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Richman et al.

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# (54) INFORMATION SYSTEM AND METHOD USING ANALYSIS BASED ON OBJECT-CENTRIC LONGITUDINAL DATA

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(58)	Field of Search	
, ,	707/102, 500, 513	; 303/126; 342/29; 600/300;
		701/29, 120, 123; 705/2, 3

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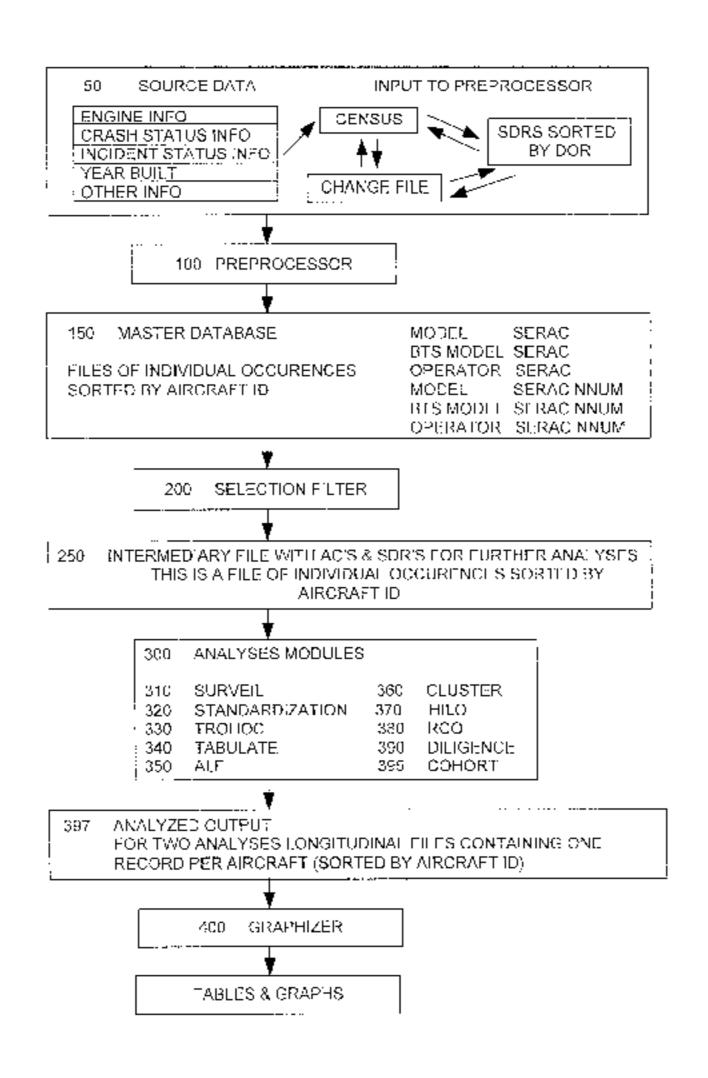
(List continued on next page.)

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## (57) ABSTRACT

A computerized information system and method using object-centric virtual database(s) and analyses advantageously suited to transportation safety applications, such as for the aviation industry (in which case aircraft may be the unit of observation i.e. the object class, and the individual object is an individual aircraft), as well as for health and other safety monitoring applications. A database comprises at least one set of data records, each data record pertaining to an event for an individual object and comprising a plurality of data fields, at least one of the data fields comprising an identifier identifying one individual object. A user interface component receives user-selected criteria for field values of the data fields and a selection component selects those data records which correspond to the userselected criteria. An analysis component identifies the selected data records according to a chronological sequence of events for individual objects covered by the selected data records, whereby each individual object covered by the selected data records is associated with a longitudinal set of data records, and derives output data for the events from the longitudinal sets of data records.

# 15 Claims, 31 Drawing Sheets



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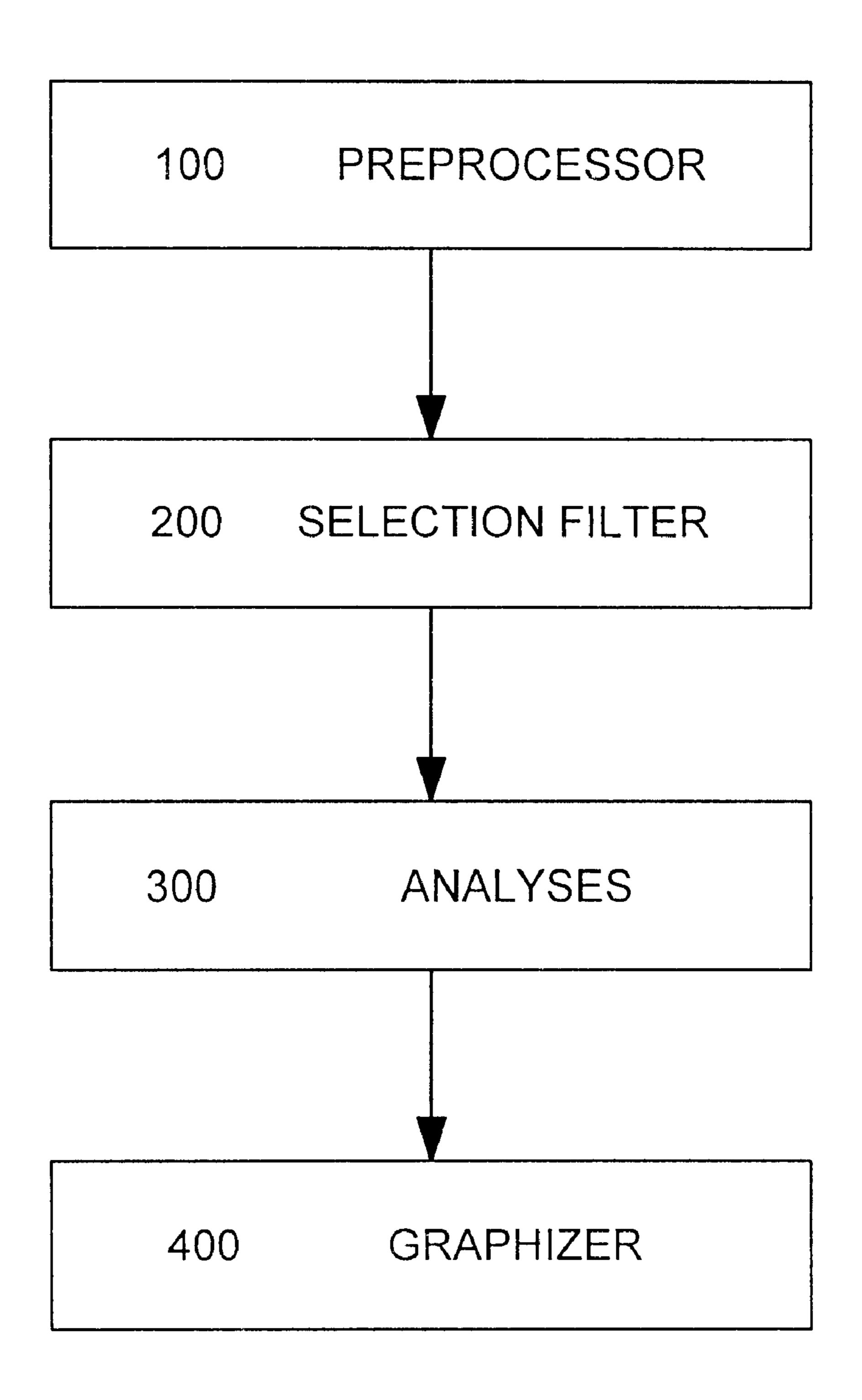
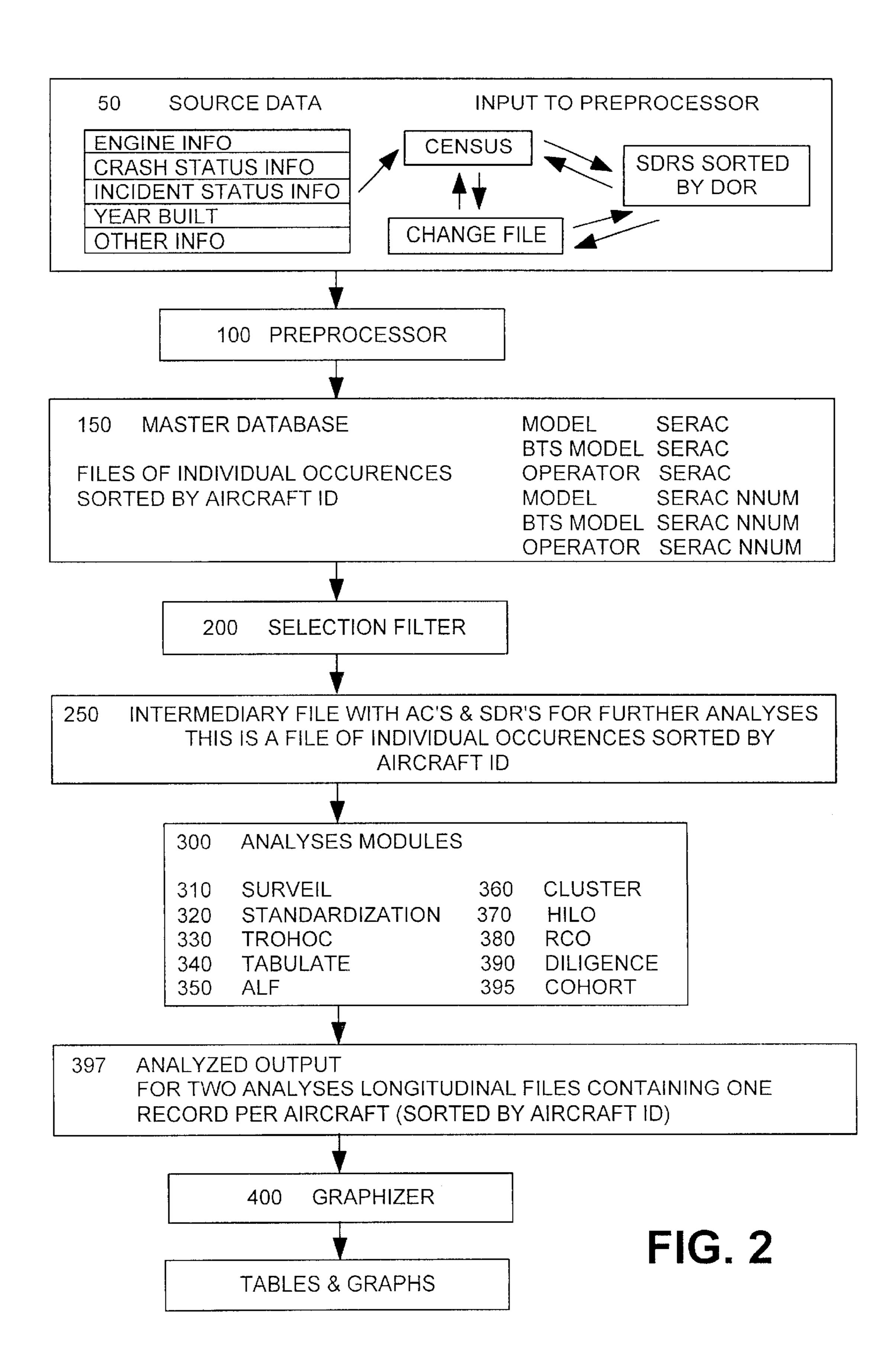
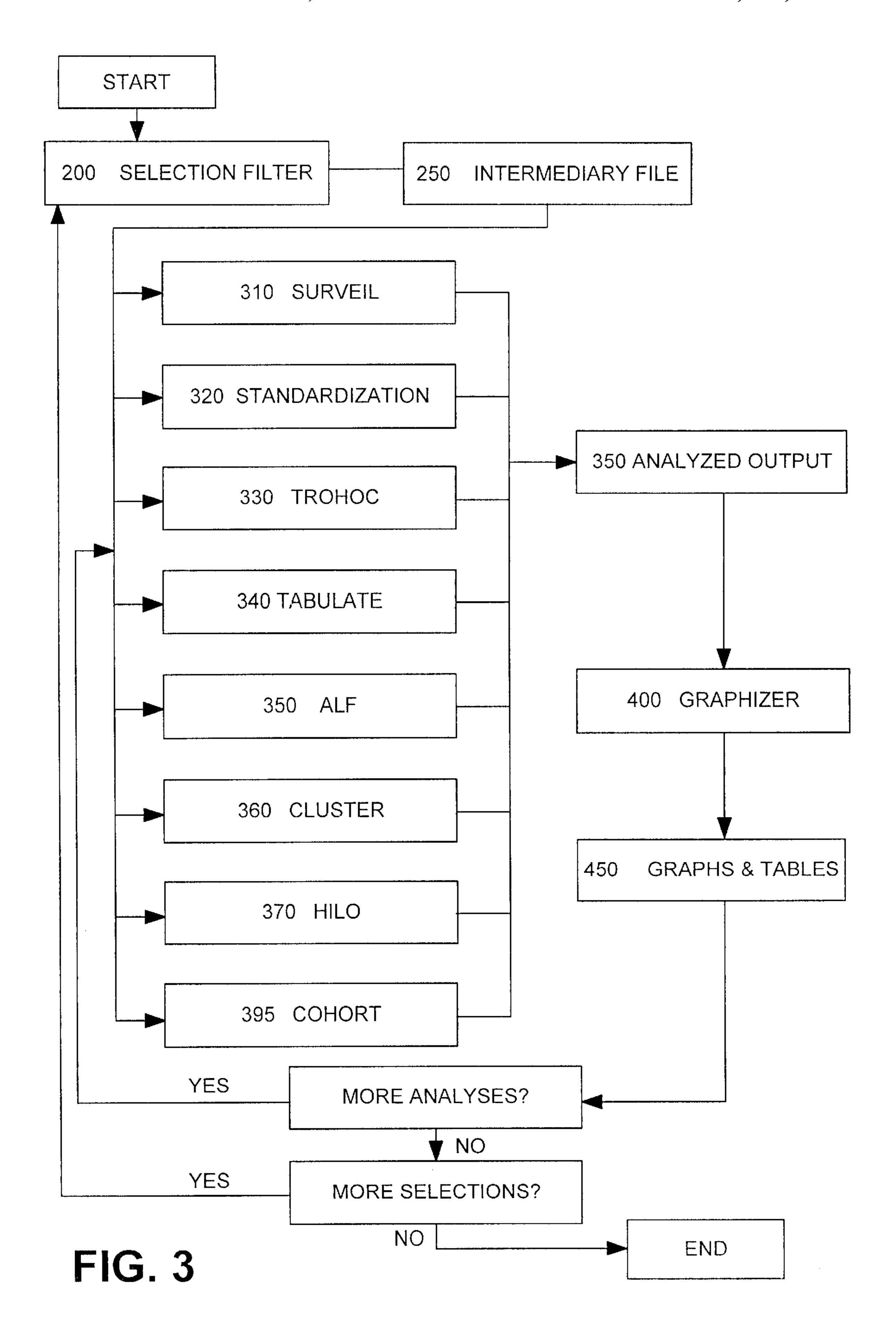
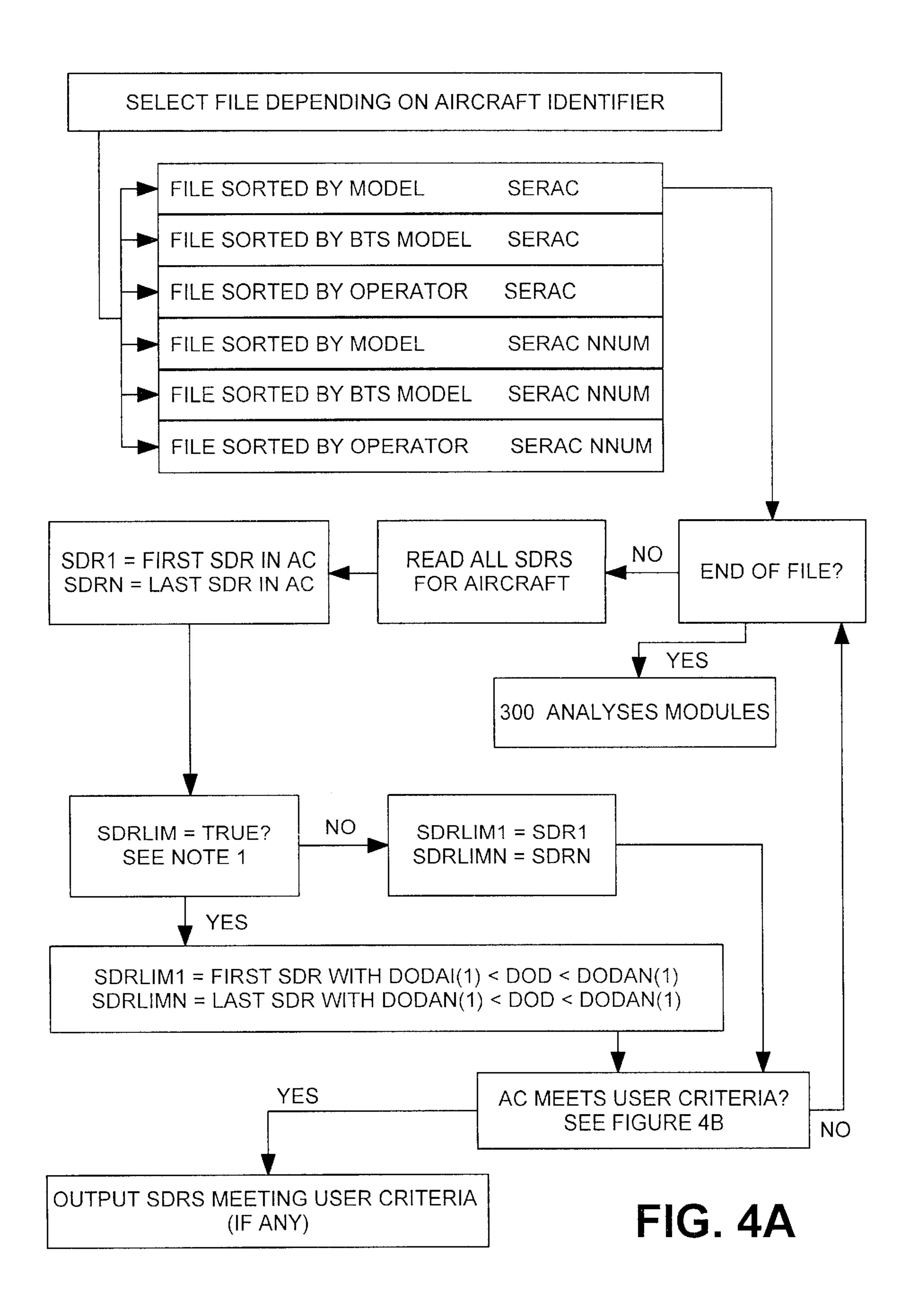
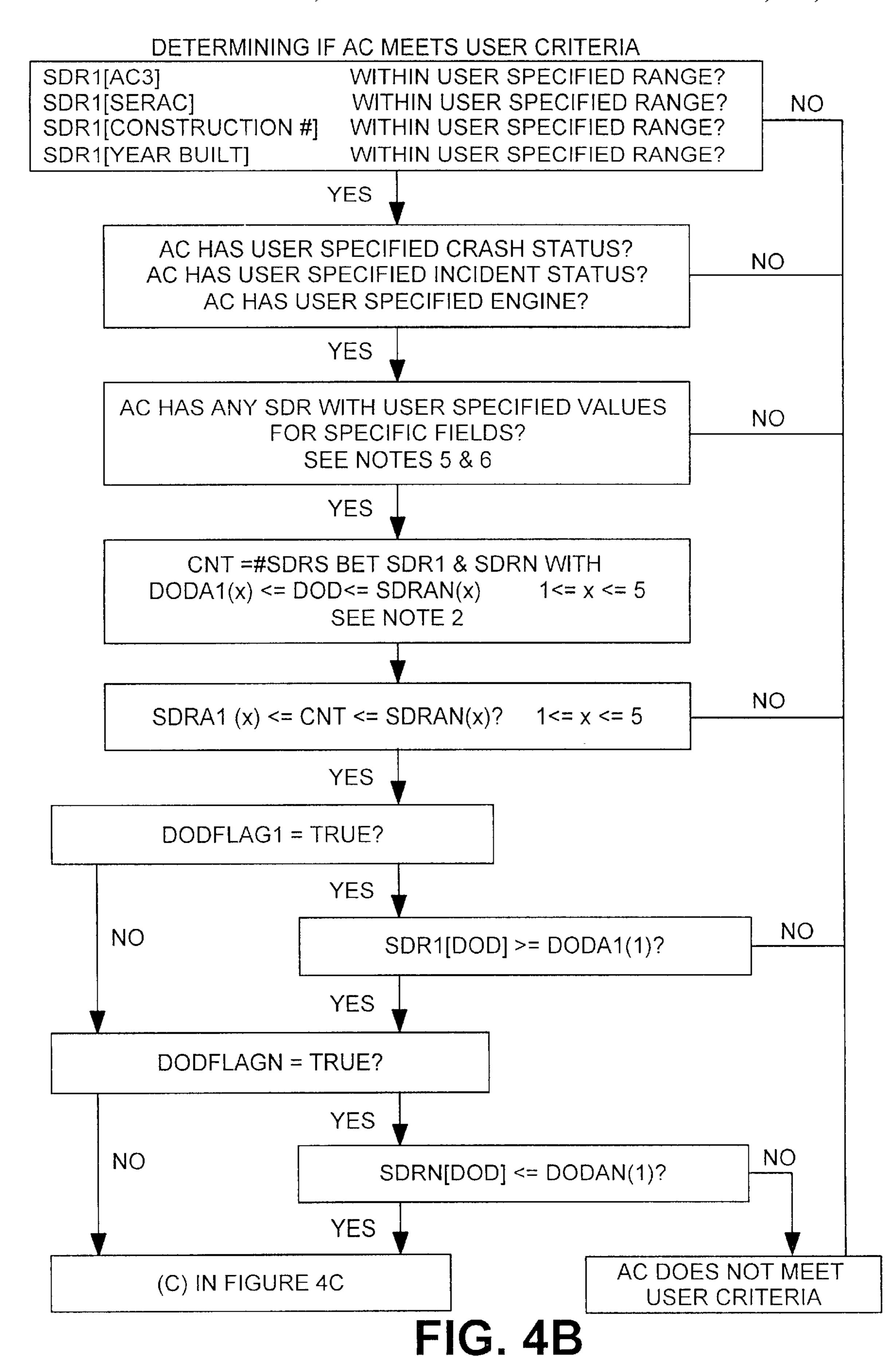


FIG. 1









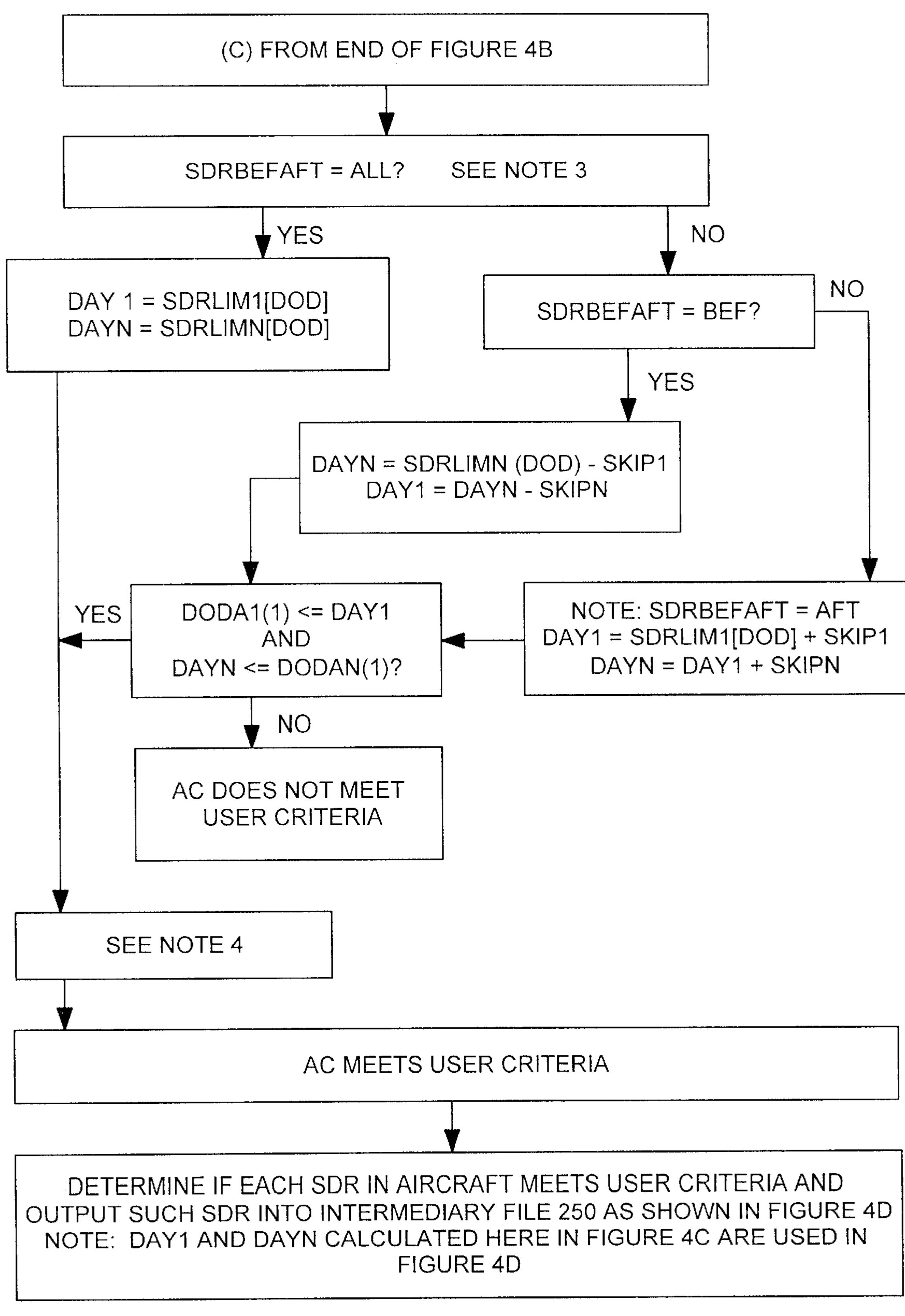


FIG. 4C

# DETERMINING IF AN SDR MEETS USER CRITERIA

NOTE: AFTER AN AIRCRAFT IS DEEMED TO HAVE MET THE USER CRITERIA AS PER FIGURES 4A AND 4B, SDRS MEETING USER CRITERIA ARE OUTPUT TO AN INTERMEDIARY FILE (250) FOR FURTHER ANALYSES (300)

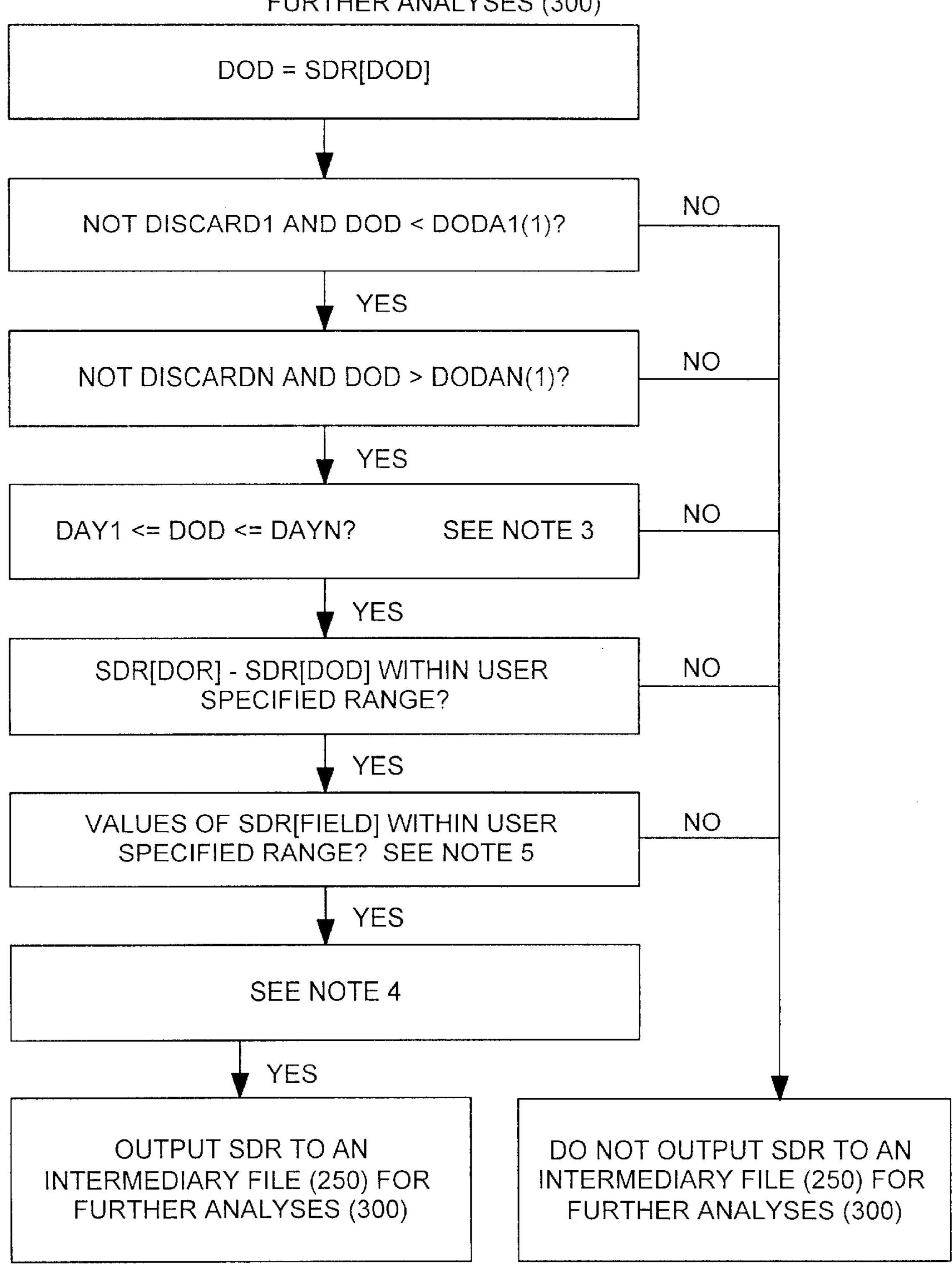


FIG. 4D

# NOTES FOR FIGURES 4A - 4D INCLUSIVE

NOTE 1 SDRLIM = TRUE => LOOK AT SDRS ONLY IN FIRST TIME PERIOD WHEN TESTING FOR USER SPECIFIED VALUES OF

FIELDS (AS PER NOTE 6 BELOW)

SDRLIM = FALSE => LOOK AT <u>ALL</u> SDRS WHEN TESTING FOR USER SPECIFIED VALUES OF FIELDS (AS PER NOTE 5 BELOW)

NOTE 2  $1 \le X \le 5$  TIME PERIODS

FIVE TIME PERIODS ARE USED FOR THIS EXAMPLE BUT THIS NUMBER CAN EASILY BE CHANGED.

