized working fluid into at least one turbine to propel a vehicle; and
  - wherein, during an extended period of regenerative braking, a first portion of the pressurized working fluid is passed through a cold side of a recuperator and a second portion of the pressurized working fluid bypasses the recuperator.

\* \* \* \* \*