NOTE 3 SDRBEFAFT ALLOWS USER TO LOOK AT A "SUBPERIOD" OF FIRST TIME PERIOD BY

A) SKIPPING SKIP1 DAYS AFTER SDRLIM1 AND LOOKING FOR SDRS FOR SKIP DAYS AFTER THAT POINT

OR

B) SKIPPING SKIP1 DAYS BEFORE SDRLIMN AND LOOKING FOR SDRS FOR SKIPN DAYS BEFORE THAT POINT

OR

C) SKIP1 & SKIPN ARE USER DEFINED

NOTE 4 OTHER USER CRITERIA MAY BE INSERTED HERE IF DESIRED

NOTE 5 SPECIFIC FIELD = OTHER, ALTERNATE, TYPE, NATURE OF CONDITION, PRECAUTIONARY PROCEDURE, SEVERITY, STAGE OF OPERATION, ATA 2 CHAR CODE AND 4 CHAR CODE

NOTE 6 CURRENT TEST INVOLVES CHECKING IF AC HAS ANY <u>SDR</u>
WITH USER SPECIFIED VALUES OF FIELDS (AS PER NOTE 5
ABOVE)

IF DESIRED, THIS COULD BE MODIFIED TO REQUIRE A MINIMUM AND 10 MAXIMUM # SDRS WITH USER SPECIFIED VALUES OF FIELDS (AS PER NOTE 5 ABOVE)

FIG. 4E

# CALCULATING REPEATS FOR AN AIRCRAFT

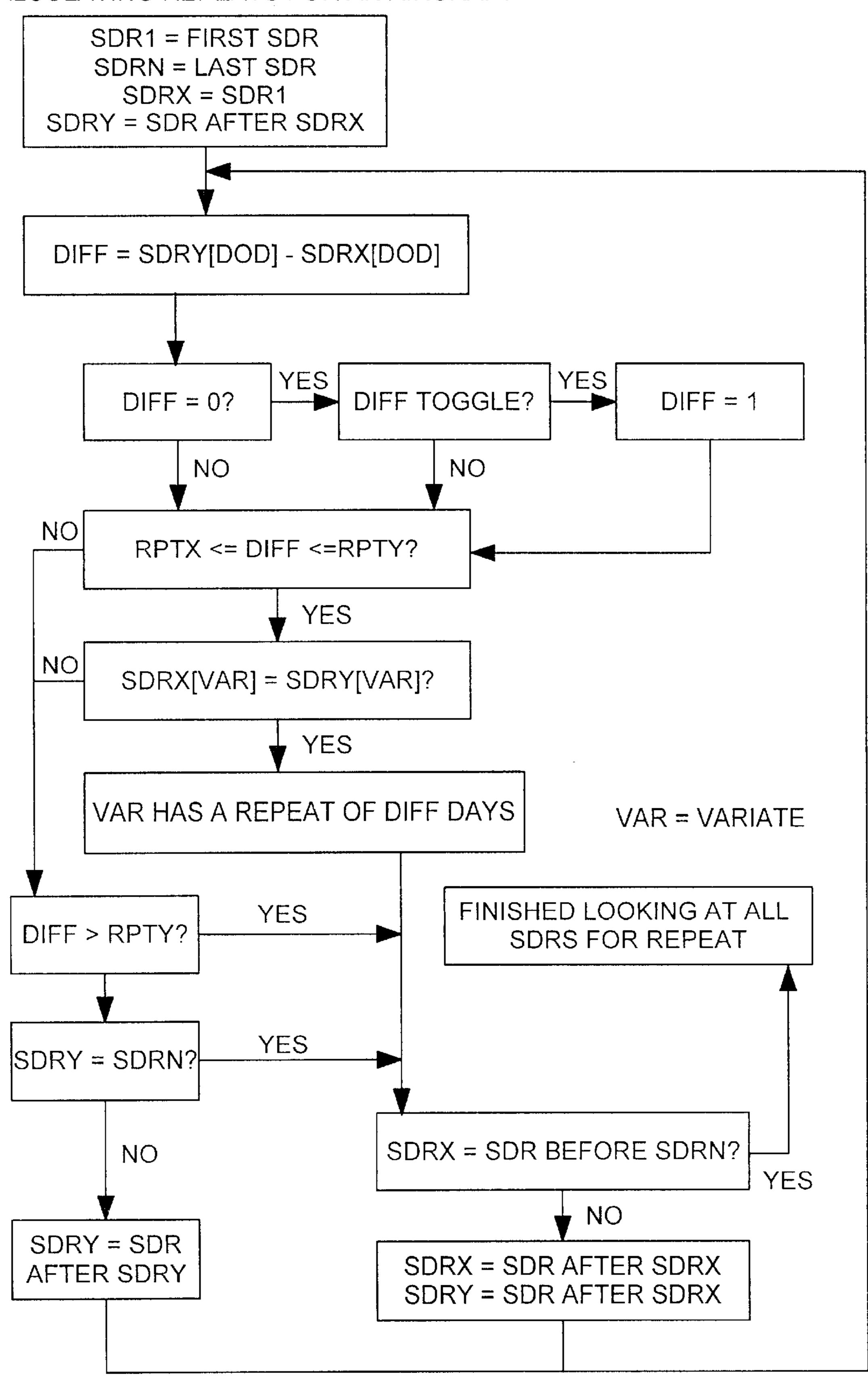
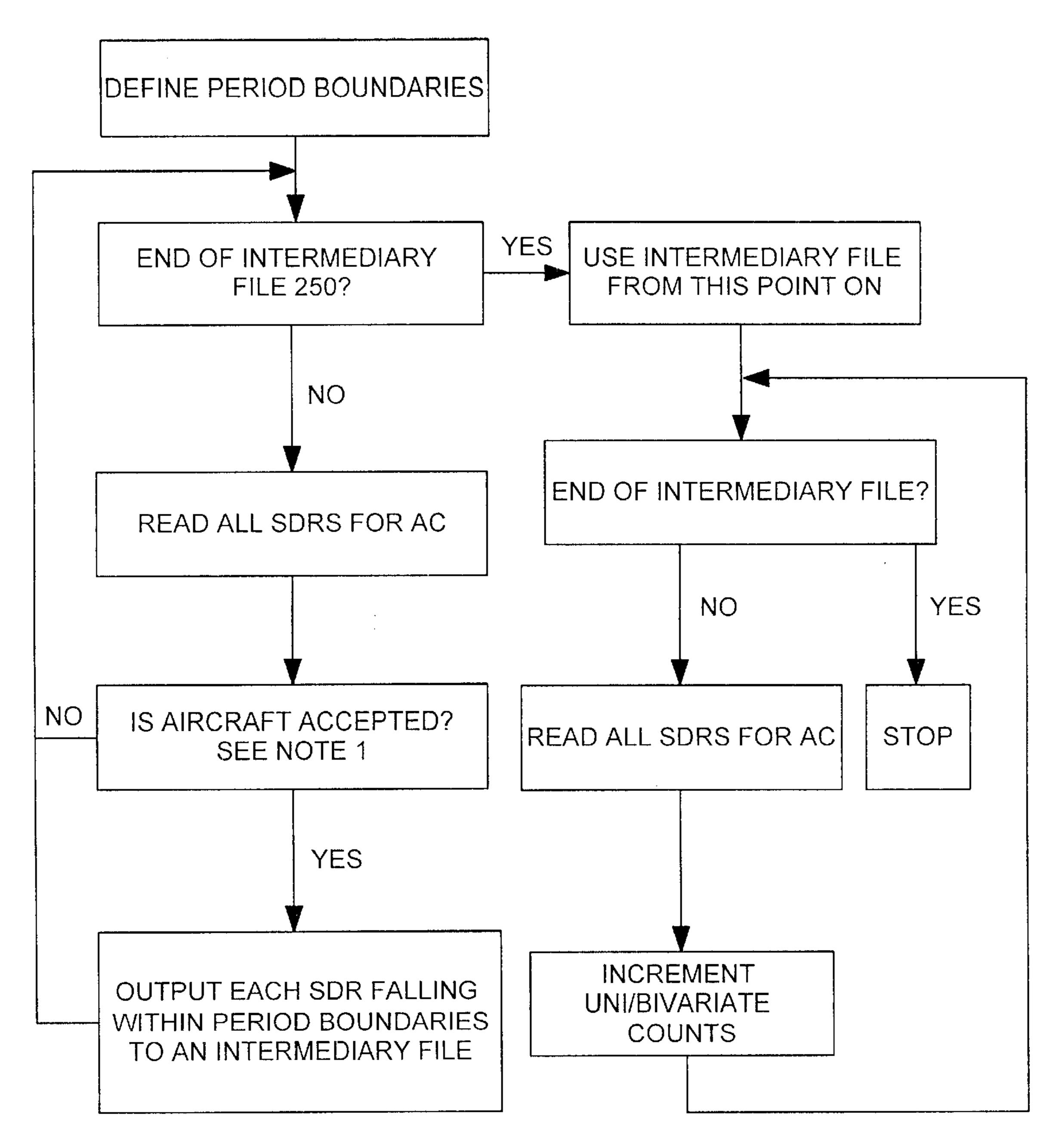


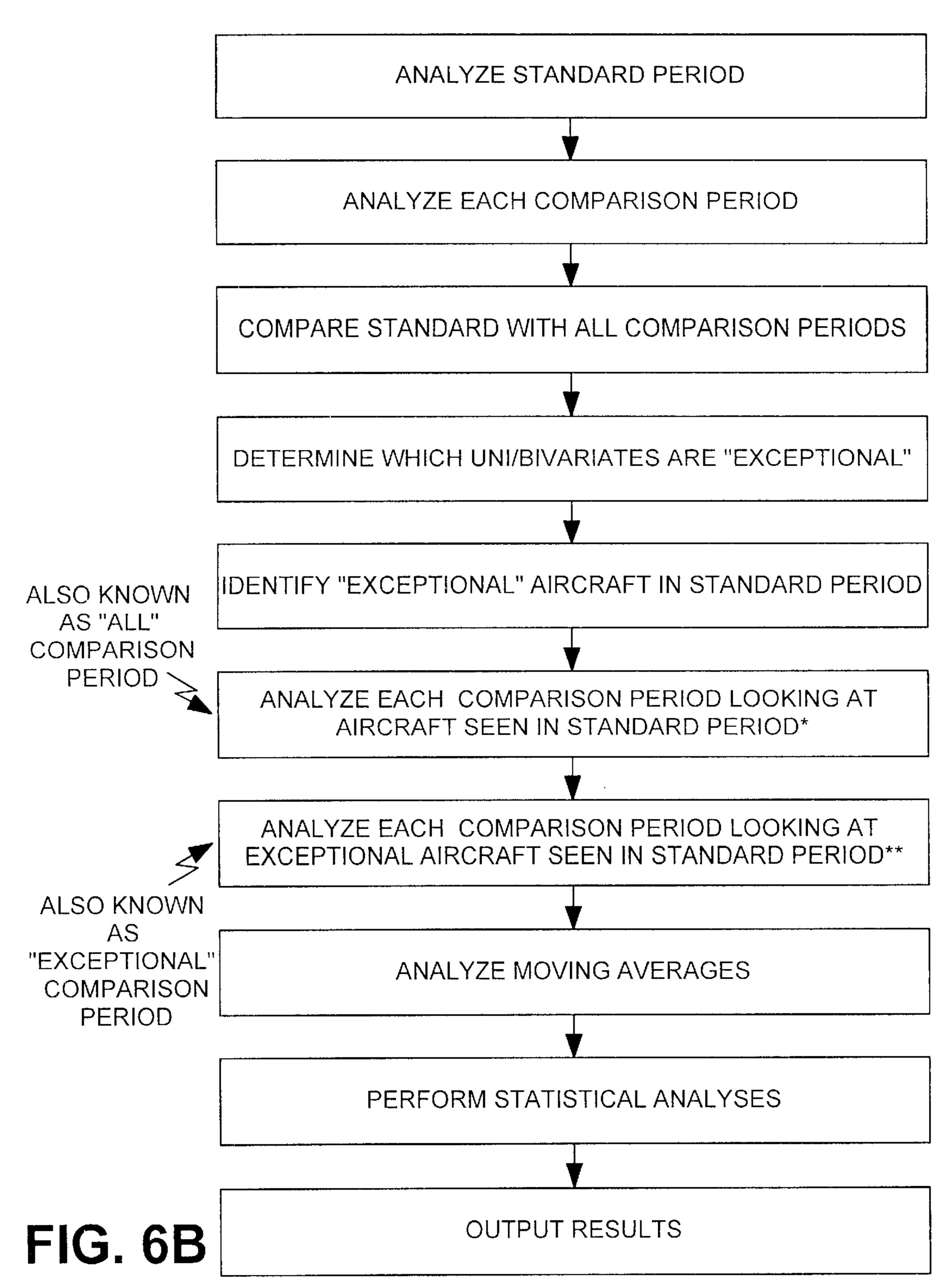
FIG. 5



#### NOTE 1

- -ALL AIRCRAFT IN THE STANDARD PERIOD ARE ACCEPTED.
- -ALL AIRCRAFT IN COMPARISON PERIOD ARE ACCEPTED.
- -AN AIRCRAFT IN AN "ALL" COMPARISON PERIOD IS ACCEPTED IF IT IS ALSO SEEN IN THE STANDARD PERIOD
- -AN AIRCRAFT IN AN 'EXCEPTIONAL' COMPARISON PERIOD IS ACCEPTED IF IT IS IS AN EXCEPTIONAL AIRCRAFT SEEN IN THE STANDARD PERIOD -ALL AC'S IN MOVING AVERAGE ARE ACCEPTED

FIG. 6A



\*IF AN AIRCRAFT IN THE EXTRA COMPARISON PERIOD IS NOT ALSO AN AIRCRAFT SEEN IN THE STANDARD PERIOD, IT IS NOT ANALYZED.

\*\*IF AN AIRCRAFT IN THE EXTRA COMPARISON PERIOD IS NOT ALSO AN EXCEPTIONAL AIRCRAFT SEEN IN THE STANDARD PERIOD, IT IS NOT ANALYZED.

# FORMING MOVING AVERAGES FROM COMPARISON PERIODS

CP1	CP2	CP3	CP4	CP5	CP6	CP7

MA1 = CP1, 2, 3

MA2 = CP2, 3, 4

MA3 = CP3, 4, 5

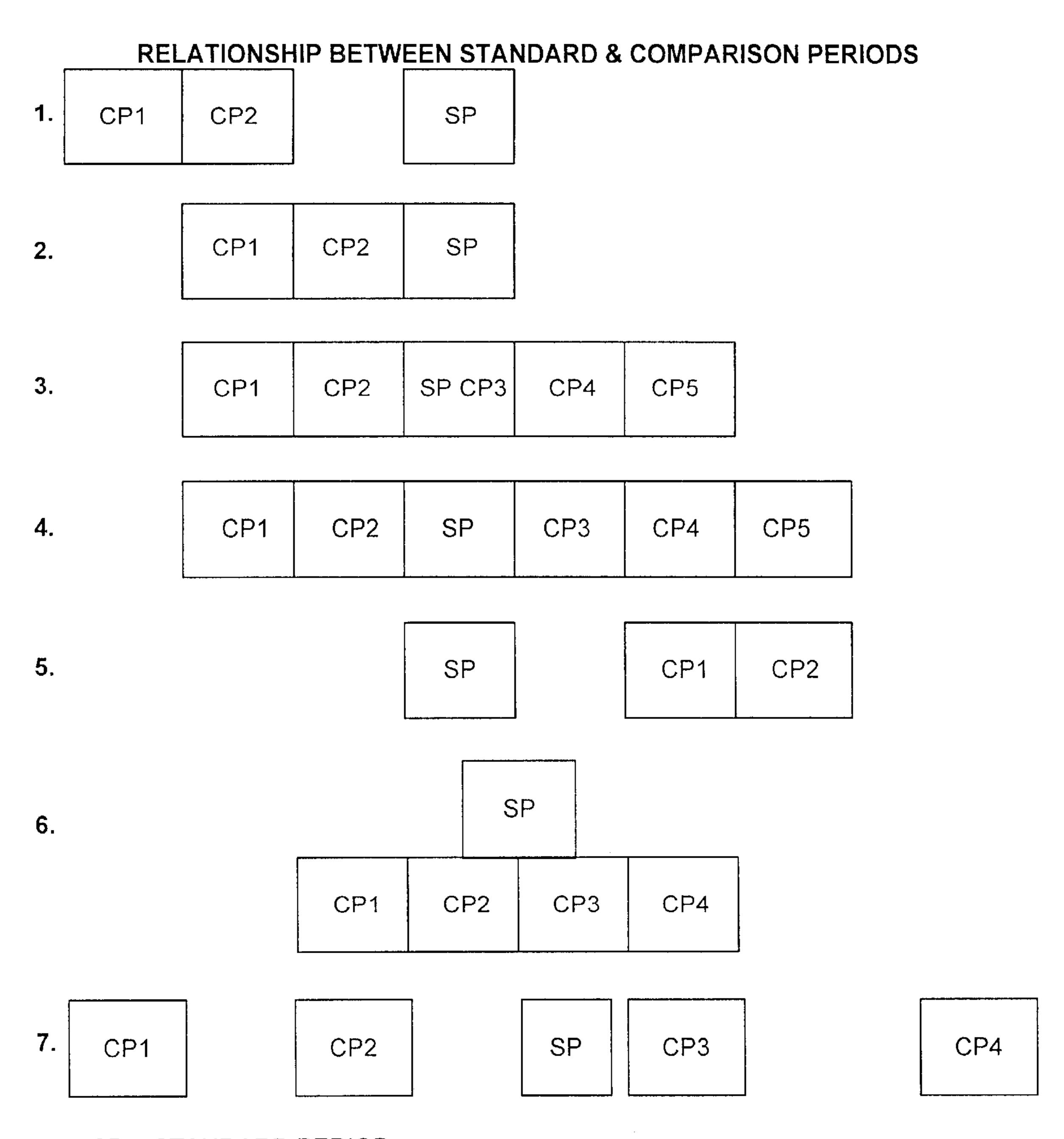
MA4 = CP4, 5, 6

MA5 = CP5, 6, 7

CP = COMPARISON PERIOD MA = MOVING AVERAGE

FIG. 6C

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SP = STANDARD PERIOD CP = COMPARISON PERIOD

- CP BEFORE SP, NONE ADJACENT TO SP
- CP BEFORE SP, ONE ADJACENT TO SP
- CP BEFORE & AFTER SP, SP DOUBLES UP AS CP
- 4. CP BEFORE & AFTER SP, SP DOES NOT DOUBLE UP AS CP
- 5. CP AFTER SP, NONE ADJACENT TO SP
- 6. CP BEFORE & AFTER SP, NONE ADJACENT TO SP
- 7. CP BEFORE & AFTER SP, CP'S WIDELY DISPERSED

FIG. 6D

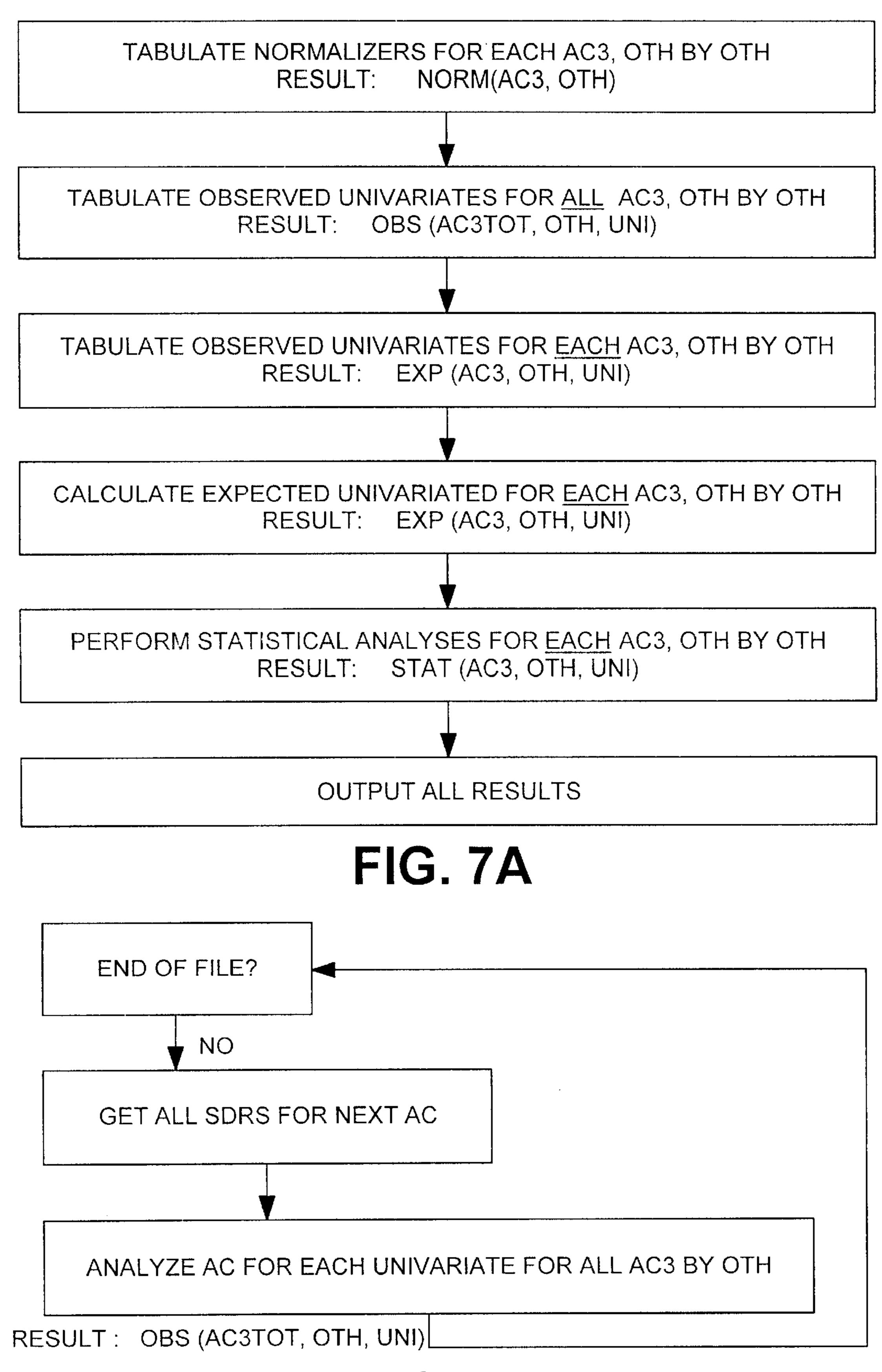
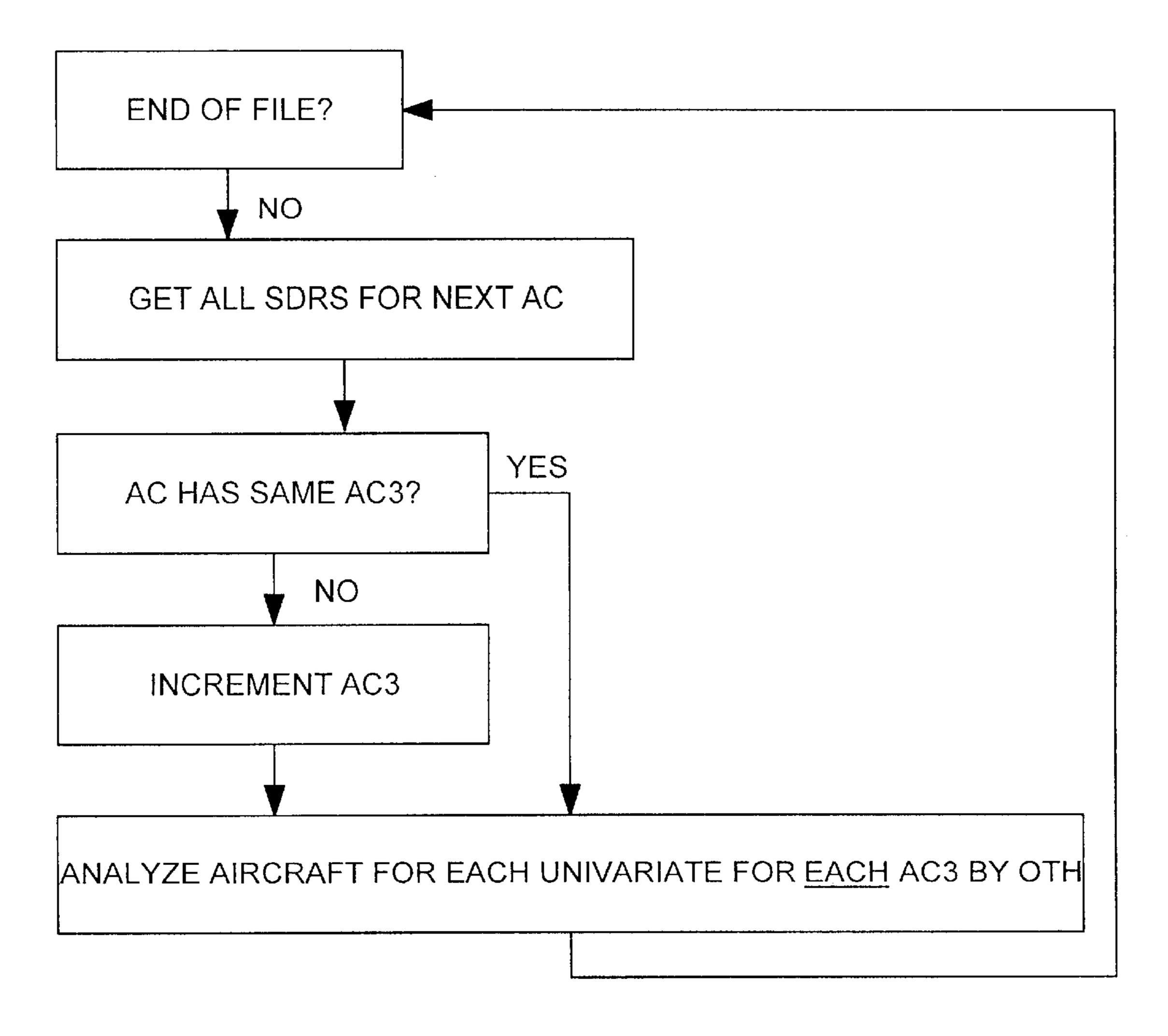


FIG. 7B



RESULT: OBS (AC3, OTH, UNI)

FIG. 7C

FOR AC3 = AC31 TO AC3N

FOR OTH = OTH1 TO OTHN

FOR UNI = UNI1 TO UNIN

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RATIO = NORM (AC3, OTH) NORM (AC3ALL, OTH)

EXP (AC3, OTH, UNI) = RATIO x OBS (AC3ALL, OTH, UNI)

**NEXT UNI** 

**NEXT OTH** 

NEXT AC3

FOR AC3 = AC31 TO AC3N

FOR OTH = OTH1 TO OTHN

FOR UNI = UNI1 TO UNIN

PRIORITY (AC3, OTH, UNI) = <u>OBS (AC3, OTH, UNI)</u> X 100 EXP (ACE, OTH, UNI

**NEXT UNI** 

**NEXT OTH** 

NEXT AC3

EG: OTHER STATISTICAL ANALYSES PERFORMED HERE LOWER/HIGHER CONFIDENCE LOWER/HIGHER CONFIDENCE LIMITS

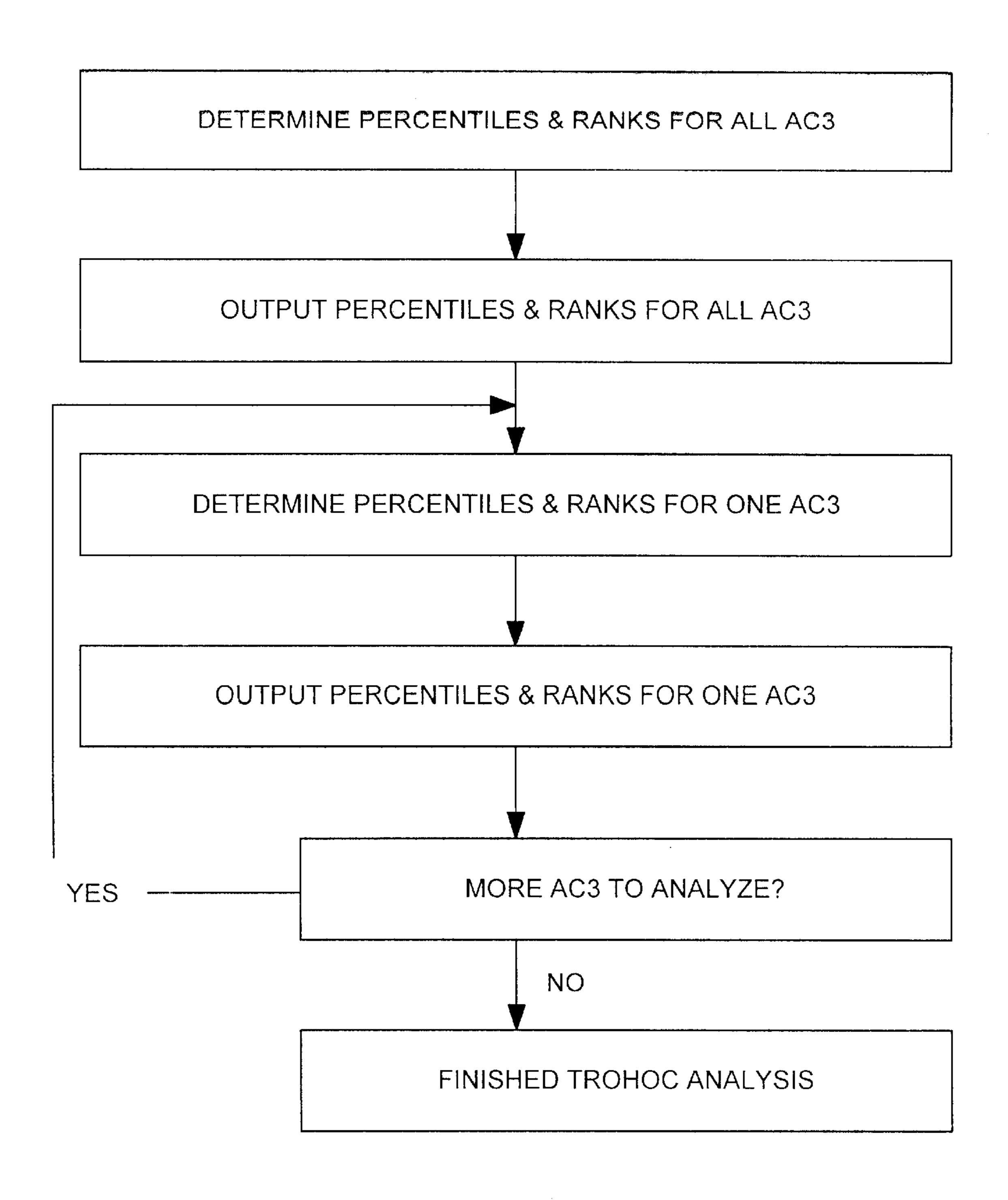
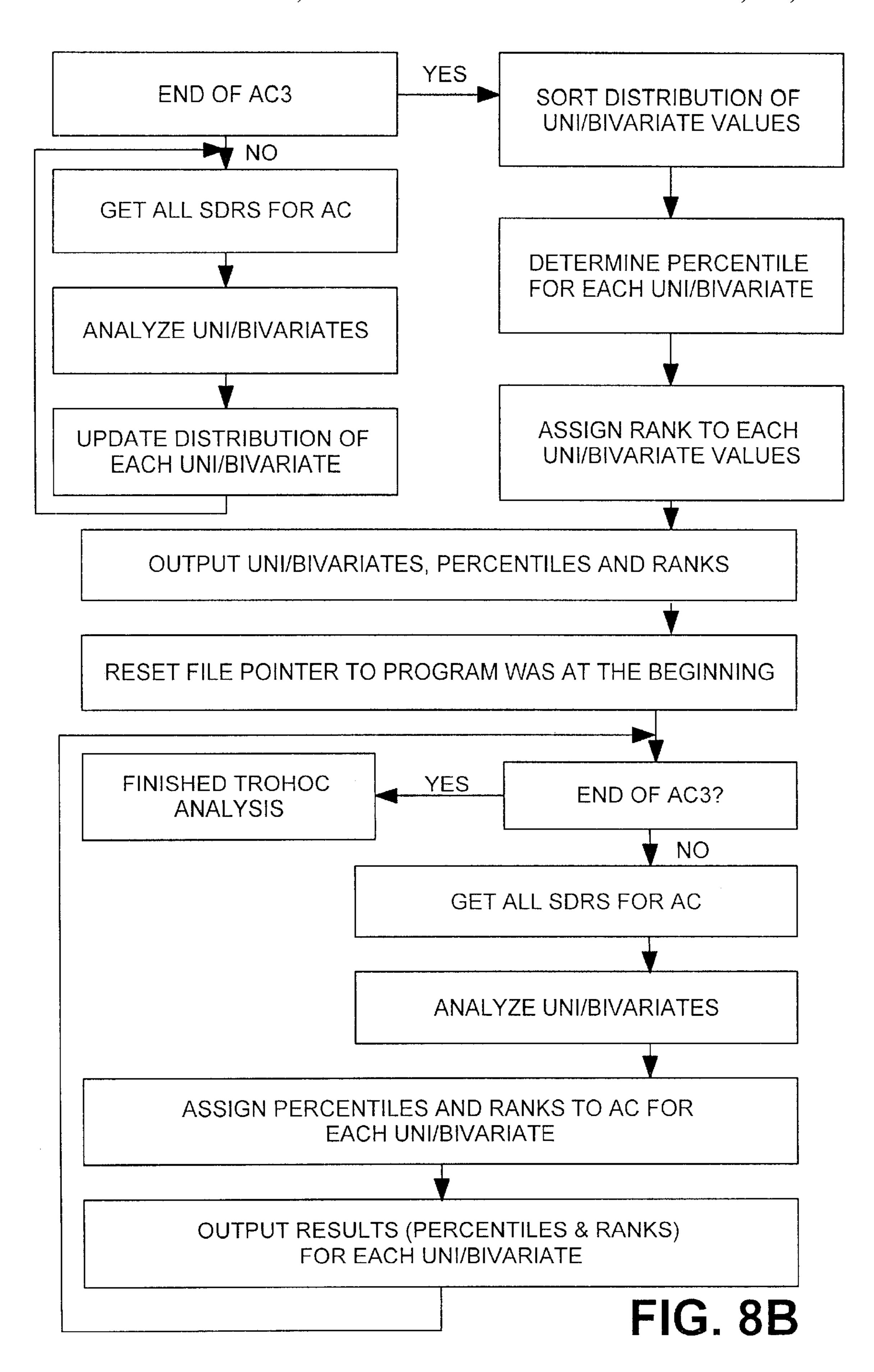


FIG. 8A



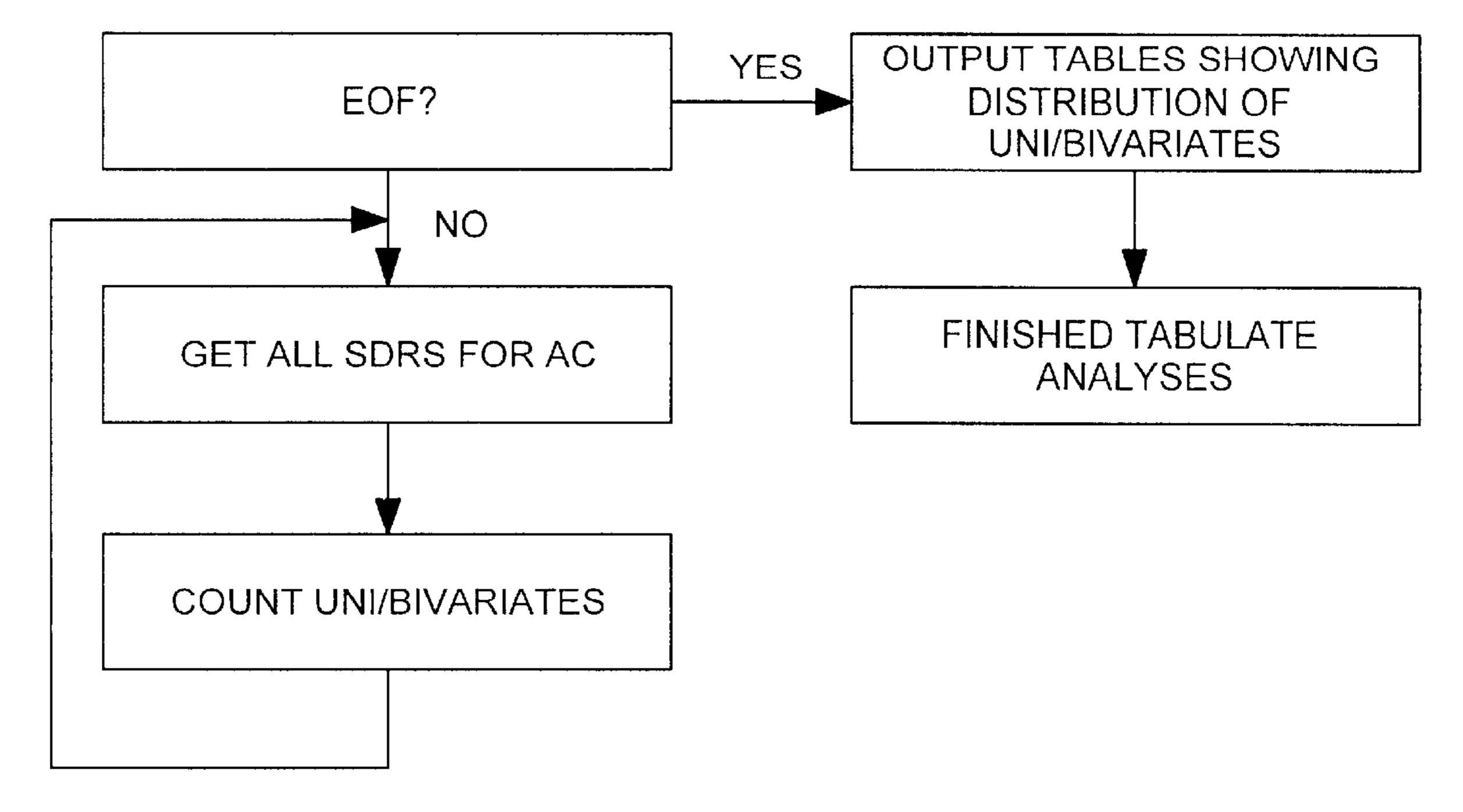
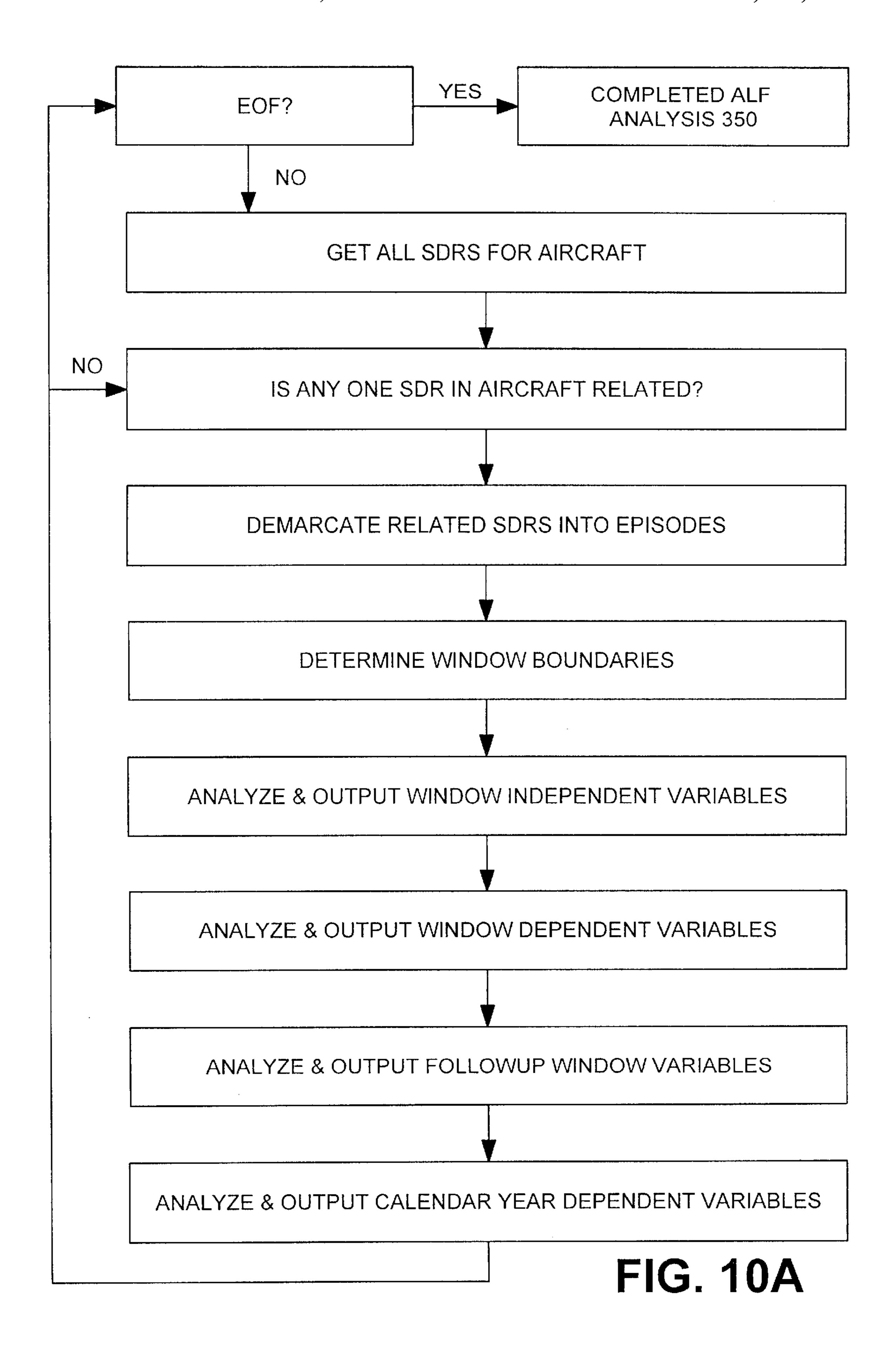


FIG. 9



NUMBER OF SDRS INAIRCRAFT

NUMBER OF SDRS WITHIN USER-SPECIFIED DAYS BEFORE OR AFTER EACH UEL NATURE OF CONDITION (NC) "COMBO CODE" FOR ALL SDRS IN AIRCRAFT PRECAUTIONARY PROCEDURE (PP) COMBO CODE FOR ALL SDRS IN AIRCRAFT

NUMBER OF UELS (TYPE = 1) IN A GIVEN MONTH OF ANY YEAR

NUMBER OF UELS (TYPE = 1) IN AIRCRAFT

SHORTEST INTERVAL BETWEEN ANY TWO UELS (TYPE = 1) IN AIRCRAFT NUMBER OF MAJORS (TYPE = 2) IN AIRCRAFT

SHORTEST INTERVAL BETWEEN ANY TWO MAJORS (TYPE = 2) IN AIRCRAFT NUMBER OF ROUTINES (TYPE = 3) IN AIRCRAFT

SHORTEST INTERVAL BETWEEN ANY TWO ROUTINES (TYPE = 3) IN AIRCRAFT NUMBER OF NMNRS (TYPE = 4) IN AIRCRAFT

SHORTEST INTERVAL BETWEEN ANY TWO NMNRS (TYPE = 4) IN AIRCRAFT NUMBER OF SDRS IN AIRCRAFT FOR USER-SPECIFIED YEAR

NUMBER OF UELS (TYPE = 1) IN AIRCRAFT FOR USER SPECIFIED CALENDAR YEAR

NUMBER OF MAJORS (TYPE= 2) IN AIRCRAFT FOR USER SPECIFIED CALENDAR YEAR

NUMBER OF ROUTINES (TYPE = 3) IN AIRCRAFT FOR USER SPECIFIED CALENDAR YEAR

NUMBER OF NMNRS (TYPE = 4) IN AIRCRAFT FOR USER SPECIFIED CALENDAR YEAR

OPERATOR NUMBER OF FIRST SDR IN USER-SPECIFIED CALENDAR YEAR NATURE OF CONDITION OF FIRST SDR IN USER-SPECIFIED CALENDAR YEAR INTERVAL BETWEEN AIRCRAFT'S 1ST SDR AND ITS LAST SDR 1ST SHORTEST INTERVAL (DAYS) BETWEEN TWO SDRS IN AIRCRAFT 2ND SHORTEST INTERVAL (DAYS) BETWEEN TWO SDRS IN AIRCRAFT 3RD SHORTEST INTERVAL (DAYS) BETWEEN TWO SDRS IN AIRCRAFT LONGEST INTERVAL (DAYS) BETWEEN TWO SDRS IN AIRCRAFT INTERVAL BETWEEN AIRCRAFT'S 1ST UEL AND ITS LAST UEL 1ST SHORTEST INTERVAL (DAYS) BETWEEN TWO UELS IN AIRCRAFT 2ND SHORTEST INTERVAL (DAYS) BETWEEN TWO UELS IN AIRCRAFT 3RD SHORTEST INTERVAL (DAYS) BETWEEN TWO UELS IN AIRCRAFT LONGEST INTERVAL (DAYS) BETWEEN TWO UELS IN AIRCRAFT INTERVAL BETWEEN AIRCRAFT'S 1ST MAJOR AND ITS LAST MAJOR 1ST SHORTEST INTERVAL (DAYS) BETWEEN TWO MAJORS IN AIRCRAFT 2ND SHORTEST INTERVAL (DAYS) BETWEEN TWO MAJORS IN AIRCRAFT 3RD SHORTEST INTERVAL (DAYS) BETWEEN TWO MAJORS IN AIRCRAFT LONGEST INTERVAL (DAYS) BETWEEN TWO MAJORS IN AIRCRAFT INTERVAL BETWEEN AIRCRAFT'S 1ST ROUTINE AND ITS LAST ROUTINE 1ST SHORTEST INTERVAL (DAYS) BETWEEN TWO ROUTINES IN AIRCRAFT 2ND SHORTEST INTERVAL (DAYS) BETWEEN TWO ROUTINES IN AIRCRAFT 3RD SHORTEST INTERVAL (DAYS) BETWEEN TWO ROUTINES IN AIRCRAFT LONGEST INTERVAL (DAYS) BETWEEN TWO ROUTINES IN AIRCRAFT INTERVAL BETWEEN AIRCRAFT'S 1ST NMNR AND ITS LAST NMNR 1ST SHORTEST INTERVAL (DAYS) BETWEEN TWO NMNRS IN AIRCRAFT 2ND SHORTEST INTERVAL (DAYS) BETWEEN TWO NMNRS IN AIRCRAFT 3RD SHORTEST INTERVAL (DAYS) BETWEEN TWO NMNRS IN AIRCRAFT LONGEST INTERVAL (DAYS) BETWEEN TWO NMNRS IN AIRCRAFT

FIG. 10B

NUMBER OF YEARS WITH 0 (1, 2, 3, 4, 5, 6 OR 7) SDRS NUMBER OF YEARS WITH 0 (1, 2, 3, 4, 5, 6 OR 7) UELS NUMBER OF YEARS WITH 0 (1, 2, 3, 4, 5, 6 OR 7) MAJORS

# EPISODE DEPENDENT VARIABLES ("X" BEING USER SELECTABLE)

DURATION IN DAYS OF EPISODE X

COMBO CODE OF EPISODE X

NATURE OF CONDITION OF FIEST RELATED SDR IN EPISODE X

PRECAUTIONARY PROCEDURE OF FIRST RELATED SDR IN EPISODE X

DOD OF FIRST RELATED SDR IN EPISODE X

INTERVAL BETWEEN FIRST AND SECOND RELATED SDR IN EPISODE X

NUMBER OF SDRS IN EPISODE X

SHORTEST INTERVAL BETWEEN 2 ANY SDRS INSIDE EPISODE X

NUMBER OF UELS IN EPISODE X

SHORTEST INTERVAL BET ANY TWO UELS IN EPISODE X

NUMBER OF MAJORS IN EPISODE X

SHORTEST INTERVAL BET ANY TWO MAJORS IN EPISODE X

NUMBER OF ROUTINES IN EPISODE X

SHORTEST INTERVAL BET ANY TWO ROUTINES IN EPISODE X

NUMBER OF NMNRS IN EPISODE IN EPISODE X

SHORTEST INTERVAL BET ANY TWO NMNRS IN EPISODE X

(NOTE: THE ABOVE VARIABLES ARE OUTPUT FOR EACH EPISODE UP TO THE XTH EPISODE.)

INTERVAL BETWEEN EPISODE X AND EPISODE X-1 (EG: IF X=3 THEN THE INTERVAL IS MEASURED BETWEEN THE FIRST EPISODE AND THE SECOND AND ANOTHER INTERVAL IS MEASURED BETWEEN THE SECOND EPISODE AND THE THIRD).

# WINDOW DEPENDENT VARIABLES

NUMBER OF SDRS IN WINDOW

SHORTEST INTERVAL BET ANY TWO SDRS IN WINDOW

NUMBER OF RELATED SDRS IN WINDOW

SHORTEST INTERVAL BET ANY TWO RELATED SDRS IN WINDOW

NUMBER OF EPISODES IN WINDOW

SHORTEST INTERVAL BET ANY EPISODES IN WINDOW

NATURE OF CONDITION OF FIRST RELATED SDR IN WINDOW

PRECAUTIONARY PROCEDURE OF FIRST RELATED SDR IN WINDOW

NUMBER OF UELS IN WINDOW

SHORTEST INTERVAL BET ANY TWO UELS IN WINDOW

NUMBER OF MAJORS IN WINDOW

SHORTEST INTERVAL BET ANY TWO MAJORS IN WINDOW

NUMBER OF ROUTINES (TYPE = 3) IN WINDOW

SHORTEST INTERVAL BET ANY TWO ROUTINES (TYPE = 3) IN WINDOW

NUMBER OF NMNRS (TYPE = 4) IN WIDNOW

SHORTEST INTERVAL BET ANY TWO NMNRS (TYPE = 4) IN WINDOW

FIG. 10C

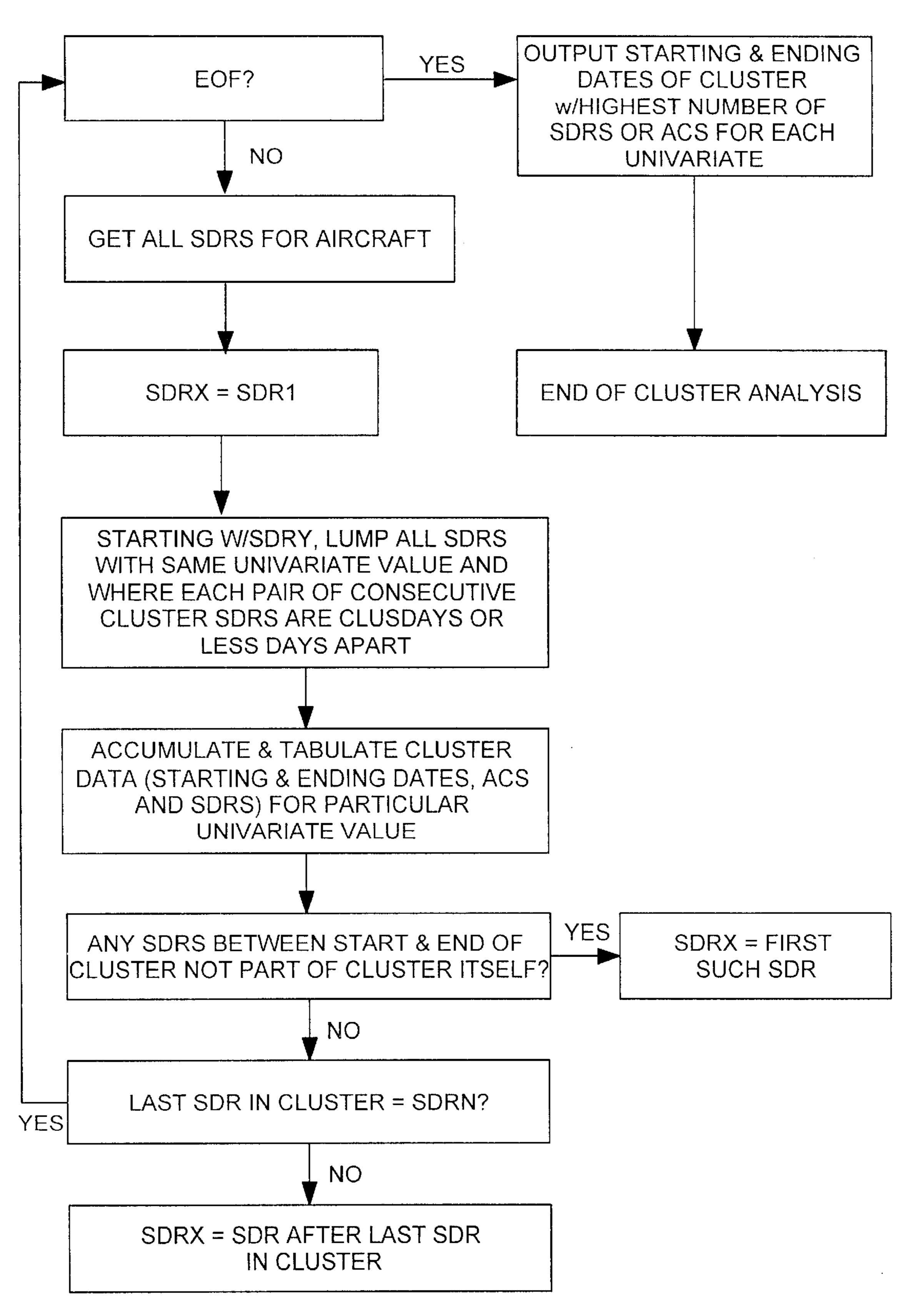
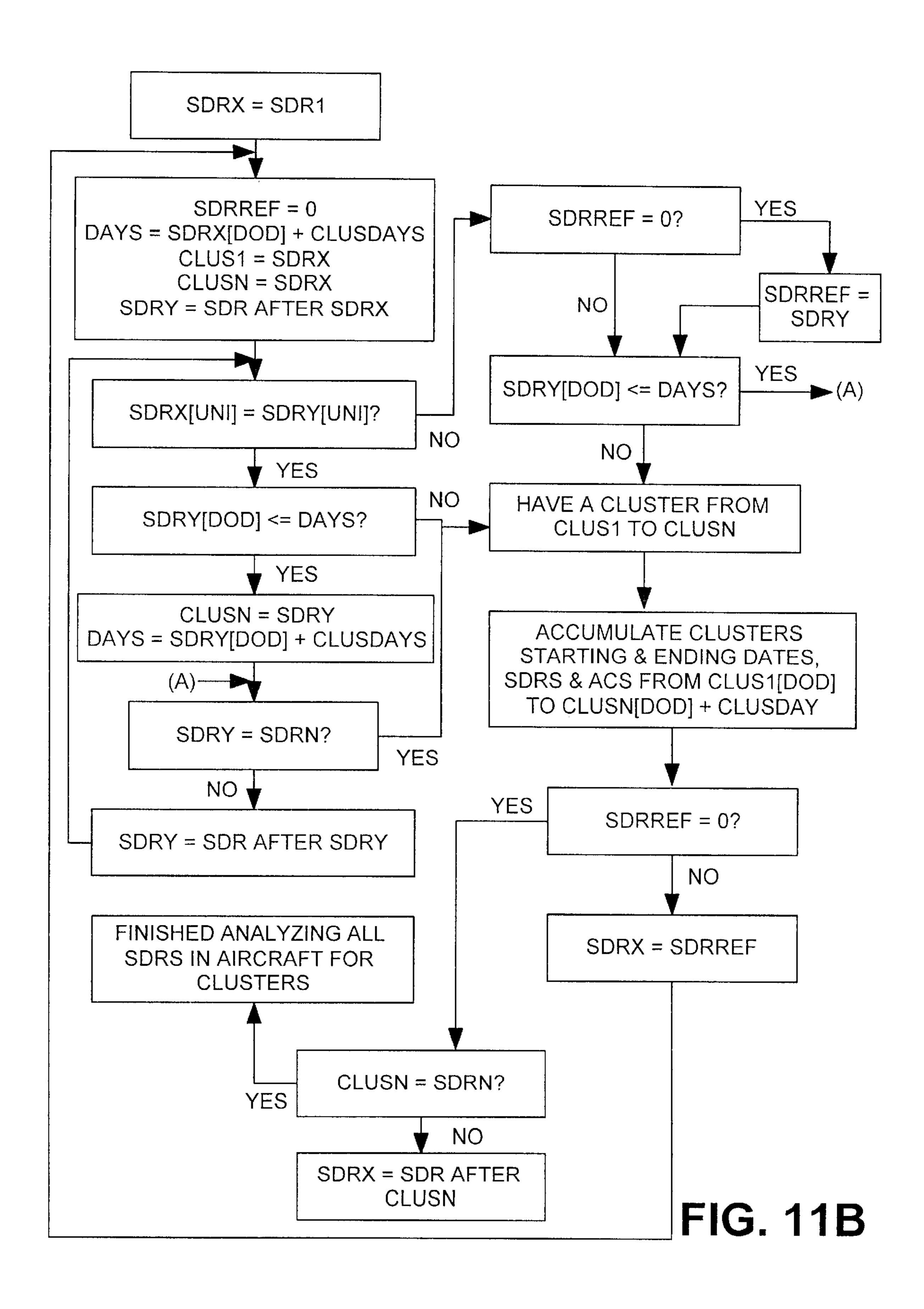


FIG. 11A



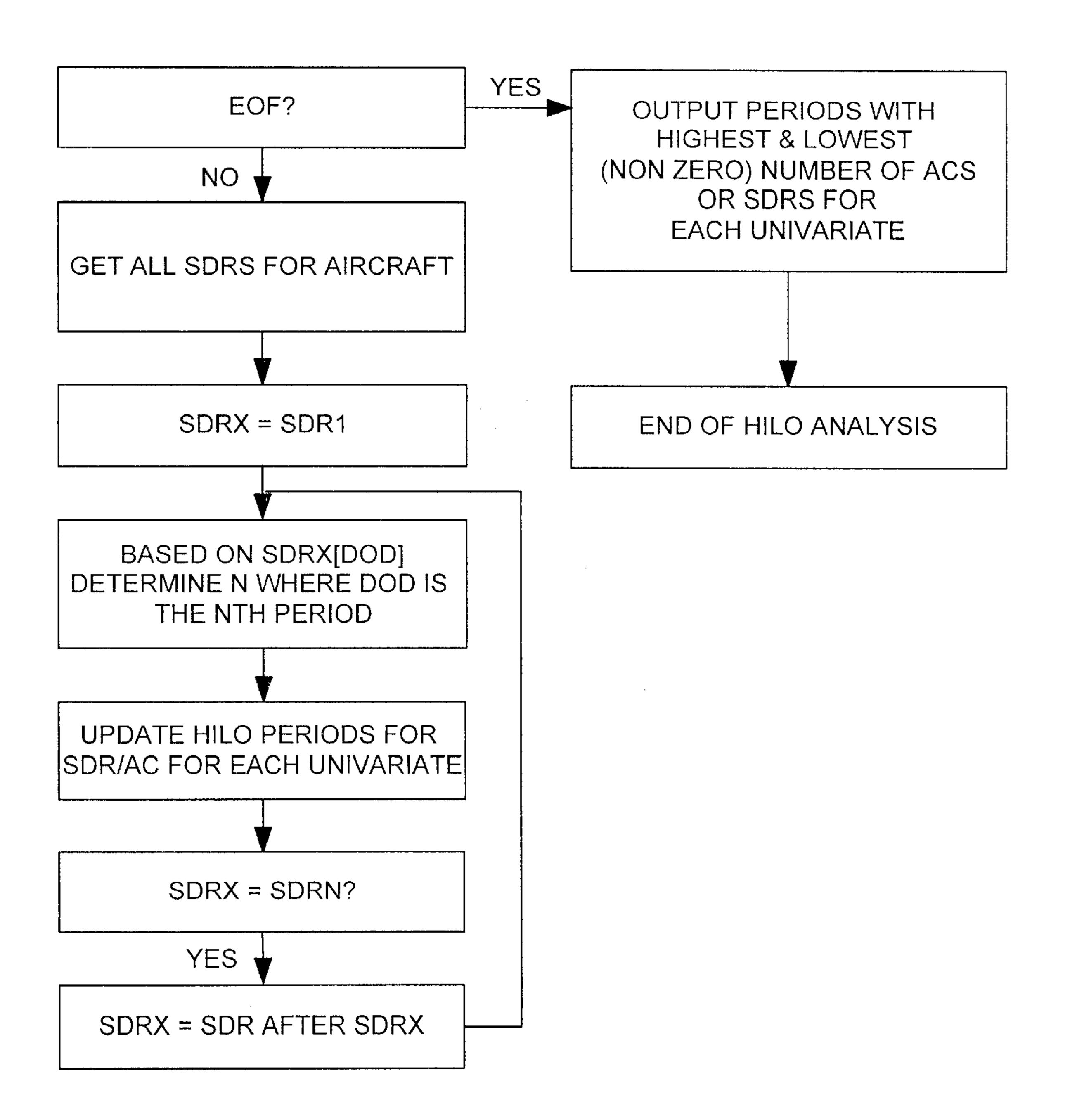


FIG. 12

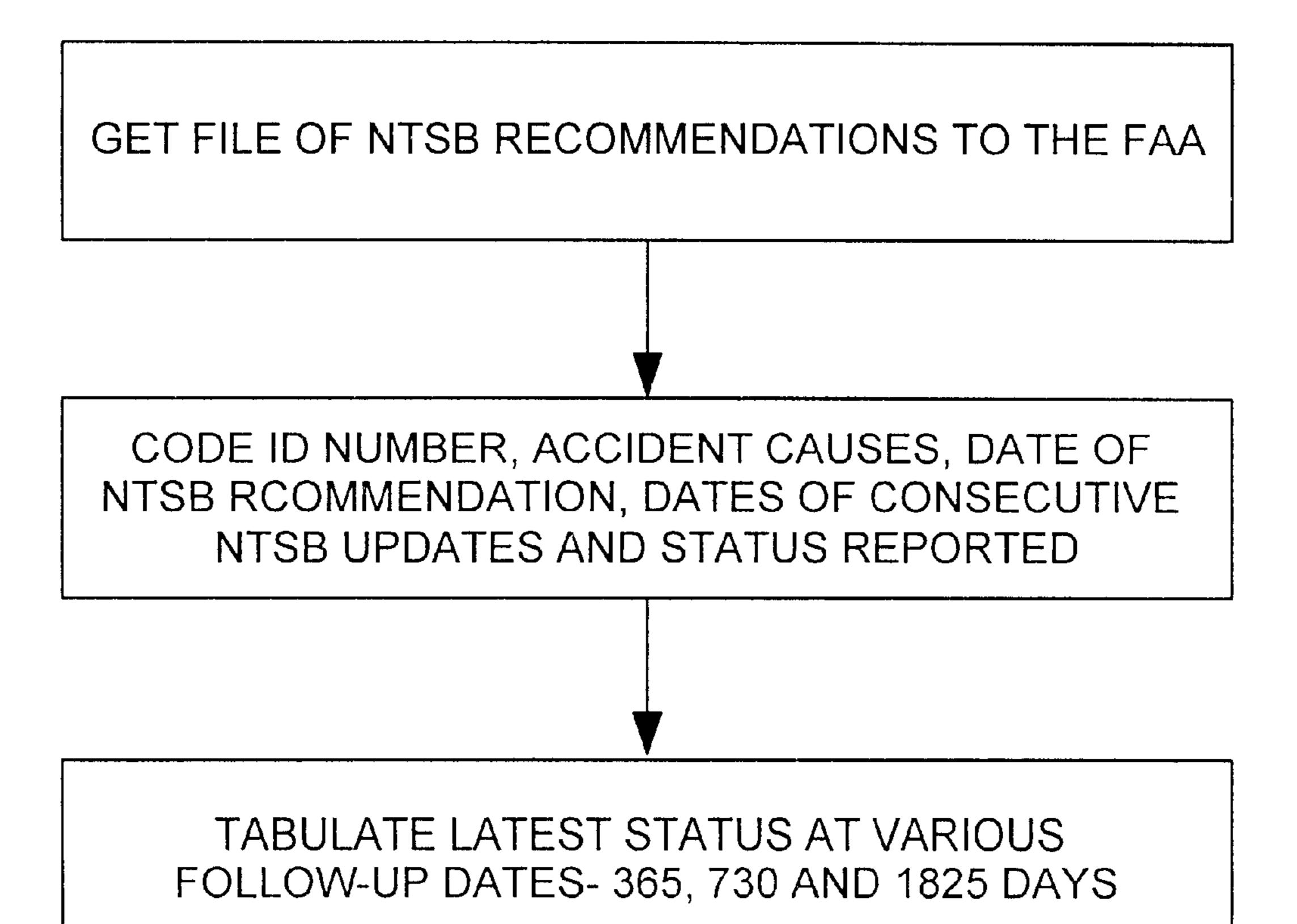


FIG. 13

GET TROHOC OUTPUT FILE (MODULE 320) FOR INDIVIDUAL AIRCRAFT

CALCULATE FOLLOWING RATIOS FOR EACH AIRCRAFT:

PEMULT- THE PROPORTION OF SDR'S WITH MULTIPLE RESPONSES FOR PRECAUTIONARY PROCEDURES AND FOR NATURE OF CONDITION SEVOCCUR- THE PROPORTION OF MORE SEVERE OCCURRENCES.

RATIO OF UNSCHEDULED LANDINGS TO ALL PRECAUTIONARY PROCEDURES

RATIO OF MORE SEVERE PART DEFECTS ("SEVERITY" VARIABLE)
TO ALL PART DEFECTS.

CDBIGNO- THE INTENSITY OF REPORTING MAINTENANCE "D" CHECKS. THIS VALUE IS DETERMINED FROM THE LARGEST NUMBER OF INSPECTION/MAINTENANCE REPORTS SUBMITTED IN ANY 60-DAY PERIOD

SDRLONGINT- THE LONGEST INTERVAL BETWEEN THE DATES OF TWO CONSECTIVE SDR'S

DAYSTOEND- THE NUMBER OF DAYS BETWEEN THE DATE OF THE LATEST SDR SUBMITTED AND THE END OF THE PERIOD ANALYZED PROPAC- THE PROPORTION OF ALL AIRCRAFT FOR WHICH ANY SDR IS SUBMITTED

CALCULATE STATISTICAL DISTRIBUTIONS FOR EACH OF THESE VARIABLES; DETERMINE CUT-OFF POINTS FROM THE TAILS OF THE DISTRIBUTIONS, ASSIGN SCORES, BASED ON THESE CUT-OFF POINTS, TO EACH OF THE 6 VARIABLES FOR EACH AIRCRAFT.

SUM SCORES FOR EACH AIRCRAFT

CALCULATE THE DISTRIBUTION OF SCORES FOR ALL AIRCRAFT FOR EACH OPERATOR

USING SCORE DISTRIBUTIONS, RANK EACH OF THE OPERATORS

FIG. 14

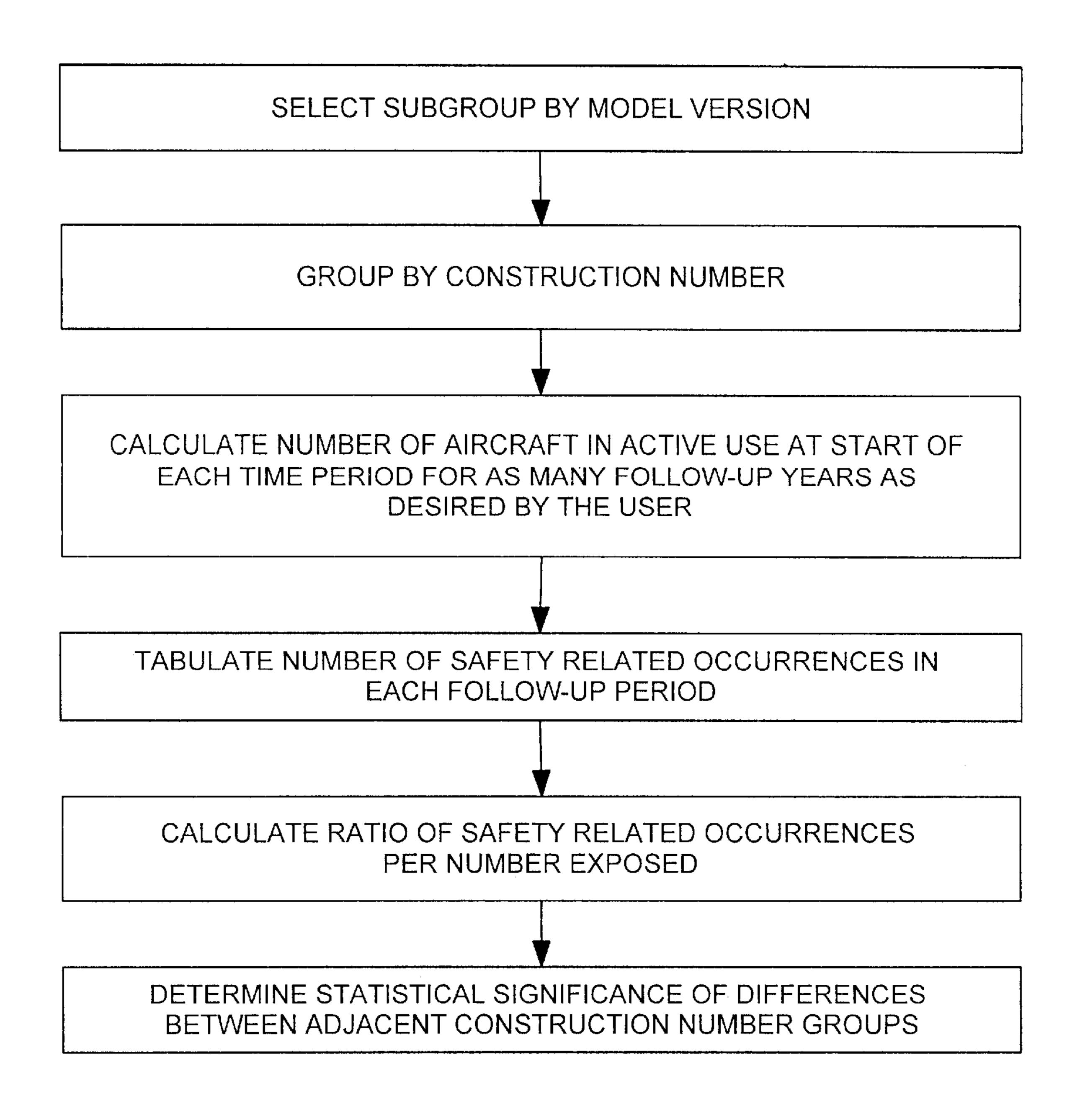
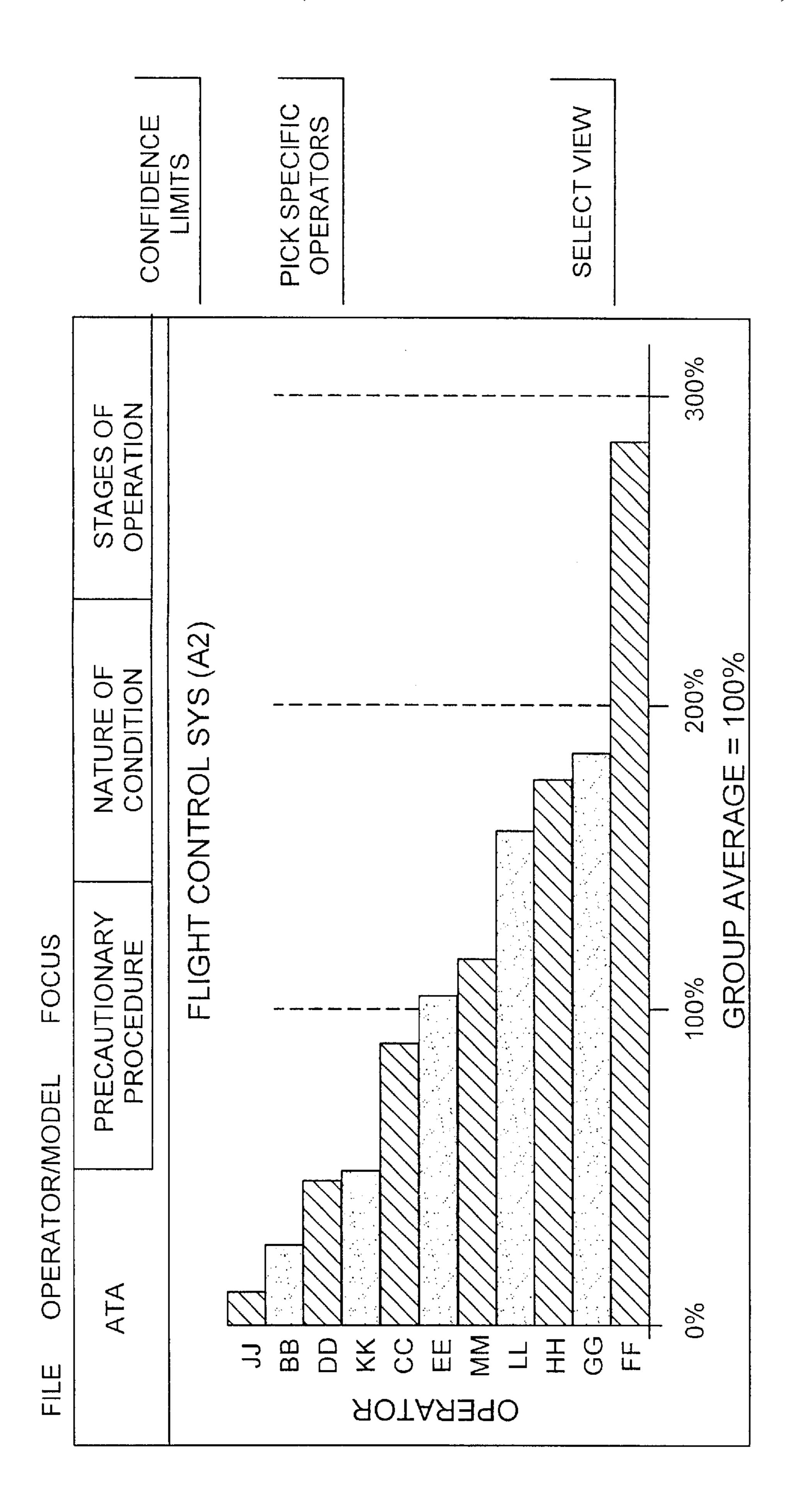


FIG. 15



**FIG.** 16A

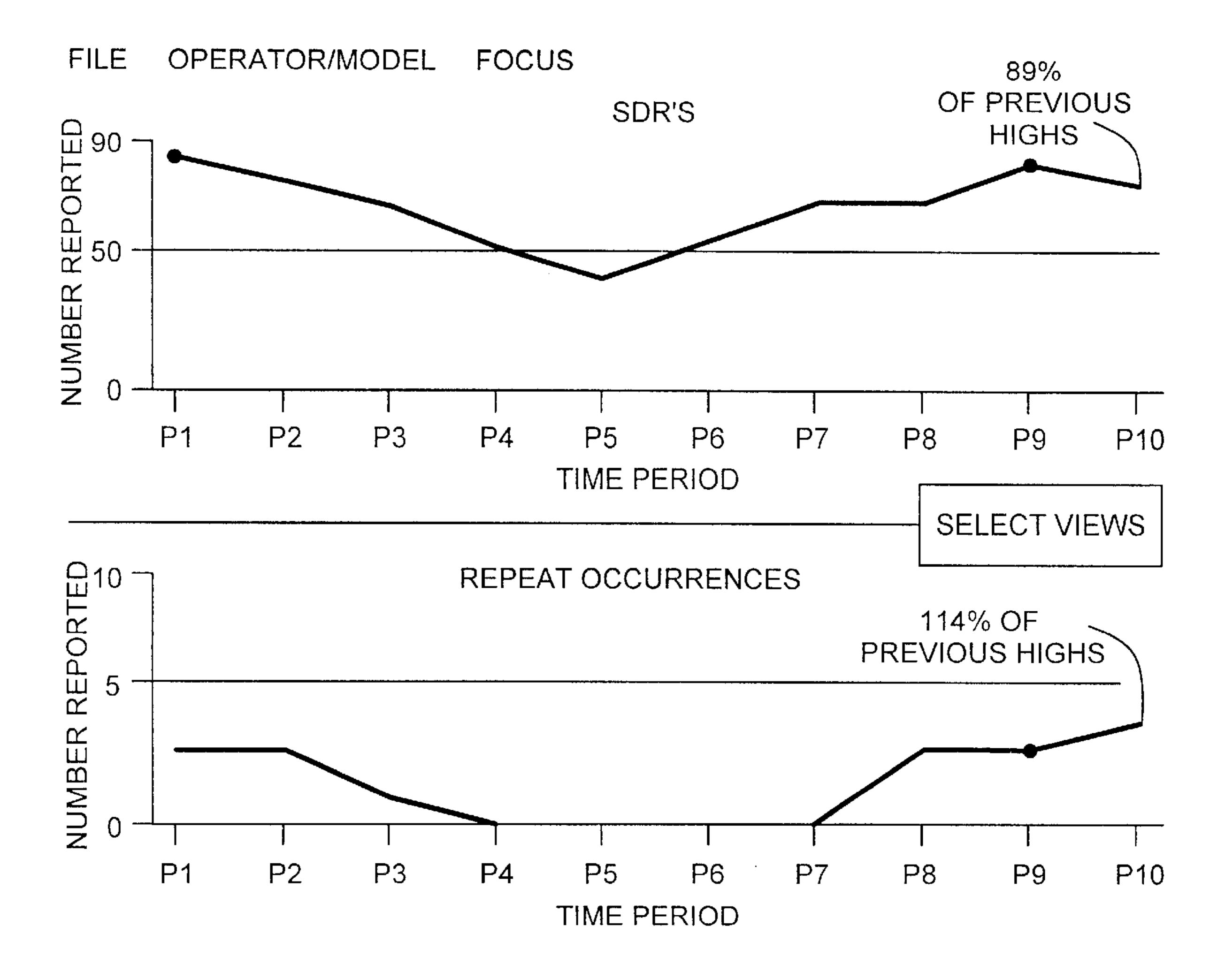
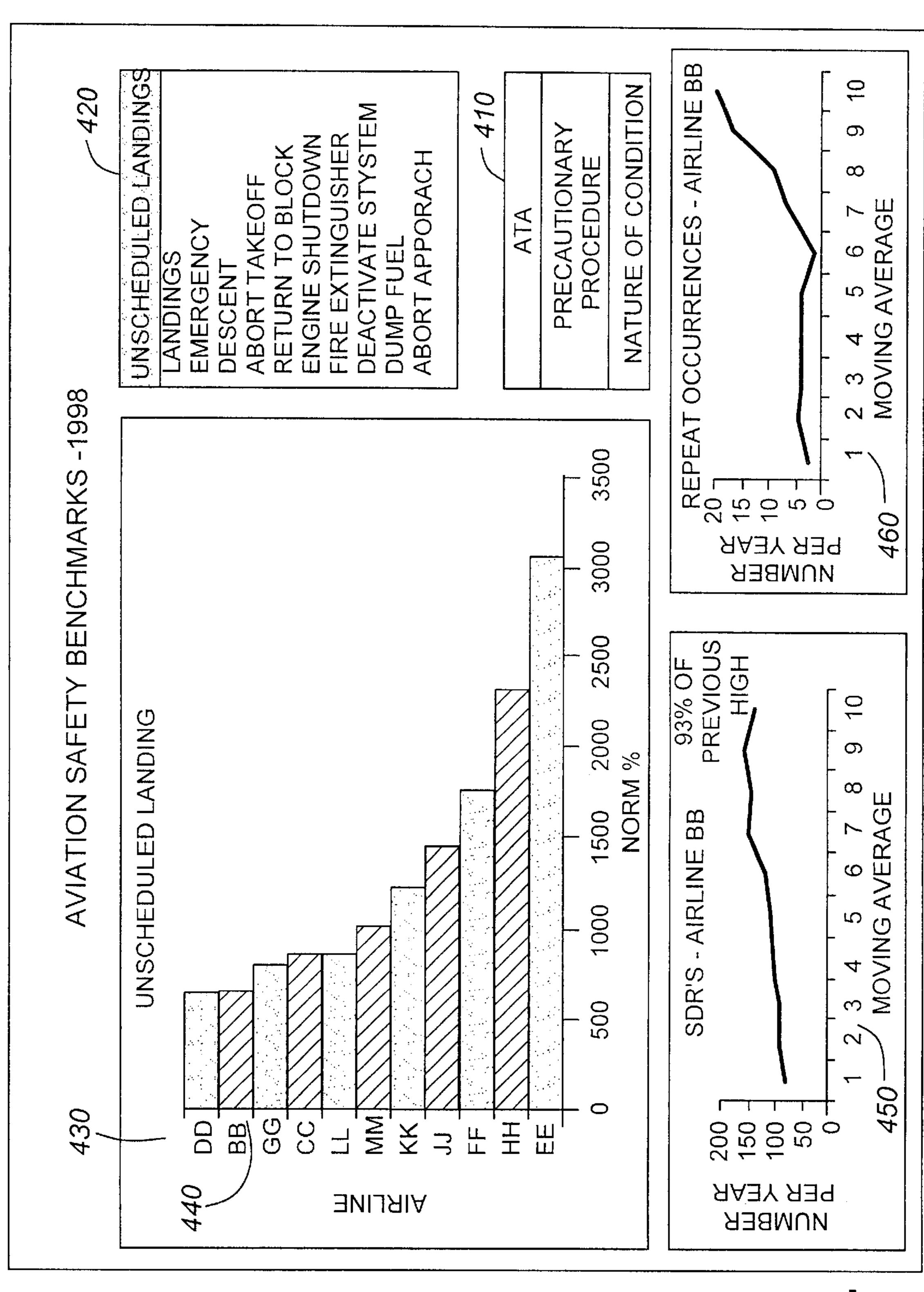


FIG. 16B



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# INFORMATION SYSTEM AND METHOD USING ANALYSIS BASED ON OBJECT-CENTRIC LONGITUDINAL DATA

#### TECHNICAL FIELD

This invention relates generally to a computerized information system and method using object-centric virtual database(s) and analyses which are advantageously suited to transportation safety applications, such as for the aviation industry (in which case the aircraft may be the unit of observation i.e. "object" referred to herein), as well as for health and other safety monitoring applications.

#### BACKGROUND

As is true for many transportation and health areas, the aviation industry generates a number of different reports relating to a variety of safety-related occurrences which are available to the public (either through the Internet or on 20 request to the source). Appendix A herein sets out a summary of several sources of these reports and the meaning of various terms used in the aviation industry in relation to reporting. Individual aircraft are identified in accident reports, incident reports, mechanical reliability reports of 25 failures (also referred to as service difficulty reports or "SDRs") and engine reliability reports. In some other available reports pertinent information is provided but not in relation to a specific aircraft. For example, as stated in Appendix A, reports in the Aviation Safety Report System 30 (ASRS) do not identify individual aircraft. In addition, Airworthiness Directives (and the automated exemption lists which may be provided therein) stipulate generally directed requirements for inspecting and/or fixing potential problems. Similarly, National Transportation Safety Board 35 (NTSB) recommendations to the Federal Aviation Authority (FAA) and their outcomes, as well as FAA enforcement data on regulatory violations, provide additional sources of safety-related occurrences. Examples of some other safetyrelated data types include pilot factors, airport and air traffic 40 management.

There is no automated means available to analyse multiple records from such publicly available sources for objects such as individual aircraft. Nor is there available any means for performing statistical analyses of trends and/or rates of non-accident occurrences (examples of which are SDRs, engine reliability reports and incidents) on such multiple records across various files let alone the same file for an individual object such as an aircraft. Consequently, the existing aviation safety systems cannot identify single aircraft with multiple problems and, because they do not identify rates, they do not provide any measure of risk which would permit a comparison of models or operators or identification of changes in trends.

Accordingly, there is a need for means to perform longitudinal analyses on individual safety/health-related objects (such as an aircraft or a patient) to enable a comparison of different groups of objects (such as aircraft models and operators). In particular, for example, there is a need for a means of standardizing public and/or private data providing non-accident safety-related information such as mechanical difficulties, as well as operational data, relating to aircraft and for analyzing such data on an object-centric basis to identify and predict potential problem areas and risks. The term "object" includes anything physical such as an aircraft, 65 nuclear reactor or other environmental hazard for which safety-related events are monitored as well as anything

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which is the subject of events such as a drug or treatment for which the events to be monitored may include reactions to the drug or a disease for which the events may include symptoms.

#### SUMMARY OF THE INVENTION

In accordance with the invention there is provided an information system for analyzing data comprised of events pertaining to an object class (e.g. aircraft). The system comprises a data access component configured for accessing a database comprising at least one set of data records (service difficulty reports) whereby each data record pertains to an event for an individual object (e.g. an individual aircraft) and comprises a plurality of data fields, at least one of the data fields comprising an identifier identifying one individual object. A user interface component receives userselected criteria for field values of the data fields. A selection component is configured for selecting those of the data records which correspond to the user-selected criteria. An analysis component is configured for analyzing the data records including identifying the selected data records according to a chronological sequence of events for individual objects covered by the selected data records, whereby each individual object covered by the selected data records is associated with a longitudinal set of data records, and deriving output data for the events from the longitudinal sets of data records.

Preferably, the data fields comprise a plurality of identifiers and a preprocessor component identifies multiple subdatabases from the selected data records whereby each sub-database comprises data records identified by one of the identifiers. Preferably, a plurality of sets of data records are provided, each being derived from a different source, whereby the sets of data records are integrated to form the database. The information provided by one set of data records may be used to supplement other data records of the database and the information provided by another one of the sets of data records may be used to correct erroneous data field values in data records obtained from any source.

A graphizer output component is preferably provided for graphically displaying the output data by multi-result interrelated graphs. The analysis component may be configured for identifying chronological trends for one or more individual objects and/or data records, for surveiling the events on a chronological basis for one or more individual objects, for standardizing the data reports and individual objects as between the different originating sources thereof (e.g. based traffic patterns for each individual aircraft and the types of aircraft models of each originating source), for comparing on a chronological basis and measured from a user-selected event the events pertaining to one individual object and the events pertaining to a user-selected comparison group of objects, for normalizing the field values for a set of data fields and outputting the resulting normalized data, for identifying related data records of the data records, for identifying a statistically high number of events for an individual object within a designated time period and/or for rating the originating sources of the data records on the basis of the diligence of the originating sources in reporting events.

#### BRIEF DESCRIPTION OF THE DRAWINGS

The invention will now be described in greater detail with reference to the attached drawings wherein:

FIG. 1 is a general block diagram of the main components of an information system in accordance with the present

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invention, the illustrated system being an aviation information system using aircraft-centric databases and analyses;

FIG. 2 is a further block diagram of the system shown by FIG. 1 with more detail shown therein with respect to the inputs and outputs of the main components;

FIG. 3 is a general flow chart diagram showing steps of the information system and method from the user's selection inputs to the presentation of the output tables and graphs;

FIGS. 4a through 4e together form a flow chart diagram of steps performed by the selection filter component 200 of the illustrated information system;

FIG. 5 is a flow chart diagram showing steps performed by the analysis modules 310, 320, 330, 340 and 360 for calculating repeats for an aircraft;

FIGS. 6a and 6b are flow chart diagrams showing steps performed by the SURVEIL module 310 and FIGS. 6c and 6d illustrate some ways comparison periods land moving averages could be constructed which in turn are analysed by the SURVEIL module 310;

FIGS. 7a, 7b, 7c and 7d are flow chart diagrams showing steps performed by the STANDARDIZATION module 320;

FIGS. 8a and 8b are flow chart diagrams showing steps performed by the TROHOC module 330;

FIG. 9 is flow chart diagram showing steps performed by the TABULATE module 340;

FIG. 10(a) is a flow chart diagram showing steps performed by the ALF module 350 and

FIGS. 10(b) and 10(c) display an exemplary descriptive  $_{30}$  listing of output produced by this module;

FIGS. 11a and 11b are flow chart diagrams showing steps performed by the CLUSTER module 360, with FIG. 11a providing an overall view of the process and FIG. 11b providing a more detailed view thereof;

FIG. 12 is a flow chart diagram showing steps performed by the HILO module 370;

FIG. 13 is a flow chart diagram showing steps performed by the RCO module 380;

FIG. 14 is a flow chart diagram showing steps performed by the DILIGENCE module 390;

FIG. 15 is a flow chart diagram showing steps performed by the COHORT module 395; and,

FIGS. 16(a)-16(c) are exemplary depictions of output 45 graphs produced by the graphizer component of the system, of which FIG. 16(a) is a single graphic output, FIG. 16(b) is a two-graph output and FIG. 16(c) is a three-graph output.

# DETAILED DESCRIPTION OF A PREFERRED EMBODIMENT

FIG. 1 shows, in general block diagram form, the main components of an aircraft-based (i.e. aircraft-centric) aviation information system and method in accordance with the present invention. In this embodiment the object class is 55 aircraft and an individual object is an individual aircraft. Events pertaining to individual aircraft are reported in service difficulty reports (SDRs), being data reports herein.

The components 100, 200, 300, 400 are in the form of computer programs. A preprocessor component 100 60 manages, massages, cleans, links and sorts the data provided by SDRs, and other safety-related data from accident/incident reports and other sources (e.g. engine reliability), into six master database files 150 (see FIG. 2), as detailed below, which are developed by the preprocessor 100 from 65 such data. A selection filter component 200 is user selected and controlled and includes a database access component for

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selecting aircraft and/or SDRs, according to user-specified criteria, from any one of the six master database files 150 for purposes of further analysis. The resulting selected data produces an intermediate (ie temporary) data file 250 which is used by an analyses component 300. The analyses component 300 comprises several analysis subcomponents 310, 320, 330, 340, 350, 360, 370, 380, 390 and 395 (FIG. 2) referred to herein as analysis modules. The modules 310, 320, 330, 340, 350, 360, 370 and 395 use the aircraft (referred to herein as "AC") as a basic unit of observation (i.e. the aircraft is the object) together with user inputted criteria to perform specific analyses. Analysis module 380 operates on the individual NTSB recommendations for which the courses and outcomes of FAA implementations are tracked. Analysis module 390 operates on the individual operators for which various statistical performance measures are used to derive an aggregate score. Two of the analysis modules viz, modules 330 and 350, produce longitudinal data files comprising one record per individual aircraft and, therefore, aircraft-based output. A graphizer component 400 outputs the results of analyses selected by the user and this output is provided in various formats including a multi-graphical display as shown, for example, by FIGS. 16(b) and (c).

SDRs form the workhorse from which most of the analysis modules derive results. In fact, each aircraft has a minimum of one SDR. Each SDR comprises the following fields: AC3 (Model, BTS Model and Opcode), SERAC (Serial number), NNUM (Nnumber), Kind, Year built, OTH (Opcode, Opcode and Model), DOD, DOR, ALT (BTS) Model, Model and BTS Model), Nature of Condition, Precautionary Procedure, Severity, Stage of Operation, ATA, Construction number, Engine, Crash report status and Incident report status. For illustration purposes, some of the values associated with some of these fields are listed in Appendix B herein. The preprocessor component 100 categorizes, sorts and identifies each SDR according to six different formats (each format corresponding to one of the six files of a master database 150) as illustrated below in Table 1 and each aircraft is thereby associated with a set of SDRs according to a given set of identifiers (i.e. per the first column of Table 1).

TABLE 1

ő	AC IDENTIFIED BY	AC CATE- GORIZED AND SORTED BY	AC3	ОТН	ALT
	MODEL, SERAC	MODEL, SERAC, DOD	MODEL	OPCODE	BTS MODEL
)	BTS MODEL, SERAC	BTS MODEL, SERAC, DOD	BTS MODEL	OPCODE	MODEL
	OPCODE, SERAC	OPCODE, SERAC, DOD	OPCODE	MODEL	BTS MODEL
	MODEL, SERAC, NNUM	MODEL, SERAC, NNUM, DOD	MODEL	OPCODE	BTS MODEL
5	BTS MODEL, SERAC, NNUM	BTS MODEL, SERAC, NNUM, DOD	BTS MODEL	OPCODE	MODEL
)	OPCODE, SERAC, NNUM	OPCODE, SERAC, NNUM, DOD	OPCODE	MODEL	BTS MODEL

The value sets associated with MODEL, BTS MODEL and OPCODE are listed in Appendix B herein. Each airplane's serial number is assigned by the aircraft's manufacturer and remains unchanged with the plane through its operational history. Its registration number (also known as the tail number) is assigned by the US Department of

Transportation. Since a plane can be owned or operated by different owners or operators over time, its registration number can change over time. Serial and registration number assignments for USA-registered aircraft also apply to non-USA registered aircraft.

The data gathered by the preprocessor component 100 may be obtained from public or private sources or a combination of both. The most important data gathered by it in the embodiment described herein are Service Difficulty Reports (SDRS) which are reports of each particular occurrence involving a mechanical difficulty pertaining to a particular aircraft. SDRs are legally required by the Federal Aviation Administration (FAA) for each and every safety related mechanical difficulty involving any USA-registered airplane. The seriousness of the mechanical difficulties reported in SDRs range from broken in-cabin passenger light bulbs at the low end of the seriousness spectrum to engine failures and cockpit smoke/fires at the other end. Appendix B herein lists some of the values for specific information fields provide by an SDR. The FAA makes SDRs readily available to the public.

For the system embodiment described herein the SDRs are compiled into a combined safety database of the preprocessor component 100. Historical SDR data was obtained from the FM's Oklahoma City office and current SDRs are downloaded directly from the FAA's website and incorporated by the preprocessor component 100 on a regular basis. The preprocessor component 100 adds certain aircraft data to each SDR such as the year it was built, its construction number and engine type all of which are stored and updated in a Census file maintained by the system.

The Census file is a master file containing validated information for aircraft (derived from SDRs and production lists), including the serial and registration numbers of the aircraft, and this information is used to validate new records received by the system. Other information kept in the Census file includes the manufacturer's construction number for the aircraft, the year in which the aircraft was built, the model land/or version identifier for the aircraft, any other identifying feature associated with the aircraft that may be desired to be tracked such as the type of wiring of the aircraft and activity history for the aircraft such as engine replacements.

A Change file keeps track of aircraft having inaccurate or garbled identifiers and their corrected identifiers. It also keeps track of inaccurate or garbled non-identifying data values and their corrected values. Other information may be added to the Census and Change datasets if desired.

In addition to gathering report data, the preprocessor component 100 harmonizes the fields in each SDR according to system definitions (model name and version as well as operator name, etc.), deletes duplicates, corrects errors and links (ie: matches) the SDR to aircraft identification data. Such verification is done by first comparing the identifier with those in the Change file and then with the Census file. Both the Change and Census datasets are updated each time a new SDR is integrated with a master SDR file of the master database 150.

Aviation Safety Report System (ASRS) reports are anonymized and unlike the SDRs or accident/incident reports, 60 do not identify the aircraft for which the report was generated (specifically, the operator and serial/registration numbers for the aircraft are not provided). For this reason the preprocessor component 100 maintains the ASRS data in a separate dataset from the master database 150.

Another smaller dataset is also maintained by the preprocessor component 100 of the illustrated embodiment,

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namely an Operator/Model Exposure and Costs (OMEC) dataset. The preprocessor component 100 tabulates departures, hours in flight, hours ramp to ramp, aircraft miles, passenger miles, aircraft days, aircraft census (mean number in service), costs and individual aircraft by age or other individual characteristic for each of the largest operators (this being about 100 operators in the embodiment described herein) and aircraft models using data collected by the Bureau of Transportation Statistics (BTS).

In addition, an Engine Reliability Dataset (ERD) is maintained by the preprocessor component 100 and this data is obtained from the ASY division of the FAA. The ERD dataset contains 1998 and 1999 engine types used for each aircraft model and by operator. It also contains engine hours and aircraft census as well as engine failures and engine changes. The ERD has two uses. Firstly, it gives more complete data on engine failures than SDRs and secondly, it gives aircraft census monthly numbers for number of aircraft and flight hours by model/version by operator.

Other public and/or private databases may be integrated into the preprocessor component 100 if and as desired. For example, private data such as Flight Oriented Quality Assurance (FOQA) data consisting of the output from the Flight Data Recorders (FDRs) (providing more data on engine performance and pilot behaviour) could be incorporated. Such added data would then be integrated into the system for use by component 300.

The selection filter component 200 selects a dataset from the database maintained by the preprocessor component 100 on the basis of user-specified criteria. The user-selectable criteria provided in the illustrated embodiment are the following:

- 1. Aircraft report history—permits report activity within five specified time periods and by minimum/maximum number of reports in each period.
- 2. Presence or absence of SDRs before the minimum date of the first specified time period from the first criteria above
- 3. Presence or absence of SDRs after the maximum date of the first specified time period from the first criteria above
- 4. Model—for example, DC9, B737, etc.
- 5. Version—for example, DC9-30, B737-300, etc.
- 6. Manufacturer—for example, Boeing, Airbus, etc.
- 7. Specific operators—for example, United Airlines, USAir, American Airlines, etc.
- 8. Operator group—major, national, regional or other.
  - 9. SDR's—only for specified operator or all SDR's for the aircraft.
  - 10. Type
  - 11. Nature of condition
- 12. Precautionary procedure
  - 13. Severity
  - 14. Stage of operation
  - 15. select 2 char ATA code on SDR basis
  - 16. select 4 char ATA code on SDR basis
- 17. select AC with user selected 2 char ATA code (and accept all other SDRs in AC)
- 18. select AC with user selected 2 char ATA code (and reject all other SDRs in AC)
- 19. select AC with user selected 4 char ATA code (and accept all other SDRs in AC)
- 20. select AC with user selected 4 char ATA code (and reject all other SDRs in AC)
- 21. minimum/maximum Year built
- 22. minimum/maximum time between DOD and DOR
- 65 23. Select all SDRs on same day or select first SDR of similar SDRs occurring on same day
  - 24. Construction number range

- 25. Crash report(s) status
- 26. Incident report(s) status
- 27. Engine manufacturer and model
- 28. Serial number range
- 29. Wiring type

Users may modify the twenty nine (29) criteria listed above through the user interface. Criteria number one allows users to compare different aircraft with similar report histories. This is pertinent because some aircraft are not in continuous use during the entire time period of observation. 10 An airplane may be withdrawn from use, stored or sold outside of the USA in which case SDRs are not legally mandated for any mechanical problems it may encounter. This first criteria provided to the user allows the user to make comparisons on the level or intensity of SDR activity 15 by specifying a minimum and maximum number of SDRs for any user definable arbitrary period of time.

Criteria numbers two and three presented to the user through the user interface allow the user to select or ignore aircraft with SDR activity before the first period's minimum 20 date or after the same period's maximum date. The same two criteria also allow the user to select the aircraft regardless of any SDR activity before the first period's minimum date or after the same period's maximum date.

A text analysis and a keyword search are also provided as 25 user definable (not selectable) fields via the user interface. Of course, different and/or additional selectable and definable fields could be provided by the system if desired. A general flow chart diagram showing steps of the information system and method from the user's selection inputs to the 30 presentation of the output tables and graphs is provided by FIG. 3.

The selection filter component **200** creates an intermediate (i.e. temporary) data file containing a collection of aircraft all meeting an input set of user-specified criteria. A 35 user interface component (e.g. a graphical user interface (GUI)) receives, from the user, a set of selectable criteria as set out above (i.e. items listed as 1–29 and the text analysis and a key word search) which the selection filter component **200** uses to select the specific file of the master database **150** which is linked to such aircraft and also the aircraft and SDRs meeting such criteria. The selection filter compiles the resulting data identified from the user-specified criteria into an intermediate data file **250**. FIGS. **4***a* through **4***e* together form a flow chart diagram of the steps performed by the 45 selection filter component **200** of this embodiment.

The analysis modules 310–370 and 395 each include a longitudinalizer function which gathers all the SDRs having the same identifying data, per column 1 of Table 1, into virtual storage. At this point the analysis module (i.e. one of 50 310–370 or 395) has a virtual aircraft with virtual SDRs, representing a chronological sequence of events for an individual aircraft. The analysis module then analyses that virtual aircraft and temporarily stores the results of the analysis in memory. The virtual aircraft with its virtual 55 SDRs are then cleared from the computer's memory and the program module repeats the cycle by gathering the SDRs for the next aircraft. The analysis results for each aircraft are collated or compared and when all of the aircraft in the intermediate file 250 have been analysed accordingly the 60 analyses of all the aircraft are output into another intermediary file 397 to be presented to the user by means of the graphizer 400.

The analyses component 300 includes a GUI which allows the user to select values for variables which are 65 specific to each analysis module which is to perform. The user selects the analysis modules which the user wishes to

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run and they are run (performed) in the system in numerical order from module 310 to module 395. Each of the analysis modules 310 through 370 and 395 analyses the intermediary file 250 produced by the selection component 200. These modules (also referred to as subcomponents) use the aircraft (AC) as the basic unit of observation and two of these also produce aircraft-centric output as illustrated in Table 2 below.

TABLE 2

SUB COMPONENT	SUBCOMPONENT NAME	AC UNIT OF OBSERVATION	AC BASED OUTPUT
310	SURVEIL	Yes	No
320	STANDARDIZATION	Yes	No
330	TROHOC	Yes	Yes
340	TABULATE	Yes	No
350	ALF	Yes	Yes
360	CLUSTER	Yes	No
370	HILO	Yes	No
380	RCO	No	No
390	DILIGENCE	No	No
395	COHORT	Yes	No

Analysis modules 310–370 and 395 are based on analysing variates which simply put, are specific values for specific fields. For example, the field TYPE has four variates, TYPE=1, TYPE=2, TYPE=3 and TYPE=4. There are two kinds of variates, namely, univariates and bivariates which are detailed below.

Univariates: These fields, which include Type, Nature of Condition, Precautionary Procedure, Stage of Operation, Severity, Year Built, AC3, OTH, ATA2 Char and ATA4 Char for this embodiment, are analysed by calculating aircraft-centric measurements for specific values of these specific fields.

Bivariates: These fields, which include Nature of Condition by ATA, Precautionary procedure by ATA, Severity by ATA, Precautionary Procedure by Nature of Condition, ATA by Nature of Condition, ATA by Precautionary Procedure, ATA by Severity and Nature of Condition by Precautionary Procedure for this embodiment, are analysed by calculating occurrences for specific values of these specific field pairings. Table 3 below identifies the analysis modules which use the univariates and bivariates.

TABLE 3

	SUB- COMPONENT	SUBCOMPONENT NAME	UNIVARIATE	BIVARIATE
)	310	SURVEIL	Yes	Yes
	320	STANDARDIZATION	Yes	Yes
	330	TROHOC	Yes	Yes
	340	TABULATE	Yes	Yes
	350	ALF	No	No
	360	CLUSTER	Yes	Yes
5	370	HILO	Yes	Yes
	380	RCO	No	No
	390	DILIGENCE	No	No
	395	COHORT	Yes	Yes

Aircraft-centire measurements include calculating SDRs, ACs, repeats and intervals for variates (either univariates or bivariates or both according to the user's selections). An aircraft's SDR count is a simple count of the number of SDRs the aircraft has for each variate. Table 4 below illustrates five SDRs all with the same identifying data (as per the first column of Table 1) which are linked to the same virtual aircraft (e.g. referred to herein as aircraft 1 or "AC1")

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by a longitudinalizer function. An analysis module then performs aircraft-centric measurements (SDRs, ACs, repeats and intervals) on this virtual aircraft.

The first aircraft-centric measurement involves SDR measurements. The SDR measurement is calculated for each 5 variate by counting the number of SDRs in the aircraft with that particular variate. For example, AC1 has three SDRs with TYPE=2 but no SDRs with TYPE=4.

The second aircraft-centric measurement also involves aircraft measurements.

The aircraft measurement is calculated for each variate by assigning the value one (1) for each variate for any number of SDRs with that particular variate. For example, AC1 will register one (1) AC for each of TYPE=1, TYPE=2 and TYPE=3 but none for TYPE=4.

Each of Tables 4 and 5 below show data linked to one specific aircraft.

TABLE 4

SE Se	OR quence	TYPE
1 <sup>st</sup> 2 <sup>nd</sup> 3 <sup>rd</sup> 4 <sup>th</sup> 5 <sup>th</sup>	d	1 2 3 2 2

TABLE 5

	SDR count	AC count
TYPE 1	1	1
TYPE 2	3	1
TYPE 3	1	1
TYPE 4	0	0

Tables 6 and 7 show how the SDR and AC counts change when the system examines two aircraft instead of one.

TABLE 6

AC ID	SDR Sequence	TYPE	
$1^{st}$	$1^{st}$	1	
$1^{st}$	$2^{\text{nd}}$	2	
$1^{\mathrm{st}}$	$3^{rd}$	3	
$1^{st}$	$4^{ ext{th}}$	2	
$1^{st}$	$5^{ ext{th}}$	2	
$2^{\mathbf{nd}}$	$1^{st}$	1	
$2^{\mathbf{nd}}$	$2^{\text{nd}}$	2	
$2^{\mathbf{nd}}$	$3^{rd}$	1	
2 <sup>nd</sup> 2 <sup>nd</sup>	4 <sup>th</sup>	4	

TABLE 7

	SDR count	AC count	55
TYPE 1	3	2	
TYPE 2	4	2	
TYPE 3	1	1	
TYPE 4	1	1	60

The third aircraft-centric measurement involves repeats. A repeat occurs when two SDRs having the same variate (examples of this are where two SDR's have the Variate TYPE=1 or where both have the variate PRECAUTION- 65 ARY PROCEDURE=4) have DODs (dates of difficulty) within a user-defined lapsed time, meaning that the SDRs in

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question are not far apart on a time line. No repeat is considered to have occurred where two SDRs have close DODs but different variates (eg where one has the variate TYPE=1 and the other has the variate TYPE=3). It is possible for a repeat to occur between an aircraft's third SDR and its eleventh SDR, for example, and this would most likely occur where an aircraft has many SDRs clustered together in a short time. To be considered a repeat the SDRs themselves are not required to be consecutive because a repeat is a measure of time and not the order of an aircraft's SDRs. FIG. 5, in the form of a flow chart, illustrates the steps performed by the analysis modules 310, 320, 330, 340 and 360 for calculating repeats for an aircraft.

The analysis component 300 permits the user, by means of the GUI, to define a minimum and maximum number of days (referred to herein as RPTX and RPTY, respectively) between the DODs for two SDRs in order to be considered a repeat. Two SDRs having DODs which are less than RPTX days apart or more than RPTY days apart are not considered to constitute a repeat. By default, in the preferred embodiment, two SDRs having DODs of the same day are defined to be exactly one day apart. However, the user may change this definition (by means of the GUI in SELECTION component 200) so that such SDRs are defined as being zero days apart. The latter definition would enable a user to distinguish between two SDRs having DODs occurring on the same day from two SDRs whose DODs occur on consecutive days.

The fourth aircraft-centric measurement involves intervals. The definition of an interval for a particular variate is defined to be the shortest repeat (applying the same boundaries as those set for repeats viz. more or equal to RPTX and less than or equal to RPTY) for that variate. As aircraft are analyzed, one by one, for a certain time period, the intervals are not accumulated. Instead, the system keeps track of the shortest interval for all aircraft for each variate.

Table 8 below illustrates some exemplary field data for an aircraft with five SDRs.

TABLE 8

SDR Sequence	DOD	TYPE	Nature of Condition	Precautionary Procedure
$1^{\mathrm{st}}$	<b>M</b> ay 01 1998	1	3	1
$2^{\mathbf{nd}}$	May 03 1998	2	3	1
$3^{rd}$	May 07 1998	1	4	1
$4^{th}$	May 14 1998	3	4	1
5 <sup>th</sup>	May 25 1998	3	4	2

For the hypothetical aircraft data of Table 8 the data calculated for the univariates, for RPTX=2 and RPTX=15, is shown in Table 9 below.

TABLE 9

UNIVARIATE	SDR	AC	REPEAT	INTERVAL
TYPE 1	2	1	1	6
TYPE 2	1	1	0	0
TYPE 3	2	1	0	0
TYPE 4	0	0	0	0
NAT. COND. 1	0	0	0	0
NAT. COND. 2	0	0	0	0
NAT. COND. 3	2	1	1	2
NAT. COND. 4	3	1	1	7
PREC. PROC. 1	4	1	3	2
PREC. PROC. 2	1	1	0	0
PREC. PROC. 3	0	0	0	0

Data calculated for an exemplary bivariate, for RPTX=2 and RPTY=15, for the aircraft of Table 8 is shown in Table 10 below.

BIVARIATE	SDR	AC	REPEAT	INTERVAL
PRE PROC 1 NAT COND 1	0	0	0	0
PRE PROC 1 NAT COND 2	0	0	0	0
PRE PROC 1 NAT COND 3	2	1	1	2
PRE PROC 1 NAT COND 4	2	1	1	7
PRE PROC 2 NAT COND 1	0	0	0	0
PRE PROC 2 NAT COND 2	0	0	0	0
PRE PROC 2 NAT COND 3	0	0	0	0
PRE PROC 2 NAT COND 4	1	1	0	0

Table 11 below shows which analysis modules use which aircraft-centric measurements (SDRs, ACs, repeats and intervals).

TABLE 11

ANALYSIS MODULE	SDR	AC	REPEAT	INTERVAL
310 - SURVEIL	Yes	Yes	Yes	Yes
320 - STANDARDIZATION	Yes	Yes	Yes	Yes
330 - TROHOC	Yes	Yes	Yes	Yes
340 - TABULATE	Yes	Yes	No	No
350 - ALF	No	No	No	No
360 - CLUSTER	Yes	Yes	Yes	Yes
370 - HILO	Yes	Yes	No	No
380 - RCO	No	No	No	No
390 - DILIGENCE	No	No	No	No
395 - COHORT	Yes	Yes	No	No

The analysis modules 310–395 will now be described. The SURVEIL module 310 produces and analyses time trends to compare events within a user-specified standard period (SP) and at least one user-specified comparison periods (CP). The standard period starts on a user specified calendar date and may be as long as the user wishes. The time element may be measured in days, weeks, months or years. All standard and comparison periods have the same length. The specified number of comparison periods can vary and, at one extreme, could cover the duration of the whole dataset from database 150 selected by the user via the user interface in selection filter 200. As an example, five years of comparison periods could comprise 5 periods of 365 days, 20 periods of 91 days or 60periods of 30 days (usually preceding the standard period).

A variate is defined to be exceptional if the aircraft-centric measurement for that variate is higher in the standard period compared to a user-defined minimum number of comparison periods. In other words, the variate's aircraft-centric measurement is "exceptionally high" (from the perspective of 50 the standard period). An exceptional aircraft is considered to be an aircraft identified in the standard period and having any such exceptional variate. A variation of this exceptionality determination involves using the term "lower" instead of "higher" in the aforementioned description. In other 55 words, for this variation the program searches for exceptionally low aircraft-centric measurements of variates (from the perspective of the standard period). The GUI interface of the analyses component 300 allows the user to toggle between these two exceptionality rules. The flow chart of 60 FIG. 6a shows the steps performed by the surveil module 310 to count univariate and bivariate fields and the flow chart of FIG. 6b shows the steps performed to identify exceptional univariates and bivariates. With reference to FIGS. 6a and 6b, FIG. 6c illustrates the manner by which 65 moving averages are formed from comparison periods and FIG. 6d illustrates the relationship between standard and

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comparison fields. In FIGS. 6a-6c a moving average is a user defined multiple (minimum of two) of consecutive comparison periods.

With respect to two adjacent comparison periods, a system default dictates that the earlier period finishes one day before the later period starts. The user may override this default via the GUI interface of the analyses component 300 so that the earlier period finishes exactly the same day the later period starts. In such latter case, however, the same SDR may be counted twice, in two different comparison periods, particularly if the SDR straddles the two periods. Another available option is to stagger the comparison periods so that there is an overlap between one comparison period and the next.

A comparison between variates is made as follows. A distribution, by either model or operator, of the foregoing univariate or bivariate tabulations is determined for output. Various statistical calculations are made on four aircraft-centric measurements (SDRs, Acs, repeats and intervals) for each variate. The SURVEIL module 310 also tabulates for output the activity history for aircraft and aircraft with reports in a standard period for variates not seen in any comparison period.

The STANDARDIZATION analysis module **320** standardizes SDR and aircraft data as between different operators, taking into account the differences in model mix and traffic pattern for each operator using an algorithm according the following. For leach time period the following steps are performed:

- 1. Normalizing data based on traffic patterns (e.g. miles, departures, aircraft hours, aircraft days or ramp to ramp hours) is tabulated in a 2-dimensional array (with subtotals for each column or row stored in appropriate array cells);
- 2. For each univariate/bivariate characteristic observed data (SDR, AC or REPEAT) is tabulated in a 2-dimensional array (operator by model) and subtotals for each column or row are stored in appropriate array cells;
- 3. For each operator and for each univariate, an aircraftcentric measurement is calculated as follows:
  - i) calculate ratio (R) of normalizing data for each model divided by normalizing data for all models; and,
  - ii) for each model, expected variate=R multiplied by the observed variate for all operators;
- 4. For each operator and for each variate characteristic the priority and statistical confidence limits are calculated as follows:

priority=observed—expected

lower confidence=(multiplier—square root of observed)<sup>2</sup>

upper confidence=(multiplier+square root of observed)<sup>2</sup>

lower confidence limit=lower confidence/expected upper confidence limit=upper confidence/expected

The multiplier is adjusted for multiple comparisons in accordance with the well-known Bonferonni method.

5. The foregoing calculations and results are output in tabular and/or graphical format.

FIGS. 7a, 7b, 7c and 7d are flow chart diagrams showing the steps performed by the STANDARDIZATION module 320 for standardizing variates for measurements of SDR, AC, repeats and intervals. The time period of interest used by this module is the same as the first time period of the selection filter component 200. If desired, however, the illustrated algorithm could be modified to cover more time

periods (such as if, for example, the user is interested to perform a standardization over a longer period of time say two or more years). As shown by these flow charts module 320 performs a standardization process for standardizing all AC3 values (i.e. for all eleven major airlines) and all OTH values (i.e. models). If desired, these steps may be modified to standardize using a specific subset of ACs (as in standardizing using nine major airlines, not all eleven) and/or a specific subset of OTH. These flow charts show the steps of standardizing AC3 based on OTH data and this, also, may be 10 modified if desired to standardize OTH based on AC3 data (in which case all references to AC3 and OTH in the flowchart would be replaced by OTH and AC3, respectively). The STANDARDIZATION module 320 can also be applied to normalize other data such as the age of an 15 aircraft fleet.

The TROHOC module 330 compares the performance of various aircraft models and peer aircraft (defined as aircraft of the same model which were manufactured in the same time period). The name "trohoc" used by the inventors for 20 this component is derived from the word "cohort" in that it is the reverse spelling of "cohort" and the reverse of the cohort process. A cohort is a group that is assembled together and followed forwards (into the future) over a period of time. For example, a cohort of patients can be 25 followed through a course of treatment starting from the present and be monitored over time into the future. A trohoc, as referred to herein, is this cohort process performed in reverse. Specifically, the trohoc process provides that after a specified event, the previous histories of those objects which 30 are affected by or involved with the event are compared with the previous histories of those objects which were not affected by or not involved with the same event.

The TROHOC module 330 compares SDR data between one aircraft and a group of peer aircraft and the comparison 35 group are of the same model/version, age and operator, when numbers permit. For accident purposes, activity is considered for data from the earliest SDR for the accident aircraft until the date of the accident itself. The comparison group is selected on the basis of having SDRs before the earliest SDR 40 for the accident aircraft and SDRs after the date of the accident. SDR reports are analysed for the same time period i.e. between the date of the earliest SDR for the accident aircraft and the date of the accident and, therefore, there is exactly the same time exposure for the accident aircraft and 45 the comparison group.

As illustrated in FIGS. 8a and 8b, the TROHOC module 330 calculates rank and percentile values for each aircraft for each variate relative to all aircraft. An exceedance percentage value is selectable by the user, for example 75%, and this 50 value identifies the number of variates for which each aircraft exceeds other aircraft. For an individual aircraft a determination is made of the number of variates above the selected exceedance value and the number of exceedances calculated for that aircraft are output for the user whereby 55 the number of exceedances for a specific aircraft is provide for statistical comparison with an exceedance distribution calculated for the comparison group. Such comparisons can also be applied to non-accident situations (e.g. for a comparative assessment of one model with others or of an 60 aircraft proposed for purchase with others).

Using the outputs of the SURVEIL and TROHOC modules 310, 330 statistical norms are determined. The user specifies the level for which norms are desired. For example, the norm may be that achieved by 90% or less of aircraft or 65 x are user-definable and selectable. the norm may be that achieved by 99% of aircraft. The highest values found in the user-defined percentage of

aircraft are determined for each variate for specific aircraft models: (i) the number of SDR's per year; (ii) the mean time between occurrences; and, (iii) the number of repeat occurrences in a user-specified number of days (for example, 30 days). These norms are useful to provide benchmarks for identifying 'exceptional' aircraft.

The TABULATE module 340 is provided to enable any operator (whether US or foreign) to apply U.S. statistical norms to identify exceptional aircraft in its own fleet of aircraft. This module provides authoritative data for assessing the mechanical reliability of an operator's fleet and provides confidentiality in that a foreign operator may utilize this module without having to divulge its own data on occurrences of mechanical problems, defects or malfunctions affecting safety. As illustrated by FIG. 9 the steps performed by this module include the following:

- 1. Input data is received from the user (through the GUI), namely, the number and type of models in the user/ operator's fleet and the level of statistical rarity sought.
- 2. The Service Difficulty Reports for the same models are analysed.
- 3. A data file is generated from the analysis for that user/ operator.
- 4. Further input data is received from the user/operator in the form of a filled-in table for each of the variates in which they have interest.
- 5. The values expected for each variate are calculated from the US statistical norms (as identified above the values are the number of SDR's per year, the mean time between occurrences and the number of repeat occurrences in a user-specified number of days such as 30 days).

The ALF module **350** identifies related SDRs on the basis of user-specified criteria including kind, nature of condition, precautionary procedure, severity, stage of operation, the time between DOD and DOR, year built, ATA 2 CHAR, ATA 4 CHAR and a key word search. SDRs meeting the userspecified criteria are classified as related SDRs while those that do not are considered to be unrelated. An episode is defined by the module to be a collection of related SDRs for which each pair of subsequent related SDRs are a given number ("x") of days apart or less (and the "x day" period is referred to as a demarcation period). The length of a demarcation period is measured by user selected time units (TU's) by days, weeks, months, quarters and/or years. The demarcation periods (also referred to herein as windows) have various starting and ending points and hence various lengths. All windows are user definable and can have "a" number of time units (TU) in days, weeks, months, quarters or years, "b" number of SDRs, "c" number of related SDRs (i.e. excluding unrelated SDRs), "d" number of unrelated SDRs (i.e. excluding related SDRs), "e" number of episodes up to and including the day immediately before the xth episode starts, "f" number of episodes up to and including the day the xth episode starts, "g" number of episodes up to and including the day the xth episode ends. Further, a window can start or finish at the following points (for any of the foregoing lengths): an arbitrary date, nth SDR, m TU before/after nth SDR, nth last SDR, m TU before/after nth last SDR, nth SDR before/after an arbitrary date, m TU before/after nth SDR before/after an arbitrary date, mth SDR of the nth episode, mth SDR of the nth episode before/after an arbitrary date, mth SDR of the nth episode before/after an arbitrary date, k TU before/after mth SDR of the nth episode, k TU before/after mth SDR of the nth episode before/after an arbitrary date, whereby the values of the dates, k, m, n and

As illustrated by FIG. 10(a) the ALF module 350 performs analyses on the basis of SDRs, ACs, episodes and

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**15** 

windows. FIGS. 10(b) and 10(c) display a descriptive listing of sample output produced by the ALF module.

The CLUSTER module 360 identifies a statistically unusually high number of mechanical problems and/or incidents within relatively short periods of time, the length of 5 which is user defined and specified, the steps performed by this cluster being illustrated by FIGS. 11a and 11b. A cluster is defined to be a number of SDRs (minimum of one) with the same variate where each SDR is no more than a user selectable number of days away from its subsequent SDR. For example, if each cluster is defined to be three days long, the aircraft covered by Table 12 below would have the clusters shown in Table 13 below. For the data of this example, the module determines and outputs "May 05–07, 1998" as the aircraft's cluster since this particular cluster contains three SDRs and three is more than any other cluster. <sup>15</sup> Similarly, the module determines and outputs for each uni/bivariate the cluster having the highest number.

TABLE 12

SDR Sequence	SDR DOD	20
1 <sup>st</sup> 2 <sup>nd</sup> 3 <sup>rd</sup> 4 <sup>th</sup> 5 <sup>th</sup>	May 01, 1998 May 02, 1998 May 05, 1998 May 06, 1998 May 07, 1998	25

TABLE 13

Cluster (each 3 days long)	Number of SDRs
May 01–03, 1998	2
May 02–04, 1998 May 03–05, 1998	1 1
May 04–06, 1998 May 05–07, 1998	2 3
May 06–08, 1998	2
<b>M</b> ay 07–09, 1998 <b>M</b> ay 08–10, 1998	1 0

The HILO module **370** determines from all aircraft in the intermediary file 250 the high and low numbers of SDRs over relatively long periods of time, and the steps performed by this module are illustrated by FIG. 12.

The National Transportation Safety Board (NTSB) inves- 45 tigations of accidents and major incidents result in recommendations. A cumulative database of NTSB recommendations to the FAA since 1963 is available to the public and this database includes the date of the occurrence, the date and identifying number of the recommendation, the date and 50 content of the FAA responses, the NTSB assessments of the response as acceptable or unacceptable and whether the NTSB considers the issue open or closed. This database covers all recommendations and their status at successive periods (i.e. 1, 2 and 5 years). The RCO module 380 55 analyses this data and outputs, in a standardized format, the time trends relating to the numbers of recommendations and also the status of the recommendations (either open or closed and satisfactory or unsatisfactory). This module also systematically compares short term and long term results. 60 FIG. 13 is a flow chart illustration of steps performed by the RCO module **380**.

Operators vary in the extent and nature of the problems they report and this is referred to herein as being variability in the diligence of reporting. The DILIGENCE module 390 65 systematically quantifies the diligence of operators. The following is a description of the operation of this module:

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- 1. The source data used by this module is a TROHOC output file (i.e. produced by the TROHOC module 320) which summarizes the reports for each aircraft.
- 2. The module considers six variables for each aircraft which reflect the operator's level of carefulness and regularity in preparing SDR's:
  - (a) PEMULT—the extent of multiple responses. The report form has 4 fields for Precautionary Procedures and 3 fields for Nature of Condition. More diligent operators will tend to complete more than one field.
  - (b) SEVOCCUR—the proportion of more severe occurrences. More diligent operators will tend to submit not only more severe occurrences but less severe occurrences. There are two components:
    - i) Ratio of unscheduled landings to all precautionary procedures.
    - ii) Ratio of more severe part defects (detailed in the "Severity" variable) to all part defects.
  - (c) CDBIGNO—the intensity of reporting maintenance checks. The "D check" is a major check every three-and-a-half years during which aircraft are essentially taken apart and put back together with many components replaced. This value is determined from the largest number of inspection/maintenance reports submitted in any 60-day period.
  - (d) SDRLONGINT—The longest interval between the dates of two consecutive SDR's.
  - (e) DAYSTOEND—The number of days between the date of the latest SDR submitted and the end of the period analyzed.
  - (f) PROPAC—The proportion of all aircraft for which any SDR is submitted.
- 3. The statistical distributions for each of these variables are calculated and cut-off points are determined from the tails of the distributions. Scores, based on these cut-off points, are assigned to each of the 6 variables for each aircraft.
- 4. The scores for each aircraft are summed.
- 5. The distribution of scores for all aircraft for a specific operator is calculated.
- 40 6. The score distributions (median and variance) are used to calculate rankings for each of the operators (i.e. the ranks being low, medium and high diligence).

FIG. 14 is a flow chart illustration of steps performed by the DILIGENCE module **390**.

The COHORT analysis module **395** counts the number of occurrences in successive time periods for equipment grouped by construction number. This module is used to address the fact that production line problems may affect equipment assembled during the same time period. These problems may be due to either labor unrest or short-term problems in component manufacture or quality control. More specifically, equipment assembled during one time period may be less reliable than equipment assembled in previous or subsequent time periods. Equipment failure may occur at various follow-up intervals, months, years or even decades in the future but the failure may not be directly identified as a cause of a problem since the original defect may contribute relatively early to a multi-factorial causal chain.

Analyses must focus on equipment categorized by their chronological dates on the production line specified by the year delivered or by their construction number. The COHORT analysis module 395 groups aircraft by construction number and examines the number of safety-related occurrences in subsequent time periods. The analysis can be world-wide for accidents (whether total hull-loss or substantial damage). U.S. data can be used for Incidents and for

Service Difficulty Reports. The purpose of this module is to detect sub-groups with significantly higher rates of safety-related occurrences in comparison to earlier or later sub-groups. The rates are calculated on the basis of the number of aircraft in active use. Aircraft withdrawn from use are 5 withdrawn from the denominator.

The following steps are performed by the COHORT module 395:

- 7. Select subgroup: Model version.
- 8. Group by construction number.
- 9. Calculate number of aircraft in active use at start of each time period for as many follow-up years as desired by the user.
- 10. Tabulate number of safety related occurrences in each follow-up period.
- 11. Calculate ratio of safety related occurrences per number exposed.
- 12. Determine statistical significance of differences between adjacent construction number groups.

The graphizer component 400 provides the calculated 20 data to the user in tabular, text and/or graphic display form (the latter being produced by condensing large data tables into graphs). A multi-result inter-related graphic display is provided by this component as illustrated in FIG. 16(c), in which three separate graphs are shown on a single screen 25 (this number of graphs being suitable for notebook computer screen sizes, but a larger number of graphs could be accommodated by larger screen sizes). The graphizer component provides to the user, within this graphic display, a menu of datasets for selection of a desired dataset (see window 410 30 which provides a choice of ATA, Precautionary Procedure and Nature of Condition datasets). Corresponding to the selected dataset 410, a further menu of data subsets 420 is provided to the user for selection of a data subset (the selected subset shown being Unscheduled Landing).

The graphizer component generates a graphical presentation (being the main graph) 430 of the global output data set produced by the analyses component 300 for the selected data set and subset. From that main graph 430 the graphizer component receives from the user, by highlighting (i.e. a 40 point and click means of selection), a selection of one or more related data subsets 440 and, in response, generates additional graphical presentations 450, 460 for each such data subset. As shown by FIG. 16(c) the additional graphical presentation(s) are displayed together on the same screen 45 display. In preparing the main graph 430 and subgraph(s) 450, 460 for display the graphizer component formats each graphical presentation to an appropriate size and position for collective viewing. As illustrated by FIG. 16(c), the main graph 430 shows the ranking of each selectable subgraph 50 data set (e.g. by operator) and the subgraph(s) shows greater detail for specific elements of the selected data subset (i.e. in this illustration, for the selected operator BB).

For the illustrated embodiment, an initial screen is provided to the user which permits the user to select criteria 55 from a number of menus including menus covering Operator/Model (e.g. 11 major operators or 15 models), Focus (e.g. Service Difficulty Reports, Accidents/incidents and Engine reliability), ATA (mechanical component), Precautionary procedure (e.g. unscheduled landing), Nature of 60 condition (e.g. smoke) and Stage of Operation (e.g. takeoff). FIG. 16(a) shows an output graph generated by the graphizer 400 for a selection by the user of the ATA menu item "Flight Control Sys (A2)", this graph representing data for the "flight control sys" reports by the eleven (11) 65 operators identified within each operator bar (e.g. the second bar from the top representing operator "BB"). In this graph

the percentages shown represent the number of smoke reports for each of the 11 operators, normalized by model and number of departures, with the group average being 100%. The two-graph display of FIG. 16(b) shows time trends for SDRs and the number of repeat occurrences, wherein the highlighted circle shows the ratio of the latest number to the two previous high numbers (and the two previous high numbers are shown by large dots).

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The foregoing description of the preferred embodiment describes aircraft-centric measurements of the present information system and method because the illustrated embodiment relates to the field of analyzing aviation safety data. However, these measurements and analyses can be equally applied to other fields such as health and product safety. For 15 example, a patient may be the object and records of doctors' reports (i.e. from patient visits to doctors or hospitalizations), rather than SDRs, might then form the basis on which analyses are conducted. In the health field, there are multiple sources of data which, without the benefit of the present system and method, are not readily collated to follow the longitudinal side effects, course and outcomes of various medical treatments. The present system and method can thereby provide the surveillance and monitoring means which is needed to detect epidemics or other undesirable outcomes and to identify risk factors.

The terms algorithm, function, module and component herein are interchangeable and are used generally herein to refer to any set of computer-readable instructions or commands such as in the form of software, without limitation to any specific location or means of operation of the same, and without limitation as to the format, style, organization or size of the command sets to be selected for any particular embodiment.

It is to be understood that the specific elements of the information system and method described herein are not intended to limit the invention defined by the appended claims. From the teachings provided herein the invention could be implemented and embodied in any number of alternative computer program embodiments by persons skilled in the art without departing from the claimed invention.

## APPENDIX A

#### DEFINITIONS AND EXPLANATIONS OF TERMS USED IN THE DESCRIPTION HEREIN AND IN THE SYSTEM'S DATA SOURCES

Accident An occurrence associated with the operation of an aircraft which takes place between the time any person boards the aircraft with the intention of flight and until such time as all persons have disembarked, and in which any person suffers death or serious injury or in which the aircraft receives substantial damage. Substantial damage means damage or failure which adversely affects the structural strength, performance or flight characteristics of the aircraft, and which would normally require major repair or replacement of the affected component. Engine failure or damage limited to the engine if only one engine fails or is damaged, bent failings or cowling, dented skin, small puncture holes in the skin or fabric, ground damage to rotor or propeller blades, and damage to landing gear, wheels, tires, flaps, engine accessories, brakes or wingtips are not generally considered "substantial damage".

Accident Rates Accident counts alone are not reliable indicators of the relative safety of airlines/operators, aircraft types, or segments of the air transportation industry because, all other things being equal, an airline with 500 airplanes

will probably have more accidents than an airline with 10 airplanes. Similarly, the more frequently used aircraft types would tend to be involved in accidents more than the less frequently used types. The method most commonly used to address these issues is to calculate accident rates in terms of accident counts divided by some measure of aviation activity; i.e., accident counts divided by flight hours, departures, miles, etc.

ASRS (Aviation Safety Report System) A voluntary, confidential and anonymous incident reporting system used 10 to identify hazards and safety discrepancies in the National Airspace System. It is also used to formulate policy and to strengthen the foundation of aviation human factors safety research. It was started in 1976 as a means for the FAA to obtain reports of deficiencies and discrepancies before accidents or incidents occur. Pilots, air traffic controllers, flight 15 attendants, mechanics, ground personnel, and others involved in aviation operations can submit reports to the ASRS when they are involved in, or observe, an incident or situation in which they believe aviation safety was compromised. The FAA provides limited immunity from regulatory 20 enforcement action to reporters as an incentive to report incidents. The ASRS identifies the airport and, since January 1994, the aircraft model, but not the registration, serial number or airline. The program encourages submissions by granting those who file reports anonymity and limited <sup>25</sup> immunity. The ASRS database contains reports submitted from 1988 to present.

ATA Refers to the Air Transport Association Joint Aviation Statistical code (statistical classification) indicating the mechanical component where the difficulty on the aircraft occurred. Also referred to as JASC—Joint Aviation Statistical Code.

ATC Refers to air traffic control.

BTS Bureau of Transportation Statistics (USA).

Category of Operation Aviation operators are categorized by the federal regulation that govern their conduct. The major categories in the NTSB database are:

Air Carriers (Scheduled Part 121 and Part 127 operators)
Charter Air Carriers (Non-scheduled Part 121 operators)
Commuter Airlines (Scheduled Part 135 operators)

Air Taxis (Non-scheduled Part 135 operators)

Departures The number of aircraft take-offs actually performed in domestic and international scheduled and non-scheduled passenger/cargo and all-cargo revenue services.

Enforcement Action The FM publishes a quarterly compilation of enforcement actions against regulated aviation entities that are closed with a civil penalty or issuance of a certificate for suspension or revocation. At the end of each quarter of the calendar year, the FAA compiles a data run for 50 all enforcement actions against aviation entities that involve safety and security issues and result in a civil penalty or issuance of a suspension or revocation order. Some security cases may be reported in the quarter following the quarter in which the action was closed. For purposes of this report, 55 cases are considered to be closed after an order has issued that is no longer appealable or when a civil penalty is imposed. However, for other record-keeping purposes in the Enforcement Information System a case is considered closed only after some other event occurs, e.g., payment of a civil 60 penalty or a determination that the penalty is not collectable or, in the case of suspension or revocation, surrender of the certificate.

Engine Utilization/Reliability The air carrier aircraft and engine utilization report covers engine shutdowns and 65 removals identified by air carrier, aircraft model and engine model.

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FAA Airworthiness Directives The FM issues alerts about unsafe conditions that are newly discovered. These directives specify the date by which implementation of corrective action is required.

FAA Incident Data System The FM Incident Data System contains incident data records for all categories of civil aviation. Incidents are events that do not meet the aircraft damage or personal injury thresholds contained in the NTSB definition of an accident. For example, the database contains reports of collisions between aircraft and birds while on approach to or departure from an airport. While such a collision may not have resulted in sufficient aircraft damage to reach the damage threshold of an NTSB accident, the fact that the collision occurred is valuable safety information that may be used in the establishment of aircraft design standards or in programs to deter birds from nesting in areas adjacent to airports. The FAA Incident Data System contains events that occurred between 1978 and the present.

Fatal injury Any injury which results in death within 30 days of the accident.

GAO General Accounting Office

Hours Flown The airborne hours in domestic and international scheduled and non-scheduled revenue service, computed from the moment an aircraft leaves the ground until it touches the ground again.

Incident A potentially hazardous event which, although not meeting the aircraft-damage or personal-injury thresholds for an accident (as defined by the NTSB), is associated with the operation of an aircraft and affects or could affect the safety of operations.

JASC See ATA

Longitudinal Analysis By following up individual objects (in this example, individual aircraft).

Miles Flown The miles (computed in airport-to-airport distances) for each inter-airport trip actually completed in domestic and international revenue services, scheduled and non-scheduled. In cases where the inter-airport distances are inapplicable, aircraft miles flown are determined by mult-plying the normal cruising speed for the aircraft type by the airborne hours.

Model Mix Refers to the factor that no two airlines fly identical aircraft fleets on identical routes. Some tend to fly larger aircraft on long routes while others fly smaller aircraft more frequently on shorter routes.

NASDAC (National Aviation Safety Data Analysis Center) The Federal Aviation Administration (FAA) promotes the open exchange of safety information in order to continuously improve aviation safety. To further this basic objective, the FAA has established the National Aviation Safety Data Analysis Center (NASDAC). It is expected that, eventually, this facility will be made available for use throughout the aviation safety community. The center is currently located in the headquarters building of the United States Federal Aviation Administration. NASDAC works towards simplifying and automating aviation data analysis processes and expands access to data and information.

Nature of Condition Symptoms or problems identified from a Service Difficulty Report and identified by 24 values e.g. warning indicator, fluid loss, etc.

Near Mid-Air Collision (NMAC) An NMAC is an incident associated with the operation of an aircraft in which a possibility of a collision occurs as a result of proximity of less than 500 feet to another aircraft or a report from a pilot or flight crew member stating that a collision hazard existed between two or more aircraft. The NMAC reporting system was revised substantially in 1992.

NTSB The National Transportation Safety Board (NTSB) is a United States independent federal agency charged by

Congress with investigating every civil aviation accident in the United States. The NTSB determines the probable cause of accidents, and issues safety recommendations aimed at preventing future accidents. The NTSB accident/incident database contains investigation reports from 1983 to present. 5

NTSB Safety Recommendations to the FAA With FAA Responses The NTSB makes safety recommendations to all elements of the transportation industry. Recipients of a recommendation have to respond formally to the recommendation and specify what action is or is not being taken 10 and why. There is an NTSB database which contains the aviation safety recommendations, the FAA responses, the subsequent correspondence and NTSB's appraisal of the response. NTSB uses the information it gathers during accident investigations and the determination of probable cause to make safety recommendations to all elements of the transportation industry. While the recipient of a recommendation does not have to implement the proposed action, it does have to formally respond to the recommendation and specify what action is or is not being taken and why. This database contains NTSB's recommendations to the FAA and the FAA's responses. It is an historical record of correspondence between the NTSB and the FAA. The NTSB Safety Recommendations database includes recommendations that were written between 1963 and the present.

Occurrence An event, other than an accident, that affects or could affect the safety of aviation operations and that is investigated by the FAA.

OMEC The Operator/Model Exposure and Costs dataset (used by the standardization analysis module **320**).

Operational Deviation (OD) An occurrence where applicable separation minima, as referenced in the operational error (OE) definition below were maintained, but:

- (1) less than the applicable separation minima existed between an aircraft and protected airspace without prior 35 approval,
- (2) an aircraft penetrated airspace that was delegated to another position of operation or another facility without prior coordination and approval,
- (3) an aircraft penetrated airspace that was delegated to another position of operation of another facility at an altitude or route contrary to the altitude or route requested and approved in direct coordination or as specified in a Letter of Agreement, pre-coordination or internal procedure,
- (4) an aircraft, vehicle, equipment, or personnel encroached upon a landing area that was delegated to another position of operation without prior coordination and approval.

Operational Error (OE) An occurrence attributable to an 50 element of the air traffic control system which results in less than the applicable separation minima between two or more aircraft, or between an aircraft and terrain or obstacles as required by Handbook 7110.65 and supplemental instructions. Obstacles include vehicles/equipment/personnel on 55 runways, or aircraft lands or departs on a runway closed to aircraft operations after receiving air traffic authorization.

Operator The individual, firm, or airline that causes the aircraft to be scheduled, moved or dispatched.

Pilot Deviation (PD) The actions of a pilot which result in 60 the violation of a Federal Aviation Regulation (FAR) or a North American Aerospace Defense Command (NORAD) Air Defense Identification Zone (ADIZ) tolerance.

Precautionary Procedure Responds to the urgent condition where there is a concern about the safety of the aircraft or people on board. A precautionary procedure differs from normal procedures, but is less than an emergency or distress

procedure which requires immediate assistance due to serious and/or imminent danger. Identified by 15 values eg. unscheduled landing, engine shutdown, etc.

Production Lists These detail aircraft characteristics (serial number, construction number, year built, engines), their operators and current status (in service, stored, written off or scrapped) and are provided in print and electronic form.

RCO Recommendation course and outcomes.

Registration Number An aircraft must be registered to be flown legally in the U.S. The FAA assigns a unique identification number to each registered aircraft called a registration number (commonly referred to a the "N Number" since all U.S. registration numbers start with "N"). The registration number is unique only among the population of currently registered aircraft and may be reused if an aircraft is destroyed or, for other reasons, its registration is allowed to lapse.

Remarks (SDR) Descriptive data provided in a Service Difficulty Report describing the conditions subsequent to or leading up to a reported problem, and may identify the cause for a malfunction and/or the emergency measures executed.

Runway Incursion Any occurrence at an airport involving an aircraft, vehicle, person, or object on the ground that creates a collision hazard or results in loss of separation with an aircraft taking off, intending to take off, landing or intending to land.

Separation Minima The minimum longitudinal, lateral, or vertical distances by which aircraft are spaced through the application of air traffic control procedures.

Serious Injury Any injury which:

- 1. requires hospitalization for more than 48 hours, commencing within 7 days from the date the injury was received;
- 2. results in a fracture of any bone (except simple fractures of fingers, toes or nose);
- 3. causes severe hemorrages, nerve, muscle or tendon damage;
- 4. involves any internal organ or;
- 5. involves second or third degree burns or any burns affecting more than 5 percent of the body surface.

Service Difficulty Report (SDR) Passenger aircraft of ten seats or more must report the occurrence or detection of each failure, malfunction or defect concerning specified events or anything else that has endangered or may endanger safe operations. The Flight Standards Service Difficulty Program objective is to achieve prompt and appropriate correction of conditions adversely affecting continued airworthiness of aeronautical products, through the collection of Service Difficulty and Malfunction or Defect Reports. The FAA's Service Difficulty Reporting program originated in the early 1970's with the earliest records currently available dating June, 1974.

Severity Identified by 7 values e.g. seldom, uncommon, frequent.

Stage of Operation Used to identify the operation state (i.e. when) and has 18 values e.g. approach, climb, cruise, descent, etc.

Traffic Refers to the Bureau of Transportation Statistics—Airline Traffic Statistics. The airline activity spreadsheet available on the Aviation Safety Information Web site contains, by airline, the number of departures, flight hours and miles performed in domestic and international, commercial service during the most recent five calendar years. These are the activity measures that are used most frequently by government and industry to calculate accident or incident rates for the principal segments of the air transportation system.

Traffic Denominators Factors which take into account the fact that more accidents occur during takeoff and landing than during other phases of flight and, therefore, the selection of the measure of aviation activity that is used to calculate rates will bias the results. For example, an airline 5 that flies longer routes has fewer takeoffs and landings per hour and is thus favored by use of hours as the measure of activity, while an airline that flies shorter routes is favored by the use of departures as the measure of activity.

Traffic Exposure Refers to days of operation, departures, 10 hours of aircraft flight, hours (gate to gate), and miles.

Turbojet Aircraft An aircraft having a jet engine in which the energy of the jet operates a turbine which in turn operates the air compressor.

Turboprop Aircraft An aircraft having a jet engine in 15 which the energy of the jet operates a turbine which drives the propeller.

Type A hierarchical classification of four values, namely, UEL, major, routine, NMNR (not major, not routine), which may also be referred to as "kind".

Unsafe or illegal Action/Condition During an Event Indicates whether the primary problem is with the aircraft, flight crew, ATC, airport, navigational aids, publications or weather.

Unscheduled Landing Types

Passenger precipitated (illness, premature labor or disruptive behavior)

Diversions due to weather

Known or suspected mechanical difficulties or malfunctions (SDR's)

Vehicle/Pedestrian Deviation (VPD) An entry or movement on an airport movement area by a vehicle operator or pedestrian that has not been authorized by air traffic control (includes aircraft operated by a non-pilot).

#### APPENDIX B

# VALUES OF THE FIELDS OF SERVICE DIFFICULTY REPORTS (SDRS):

TYPE (4 values)

UEL (Unscheduled Emergency Landing)

MAJOR

**ROUTINE** 

NMNR (Not major, not routine)

NATURE OF CONDITION (23 values)

VISIBLE FLAME

**SMOKE** 

FOREIGN OBJECT

INFLIGHT SEPARATION

VIBRATION

**CONTROLS AFFECTED** 

MULTIPLE FAILURES

POWER SOURCE LOSS

FLIGHT ALTITUDE INDICATOR

WARNING INDICATION

FLUID LOSS

SYSTEM TEST BAD

OVERTEMP

FALSE WARNING

OTHER

NO WARNING INDICATION

ENGINE POWER LOSS

OTHER SYSTEM AFFECTED

CASE PENETRATION
INADEQUATE QUALITY

**FLAMEOUT** 

STOPPAGE

SIGNIFICANT FAILURE

PRECAUTIONARY PROCEDURE (14 values)

UNSCHEDULED LANDING

EMERGENCY DESCENT

ABORT TAKEOFF

RETURN TO BLOCK

**ENGINE SHUTDOWN** 

FIRE EXTINGUISHER

**OXYGEN MASK** 

DEACTIVATE SYSTEM

**DEPRESSURIZATION** 

DUMP FUEL

INONE

ABORT APPROACH

**OTHER** 

AUTOROTATION

SEVERITY (6 values)

**SELDOM** 

UNCOMMON

OCCASSIONAL

COMMON

FREQUENT

OVER 24 MO

STAGE OF OPERATION (17 values)

AIR AMBULANCE

**AEROBATIC** 

AGRICULTURE

APPROACH

CLIMB

40 CRUISE

35

50

DESCENT

EXTERNAL LOAD

FIRE FIGHTING

45 HOVERING

INSPECTION/MAINTENANCE

LANDING

MAPPING/SURVEY

NOT REPORTED

TAKEOFF

TAXI

UNKNOWN

YEAR BUILT (5 values)

<sup>55</sup> 1900–1901 YEAR BUILT

1902–1968 YEAR BUILT

1969–1978 YEAR BUILT

1979–1988 YEAR BUILT

60 1989–1999 YEAR BUILT

DOD-DOR (7 values)

DOR-DOD

0 DOR-DOD

<sub>65</sub> 1–3 DOR-DOD

4–11 DOR-DOD

12-26 DOR-DOD

27–55 DOR-DOD		DEHAVILLAND DHC8
56–112 DOR-DOD		AEROSPATIALE AS262
113+DOR-DOD		AEROSPATIALE AS355
PARETO (18 values)	5	AEROSPATIALE AS365
0 SDRS	3	AEROSPATIALE ATR42
1 SDRS		AEROSPATIALE ATR72
2 SDRS		AEROSPATIALE CONCORDE
3 SDRS		BAE ONE-ELEVEN
4 SDRS	10	BAG BAE146
5 SDRS		BAE HP137
6 SDRS		BAG JETSTM
7 SDRS		BAE ATP
8 SDRS	15	BOLKMS BK117
9 SDRS		
10 SDRS		BOLKMS BO105
11–14 SDRS		EMBRAER 110
15–14 SDRS 15–24 SDRS	20	EMBRAER 120
	20	FOKKER F27 FRIENDSHIP
25+SDRS		FOKKER F28 FELLOWSHIP
4+SDRS		SAAB SF340
3+SDRS		SHORTS (STBROS) SD3
2+SDRS	25	SWEARINGEN SA226
1+SDRS		SWEARINGEN SA227
MODEL (112 values)		AMD DASSAULT FALCON
BOEING 107		BEECH 1900
BOEING 234	30	BEECH C45
BOEING 707		BEECH 58
BOEING 720		BEECH 99
BOEING 727		BEECH 100
BOEING 737	25	BEECH 200
BOEING 747	35	CASA C212
BOEING 757		CANADAIR CL600
BOEING 767		CANADAIR CL44
BOEING 777		CURTISS C46
DOUGLAS C118	40	CONVAIR 240
DOUGLAS DC3		CONVAIR 340
DOUGLAS C54 (DC4)		CONVAIR 440
DOUGLAS DC6		CONVAIR 440 CONVAIR APJC
DOUGLAS DC7	45	
DOUGLAS DC8		DORNIER 228
DOUGLAS DC9		DORNIER 328
DOUGLAS MD88 (DC9)		FAIRCHILD FH27
DOUGLAS DC10	50	FAIRCHILD FH227/SA227
DOUGLAS MD11 (DC11)	50	NIHON YS11
DOUGLAS MD90		AGUSTA A109
AIRBUS 300		BNORM BN2A
AIRBUS 310		BRAERO BAE125
AIRBUS 320	55	BRAERO HS748
AIRBUS 330		CESSNA 172
AIRBUS 340		CESSNA 207
LOCKHEED 1011 TRISTAR		CESSNA 208
LOCKHEED 132	60	CESSNA 402
LOCKHEED 188		CESSNA 404
LOCKHEED 382		CESSNA 550
DEHAVILLAND DHC2		HUGHES 369
DEHAVILLAND DHC3	سو سر سو سر	HWKSLY DH114
DEHAVILLAND DHC6	65	LEAR 24
DEHAVILLAND DHC7		LEAR 25

LEAR 35		DOUG DC987
MARTIN 404		DOUG DC98 DEFAULT & MD88
MTSBSI MU2B		DOUG MD9
PIPER PA232	5	AIRBUSA300B4 & 300F
PIPER PA313	3	AIRBUSA3102
PIPER PA31		AIRBUSA3103
SNIAS AS350		AIRBUSA3202
WESTLD WESTLAND		AIRBUSA300B2
SKRSKY S76 COPTER	10	AIRBUSA330 (EXPECT NONE)
BELL 205 COPTER		BOEING7271XX W/O C
BELL 206 COPTER		BOEING7271XXW/C
BELL 212 COPTER		BOEING7271XX BOEING7272XX
BELL 212 COLTER BELL 214 COPTER	15	DOUG DC101
BELL 214 COLTER BELL 222 COPTER		DOUG DC101 DOUG DC101 DOUG DC101
BELL 230 COPTER		DOUG DC103 W/O CF
BELL 407 COPTER	20	DOUG DC104
BELL 412 COPTER	20	DOUG DC103 W/CF (EXPECT NONE
GULFSTREAM G1159		DOUG MD11
GULFSTREAM G159		BOEING7071XX W/O B
AIRBUS 319		BOEING7071XX W/B
EMBRAER 145	25	BOEING7072XX (EXPECT NONE)
DEFAULT		BOEING7073XX W/O B W/O C
BTS MODEL (118 values)		BOEING7073XX W/B
DHAV DHC2		BOEING7073XX W/C
DHAV DHC3	30	BOEING7074XX
DHAV DHC4		BOEING7200 W/O B
DOUG DC2 (EXPECT NONE)		BOEING7200 W/B
DOUG DC3 & C47		BOEING7471XX
DOUG DC4 & C54	2.5	BOEING7472XX & 3XX
DOUG DC6 DEFAULT	35	BOEING747 W/C
DOUG DC6A		BOEING7474XX
DOUG DC6B		BOEING747 W/F
DOUG DC7B		BOEING747 W/SP
DOUG DC7C	40	DOUG DC81 (EXPECT NONE)
LKHEED1049 DEFAULT		DOUG DC61 (LAI LC1 NONL)  DOUG DC82
LKHEED1049H		
DHAV DHC81		DOUG DC84
DHAV DHC83	45	DOUG DC84
DHAV DHC65		DOUG DC85
LKHEED188		DOUG DC8F5
LKHEED382		DOUG DC861
DHAV DHC7	<b>~</b> 0	DOUG DC863F
BOEING7375XX	50	DOUG DC862
BOEING7374XX		DOUG DC863 DEFAULT
BOEING7373XX W/LR (EXPECT NONE)		DOUG DC871
BOEING7373XX W/O LR		DOUG DC872
BOEING7372XX W/O C & 1XX	55	DOUG DC873 DEFAULT
BOEING7372XX W/C BOEING7372XX W/C		DOUG DC873F
BOEING7572XX		AIRBUSA340
BOEING7672XX		BOEING107 & 114
BOEING7673XX	60	BOEING234
BOEING7073AA BOEING777		BOEING7070XX
DOUG DC91 DEFAULT		BOEING707 DEFAULT
DOUG DC91 DLIACLI DOUG DC915F		BOEING720 DEFAULT
DOUG DC9131 DOUG DC93		BOEING7270XX
DOUG DC93 DOUG DC94	65	BOEING727 DEFAULT
DOUG DC94 DOUG DC95		BOEING727 DEFFICE BOEING7370XX

**29 30** BOEING737 DEFAULT IXKA KIWI INT ORJA RENO BOEING7470XX VJ6A VALUJET (NOW AIRTRAN) BOEING747 DEFAULT MZZA CONTINENTAL MICRONESIA BOEING7570XX BOEING757 DEFAULT EALA EASTERN BOEING7670XX HEEA PETROLEUM HELICOPTER ZZDA AIRTRAN (WAS VALUJET) BOEING767 DEFAULT BOEING DEFAULT GTIA SPIRIT 10 W8PA OKLAHOMA MAY 99 AIRBUSA300 DEFAULT AIRBUSA310 DEFAULT P2EA VANGUARD AAIRBUSA319 AAA ALLEGHENY KT3R OKLAHOMA MAY 99 AIRBUSA320 DEFAULT AIRBUS DEFAULT (EXPECT NONE) AWI HUGHES AIR WEST DHAV DHC1 F3LA FRONTIER DHAV DHC82 VFSA OKLAHOMA MAY 99 DHAV DHC8 DEFAULT NCA NORTH CENTRAL 20 DHAV DEFAULT GUUA GULFSTREAM INT DOUG DC7 DEFAULT K3HA KIWI INT DOUG DC8 DEFAULT SABA OKLAHOMA MAY 99 DOUG DC92 TC8A TRANS CONTINENTAL DOUG DC9 DEFAULT H3R OKLAHOMA MAY 99 25 DOUG DC10 DEFAULT" PNSA PENINSULA DOUG DEFAULT" EQGA OKLAHOMA MAY 99 LKHEED10113 SOU SOUTHERN AW LKHEED1011 DEFAULT WAA WIEN AIR ALASKA 30 LKHEED132 BDHA OKLAHOMA MAY 99 LKHEED DEFAULT FXLA FINE UNDEF MANUFACTURER APN OKLAHOMA OPERATOR CODE (225 values) BJNA TEM ENTERPRISES 35 AALA AMERICAN PAIA PIEDMONT AL UALA UNITED PAAA PAN AMERICAN WORLD DALA DELTA FALA FRONTIER AL USAA USAIR **BNFA BRANIFF AW** NWAA NORTHWEST TXIA TEXAS INT AL CALA CONTINENTAL OZAA OZARK AL FDEA FEDERAL EXPRESS WALA WESTERN AL TWAA TRANS WORLD NALA NATIONAL AL SWAA SOUTHWEST ANEA AIR NEW ENGLAND AWXA AMERICA WEST WTAA WESTAIR COMMUTER A IPXA UPS ABXA ABX A COMA COMAIR MALA MESABA AV ASAA ALASKA 50 PCAA PENNSYLVANIA COMMUTE HZMA HORIZON RMXA OMNIFLIGHT HELICOPTERS AMTA AMERICAN TRANS SIMA SIMMONS AL EIAA EVERGREEN INT BRIA BRITT AW SRAA SOUTHERN AIR TRANSPORT ZIAA ZANTOP INT 55 HALA HAWAIIAN RAIA TRANS STATES AL MWEA MIDWEST EXPRESS AWAA AIR WISCONSIN AL TSAA ALOHA RAAA REEVE ALEUTIAN AW TWRA TOWER EISA EXPRESS ONE INT MRKA MARKAIR P5CA POLAR AC TAOA USAIR SHUTTLE" MIDA MIDWAY AL WRLA WORLD AW

CLCA CHALLENGE AC

DHLA DHL AW

RRXA EMERY WORLDWIDE AL

65

CKSA AMERICAN INT

PAYA TRANS WORLD EXPRESS

VTZA ATLANTIC COAST

RIVA CARNIVAL AL

PLGA BUSINESS EXPRESS

**31 32 NVEA AMERICAN EAGLE** ACLA AIRCAL ARWA ARROW A TIAA TRANSAMERICA AL SCNA SUN COUNTRY AL SALA SUBURBAN AL MGMA MGM GRAND A ASOA ATLANTIC SOUTHEAST AL ASPA ASPEN AW PSAX OKLAHOMA RYNA RYAN INT PEXA PEOPLE EXPRESS PEXA PEOPLE EXPRESS EMAA EMERALD A REXA EXPRESS AL VNAA JETSTREAM INT 10 RIAA RICH INT FAVA AVAIR NACA NORTHERN AC EMXA METRO EXPRESS RAXA ROSENBALM AV TAGA ORION AIR CHQA CHAUTAUQUA AL KTIA KEY AL 15 TCAA TRANS CONTINENTAL AL TIAS TRANSAMERICA PREA PRECISION VALLEY AVI CAIA COMMAND AW SPAA SIERRA PACIFIC AL FTLS FLYING TIGER ERAA ERA AV CPLA CHAPARRAL AL 20 FWTA FLORIDA WEST AL WWMA WINGS WEST AL A6WA AIR WISCONSIN AL PBAA PROVINCETOWN BOSTON AL GLBA GREAT LAKES AV IXXA AIR TRANSPORT INT BUFA BUFFALO AW NYAA NEW YORK AL S2YA ASTRAL CROA CROWN AW 25 GRAA GREAT AMERICAN AW SAAS OKLAHOMA GAIA KITTY HAWK AC WRNA TRADEWINDS MASA MESA AL SUNA SUN AIRE LINES COEA EMPIRE AL ACAX OKLAHOMA 30 TALA TRANS AIR LINK MEJA MERLIN EXPRESS RAYA OKLAHOMA ISAA INTERSTATE AL PDLA PRESIDENTIAL AW ZIAX OKLAHOMA GATA GULF AIR TRANSPORT RMAA ROCKY MOUNTAIN AW 35 MVAA MISSISSIPPI VALLEY AL RIOA RIO AW HERA HERMENS A IASA INTERNATIONAL AIR SERV FLEA FLORIDA EXPRESS IMPA IMPERIAL AL AIAX OKLAHOMA CAPS OKLAHOMA AFLA MIDWAY AL BRTA BRITT AL FLTA FLIGHT TRAILES FECT OKLAHOMA GWAT OKLAHOMA R7MA ROCKY MOUNTAIN HOLIDAYS AORA OKLAHOMA UF6A UFS I FLAA CAMAIR INT ALTT OKLAHOMA ALGA OFFSHORE LOGISTICS POAA PUERTO RICO INT AL AIAA ATORIE A SMBA SEDALIA MARSHALL BOONVIL PATX OKLAHOMA ONAS OKLAHOMA 50 JJBA CHAMPLAIN ENTERPRISE CCDA CASCADE AW HPJA APOLLO AW WRTA WRIGHT AL ONEA AIR ONE RANA RANSOME A EIAS OKLAHOMA TRBA EXECUTIVE REPA REPUBLIC AL MAAA MIDSTATE AL 55 AMWA AIR MIDWEST SWAX OKLAHOMA NAXA OKLAHOMA RDLS OKLAHOMA HNAA HENSON AV ANAA ANA SWIA SKYWEST AL GPPA OKLAHOMA PCTA PORTS OF CALL RAIX OKLAHOMA FTLA FLYING TIGER LINE RAHT OKLAHOMA MTRA METROFLIGHT FECZ OKLAHOMA

SCIA SCENIC AL

PCTX PORTS OF CALL

SMMA SUMMIT AL

65

PSAA PACIFIC SOUTHWEST

BHAA BAR HARBOR AW

C2XA CONTINENTAL EXPRESS

**33** 34 A4JA OKLAHOMA MAY 99 NACELLE/PYLON STRUCTURE ACA OKLAHOMA MAY 99 EMPENNAGE STRUCTURE ACZA OKLAHOMA MAY 99 WINDOW/WINDSHIELD SYSTEM AETA OKLAHOMA MAY 99 WING STRUCTURE PROPELLER SYSTEM ARWR OKLAHOMA MAY 99 ASRA OKLAHOMA MAY 99 MAIN ROTOR SYSTEM MAIN ROTOR DRIVE B4MR OKLAHOMA MAY 99 CAPA OKLAHOMA MAY 99 TAIL ROTOR 10 CICA PARADISE ISLAND TAIL ROTOR DRIVE SYSTEM ROTORCRAFT FLIGHT CONTROL EF2R OKLAHOMA MAY 99 GEMA OKLAHOMA MAY 99 POWER PLANT HCMA HELICOPTER CONSULTANT ENG TURBINE/TURBOPROP 15 ENG FUEL & CONTROL IU6R OKLAHOMA MAY 99 JAMA OKLAHOMA MAY 99 IGNITION SYS MACA OKLAHOMA MAY 99 ENG BLEED AIR SYSTEM MIAS OKLAHOMA MAY 99 ENG CONTROLS 20 PCSA OKLAHOMA MAY 99 ENG INDICATING SYSTEM PIOA OKLAHOMA MAY 99 ENG EXHAUST PSA OKLAHOMA MAY 99 ENG OIL SYSTEM RDLA OKLAHOMA MAY 99 **ENG START** TURBINE SYSTEM RECIP ONLY RJEF OKLAHOMA MAY 99 25 SBWS OKLAHOMA MAY 99 WATER INJECTION ACCESSORY GEAR BOXES SWJA OKLAHOMA MAY 99 SWXA OKLAHOMA MAY 99 ENGINE RECIPROCATING 30 ATA 4 CHARACTER CODES (448 values) TWUA OKLAHOMA MAY 99 PLACARDS AND MARKINGS VG5R OKLAHOMA MAY 99 FUEL SERVICING VSTA OKLAHOMA MAY 99 OIL SERVICING Y2PA USA JET HYDRAULIC FLUID SERVICING ZYWA WEST MICHIGAN AIR CARE 35 COOLANT SERVICING DFLT DEFAULT ATA 2 CHARACTER CODES (50 values) HELICOPTER VIB NOISE ANALYSIS SERV PLACARD MARKING HELICOPTER VIBRATION ANALYSIS SERV FUEL OIL COOLANT HELICOPTER NOISE ANALYSIS AIR CONDITIONING SYSTEM HELICOPTER VIBRATE AIR CONDITIONING COMPRESSOR SYSTEM AIR DISTRIBUTION SYSTEM JAUTO PILOT COMMUNICATIONS SYSTEM AIR DISTRIBUTION FAN ELECTRICAL POWER SYSTEM PRESSURIZATION CONTROL SYSTEM INTERIOR EQUIPMENT PRESSURE CONTROLLER FIRE PROTECTION PRESSURE INDICATOR FLIGHT CONTROL SYSTEM REGULATOR-OUTFLOW VALVE FUEL SYSTEM PRESSURE SENSOR HYDRAULIC SYSTEM HEATING SYSTEM COOLING SYSTEM (AIR VAPOR CYCLE) ANTI-ICE SYSTEM INSTRUMENTS TEMPERATURE CONTROL SYSTEM TEMPERATURE CONTROLLER LANDING GEAR 55 CABIN TEMPERATURE INDICATOR LIGHTING SYSTEM CABIN TEMPERATURE SENSOR NAVIGATION SYSTEM OXYGEN SYSTEM HUMIDITY CON OL SYSTEM PNEUMATIC SYSTEM AUTO FLIGHT SYSTEM VACUUM SYSTEM AUTOPILOT SYSTEM 60 WATER/WASTE SYSTEM AUTOPILOT COMPUTER CENTRAL MAINT COMPUTER ALTITUDE CONTROLLER AIRBORNE APU SYSTEM FLIGHT CONTROLLER AUTOPILOT TRIM INDICATOR PRACTICE/STRUC/BALLOON DOORS AUTOPILOT MAIN SERVO

AUTOPILOT TRIM SERVO

FUSELAGE

SPEED-ATTITUDE CORRECTION SYSTEM AUTO THROTTLE SYSTEM AERODYNAMIC LOAD ALLEVIATING COMMUNICATIONS SYSTEM HF COMMUNICATION SYSTEM UHF COMMUNICATION SYSTEM VHF COMMUNICATION SYSTEM DATA TRANSMISSION AUTO CALL PA & ENTERTAINMENT SYSTEM INTERPHONE AUDIO INTEGRATING STATIC DISCHARGE SYSTEM AUDIO & VIDEO MONITORING ELECTRICAL POWER SYSTEM ALTERNATOR-GENERATOR DRIVE SYSTEM AC GENERATION SYSTEM AC GENERATOR-ALTERNATOR AC INVERTER PHASE ADAPTER AC REGULATOR AC INDICATING SYSTEM DC GENERATION SYSTEM BATTERY OVERHEAT WARNING SYSTEM BATTERY CHARGER DC RECTIFIER-CONVERTER DC GENERATOR-ALTERNATOR STARTER-GENERATOR DC REGULATOR DC INDICATING SYSTEM EXTERNAL POWER SYSTEM AC POWER DISTRIBUTION SYSTEM DC POWER DISTRIBUTION SYSTEM **EQUIPMENT FURNISHINGS** FLIGHT COMPARTMENT EQUIPMENT PASSENGER COMPARTMENT EQUIP BUFFET GALLEYS LAVATORIES CARGO COMPARTMENTS AGRICULTURAL EXTERNAL LOAD SYSTEM EMERGENCY EQUIPMENT LIFE JACKET EMERGENCY LOCATOR BEACON PARACHUTE LIFE RAFT ESCAPE SLIDE ACCESSORY COMPARTMENTS BATTERY BOX STRUCTURE ELECTRONIC SHELF SECTION FIRE PROTECTION SYSTEM DETECTION SYSTEM SMOKE, DETECTION FIRE, DETECTION OVERHEAT, DETECTION EXTINGUISHING SYSTEM FIRE BOTTLE, FIXED

FIRE BOTTLE, PORTABLE

FLIGHT CONTROL SYSTEM CONTROL COLUMN SECTION AILERON CONTROL SYSTEM AILERON TAB CONTROL SYSTEM RUDDER CONTROL SYSTEM RUDDER TAB CONTROL SYSTEM RUDDER ACTUATOR ELEVATOR CONTROL SYSTEM 10 ELEVATOR TAB CONTROL SYSTEM STABILIZER CONTROL SYSTEM STABILIZER POSITION IND SYSTEM STABILIZER ACTUATOR TE FLAP CONTROL SYSTEM TE FLAP POSITION IND SYSTEM TE FLAP ACTUATOR DRAG CONTROL SYSTEM DRAG CONTROL ACTUATOR GUST LOCK-DAMPER SYSTEM LE FLAP CONTROL SYSTEM LE FLAP POSITION IND SYSTEM LE FLAPACTUATOR 25 FUEL SYSTEM FUEL STORAGE FUEL DISTRIBUTION FUEL FILTER-STRAINER FUEL PUMPS FUEL SELECTOR SHUTOFF VALVE FUEL TRANSFER VALVE FUEL DUMP 35 FUEL INDICATION SYSTEM FUEL QUANTITY INDICATOR FUEL QUANTITY SENSOR FUEL TEMPERATURE FUELPRESSURE HYDRAULIC POWER SYSTEM HYDRAULIC, MAIN SYSTEM ACCUMULATOR, MAIN

FILTER, MAIN

PUMP, MAIN HANDPUMP, MAIN PRESSURE RELIEF, MAIN

RESERVOIR, MAIN PRESSURE REGULATOR, MAIN HYDRAULIC, AUXILIARY SYSTEM ACCUMULATOR, AUXILIARY

FILTER, AUXILIARY 55 PUMP, AUXILIARY HANDPUMP, AUXILIARY PRESSURE RELIEF, AUXILIARY

RESERVOIR, AUXILIARY 60 PRESSURE REGULATOR, AUXILIARY HYDRAULIC, INDICATING SYSTEM PRESSURE INDICATOR

PRESSURE SENSOR 65 QUANTITY INDICATOR QUANTITY SENSOR

ICE RAIN PROTECTION SYSTEM AIRFOIL ANTI-ICE, DEICE

INTAKE ANTI-ICE, DEICE PITOT STATIC ANTI-ICE

WINDOWS WINDSHIELDS & DOORS

ANTENNA RADOME ANTI-ICE PROP ROTOR ANTI-ICE, DEICE

WATER LINE ANTI-ICE

ICE DETECTION

INDICATING RECORDING SYSTEM

INSTRUMENT PANEL

INDEP INSTRUMENT (CLOCK,ETC)
DATA RECORDERS (FLIGHT MAINT)

CENTRAL COMPUTERS (EG EICAS)

CENTRAL WARNING

CENTRAL DISPLAY

AUTOMATIC DATA

LANDING GEAR SYS

LANDING GEAR WHEEL FAIRING

MAIN LANDING GEAR

MAIN LANDING GEAR ATTACH SEC

EMERGENCY FLOATATION SECTION

MAIN GEAR STRUT AXLE TRUCK

NOSE TAIL LANDING GEAR NOSE TAIL GEAR ATTACH SECTION

NOSE TAIL GEAR STRUT AXLE

GEAR EXTENSION AND RETRACT SYSTEM

LANDING GEAR DOOR RETRACT SEC

LANDING GEAR DOOR ACTUATOR

LANDING GEAR ACTUATOR

LANDING GEAR SELECTOR

LANDING GEAR BRAKES SYSTEM

ANTI-SKID SECTION

**BRAKE** 

MASTER CYLINDER BRAKE VALVE

TIRE CASING

TUBE

WHEEL SKI FLOAT

LANDING GEAR STEERING SYSTEM

STEERING UNIT

SHIMMY DAMPER

LG POSITION AND WARNING

AUX GEAR (TAIL ROTORCRFT SKID)

LIGHTING SYSTEM

FLIGHT COMPARTMENT LIGHTING

PASSENGER COMPARTMENT LIGHTING

CARGO COMPARTMENT LIGHTING

EXTERIOR LIGHTING

EMERGENCY LIGHTING

NAVIGATION SYSTEM

FLIGHT ENVIRONMENT DATA

PITOT STATIC SYSTEM

AIR TEMPERATURE THRUST LIMIT

RATE OF CLIMB

AIRSPEED MACH INDICATING

HIGH SPEED WARNING

ALTIMETER, BAROMETRIC ENCODER

AIR DATA COMPUTER

STALL WARNING SYSTEM

ATTITUDE & DIRECTION

ATTITUDE GYRO & INDICATION
DIRECTIONAL GYRO & INDICATION

MAGNETIC COMPASS

TURN AND BANK RATE OF TURN

10 INTEGRATED FLIGHT DIRECTOR SYSTEM

LANDING AND TAXI AIDS LOCALIZER VOR SYSTEM

GLIDE SLOPE SYSTEM

<sup>15</sup> MICROWAVE LANDING SYSTEM

MARKER BEACON SYSTEM

HEADS UP DISPLAY SYSTEM

WIND SHEAR DETECTION SYSTEM

20 INDE POSITION DETERMINING SYSTEM

INERTIAL GUIDANCE SYSTEM
WEATHER RADAR SYSTEM

DOPPLER SYSTEM

25 GROUND PROXIMITY SYSTEM

AIR COLLISION AVOIDANCE (TCAS)

NON RADAR WEATHER SYSTEM

DEPENDENT POSITION DETERMINING

30 DME/TACAN SYSTEM

ATC TRANSPONDER SYSTEM

LORAN SYSTEM

VOR SYSTEM

35 ADF SYSTEM

OMEGA NAVIGATION SYSTEM

GLOBAL POSITIONING SYSTEM

FLT MANAGEMENT COMPUTING SYSTEM

OXYGEN SYSTEM

CREW OXYGEN SYSTEM

PASSENGER OXYGEN SYSTEM

PORTABLE OXYGEN SYSTEM

PNEUMATIC SYSTEM

PNEUMATIC DISTRIBUTION SYSTEM
PNEUMATIC INDICATING SYSTEM

VACUUM

VACUUM DISTRIBUTION SYSTEM

O VACUUM INDICATING SYSTEM

WATER AND WASTE SYSTEM

POTABLE WATER SYSTEM

WASH WATER SYSTEM

55 WASTE DISPOSAL SYSTEM

AIR SUPPLY (WATER PRESS SYSTEM)

CENTRAL MAINTENANCE COMPUTER

AIRBORNE APU SYSTEM

APU ASSEMBLY AND COWL

APU CORE ENGINE

APU ENGINE FUEL AND CONTROL

APU START IGNITION SYSTEM

APU BLEED AIR SYSTEM

APU CONTROLS

APU INDICATING SYSTEM

APU EXHAUST SYSTEM
APU OIL SYSTEM
STANDARD PRACTICES STRUCTURES
AIRCRAFT STRUCTURES
BALLOON/DIRIGIBLE REPORTS

DOORS/

PASSENGER/CREW DOORS

**EMERGENCY EXIT** 

CARGO/BAGGAGE DOORS

SERVICE DOORS
GALLEY DOORS

ELECT/ELECTRONIC COM DOORS
HYDRAULIC COMPARTMENT DOORS
ACCESSORY COMPARTMENT DOORS

AIR CONDITIONING COMP DOORS

FLUID SERVICE DOORS

AUXILIARY POWER UNIT DOOR

TAIL CONE DOOR
FIXED INNER DOORS
ENTRANCE STAIRS
DOOR WARNING

LANDING GEAR DOORS

FUSELAGE

**AERIAL TOW EQUIPMENT SECTION** 

ROTORCRAFT TAIL BOOM FUSELAGE MAIN STRUCTURE

FRAMES (MAIN FUSELAGE)

BULK HEADS (MAIN FUSELAGE) LONGERONS STRINGERS (MAIN FUS)

**KEELS (MAIN FUSELAGE)** 

FLOOR BEAMS (MAIN FUSELAGE) FUSELAGE AUXILIARY STRUCTURE

FLOOR PANELS (AUX FUSELAGE)

INTERNAL MOUNT STRUC (AUX FUS)

INTERNAL STAIRS (AUX FUS)
FIXED PARTITIONS (AUX FUS)

PLATES SKINS (AUX FUSELAGE)

FUSELAGE ATTACH FITTINGS SYS

WING ATTACH FITTINGS (ON FUS)

STAB ATTACH FITTINGS (ON FUS)

GEAR ATTACH FITTINGS (ON FUS)

DOOR HINGE (ON FUSELAGE)

EQUIP ATTACH FITTINGS (ON FUS)

ENGINE ATTACH FITTING (ON FUS)

SEAT/CARGO ATTACH FITTING

AERODYNAMIC FAIRINGS STRUCTURE

NACELLES/PYLONS STRUCTURE

MAIN FRAME (ON NACELLES PYLON)

FRAMES (ON NACELLES/PYLON)

BULKHEAD FIREWALLS (NAC PYLON)

LONGERON STRINGERS (NAC PYLON)

PLATES/SKINS (NACELLE/PYLON)

ATTACH FITTINGS (NAC PYLON)

EMPENNAGE STRUCTURE

HORIZONTAL STABILIZER

SPARS/RIBS (HORIZONTAL STAB)

PLATES/SKINS (HORIZONTAL STAB)
TAB STRUC (HORIZONTAL STAB)

**ELEVATORS** 

SPARS/RIBS (ON ELEVATOR)
PLATES/SKINS (ON ELEVATOR)
TAB STRUCTURE (ON ELEVATOR)

VERTICAL STABILIZER

SPARS/RIBS (ON VERT STAB)

PLATES/SKINS (ON VERT STAB)
VENTRAL STRUC (ON VERT STAB)

**RUDDER** 

SPARS/RIBS (ON RUDDER)

PLATES/SKINS (ON RUDDER)
TAB STRUCTURE (ON RUDDER)
ATTACH FITTINGS (FLT CONTROLS)
HORIZONTAL STAB ATTACH FITTING

20 ELEVATOR/TAB ATTACH FITTING
VERTICAL STAB ATTACH FITTING
RUDDER/TAB ATTACH FITTING
WINDOWS-WINDSHIELD SYSTEM

FLIGHT COMPARTMENT WINDOWS
PASSENGER COMPARTMENT WINDOWS
DOOR WINDOWS

INSPECTION WINDOWS

WING STRUCTURE

MAIN FRAME (ON WING)

SPAR (ON WING)

RIBS/BULKHEAD S (ON WING)

LONGERONS/STRINGERS (ON WING)
CENTER WING BOX (ON WING)
AUXILIARY STRUCTURE (ON WING)

PLATES/SKINS (ON WING)

ATTACH FITTINGS (ON WING)

FUS (ATTACH FITTING ON WING)

NAC/PYLON (FITTING ON WING)

LANDING GEAR (FITTING ON WING)
CONT SURFACE (FITTING ON WING)

FLIGHT SURFACE S (WING)

AILERONS AILERON TABS

TRAILING EDGE FLAPS

LEADING EDGE DEVICES
SPOILERS

PROPELLER SYSTEM

PROPELLER ASSEMBLY

PROPELLER BLADE SECTION
PROP DEICE BOOT SECTION
PROP/SPINNER SECTION

PROPELLER HUB SECTION

PROPELLER CONTROLLING SYSTEM
PROP SYNCHRONIZER SECTION
PROPELLER GOVERNOR
PROPELLER FEATHER/REVERSING

PROPELLER BRAKING

PROPELLER INDICATING SYSTEM

MAIN ROTOR SYSTEM

MAIN ROTOR BLADE SYSTEM
MAIN ROTOR HEAD SYSTEM
MAIN ROTOR MAST/SWASHPLATE
MAIN ROTOR INDICATING SYSTEM

MAIN ROTOR DRIVE

ENGINE/TRANSMISSION COUPLING

MAIN ROTOR GEARBOX MAIN ROTOR BRAKE

ROTORCRAFT COOLING FAN SYSTEM

MAIN ROTOR TRANS MOUNT

ROTOR/DRIVE INDICATING SYSTEM

TAIL ROTOR

TAIL ROTOR BLADE

TAILROTOR

TAIL ROTOR INDICATING SYSTEM

TAIL ROTOR DRIVE SYSTEM

TAIL ROTOR DRIVE SHAFT

TAIL ROTOR GEARBOX

TAIL ROTOR DRIVE IND SYSTEM ROTORCRAFT FLIGHT CON OL

MAIN ROTOR CONTROL

TILT ROTOR FLIGHT CONTROL

TAIL ROTOR CONTROL SYSTEM

ROTORCRAFT SERVO SYSTEM

POWER PLANT

ENGINE COWLING SYSTEM

COWL FLAP SYS (RECIP. ONLY)

ENG AIR BAFFLE SEC (RECIP.)

MOUNTS

**FIRESEALS** 

AIR INTAKE

ENGINE DRAINS

ENGINE (TURBINE TURBOPROP)
REDUCTION GEAR AND SHAFT

AIR INLET SECTION (CORE ENG)

COMPRESSOR SECTION

COMBUSTION SECTION

TURBINE SECTION

ACCESSORY DRIVES

OIL SYSTEM

**BYPASS SECTION** 

ENGINE FUEL AND CONTROL

FUEL DISTRIBUTION

FUEL OIL COOLER

**FUEL HEATER** 

FUEL INJECTOR NOZZLE

FUEL PUMP

FUEL CONTROLLING SYSTEM

FUEL CONTROL ELECTRONIC

FUEL CONTROL CARBURETOR

TURBINE GOVERNOR

FUEL DIVIDER

FUEL INDICATING SYSTEM

FUEL FLOW INDICATING

FUEL PRESSURE INDICATING

FUEL FLOW SENSOR

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FUEL PRESS SENSOR IGNITION SYSTEM

**IGNITION POWER SUPPLY** 

LOW TENSION COIL

EXCITER

INDUCTION VIBRATOR
MAGNETO DISTRIBUTOR

DIST (IGNITION HARNESS)

SPARK PLUGS IGNITERS

SWITCHING

ENGINE BLEED AIR SYSTEM

ENGINE ANTI-ICING SYSTEM

ENGINE COOLING SYSTEM

COMPRESSOR BLEED CONTROL

COMPRESSOR BLEED GOVERNOR

COMPRESSOR BLEED VALVE

20 INDICATING SYSTEM

ENGINE CONTROLS

ENGINE SYNCHRONIZING

MIXTURE CONTROL

25 POWER LEVER

EMERGENCY SHUTDOWN SYSTEM

ENGINE INDICATING SYSTEM

POWER INDICATING SYSTEM

30 ENGINE PRESSURE RATIO (EPR)

BME TORQUE

MANIFOLD PRESSURE (MP)

RPM

TEMPERATURE INDICATING SYSTEM

CYLINDER TEMP (CHT)

EGT TIT

ANALYZERS

**IGNITION ANALYZER** 

VIBRATION ANALYZER

INTEGRATED ENGINE INST SYSTEM

ENGINE EXHAUST

COLLECTOR NOZZLE

NOISE SUPPRESSOR

THRUST REVERSER

ENG OIL SYSTEM (AIRFRAME FURNISH)

OIL STORAGE (AIRFRAME FURNISH)

50 ENG OIL DIST (AIRFRAME FURN)

COOLER

TEMPERATURE REGULATOR

OFF VALVE

55 ENGINE OIL INDICATING SYSTEM

PRESSURE

QUANTITY

TEMPERATURE

60 ENGINE STARTING

CRANKING

STARTER

VALVES CONTROLS

TURBINE SYSTEM (RECIP. ONLY)

POWER RECOVERY TURBINE

TURBO SUPERCHARGER

WATER INJECTION

ACCESSORY GEAR-BOXES

ENGINE (RECIPROCATING)

**ENGINE FRONT SECTION** 

ENGINE POWER SECTION

ENGINE CYLINDER SECTION

**ENGINE REAR SECTION** 

**ENGINE OIL SYSTEM** 

What is claimed is:

- 1. An information system for analysing data comprised of events pertaining to an object class wherein the object class is aircraft and an individual object is an individual aircraft having one or more identifiers unique to that individual aircraft, the system comprising:
  - (a) a data access component configured for accessing a database comprising multiple, integrated sets of data records derived from different sources whereby each data record pertains to an event for an individual object and comprises data fields, the data fields comprising multiple identifiers identifying said individual object;
  - (b) a user interface component for receiving user-selected criteria for field values of the data fields;
  - (c) a pre-processor component configured for identifying 25 multiple sub-databases from the data records of the database whereby each sub-database comprises data records identified by a different identifier, or a different set of identifiers, than those of the other sub-databases;
  - (d) a selection component configured for selecting from <sup>30</sup> one or more of the sub-databases the data records which correspond to the user-selected criteria; and,
  - (e) an analysis component configured for:
    - analysing the selected data records including identifying the selected data records according to a chronological sequence of events for individual objects covered by those selected data records, whereby each individual object covered by the selected data records is associated with a longitudinal set of data records; and
    - deriving output data for the events from the longitudinal sets of data records.
- 2. An information system according to claim 1 wherein the information provided by one said set of data records is used to supplement other data records of said database.
- 3. An information system according to claim 2 wherein the information provided by another one of said sets of data records is used to correct erroneous data field values in data records obtained from any said source.
- 4. An information system according to claim 3 wherein data records are service difficulty reports each said report pertaining to one said individual aircraft.
- 5. An information system according to claim 4 wherein said analysis component is configured for standardizing said data reports and individual objects as between the different originating sources thereof.
- 6. An information system according to claim 5 wherein said standardizing is performed on the basis of the traffic

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patterns for each aircraft and the types of aircraft models of each said originating source.

- 7. An information system according to claim 2 and further comprising a graphizer output component configured for graphically displaying said output data by multi-result interrelated graphs.
- 8. An information system according to claim 2 wherein said analysis component is configured for identifying chronological trends for one or more said individual objects and/or data records.
- 9. An information system according to claim 1 wherein said analysis component is configured for surveiling said events on a chronological basis for one or more said individual objects.
- 10. An information system according to claim 1 wherein said analysis component is configured for comparing, on a chronological basis and measured from a user-selected event, the events pertaining to one said individual object and the events pertaining to a user-selected comparison group of said objects.
- 11. An information system according to claim 1 wherein said analysis component is configured for normalizing the field values for a set of data fields and outputting the resulting normalized data.
- 12. An information system according to claim 1 wherein said analysis component is configured for identifying related data records of said data records.
- 13. An information system according to claim 1 wherein said analysis component is configured for identifying a statistically high number of events for an individual object within a designated time period.
- 14. An information system according to claim 1 wherein said analysis component is configured for rating said originating sources of said data records on the basis of the diligence of said originating sources in reporting events.
- 15. A method for analyzing data comprised of events pertaining to an object class, said method comprising the steps:
  - (a) accessing a database comprising at least one set of data records whereby each data record pertains to an individual said object and comprises a plurality of data fields, at least one of said data fields comprising an identifier identifying one said individual object;
  - (b) receiving user-selected criteria for field values of said data fields;
  - (c) selecting those of said data records which correspond to said user-selected criteria; and,
  - (d) analyzing said data records including identifying said selected data records according to a chronological sequence of events for individual objects covered by said selected data records, whereby each said individual object covered by said selected data records is associated with a longitudinal set of data records, and deriving output data for said events from said longitudinal sets of data records.

\* \* \* \* \*

## UNITED STATES PATENT AND TRADEMARK OFFICE

# CERTIFICATE OF CORRECTION

PATENT NO. : 6,631,384 B1

APPLICATION NO.: 09/654911 DATED: October 7, 2003

INVENTOR(S) : Alex Richman, Elliott M. Richman and Mark Hudak

It is certified that error appears in the above-identified patent and that said Letters Patent is hereby corrected as shown below:

### IN THE CLAIMS

Claim 2, col. 43, lines 43 and 44, delete "wherein the" and insert -- configured for using --

Claim 2, col. 43, lines 44 and 45, "is used" should be deleted;

Claim 3, col. 43, lines 46 and 47, delete "wherein the" and insert -- configured for using --;

Claim 3, col. 43, line 48, "is used" should be deleted;

Claim 3, col. 43, line 49, -- of -- should be inserted between "any" and "said" and "source" should be replaced with -- sources --;

Claim 4, col. 43 line 50, "3" should be replaced with --1 -- and -- said -- should be inserted between "wherein" and "data";

Claim 4, col. 43, line 51, delete "are" and insert -- comprise --;

Claim 7, col. 44, line 3, "2" should be replaced with -- 1 --;

Claim 8, col. 44, line 7, "2" should be replaced with --1 --;

Claim 15, col. 44, line 37, -- whereby the object class is aircraft and an individual object is an individual aircraft having one or more identifiers unique to that individual aircraft -- should be inserted between "object class" and "said method";

Claim 15, col. 44, line 39 and 40, "at least one set of data records" should be deleted;

Claim 15, col. 44 line 39 -- multiple, integrated sets of data records derived from different sources -- should be inserted between "comprising" and "whereby each";

Claim 15, col. 44, line 40, -- event for an -- should be inserted between "to an" and "individual";

Claim 15, col. 44, line 41, "a plurality of" should be deleted;

Claim 15, col. 44, lines 42 and 43 "at least one of said data fields comprising an identifier indentitying one said individual object" should be deleted;

Claim 15, col. 44, line 42 -- the data fields comprising multiple identifiers indentifying said individual object -- should be inserted after "data fields"

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APPLICATION NO.: 09/654911 DATED: October 7, 2003

INVENTOR(S) : Alex Richman, Elliot M. Richman and Mark Hudak

It is certified that error appears in the above-identified patent and that said Letters Patent is hereby corrected as shown below:

IN THE CLAIMS (cont'd)

Claim 15, col. 44, line 46, -- identifying multiple sub-databases from the data records of the database whereby each sub-database comprises data records identified by a different identifier, or a different set of identifiers, than those of the other sub-databases; -- should replace (c) as issued

Claim 15, col. 44, line 46, replace "(c)" with -- (d) --;

Claim 15, col. 44, line 46, delete "those of";

Claim 15, col. 44, line 48, replace "(d)" with -- (e) --;

Claim 15, col. 44, line 48, delete the first instance of "said" and replace it with -- the selected --;

Claim 15, col. 44, line 48, delete the second instance of "said" and replace it with -- the --;

Claim 15, col. 44, line 51, delete the first instance of "said" and replace it with -- those --

Claim 15, col. 44, line 51, delete the second instance of "said";

Claim 15, col. 44, line 52, delete "said" and replace it with -- the --;

Claim 15, col. 44, lines 53 - 55, delete ", and deriving output data for said events form said longitudinal sets of data records." and insert -- ; and --

# UNITED STATES PATENT AND TRADEMARK OFFICE CERTIFICATE OF CORRECTION

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APPLICATION NO.: 09/654911 DATED: October 7, 2003

INVENTOR(S) : Alex Richman, Elliot M. Richman and Mark Hudak

It is certified that error appears in the above-identified patent and that said Letters Patent is hereby corrected as shown below:

IN THE CLAIMS (cont'd)

Claim 15, col. 44, line 56, insert -- (f) deriving output data for said events from the longitudinal sets of data records. --

Signed and Sealed this

Twenty-second Day of May, 2007

JON W. DUDAS

Director of the United States Patent and Trademark Office